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A FINITE ELEMENT HEAD INJURY MODEL. VOLUME II. COMPUTER PROGRAM--ETC(U)

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Technical Report

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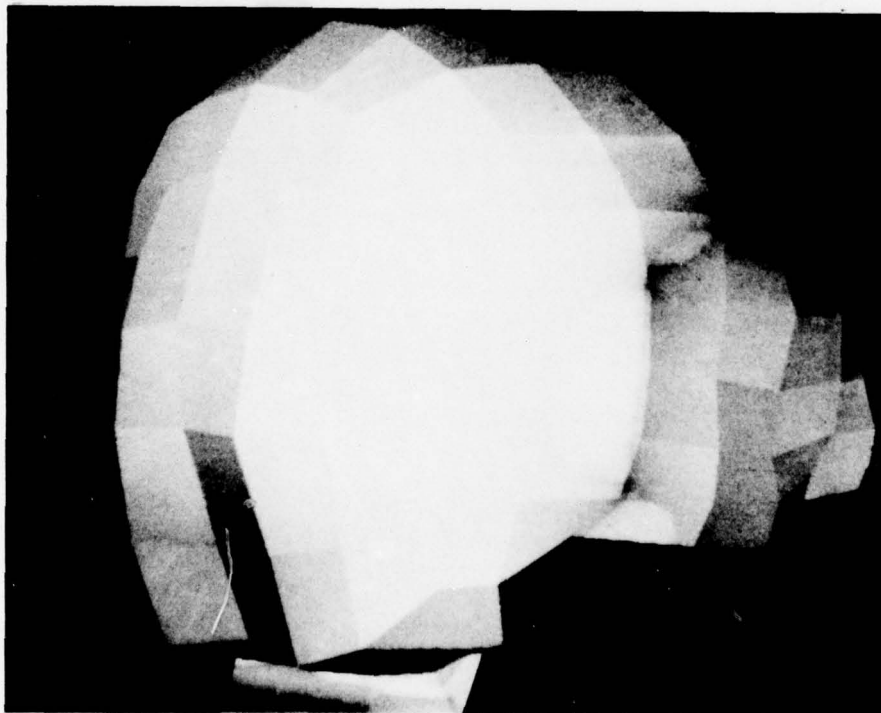
DEPARTMENT OF TRANSPORTATION,  
NAVAL AEROSPACE MEDICAL RESEARCH LABORATORY,  
and OFFICE OF NAVAL RESEARCH

July 1977

**CIVIL ENGINEERING LABORATORY**

Naval Construction Battalion Center

Port Hueneme, California 93043



**A FINITE ELEMENT HEAD INJURY MODEL - Volume II:**  
Computer Program Documentation

by T. A. Shugar

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A FINITE ELEMENT HEAD INJURY MODEL - VOLUME II:  
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Volume II contains necessary information and documentation for executing the HIM computer program. Documentation includes a user's manual, a flow chart, CDC 6600 control cards, sample input data, and a FORTRAN IV source code listing of the HIM program. In addition, listings are provided for a preprocessor (skull mesh generator), a bandwidth minimizer, and a subroutine for an improved finite element for simulating the load-deformation response of the skull.

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Volume II of this report contains Appendixes B, C, D, E, F, and G HIM Code documentation. Appendix A can be found in Volume I.

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## Appendix B

### SUMMARY OF FEAP MODIFICATIONS FOR IMPROVED EFFICIENCY

The major changes made to the FEAP program are listed below:

1. The entire program was recompiled using FTN (OPT=2). Several FORTRAN errors were found and corrected.
2. The PUNCH, PUNCHB, TAPE8 and TAPE12 files were removed. Buffer sizes were increased to 3075 for TAPE7 and TAPE10 and to 2050 for TAPE13 and TAPE14. TAPE11 becomes a direct access (QUICKIO) file with a buffer size of 65 words. QUICKIO is a FORTRAN callable subroutine which decreases I/O time.
3. All files were binary blocked using FTNBIN.
4. The global stiffness matrix is saved on a direct-access file in subroutine MAKEA. During forward reduction in USOL, no additional files are used for spill-over processing. All intermediate storage required is written in-place as an appropriate record on the QUICKIO file. This reduces the appropriate record on the QUICKIO file, which in turn reduces the amount of scratch storage required by more than half.
5. USOL was completely rewritten to allow direct-access equation processing. A COMPASS routine was included to increase the solution efficiency.
6. RESOLVE was rewritten to accommodate direct-access processing during the back-substitution phase.
7. Subroutines PRTMAT, COMPACT, CKQUAD, PACKD, SLD04, and QSHP8 were removed.
8. Subroutines FELMT, MESHCK, TSOLVE, and RESET were modified to accommodate the node resequencing (bandwidth minimizing) changes. Primarily, this involved adding named COMMON/MINBW/NEW(1500) and internally altering nodal references. Note that fixing NEW at 1,500 words inherently limits the program to 1500 nodes maximum. This COMMON would have to be extended if larger problems are anticipated.
9. Two input changes were made, neither of which affect the execution of existing decks; i.e., no actual changes were made to input processing unless the new features are desired. An optional first card may be included with the entry ISZDT. Blank COMMON will be set to this value allowing execution at reduced core size for smaller problems. The entry is read using I5 format. The default value is 58000. The second option is activated by entering "1" in cc 79 of the FEAP card. If activated, the program expects to encounter the PREFEAP punched output immediately following the FEAP card. PREFEAP punched output contains the optionally reordered nodal point numbering system.



## Appendix C

### USER INSTRUCTIONS FOR THE HIM CODE

## 1 CONTROL CARDS

### Card 1. (15)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	ISZDT	Blank COMMON will be set to this value. Default value is 58000	(1)

### Card 2. (6X, 12A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
7-10	—	Must contain the word FEAP 73	
13-78	—	Output page header	(2)
79	—	Enter 1 if optimized nodal point numbering is employed. Otherwise leave blank	(3)

Card Group 2a. If Card 2 contains a 1 in cc 79, the program expects to encounter a group of cards specifying an optimally reordered nodal point numbering system prior to encountering Card 3.

### Card 3. (15,1X,3A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	NOIM	Spatial dimension of problem. Use 3	
7-12	—	X coordinate designation	
13-18	—	Y coordinate designation	
19-24	—	Z coordinate designation	

### Card 4. (15,1X,3A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	NDF	Number of unknown displacements per node. Use 3.	
7-12	—	X displacement designation	
13-18	—	Y displacement designation	
19-24	—	Z displacement designation	



Card 5. (I5,10X,15,30X,3F10.0)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	NEN	Maximum number of nodes connected to any one element. Use 8.	
16-20	MBAN	Maximum expected half-bandwidth	(4)
51-60	CON3 (1)	User defined X scale factor	(5)
61-70	CON3 (2)	User defined Y scale factor	
71-80	CON3 (3)	User defined Z scale factor	

## Notes

- (1) For typical HIM code simulations this card can be left blank thereby specifying 58000 as the blank COMMON allocation.
- (2) Any description that the user wishes to have printed at the top of each output page can be entered.
- (3) Large, three-dimensional, dynamic problems can be expensive. A bandwidth minimizer has been employed to resequence the nodal point numbering scheme and optimally reorder the system of equations to provide economy in computational costs. A set of cards has been provided for this purpose which map the old and new numbering schemes. The mapping is transparent to the user; he refers always to the old system. Use of this option is highly recommended.
- (4) Typically this value is 150. The default value is 100 and is not sufficient. Therefore 150 should be specified.
- (5) These scale factors can be used to alter the size and shape of the HIM skull configuration as provided. Their default values are unity. Also these scale factors can be employed to change the provided geometrical configurations from inches to any appropriate unit of measure; *i.e.*, *centimeters*.

## II MATERIAL PROPERTY SPECIFICATION

### Card 1. (I5,1X,12A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	NUMMAT	Number of different material characterizations to follow.	(1)
7-12	—	Must contain the word MATERI	

One group of cards follows for each material. There must be NUMMAT card groups. There are at least two cards for each group; card (a) and card (b). For viscoelastic characterization, additional cards are required within the group as described below.

### Card (a). (I5,1X,I5,11A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	—	Material number (1 to NUMMAT)	
7-11	—	Element type	(2)
12-77	—	Alphanumeric description of each material	

### Card (b). (2I5,2F10.0)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	NG	Number of viscoelastic shear terms. If not viscoelastic, leave blank.	(3)
6-10	NK	Number of viscoelastic bulk terms. If not viscoelastic, leave blank.	
11-20	—	The infinite shear modulus or elastic shear modulus, $G_0$ .	
21-30	—	The infinite bulk modulus or elastic bulk modulus, $K_0$ .	
31-40	—	Mass density	

### Card (c). (8F10.)

Omit this and the next card if material is not viscoelastic.

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-10	—	First shear modulus term $G_1$	
11-20	—	First relaxation time term, $A_1$	
21-30	—	Second shear modulus term, $G_2$	
etc., until NG terms are identified.			

Card (d). (8F10.)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-10	—	First bulk modulus term, $K_1$	
11-20	—	First bulk relaxation time term, $B_1$	
21-30	—	Second bulk modulus term, $K_2$	
etc., until NK terms are identified.			

### Notes

(1) NUMMAT is usually assigned a value of 5. This allows the option for a different material specification for each of the following skull-brain regions.

<u>Material Number</u>	<u>Region</u>
1	Outer table bone
2	Diploe
3	Inner table bone
4	Subarachnoid space
5	Brain

(2) For the elastic skull materials (1, 2, and 3) the word ELM21 should be entered. For the subarachnoid space material (4) and brain material (5), an element possessing reduced integration is recommended, and the word ELM22 should be entered. Otherwise ELM21 may be entered for all materials.

(3) Viscoelastic material characterization is specified by the following equations:

$$G(t) = G_0 + \sum_{i=1}^{NG} G_i e^{-t/A_i}$$

$$K(t) = K_0 + \sum_{j=1}^{NK} K_j e^{-t/B_j}$$

Recommended units are the inch-pound-second system, although any consistent system can be used. (See note I(5)).



### III SKULL-BRAIN GEOMETRICAL INPUT

#### Card 1. (I5,1X,A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	NUMNP	Number of nodal points in the skull-brain system	(1)
7-12	—	Must contain the word NODAL	

There must follow NUMNP nodal point cards in numerical order.

#### Typical Nodal Point Card (I5,I15,3F10.0)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	—	Nodal point number	
15	—	Insert 0 for free node or 1 for restraint in X direction	(2)
16	—	Insert 0 for free node or 1 for restraint in Y direction	
17	—	Insert 0 for free node or 1 for restraint in Z direction	
21-30	—	X coordinate value	
31-40	—	Y coordinate value	
41-50	—	Z coordinate value	



## Notes

(1) For the six-cubed, half-skull model configuration provided, this value is 716. This is the model which has been most useful in the development of the HIM code. A complete set of nodal point cards whose coordinates are entered in inches has been provided for this configuration.

(2) The boundary conditions to be used for the HIM simulation are activated in cc 15, 16, and 17. Entering a 0 allows the node to move in any one of three possible directions. A zero in cc 15, 16, and 17 means the node is completely free. A 1 means that the node's displacement is constrained to move in a manner to be specified in subsequent input cards (see Section VI). The sets of nodal point cards which have been provided do not have any boundary condition preset with the following exceptions. The nodal point cards contain a 1 in cc 16 for all nodal points located in the midsagittal plane. Thus Y component displacements for half skull simulation are automatically constrained to zero in the midsagittal plane for the user's convenience.

#### IV ELEMENT CARDS

##### Card 1. (15,1X,A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	NUMEL	Number of elements in the skull-brain system	(1)
7-12	—	Must contain the word ELEMEN	

There must follow 2 cards for each element, taking the elements in numerical order.

##### Typical Element Card Set

##### Card (a) (215)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	—	Element number	
6-10	—	Material number	(2)

##### Card (b) (814)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-4	—	Node 1	
5-8	—	Node 2	
9-12	—	Node 3	
.	.		
.	.		
.	.		
29-32	—	Node 8	

### Notes

- (1) For the six-cubed, half-skull model configuration, the value is 555. A complete set of element cards has been provided for this configuration.
- (2) A material number for which properties have been identified in Section II is entered. The corresponding element is then assigned those properties.

## V INITIAL CONDITIONS

These cards can be omitted if the initial displacement velocity and acceleration of the skull-brain system are zero.

### Card 1. (15,1X,A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	NICD	Number of initial condition vectors. Specify a "3".	(1)
7-12	—	Must contain the word INITIA	

### Card 2. (6X,2A6) Repeat NICD times

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
7-18		Descriptive title for initial conditions	(2)

### Card 3. (215,7F10.0) Repeat for each degree of freedom

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	—	Degree of freedom number for X; 3*NODE -2 = entry for Y; 3*NODE -1 = entry for Z; 3*NODE -0 = entry	(3)
6-10	—	Generator increment	(4)
11-20	—	Initial displacement value	
21-30	—	Initial velocity value	
31-40	—	Initial acceleration value	

### Card 4. (15)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5		Largest node times 3 plus 1 (NUMNP*3+1)	(5)

## Notes

- (1) When initial conditions of the skull-brain system are specified, the initial displacement, velocity, and acceleration (three vectors) are supplied as input.
- (2) Three cards follow Card 1 and merely describe each of the three initial condition vectors. These descriptions will appear above each vector in the subsequent printout.
- (3) Any or all nodes may be assigned an initial condition in the X, Y, and Z directions. But the initial condition must correspond to a degree of freedom number associated with the node and not the node number itself. The indicated equation is employed to obtain the degree of freedom number.
- (4) The generator increment is used when the initial condition is uniformly applied over the entire skull-brain system, and this is generally the case. It avoids the necessity for inputting all but a few cards. For example, if the skull-brain system is initially moving at 240 in./sec in the +X direction, two cards are required to assign the initial velocity to all nodes. The first would be for node 1 and the second for node NUMNP (largest node number in the system). The first card would contain a degree of freedom number of  $1(3*1-2)$ , a generator increment of 2 (assigns the initial condition to only the degree of freedom numbers associated with the X-direction), and a 240 in cc 21-30. The second card would contain the value of  $3*NUMNP-2$  in cc 1-5 and 240 in cc 21-30.
- (5) This card ends the initial condition data.



## VI PROPORTIONAL LOAD OR DISPLACEMENT SPECIFICATION USING ANALYTICAL FUNCTIONS

This section should be skipped if using point-by-point input loading. Such loading is covered in Section VIII. But this section is applicable to simulated dynamic loads or displacements which rise and subside according to a single time function (see Section VIII for the time function). They may all be assigned different magnitudes, but otherwise at any given instant the loads are proportional to one another.

### Card 1. (15,1X,A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	—	Last node to which a force or displacement is to be specified. Must contain the word FORCE.	(1)

The following group of cards is composed of one card for each node to which a specified force or displacement is assigned.

### Typical Card (15,5X,6F10.0)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5		Node to which a force or displacement is specified	
11-20		X component of force or displacement	(2)
21-30		Y component of force or displacement	
31-40		Z component of force or displacement	



### Notes

(1) Merely means the largest node number to which a force or displacement is applied. Forces and displacements must be applied at nodal points.

(2) This value is the ratio between the simulated peak load component at this node and the largest peak simulated load considering all nodes. It may also be a direction cosine. Only ratios from 0 to 1 are assigned here. The magnitude of the resultant for the largest peak load is prescribed in Section VIII (see Note (9)).

## VII PLOT CARD

This card is to be used only when diverting computed response data to an output tape. This action does not suppress the printing of output data.

### Card 1. (I5,I1X,A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	—	Number of time steps plus one for which data is output	(1)
7-12		Must contain the word PLOT3D	

**Note**

(1) Data will be placed on an output tape for subsequent postprocessing for each time step up to, but not including the value indicated. Usually data for all time steps is desired. Therefore this entry should be one more than the number of time steps prescribed in the dynamic analysis. See the variable NTS in the next section.

## VIII IMPLICIT TIME-INTEGRATION SPECIFICATION

This card is required on all static or dynamic runs.

### Card 1. (I5,1X,A6)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5	NSEQ	Number of time sequences, usually 1	(1)
7-12		Must contain the word IMPLIC for dynamic problems. Must contain the word VISCOE for static problems.	

NSEQ sets of the following cards are required

### Card (a) (F10.0,8I5,2F10.0)

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-10	DT	Time step size	(2)
11-15	NTS	Number of time steps in the sequence(s) plus 1	(3)
16-20	INT	Print interval	(4)
21-25	NNI	First node printed, generally 1	(5)
26-30	NNE	Last node printed, generally NUMNP	
31-35	NEI	First element stress printed, generally 1	
36-40	NEF	Last element stress printed, generally NUMEL	
41-45	NPROP	Number of proportional loads	(6)
46-50	NFORC	Largest node to which a force or displacement has been specified in a point-by-point specification of input loading	(7)
51-60	BETA	Newmark integration parameter, usually specify 0.25	(8)
61-70	DEL	Newmark integration parameter, usually specify 0	

**Card (b). (2I5,7F10.0) One for each NPROP.** Use only when specifying analytical function input. If using point-by-point input skip to card (c).

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5		Proportional load type, either 1 or 2	(9)
6-10	K	Exponent for type 2	
11-20	TMIN	Start time for loading	
21-30	TMAX	End time for loading	
31-40		AO	
41-50		A1	
51-60		A2	
61-70		A3	
71-80		A4	

**Card group (c) (15,5X,6F10.0).** This group of cards is comprised of one card for each node to which an input value is assigned. Node NFORC is the last node from which a card is provided. The group represents a loading vector, and generally there should be one group for each time step in the dynamic analysis. Omit this card group if NPROP is non-zero.

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5		Node to which force or displacement is specified	
11-20		X component value of force or displacement	
21-30		Y component value of force or displacement	
31-40		Z component value of force	

**Card (3) (A5) last data card**

<u>Columns</u>	<u>Variable</u>	<u>Description</u>	<u>Note</u>
1-5		Must contain the word STOP	



## Notes

- (1) Little experience has been gained with values other than 1. Potentially, NSEQ can be used to specify a broader range of different loading functions. (Card (a) describes how different functions can be specified with  $NSEQ > 1$ ). For example, one analysis could consist of two sequences where the first is the onset of loading requiring a small time step and the second is the loading decay which may be slower and during which a larger, more economical, time step can be employed.
- (2) This value should be chosen with caution. It generally depends on the rise time and should be no greater than one-fifth of that value. A value of one-tenth would be sufficiently small.
- (3) This is the number of time steps plus 1 for which the dynamic analysis is to proceed. Generally, it is chosen so that the entire rise and decay of the loading function can be simulated with the prescribed time step.
- (4) Several inches of paper may be required, depending upon NTS, for the printing of output information. A blank in cc 16-20 will cause the printing of data for each time step; issuing a 1 will cause the printing of data for every other time step, etc.
- (5) The next four entries also provide for reductions in the amount of output. If a few nodes and elements can be identified as key check points in the analysis it is only necessary to monitor their output. However, data elsewhere should at least be saved on tape (see Section VII).
- (6) This value is employed to distinguish among the specification of different input load functions in the dynamic analysis. For example if it is desired that the prescribed load vary as a haversine function and subsequently change to a linear function, then  $NPROP=2$ . This is also true of a triangular function. If a single function, such as a ramp function or haversine function, is desired then  $NPROP=1$ . If a point-by-point input loading is desired then  $NPROP$  is left blank.
- (7) This entry is for point-by-point load specification only. If proportioned load functions are employed,  $NFORC$  is left blank.
- (8)  $BETA$  and  $DEL$  are the Newmark integration parameters (implicit). Investigators have determined that  $BETA = 0.25$  and  $DEL = 0$  are good choices. If  $DEL > 0$  artificial positive damping is introduced and if  $DEL < 0$  artificial negative damping is introduced. The numerical integration will be unconditionally stable if

$$BETA \geq \frac{4 (DEL + 1)^2}{16}$$

The higher and least important nodes may be damped out by choosing  $BETA = 0.276$  and  $DEL = 0.05$ .



(9) Proportional load type 1 is a polynomial function,

$$f(t) = A_0 + A_1 \cdot t + A_2 \cdot t^2 + A_3 \cdot t^3 + A_4 \cdot t^4$$

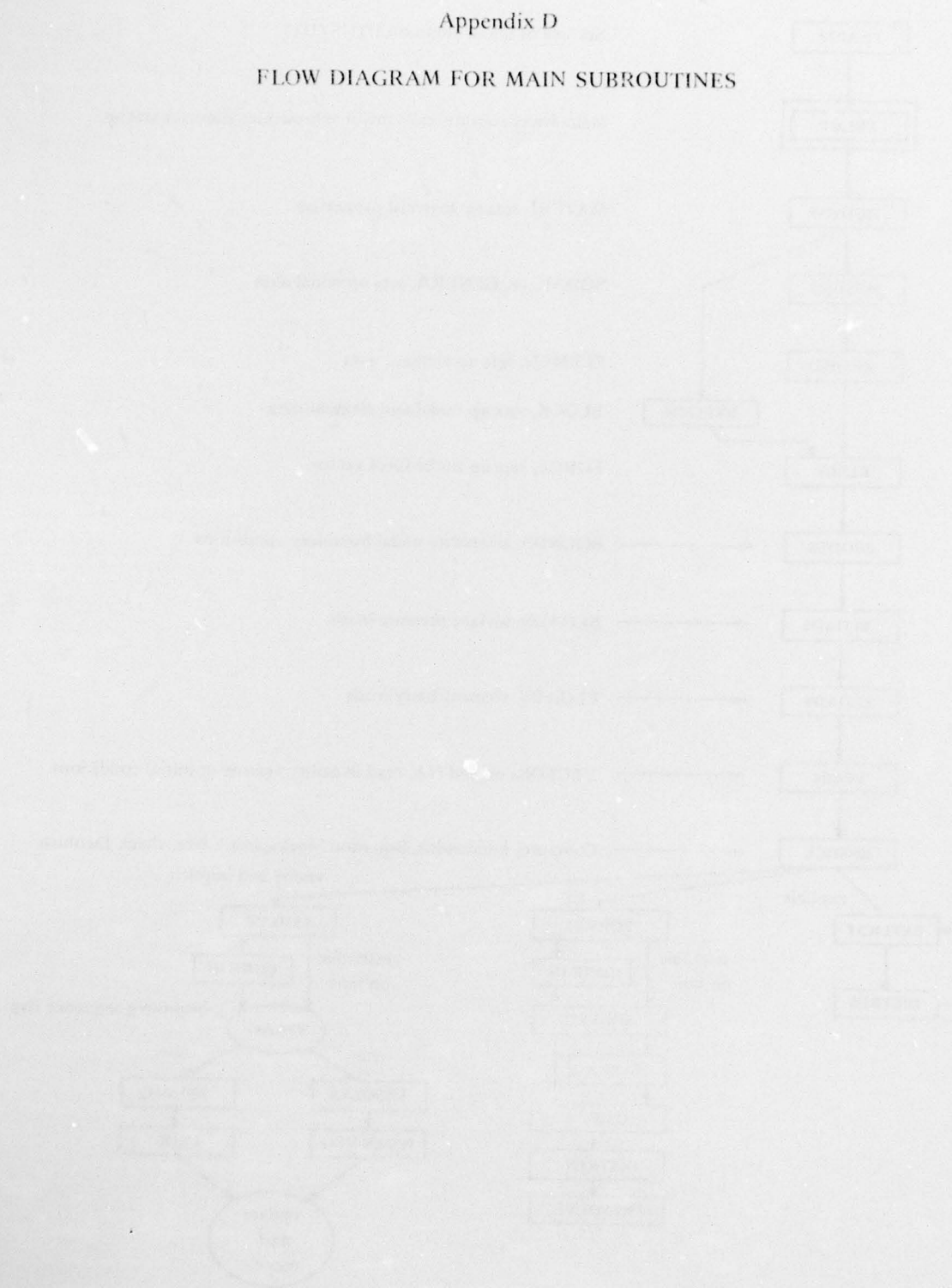
Proportional load type 2 is a transcendental function,

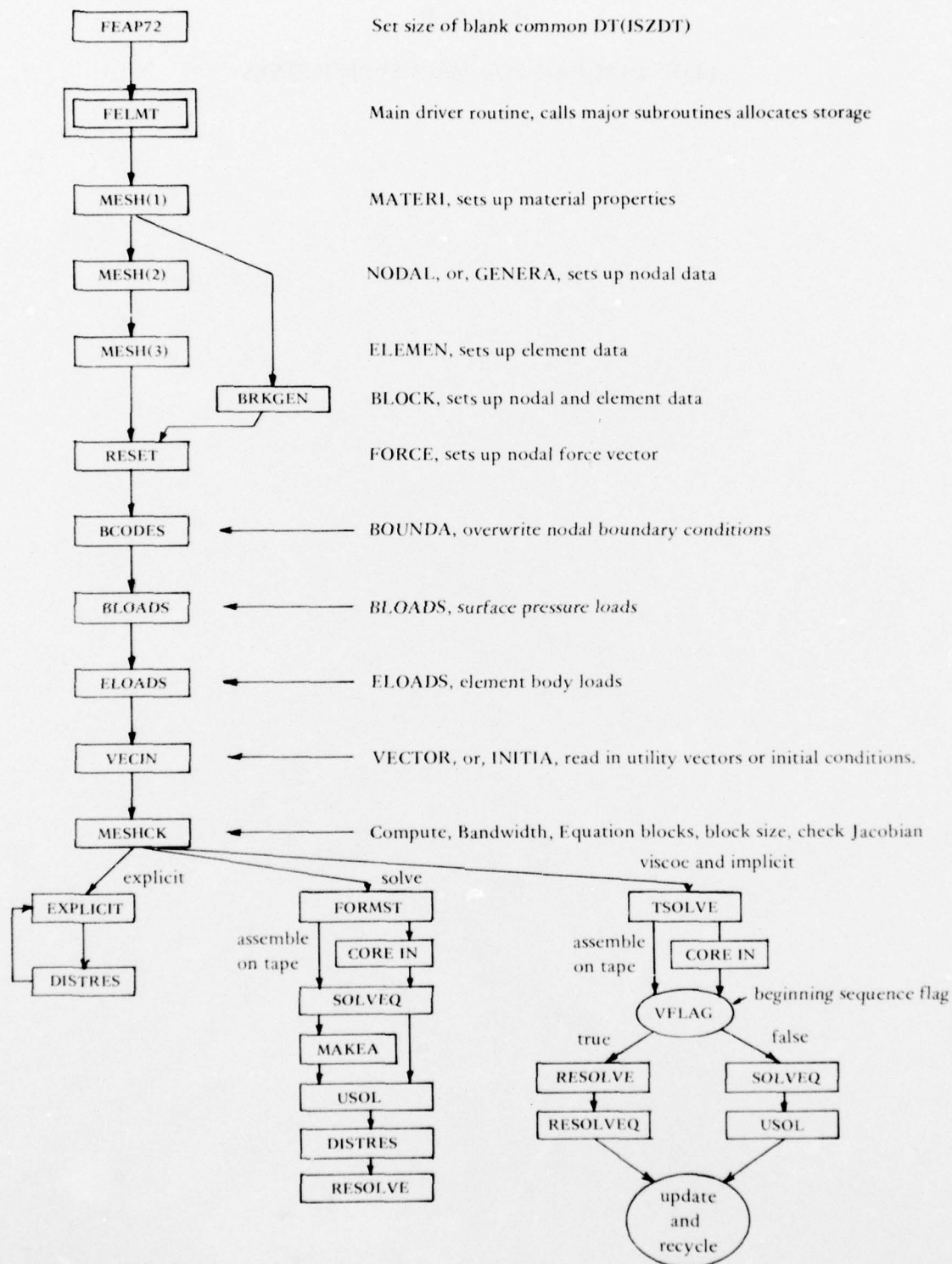
$$f(t) = A_0 \cdot \sin^K(A_1 \cdot t) + A_2 \cdot \cos^K(A_3 \cdot t) + A_4$$

A haversine function has been found to be particularly useful in the specification of impact forces,  $F(t) = A_0 \cdot \sin^K(A_1 \cdot t)$ , where  $A_0$  is the peak force magnitude,  $A_1$  is pi divided by the load duration in seconds, and  $K = 2$ . Force magnitude must be halved when employing a half-skull model.

Appendix D

FLOW DIAGRAM FOR MAIN SUBROUTINES





Appendix E  
SAMPLE JOB CONTROL CARDS



\$CHARGE,L2317\*X=008.  
MONKEY,CM15000,CL262000,T4000,I04000,TP1,P6.  
REQUEST FEAP,HI. (L2313)  
SKIPF (FEAP,1,17,H)  
COPYHF (FEAP,LGO)  
REWIND (FEAP)  
UNLOAD (FEAP)  
REQUEST TAPE30,HI. (SAVE)  
ATTACH (IO,QUICKIO)  
HFL 150000.  
SET (0)  
LOAD (LGO)  
LOAD (IO)  
EXECUTE.  
EXIT.  
EXIT.



Appendix F

HIM CODE LISTING

\$CHARGE,L231744=008.  
MONKEY,CM1500,CL262000,14000,104000,TP1,P6.  
REQUEST FEAP,M1. (L2313)  
SKIPF (FEAP,1,17,B)  
COPYMF (FEAP,L60)  
REWIND (FEAP)  
UNLOAD (FEAP)  
REQUEST, TAPE 30,M1. (SAVE)  
ATTACH (IO,QUICKIO)  
HFL,150000.  
SET (0)  
LOAD (L60)  
LOAD (IO)  
EXECUTE.  
EXIT.  
EXIT.

PROGRAM FEAP72 (INPUT=65,OUTPUT,TAPE5=INPUT,TAPE6=OUTPUT,  
 • TAPE7=3075,TAPE9,TAPE10=3075,TAPE11=65,TAPE13=2050,  
 • TAPE14=2050,TAPE30)

```

C
CXXXX NOTE NOTE NOTE NOTE
C TWO ADDL. INPUT FEATURES ARE INCLUDED IN THIS VERSION
C
C 1. AN OPTIONAL CARD MAY BE INCLUDED IMMEDIATELY PRECEDING
C THE FIRST FEAP CARD. THIS CARD HAS THE ENTRY ISZDT (15).
C ADMISSABLE RANGE IS 500.LE. ISZDT .LE.65000.
C THE PROGRAM WILL ASSIGN THIS VALUE TO THE SIZE OF BLANK
C COMMON AND RFL UP TO THIS LIMIT. THIS WILL ALLOW THE
C PROGRAM TO LOAD AND EXECUTE IN ONLY THE AMOUNT OF CORE REQD.
C IF OMITTED, A DEFAULT VALUE OF 58000 IS ASSIGNED TO ISZDT.
C
C 2. A 1 ENTERED IN CC 79 OF THE FEAP CARD INDICATES THAT
C THE NODE RESEQUENCING CARDS PUNCHED BY THE FEAP PREPROCESSOR
C ARE TO BE INPUT IMMEDIATELY FOLLOWING THE FEAP CARD.
C
CXXXX NOTE NOTE NOTE NOTE
C
C..... FINITE ELEMENT ASSEMBLY PROGRAM * STORAGE ALLOCATION
C
COMMON /TIMING/ TRICP,NTRI,HACKCP,NBACK
COMMON DT(1)
DIMENSION IHIN(6)
DATA IHIN/7,9,10,13,14,30/
DATA JFEAP/4,HFEAP/
CALL FINKIN (1,6,IHIN)
CALL BLKDAT
TRICP=0.
HACKCP=0.
NTRI=J
NBACK=0
I1=0.
I2=0.
C
C..... ISZDT MUST COINCIDE WITH SIZE OF ARRAY DT
C
CXXXX
C CHECK FOR OPTIONAL FIRST CARD
CXXXX
ISZDT=58000
READ (5,100) IFEAP
100 FORMAT (A4)
HACKSPACE 5
IF (IFEAP.EQ.JFEAP) GO TO 50
READ (5,200) ISZDT
200 FORMAT (15)
IF (ISZDT.LT.500) ISZDT=500
IF (ISZDT.GT.65000) ISZDT=65000
50 CALL XREF (LOC(DT(ISZDT)))
C
C..... REASONABLE MAXIMUM BAND WIDTH
C
MBAND=1000
CALL FELMT(DT,ISZDT,MBAND)
CXXXX
C PRINT PROGRAM TIMINGS
C NOTE THAT THE TIMINGS ARE NOT MEANINGFUL
C IF STACKED JOBS WERE PROCESSED
CXXXX
IF (NTRI.GT.0) T1=TRICP/NTRI
IF (NBACK.GT.0) T2=HACKCP/NBACK
WRITE (6,500) NTRI,TRICP,T1,NBACK,HACKCP,T2
500 FORMAT ( //
*46H PROGRAM TIMINGS //
*46H NUMBER OF TOTAL TIME FOR EACH /
*46H TRIANGULAR TIME FACTORIZATION /
*46H FACTORIZATIONS (CP) (CP) //
* I15,F10.3,F20.3//

```

FEA 4C  
 FEA 5C  
 FEA 6C

FEA 8C  
 FEA 9C  
 FEA 10C

FEA 12C  
 FEA 13C  
 FEA 14C  
 FEA 16C

*46M	NUMBER OF	TOTAL	TIME FOR EACH /	
*46M	BACKWARD	TIME	REDUCTION /	
*46M	REDUCTIONS	(CP)	(CP)	//
• 115.F10.3.F20.3// )				
STUP				FEA 17C
END				FEA 18C
*DECK MANUAL				
SUBROUTINE MANUAL				MAN 1C
C				MAN 2C
C	***** USER INSTRUCTIONS AND INPUT FORMATS FOR FEAP-73 *****			MAN 3C
C				MAN 4C
C	FINITE ELEMENT ANALYSIS PROGRAM			MAN 5C
C				MAN 6C
C				MAN 7C
C				MAN 8C
C	FEAP73 IS A GENERAL (F)INITE (E)LEMENT (A)NALYSIS (P)ROGRAM			MAN 9C
C	WHICH FURNISHES TO THE USER MESH INPUT/OUTPUT, ELEMENT ASSEMBLY			MAN 10C
C	AND SOLUTION OF EQUATIONS (LINEAR, IMPLICIT AND EXPLICIT TIME			MAN 11C
C	DEPENDENT, NONLINEAR), PRESCRIBED GENERALIZED NODAL FORCES,			MAN 12C
C	PRESCRIBED NODAL AND ELEMENT DATA, AND OUTPUT OF THE			MAN 13C
C	GENERALIZED DISPLACEMENTS AND FORCES. ELEMENT MATRICES FOR TWO			MAN 14C
C	AND THREE DIMENSIONAL LINEAR ELASTICITY, SHELLS, PLATES, AND			MAN 15C
C	FIELD (LAPLACE EQUATION) PROBLEMS ARE AVAILABLE. ALTERNATIVELY			MAN 16C
C	USERS MAY SUPPLY THEIR OWN ELEMENT LIBRARY BY PROVIDING A			MAN 17C
C	SUBROUTINE CALLED ELMTNN, WHERE NN IS A TWO DIGIT NUMBER			MAN 18C
C	IDENTIFYING THE ELEMENT SUBROUTINE. EACH ELEMENT SUBROUTINE			MAN 19C
C	HAS AT LEAST FOUR BASIC FUNCTIONS WHICH ARE DELINEATED BY A			MAN 20C
C	SWITCHING PARAMETER, ISW, IN THE SUBROUTINE.			MAN 21C
C				MAN 22C
C	ELMTNN(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,			MAN 23C
C	IX,IF,FORCE,ESTIF,U,VECT,ISW)			MAN 24C
C				MAN 25C
C	N IS ELEMENT NUMBER.			MAN 26C
C	MA IS THE MATERIAL NUMBER.			MAN 27C
C	NDIM IS SPATIAL DIMENSION, 1,2,OR 3.			MAN 28C
C	NDF IS NUMBER OF DEGREES OF FREEDOM PER NODE.			MAN 29C
C	NEL IS THE NUMBER OF EXTERNAL NODES PER ELEMENT			MAN 30C
C	NEL1 IS DIMENSION OF ELEMENT PROPERTY ARRAY.			MAN 31C
C	NSTF IS THE SIZE OF THE ELEMENT STIFFNESS.			MAN 32C
C	NSIZV IS THE SIZE OF UTILITY VECTORS.			MAN 33C
C	NVEC IS THE NUMBER OF UTILITY VECTORS.			MAN 34C
C	MCT IS A PRINTER LINE COUNTER.			MAN 35C
C	DM IS A PARAMETER FOR MATERIAL IDENTIFICATION.			MAN 36C
C	DIL(1) IS MATERIAL PROPERTY MATRIX (63 CELLS).			MAN 37C
C	XYZ(NDIM,1) ARE NODAL COORDINATES.			MAN 38C
C	IX(NEL1,1) ARE ELEMENT PROPERTIES, NODES, ETC.			MAN 39C
C	F(NDF,1) ARE NODAL GENERALIZED FORCES.			MAN 40C
C	FORCE(NSTF,2) IS ELEMENT FORCE VECTOR TO BE			MAN 41C
C	COMPUTED. COLUMN 2 IS LUMPED MASS			MAN 42C
C	ESTIF(NSTF,NSTF) IS ELEMENT MATRIX TO BE			MAN 43C
C	COMPUTED.			MAN 44C
C	VECT(NSIZV,1) ARE PRESCRIBED NODAL OR ELEMENT			MAN 45C
C	QUANTITIES, TEMPERATURES ETC.			MAN 46C
C	U(NDF,1) IS SOLUTION VECTOR.			MAN 47C
C	ISW IS SWITCHING PARAMETER.			MAN 48C
C				MAN 49C
C	ISW=1: ** MATERIAL CHARACTERIZATION**			MAN 50C
C	ISW=2: ** CHECK ELEMENT FOR POSITIVE AREA *			MAN 51C
C	ISW=3: ** ELEMENT STIFFNESS COMPUTATION**			MAN 52C
C	ISW=4: ** ELEMENT STRESSES AND PRINTOUT**			MAN 53C
C	ISW=5: ** ELEMENT LOAD COMPUTATION			MAN 54C
C	ISW=6: ** NONLINEAR GENERALIZED FORCES			MAN 55C
C	OTHER ISW MAY BE USED FOR SPECIAL PURPOSES.			MAN 56C
C				MAN 57C
C	USERS CAN GENERATE SURFACE LOADINGS BY PROVIDING SLONN			MAN 58C
C	SUBROUTINES (WHERE NN IS A TWO DIGIT NUMBER BETWEEN 01 AND 05)			MAN 59C
C	THAT SPECIFY THE LOAD ROUTINE. THE SUBROUTINE IS ACCESSED BY			MAN 60C
C	THE CALL TO			MAN 61C
C				MAN 62C
C	SLONN(NDIM,NDF,NOP,NPRES,IPRES,PR,XYZ,FS)			MAN 63C
C				MAN 64C
C	WHERE IN ADDITION TO QUANTITIES DEFINED ABOVE			MAN 65C



C	FOR ELMTNN.	MAN 66C
C	NUP IS THE DIMENSION OF LOADED SURFACE	MAN 67C
C	NPRES IS NUMBER OF LOADED NODES (MAX 8)	MAN 68C
C	IPRES(R) ARE NODE NUMBERS OF LOADED NODES.	MAN 69C
C	PR(R) ARE LOAD VALUES AT CORRESPONDING IPRES	MAN 70C
C	NODES.	MAN 71C
C	FS(6,8) ARE THE COMPUTED GENERALIZED (NODAL)	MAN 72C
C	FORCES FOR EACH DEGREE OF FREEDOM AT EACH	MAN 73C
C	IPRES NODE.	MAN 74C
C	SEE SECTION 7.1 FOR DATA INPUT DETAILS.	MAN 75C
C		MAN 76C
C		MAN 77C
C	INTEGRATION TABLE IS ACCESSED BY THE CALL	MAN 78C
C	CALL INTEGL(LIM,NCI,NDIM,LINT,STUW)	MAN 79C
C		MAN 80C
C	STUW(4,M) INTEGRATION POINTS AND WEIGHTS.	MAN 81C
C	**NOTE** M MUST BE SET EXPLICITLY AND BE LARGER	MAN 82C
C	THAN OR EQUAL TO LINT.	MAN 83C
C	LINT = RETURNS WITH NUMBER INTEGRATION POINTS.	MAN 84C
C	NCI = 0 RETURNS GAUSS POINTS AND WEIGHTS IN	MAN 85C
C	STUW.	MAN 86C
C	LIM = 1 TO 5 IS NUMBER OF GAUSS POINTS/DIM-	MAN 87C
C	ENSION.	MAN 88C
C	NCI = 1 RETURNS A SPECIAL 3-D GAUSS FORMULA.	MAN 89C
C	SET LIM = 1 FOR 6 PT. CUBIC ACCURACY.	MAN 90C
C	SET LIM = 2 FOR 14 PT. QUINTIC ACCURACY.	MAN 91C
C	NCI = 2 RETURNS TRIANGULAR INTEGRATION FORMULA.	MAN 92C
C	SET LIM = 1 FOR 1 PT. LINEAR ACCURACY.	MAN 93C
C	SET LIM = 2 FOR 3 PT. QUADRATIC ACCURACY.	MAN 94C
C	SET LIM = 3 FOR 7 PT. QUARTIC ACCURACY.	MAN 95C
C		MAN 96C
C		MAN 97C
C		MAN 98C
C		MAN 99C
C	1.) DATA TYPE IDENTIFICATION CARDS (IS,IX,IZAB).	MAN100C
C		MAN101C
C	EACH DATA SEGMENT IS PRECEDED BY A CARD WHICH IDENTIFIES THE	MAN102C
C	TYPE OF DATA AND LIMITS ON THE AMOUNT OF DATA WHICH IMMEDIATELY	MAN103C
C	FOLLOWS THE CARD. EXCEPT AS NOTED THE DATA SEGMENTS MAY APPEAR	MAN104C
C	IN ANY ORDER. THE IDENTITY CARDS MAY ALSO AID THE USER IN	MAN105C
C	INTERPRETING THE INPUT DATA CARDS. AS SUPPLIED THERE ARE	MAN106C
C	TWENTY-FIVE DIFFERENT DATA IDENTIFICATION CARDS. THESE ARE	MAN107C
C		MAN108C
C	COL 7 TO 12 IDENTITY(RESTRICTIONS)	MAN109C
C		MAN110C
C	FEAP73 START OF EACH PROBLEM (MUST PRECEDE ALL OTHER	MAN111C
C	DATA).	MAN112C
C	FILE CHANGE OUTPUT PAGE HEADINGS	MAN113C
C	REMARK COMMENTS ON OUTPUT	MAN114C
C	MATERI MATERIAL CHARACTERIZATION.	MAN115C
C	NODAL NODAL CARDS	MAN116C
C	PULAH PULAH CONVERSION. (PRECEDE BY NODAL, GENERA, OR	MAN117C
C	BLOCK)	MAN118C
C	ELEMEN ELEMENT CONNECTION CARDS.	MAN119C
C	GENERA GENERATE NODES IN A LINEAR PATH BY ANY INCREMENT	MAN120C
C	BLOCK GENERATE ALL MESH DATA (BOTH NODAL AND ELEMENT)	MAN121C
C	FOR A 2, (OR 3) DIMENSIONAL REGION WHOSE BOUNDARY	MAN122C
C	MAY BE DEFINED BY 4(8) OR 8(20) COLLOCATED POINTS	MAN123C
C	BOUNDARY CODE PRESCRIPTION (PRECEDE BY NODAL OR	MAN124C
C	GENERATE OR BLOCK)	MAN125C
C	VECTOR PRESCRIBED NODAL OR ELEMENT DATA (PRECEDE BY	MAN126C
C	NODAL OR PULAH AND ELEMEN)	MAN127C
C	FORCE NODAL GENERALIZED FORCES (PRECEDE BY NODAL OR	MAN128C
C	GENERA OR BLOCK).	MAN129C
C	HLAUS SURFACE LOADINGS (SAME AS FORCE).	MAN130C
C	ELAUS ELEMENT LOADINGS (SAME AS FORCE).	MAN131C
C	MESH CHECK CONSISTENCY OF MESH ONLY (SAME AS SOLVE)	MAN132C
C	PLOT PLOT MESH, RESULTS (SAME AS SOLVE)	
C	INITIA INITIAL CONDITION PRESCRIPTION FOR DYNAMIC	MAN134C
C	ANALYSIS (PRECEDE BY NODAL, GENERA OR BLOCK)	MAN135C
C	SOLVE COMPLETE FORMULATION AND SOLUTION FROM ELEMENTS	MAN136C
C	(PRECEDE BY MATERI, NODAL OR GENERA, AND ELEMEN	MAN137C



C	RESOLV	OR PRECEDE BY MATER1 AND BLOCK 1	MAN138C
C		USE PREVIOUS PROBLEM DESCRIPTION WITH NEW LOAD	MAN139C
C	EXPLIC	ONLY (PRECEDE BY SOLVE AND NEW LOADING CARDS).	MAN140C
C		DYNAMIC SOLUTION BY EXPLICIT INTEGRATION, (SAME	MAN141C
C		AS SOLVE)	MAN142C
C	IMPLIC	IMPLICIT INTEGRATION OF DYNAMIC PROBLEMS	MAN143C
C		(PRECEDE BY SAME DATA AS FOR SOLVE)	MAN144C
C	VISCOE	QUASI-STATIC LINEAR VISCOELASTIC INTEGRATION	MAN145C
C		(PRECEDE BY SAME DATA AS FOR SOLVE)	MAN146C
C	FOURIE	FOURIER COMPOSITION (SAME AS SOLVE)	MAN147C
C	ADDDUP	ACCUMULATE FOURIER SOLUTION (AFTER FOURIE)	MAN148C
C	STOP	NORMAL EXIT (MUST FOLLOW ALL DATA)	MAN149C
C			MAN150C
C			MAN151C
C	***NOTE*** EACH IDENTIFIER IS PUNCHED STARTING IN COL 7 (LEFT JUSTIFIED).		MAN153C
C	EXCESS CARDS MAY EXIST BETWEEN EACH SECTION OF DATA, HOWEVER, THE DATA TO BE USED MUST IMMEDIATELY FOLLOW THE TYPE CARD AND MUST BE IN PROPER ORDER. NO PARTICULAR ORDER OF THE TYPE CARDS IS NECESSARY EXCEPT THAT THE FEAP73 CARD MUST ALWAYS BE THE FIRST CARD IN EACH SET OF DATA, AND RESTRICTIONS MUST BE OBSERVED.		MAN154C
C			MAN155C
C			MAN156C
C			MAN157C
C			MAN158C
C			MAN159C
C			MAN160C
C			MAN161C
C	2.) PROBLEM INITIATION AND CONTROL CARDS		MAN162C
C	CARD 1. (6X,12A6)		MAN163C
C			MAN164C
C	COL 7 TO 12	MUST CONTAIN WORD FEAP73	MAN165C
C	COL 13 TO 74	OUTPUT PAGE HEADER	MAN166C
C			MAN167C
C	CARD 2. (15,1X,3A6)		MAN168C
C			MAN169C
C	COL 1 TO 5	NDIM - SPATIAL DIMENSION OF PROBLEM (1 TO 3)	MAN170C
C	COL 7 TO 12	NAMES TO BE PRINTED AS OUTPUT HEADERS TO	MAN171C
C	COL 13 TO 18	COORDINATES - IF BLANK SET TO 1,2,3 AS NEEDED.	MAN172C
C	COL 19 TO 24		MAN173C
C			MAN174C
C	CARD 3. (15,1X,6A6)		MAN175C
C			MAN176C
C	COL 1 TO 5	NDF - NUMBER OF UNKNOWN PER NODE (1 TO 6)	MAN177C
C	COL 7 TO 12	NAMES TO BE PRINTED AS OUTPUT HEADERS OF THE	MAN178C
C	COL 13 TO 18	GENERALIZED DISPLACEMENTS AND FORCES - IF	MAN179C
C	.....	BLANK SET TO 1,2,3,4,5,6 AS NECESSARY	MAN180C
C	COL 37 TO 42		MAN181C
C			MAN182C
C	CARD 4. (615,5F10.0)		MAN183C
C			MAN184C
C	COL 1 TO 5	NEN - MAXIMUM NUMBER OF NODES CONNECTED TO ANY	MAN185C
C	COL 6 TO 10	ELEMENT (1 TO 20).	MAN186C
C		NEXTRA - INCREASES ELEMENT MATRIX SIZE FROM	MAN187C
C		NDF*NEN TO NDF*NEN + NEXTRA	MAN188C
C	COL 11 TO 15	IREC - COMPUTE GENERALIZED FORCE CHECK IF	MAN189C
C		NONZERO (FOR TIME INVARIANT ANALYSIS ONLY)	MAN190C
C	COL 16 TO 20	MBAN - MAXIMUM EXPECTED BANDWIDTH, DEFAULT IS	MAN191C
C		SET TO 100, USED AS AN ERROR CHECK TO PREVENT	MAN192C
C		RUNNING WITH AN OBVIOUS ERROR.	MAN193C
C	COL 21 TO 25	IBUF - BUFFER SIZE FOR STORAGE OF HISTORY	MAN194C
C		EFFECTS IN TIME DEPENDENT ANALYSIS, DEFAULT IS	MAN195C
C		IBUF = 15401/20	MAN196C
C	COL 26 TO 30	NC1 - USER INTEGER CONSTANT	MAN197C
C	COL 31 TO 40	CON1 - USER DEFINED CONSTANT	MAN198C
C	COL 41 TO 50	CON2 - USER DEFINED CONSTANT	MAN199C
C	COL 51 TO 60	CON3(1), 1-COORD. MULTIPLIER, DEFAULT = 1.0	MAN200C
C	COL 61 TO 70	CON3(2), 2-COORD. MULTIPLIER, DEFAULT = 1.0	MAN201C
C	COL 71 TO 80	CON3(3), 3-COORD. MULTIPLIER, DEFAULT = 1.0	MAN202C
C	2.1) *REMARK* USER COMMENTS ON OUTPUT. (6X,12A6)		MAN203C
C			MAN204C
C	SUBSEQUENT CARDS		MAN205C
C			MAN206C
C	COL 7 TO 12	MUST CONTAIN REMARK	MAN207C
C			MAN208C
C			MAN209C

C	COL 13 TO 78	STATEMENTS TO BE OUTPUT + USE AS MANY REMARK	MAN210C
C		CARDS AS DESIRED. INSERT BEFORE ANY TYPE CARD.	MAN211C
C	2.2) TITLE CHANGE ON OUTPUT (6X+12A6)		MAN212C
C			MAN213C
C			MAN214C
C	COL 7 TO 12	MUST CONTAIN TITLE	MAN215C
C	COL 13 TO 78	NEW TITLE DESCRIPTOR	MAN216C
C	2.3) EXECUTION TERMINATION (6X+A4)		MAN217C
C			MAN218C
C			MAN219C
C	COL 7 TO 10	MUST CONTAIN STOP, INSERT AFTER LAST PROBLEM.	MAN220C
C			MAN221C
C			MAN222C
C	3.) MATERIAL CHARACTERIZATION (15+1X+12A6)		MAN223C
C			MAN224C
C	COL 1 TO 5	NUMMAT = NUMBER OF DIFFERENT MATERIAL CHARACTERIZATIONS TO FOLLOW.	MAN225C
C	COL 7 TO 12	MUST CONTAIN WORD MATERI	MAN226C
C			MAN227C
C			MAN228C
C		THE FOLLOWING CARDS ARE SUPPLIED FOR EACH MATERIAL TO BE CHARACTERIZED (MUST BE EXACTLY NUMMAT SETS OF CARDS)	MAN229C
C			MAN230C
C			MAN231C
C	CARD 1.) ELEMENT SELECTOR CARD (15+1X+A5+11A6)		MAN232C
C			MAN233C
C	COL 1 TO 5	MATERIAL NUMBER (1 TO NUMMAT)	MAN234C
C	COL 7 TO 11	ELMNN = WHERE NN IS NUMBER OF ELEMENT CLASS (01 TO 30) TO WHICH THE CHARACTERIZATION BELONGS.	MAN235C
C	COL 12 TO 77	ALPHANUMERIC INFORMATION TO BE OUTPUT.	MAN236C
C			MAN237C
C			MAN238C
C		CARD 2.), ETC. ** USER DEFINED FOR EACH ELEMENT TYPE PROVIDED.	MAN239C
C			MAN240C
C	4.) NODAL CARDS (15+1X+A6)		MAN241C
C			MAN242C
C	COL 1 TO 5	NUMNP = NUMBER OF NODAL POINTS	MAN243C
C	COL 7 TO 12	MUST CONTAIN NODAL	MAN244C
C			MAN245C
C		SUBSEQUENT CARDS LAST NODAL CARD MUST NOT BE GENERATED.	MAN246C
C		(15+115+3F10.0)	MAN247C
C			MAN248C
C	COL 1 TO 5	NODE NUMBER	MAN249C
C	COL 15	1 IF 1 DISPLACEMENT IS SPECIFIED	MAN250C
C	COL 16	1 IF 2 DISPLACEMENT IS SPECIFIED	MAN251C
C	COL 17	1 IF 3 DISPLACEMENT IS SPECIFIED	MAN252C
C	COL 18	1 IF 4 DISPLACEMENT IS SPECIFIED	MAN253C
C	COL 19	1 IF 5 DISPLACEMENT IS SPECIFIED	MAN254C
C	COL 20	1 IF 6 DISPLACEMENT IS SPECIFIED	MAN255C
C	COL 21 TO 30	1 COORDINATE VALUE	MAN256C
C	COL 31 TO 40	2 COORDINATE VALUE * AS REQUIRED	MAN257C
C	COL 41 TO 50	3 COORDINATE VALUE	MAN258C
C			MAN259C
C		NODAL CARDS MUST BE IN ORDER. MISSING NODES ARE INTERPOLATED LINEARLY FROM INPUT NODES. IF SUCCEEDING CARDS HAVE IDENTICAL BOUNDARY CODES, THIS BOUNDARY CODE WILL BE ASSIGNED TO THE INTERVENING NODES. IN ALL OTHER CASES THE BOUNDARY CODE IS SET TO ZERO.	MAN260C
C		* TERMINATE ON NODE NUMNP OR A BLANK CARD.	MAN261C
C			MAN262C
C			MAN263C
C			MAN264C
C			MAN265C
C	4.1) NON SEQUENTIAL NODAL GENERATOR OPTION. (15+1X+A6)		MAN266C
C			MAN267C
C	COL 1 TO 5	NUMBER OF NODAL POINTS	MAN268C
C	COL 7 TO 12	MUST CONTAIN GENERA	MAN269C
C			MAN270C
C		SUBSEQUENT CARDS (215+110+3F10.0)	MAN271C
C			MAN272C
C	COL 1 TO 5	NODE=NUMBER	MAN273C
C	COL 6 TO 10	NODE=NUMBER-INCREMENT WHICH WILL BE SUCCESSIVELY ADDED TO NODE=NUMBER UNTIL SUM IS GREATER THAN NODE=NUMBER ON FOLLOWING CARD (ALGEBRAIC).	MAN274C
C		BOUNDARY CODE. SAME AS INPUT FOR NODAL.	MAN275C
C	COL 15 TO 20	IF SUCCEEDING CARDS HAVE IDENTICAL BOUNDARY CODES, THIS BOUNDARY CODE WILL BE ASSIGNED TO THE INTERVENING NODES. IN ALL OTHER CASES THE BOUNDARY CODE IS SET TO ZERO.	MAN276C
C			MAN277C
C			MAN278C
C			MAN279C
C			MAN280C
C			MAN281C

C	COL 21 TO 30	1 COORDINATE VALUE *	MAN282C
C	COL 31 TO 40	2 COORDINATE VALUE * AS REQUIRED *	MAN283C
C	COL 41 TO 50	3 COORDINATE VALUE *	MAN284C
C			MAN285C
C	* TERMINATE WITH BLANK CARD *		MAN286C
C	4.2) BOUNDARY CODE PATCH-UP OPTION. (6X,46)		MAN287C
C	COL 6 TO 12	MUST CONTAIN BOUNDARY	MAN288C
C	SUBSEQUENT CARDS. (R15)		MAN289C
C	COL 1 TO 5	N, NODE NUMBER TO HAVE REDEFINED BOUNDARY CODE.	MAN290C
C	COL 6 TO 10	NX, GENERATOR INCREMENT TO BE ADDED ALGEBRAICALLY TO N, UNTIL SUM EXCEEDS (MAX OR MIN) THE N OF THE FOLLOWING CARD.	MAN291C
C	COL 11 TO 15	IBC(1), (I=1,2,...,NDF) CODE FOR SPECIFYING FORCE	MAN292C
C	COL 16 TO 20	OR DISPLACEMENT BOUNDARY CONDITIONS.	MAN293C
C	COL ...		MAN294C
C		IBC(1) .EQ. 0, FORCE SPECIFIED.	MAN295C
C		IBC(1) .GT. 0, DISPLACEMENT SPECIFIED, NO	MAN296C
C		INTERVENING GENERATION.	MAN297C
C		IBC(1) .LT. 0, DISPLACEMENT SPECIFIED.	MAN298C
C		GENERATE BETWEEN MISSING NODES IN ALGEBRAIC	MAN299C
C		INCREMENTS OF NX.	MAN300C
C	* TERMINATE WITH A BLANK CARD. *		MAN301C
C	4.3) POLAR OR CYLINDRICAL COORDINATE CONVERSION TO CARTESIAN		MAN302C
C	COORDINATES (6X,46)		MAN303C
C	COL 7 TO 12	MUST CONTAIN POLAR (LEFT JUSTIFIED)	MAN304C
C	CARD 1. (15,5X,2F10,0)		MAN305C
C	COL 11 TO 15	N3, INCREMENT ADDED (ALGEBRAICALLY), N1 TO N2	MAN306C
C	COL 1 TO 5	N1, FIRST NODE TO BE CONVERTED	MAN307C
C	COL 6 TO 10	N2, LAST NODE TO BE CONVERTED	MAN308C
C	COL 21 TO 30	X0, ORIGIN OF POLAR X-COORDINATE	MAN309C
C	COL 31 TO 40	Y0, ORIGIN OF POLAR Y-COORDINATE	MAN310C
C	5.) ELEMENT CARDS (15,1X,46)		MAN311C
C	COL 1 TO 5	NUMEL - NUMBER OF ELEMENTS	MAN312C
C	COL 7 TO 12	MUST CONTAIN ELEMENT	MAN313C
C	SUBSEQUENT CARDS (415,2013/2014)		MAN314C
C	CARD 1.		MAN315C
C	COL 1 TO 5	ELEMENT NUMBER	MAN316C
C	COL 6 TO 10	MATERIAL NUMBER	MAN317C
C	COL 11 TO 15	NUMBER OF SUBSEQUENT ELEMENTS USING SAME	MAN318C
C		STIFFNESS MATRIX * SAVES RECOMPUTATION OF	MAN319C
C		SIMILAR MATRICES. ELEMENT MUST ALSO HAVE	MAN320C
C		SAME ELEMENT FORCE VECTOR * IF THESE ARE	MAN321C
C		IN THE STIFFNESS SUBROUTINE *	MAN322C
C	COL 16 TO 20	PRINT ELEMENT MATRIX IF NONZERO.	MAN323C
C	COL 21 TO 23	IXD(1) ELEMENT INCREMENT ARRAY ON NODE 1.	MAN324C
C	COL 24 TO 26	IXD(2) * IF NOT INPUT IS SET AUTOMATICALLY	MAN325C
C		UP TO	MAN326C
C	COL 78 TO 80	IXD(20) FOR SERENDIPITY ELEMENTS * SEE REPORT	MAN327C
C	CARD 2.		MAN328C
C	COL 1 TO 4	NODE 1	MAN329C
C	COL 5 TO 8	NODE 2	MAN330C
C	COL 9 TO 12	NODE 3	MAN331C
C		IX(INEL,1) ARRAY	MAN332C
C	COL 77 TO 80	CONTINUE IN I4 FORMAT TO A MAXIMUM	MAN333C
C		NODE 2)	MAN334C
C	NODES MUST BE NUMBERED AROUND ELEMENT BY RIGHT HAND SCREW RULE.		MAN335C
C			MAN336C
C			MAN337C
C			MAN338C
C			MAN339C
C			MAN340C
C			MAN341C
C			MAN342C
C			MAN343C
C			MAN344C
C			MAN345C
C			MAN346C
C			MAN347C
C			MAN348C
C			MAN349C
C			MAN350C
C			MAN351C
C			MAN352C
C			MAN353C

C ELEMENT CARDS MUST BE IN ORDER. MISSING ELEMENTS ARE GENERATED BY INCREMENTING NODES. MAN354C  
 C LAST ELEMENT CARD MUST NOT BE GENERATED. MAN355C  
 C \* TERMINATE ON ELEMENT NUMEL OR A BLANK CARD \* MAN356C  
 C MAN357C  
 C MAN358C  
 C 5.1) BLOCK GENERATOR: GENERATES ALL MESH DATA. (6X,A6) MAN359C  
 C COL 1 TO 5 NUMBER OF NODAL POINTS TO BE GENERATED. MAN360C  
 C COL 7 TO 12 MUST CONTAIN BLOCK MAN361C  
 C MAN362C  
 C SUBSEQUENT CARDS (1015/615/(10X,3F10.0)) MAN363C  
 C MAN364C  
 C MAN365C  
 C CARD 1: MAN366C  
 C COL 1 TO 5 NN, NUMBER OF POINTS REQUIRED TO DEFINE MAN367C  
 C BOUNDARY OF REGION. FOR 2-DIM., NN=4 OR 8. MAN368C  
 C FOR 3-DIM., NN= 8 OR 20. MAN369C  
 C COL 6 TO 10 NUMBER OF ELEMENTS IN X-DIRECTION. MAN370C  
 C COL 11 TO 15 NUMBER OF ELEMENTS IN Y-DIRECTION. MAN371C  
 C COL 16 TO 20 NUMBER OF ELEMENTS IN Z-DIRECTION. MAN372C  
 C COL 21 TO 25 INITIAL NODE NUMBER, DEFAULT = 1. MAN373C  
 C COL 26 TO 30 INITIAL ELEMENT NUMBER, DEFAULT = 1. MAN374C  
 C COL 31 TO 35 MATERIAL NUMBER OVER REGION, DEFAULT = 1. MAN375C  
 C COL 36 TO 40 BOUNDARY CODE SKIP. A NON-ZERO ENTRY WILL OMIT MAN376C  
 C SETTING ALL INTERIOR BOUNDARY CODES TO ZERO. MAN377C  
 C COL 41 TO 45 IREUSE - REUSE ELEMENT STIFFNESS OPTION - USES MAN378C  
 C EACH ELEMENT STIFFNESS IREUSE TIMES BEFORE MAN379C  
 C GENERATING A NEW ELEMENT STIFFNESS MATRIX. MAN380C  
 C COL 46 TO 50 ELEMENT STIFFNESS-PRINT. A NON-ZERO ENTRY WILL MAN381C  
 C CAUSE PRINT-OUT OF FIRST ELEMENT. MAN382C  
 C COL 51 TO 55 INSH - IF NONZERO SUPPRESSES PRINT OF NODES. MAN383C  
 C COL 56 TO 60 IELM - IF NONZERO SUPPRESSES PRINT OF ELEMENTS. MAN384C  
 C MAN385C  
 C CARD 2: (BOUNDARY CODE AS DEFINED IN NODAL CARD.) MAN386C  
 C MAN387C  
 C COL 1 TO 10 BOUNDARY CODE OVER FACE +X. MAN388C  
 C COL 11 TO 20 BOUNDARY CODE OVER FACE +Y. MAN389C  
 C COL 21 TO 30 BOUNDARY CODE OVER FACE +Z. MAN390C  
 C COL 31 TO 40 BOUNDARY CODE OVER FACE -X. MAN391C  
 C COL 41 TO 50 BOUNDARY CODE OVER FACE -Y. MAN392C  
 C COL 51 TO 60 BOUNDARY CODE OVER FACE -Z. MAN393C  
 C MAN394C  
 C MAN395C  
 C CARD 3: (REPEAT NN TIMES.) MAN396C  
 C COL 11 TO 20 1-COORDINATE OF BOUNDARY-DEFINING-POINT. MAN397C  
 C COL 21 TO 30 2-COORDINATE OF BOUNDARY-DEFINING-POINT. MAN398C  
 C COL 31 TO 40 3-COORDINATE OF BOUNDARY-DEFINING-POINT. MAN399C  
 C MAN400C  
 C NOTE: BLOCK GENERATES ONLY 4 PT. QUADRILATERALS OR 8 PT. BRICKS. MAN401C  
 C INPUT OF CARDS 3.) FOLLOW ORDER RULES FOR ELEMENT INPUT (SEE 5.) MAN402C  
 C R-S-T ARE LOCAL COORDINATES, I.E. (-1 .LE. R+S+T .LE. 1.) MAN403C  
 C WHERE R IS DIRECTED FROM NODE 1 TO 2, S IS IN PLANE OF FIRST MAN404C  
 C THREE NODES, AND T IS NORMAL TO R-S PLANE. MAN405C  
 C MAN406C  
 C MAN407C  
 C 6.) VECTOR CARDS, I.E. USER DEFINED INPUT (15,1X,A6) MAN408C  
 C COL 1 TO 5 NVEC, NUMBER OF DIFFERENT VECTORS (7 MAX) MAN409C  
 C COL 7 TO 12 MUST CONTAIN VECTOR MAN410C  
 C MAN411C  
 C SUBSEQUENT CARDS MAN412C  
 C MAN413C  
 C MAN414C  
 C CARD 1: (215) MAN415C  
 C COL 1 TO 5 NSIZV, VECTOR LENGTH, COMMON TO ALL NVEC VECTORS MAN416C  
 C COL 6 TO 10 IPICK, CODED PARAMETER. MAN417C  
 C MAN418C  
 C IPICK = 0, VECTORS ASSOCIATED WITH NODES MAN419C  
 C IPICK = 1, VECTORS ASSOCIATED WITH DEG. FREEDOM MAN420C  
 C IPICK = 2, VECTORS ASSOCIATED WITH ELEMENTS MAN421C  
 C MAN422C  
 C MAN423C  
 C CARD 2: (6X, 2A6) REPEAT NVEC TIMES MAN424C  
 C MAN425C



C COL 7 TO 18 DESCRIPTIVE TITLE FOR VECTOR MAN426C  
 C CARD 3. (215,7F10,0) MAN427C  
 C MAN428C  
 C COL 1 TO 5 POSITION NUMBER OF VECTOR ELEMENT, 1 TO NSIZV MAN429C  
 C COL 11 TO 20 VECTOR ELEMENT VALUE OF VECTOR 1 MAN430C  
 C COL 6 TO 10 GENERATOR INCREMENT MAN431C  
 C COL 21 TO 30 VECTOR ELEMENT VALUE OF VECTOR 2 MAN432C  
 C COL ..... AS REQUIRED FOR NVEC VECTORS MAN433C  
 C MAN434C  
 C LINEAR INTERPOLATION IS PERFORMED ON ALL VECTORS BETWEEN MAN435C  
 C NON-CONSECUTIVE POSITION NUMBERS SPECIFIED IN COL 1 TO 5 IF MAN436C  
 C INCREMENT IS NONZERO. MAN437C  
 C IF DESCRIPTIVE TITLES OF ALL VECTORS ARE BLANK CARDS, PRINTING MAN438C  
 C OF THE VECTOR VALUES IS SUPPRESSED. MAN439C  
 C MAN440C  
 C \* TERMINATE ON BLANK CARD \* MAN441C  
 C MAN442C  
 C MAN443C  
 C 6.1) INITIAL CONDITIONS FOR TIME DEPENDENT ANALYSIS. MAN444C  
 C COL 1 TO 5 NICO, NUMBER OF INITIAL CONDITION VECTORS MAN445C  
 C COL 7 TO 12 MUST CONTAIN INITIA MAN446C  
 C MAN447C  
 C SUBSEQUENT CARDS MAN448C  
 C CARD 1. (6X,2A6) REPEAT NICO TIMES MAN449C  
 C MAN450C  
 C COL 7 TO 18 DESCRIPTIVE TITLE FOR INITIAL CONDITIONS MAN451C  
 C CARD 2. (215,7F10,0) MAN452C  
 C COL 1 TO 5 POSITION NUMBER, AS IN VECTOR CARDS FOR IPICK=1 MAN453C  
 C COL 6 TO 10 GENERATOR INCREMENT MAN454C  
 C COL 11 TO 20 INITIAL CONDITION 1 MAN455C  
 C COL 21 TO 30 INITIAL CONDITION 2 MAN456C  
 C COL ..... AS REQUIRED FOR NICO INITIAL CONDITIONS MAN457C  
 C MAN458C  
 C INTERPOLATION BETWEEN INPUT VALUES AS DESCRIBED IN VECTOR INPUT. MAN459C  
 C \*\*\* NOTE \*\*\* IF MISSING THE INITIAL CONDITIONS ARE SET ZERO MAN460C  
 C MAN461C  
 C 6.2) TWO AND THREE DIMENSIONAL PLOTTING PORTHOLE. (TAPE30) MAN462C  
 C COL 1 TO 5 KSTEP, NUMBER OF TIME STEPS TO BE PLOTTED MAN463C  
 C COL 7 TO 12 MUST CONTAIN PLOT MAN464C  
 C THIS PLOT CONTROL CARD WRITES ALL INFORMATION TO TAPE30  
 C FOR USE IN PROGRAM PLOT2D) IF NDIM = 2. IF NDIM = 3, A THREE  
 C DIMENSIONAL PORTHOLE RECORD IS WRITTEN IN SUBROUTINE PLOT2D.  
 C  
 C 7.1) FORCE CARDS (15,1X,A6) MAN465C  
 C COL 1 TO 5 LAST NODE TO WHICH A FORCE IS TO BE SPECIFIED MAN466C  
 C COL 7 TO 12 MUST CONTAIN FORCE MAN467C  
 C SUBSEQUENT CARDS (15,5X,6F10,0) MAN468C  
 C MAN469C  
 C THE FOLLOWING VALUES ARE EACH INTERPRETTED AS FORCES IF THE  
 C CORRESPONDING BOUNDARY CODE IS A 0 \*ZERO\* AND AS A DISPLACEMENT  
 C IF THE CORRESPONDING BOUNDARY CODE IS 1 \*ONE\*. MAN470C  
 C MAN471C  
 C COL 1 TO 5 NODE TO WHICH FORCE OR DISPLACEMENT IS APPLIED MAN472C  
 C COL 11 TO 20 VALUE OF 1 FORCE/DISPLACEMENT MAN473C  
 C COL 21 TO 30 VALUE OF 2 FORCE/DISPLACEMENT \* AS \* MAN474C  
 C COL 31 TO 40 VALUE OF 3 FORCE/DISPLACEMENT \* REQUIRED \* MAN475C  
 C COL 41 TO 50 VALUE OF 4 FORCE/DISPLACEMENT MAN476C  
 C COL 51 TO 60 VALUE OF 5 FORCE/DISPLACEMENT MAN477C  
 C COL 61 TO 70 VALUE OF 6 FORCE/DISPLACEMENT MAN478C  
 C MAN479C  
 C 7.1) SURFACE LOAD CARDS (15,1X,A6) MAN480C  
 C MAN481C  
 C MAN482C  
 C MAN483C  
 C MAN484C  
 C MAN485C  
 C MAN486C  
 C MAN487C



C COL 1 TO 5 NUMBER OF LOADED FACE CARDS MAN488C  
 C COL 7 TO 12 MUST CONTAIN BLOADS MAN489C  
 C  
 C CARD 1. (15,1X,A5,14,8(5,813)) MAN490C  
 C  
 C COL 1 TO 5 DIMENSION OF LOADING SURFACE, (1 OR 2). MAN491C  
 C COL 7 TO 11 SLD(NN), ALPHA-NUMERIC NAME OF SURFACE LOADING MAN492C  
 C SURROUTINE (NN IS BETWEEN 1 AND 5) MAN493C  
 C COL 12 TO 15 NMT, NUMBER OF ADDITIONAL ELEMENT LOAD MAN494C  
 C SURFACES TO BE GENERATED FROM CURRENT MODEL. MAN495C  
 C COL 16 TO 20 IPRES(N), NODE NUMBERS DEFINING LOADING SURFACE MAN496C  
 C OF CURRENT ELEMENT. MAN497C  
 C COL ... (IDENTIFY FROM 2 TO 8 AS REQUIRED) MAN498C  
 C COL 51 TO 55 INC(N), INCREMENT VALUE ADDED TO IPRES(N) TO MAN499C  
 C COL 56 TO 58 IDENTIFY NODE NUMBERS OF A GENERATED SEQUENCE. MAN500C  
 C COL 59 TO 61 (IDENTIFY FROM 2 TO 8 AS REQUIRED) MAN501C  
 C COL ... MAN502C  
 C  
 C CARD 2. (8F10.0) MAN503C  
 C  
 C COL 1 TO 80 LOAD AT NODES GIVEN ON PREVIOUS CARD. MAN504C  
 C MUST CORRESPOND IN SEQUENCE TO THE NODE NUMBERS. MAN505C  
 C  
 C 7.2) ELEMENT LOAD CARDS (15,1X,A6 ) MAN506C  
 C  
 C COL 1 TO 5 NLD, NUMBER OF ELEMENT LOAD CARDS. MAN507C  
 C COL 7 TO 11 MUST CONTAIN ELLOADS MAN508C  
 C  
 C SUBSEQUENT CARDS (15,1X,A5,14,15,6F10.0) MAN509C  
 C  
 C COL 1 TO 5 IEL, INITIAL ELEMENT OF A GENERATED SEQUENCE. MAN510C  
 C COL 7 TO 11 ELM(NN), ALPHA-NUMERIC NAME OF ELEMENT MAN511C  
 C SURROUTINE WHERE ELEMENT LOADS ARE COMPUTED. MAN512C  
 C USED AS CHECK TO INSURE IEL, ET, Z ARE PROPER MAN513C  
 C ELEMENTS. MAN514C  
 C COL 12 TO 15 INC, INCREMENT NUMBER IN A GENERATED SEQUENCE. MAN515C  
 C (DEFAULT = 1). MAN516C  
 C COL 16 TO 20 JEL, TERMINAL ELEMENT NUMBER IN A GENERATED MAN517C  
 C SEQUENCE, IF JEL = 0, ONLY IEL IS COUNTED. MAN518C  
 C COL 21 TO 80 USER DEFINED VALUES FOR DETERMINING BODY LOADS MAN519C  
 C IN THE ISW=5 PORTION OF ELM(NN). MAN520C  
 C  
 C NOTE: USER MUST PROVIDE COMPUTATION OF LOADS IN ELM(NN). MAN521C  
 C OR IS TRANSFERRED TO SURROUTINE ELM(NN) IN THE U VECTOR. MAN522C  
 C WHEN ISW = 5, ONLY. MAN523C  
 C  
 C 7.3) PROPORTIONAL LOADS FOR TIME DEPENDENT ANALYSIS MAN524C  
 C  
 C TRANSFER TO THIS OPTION OCCURS ONLY FOR TIME ANALYSES. MAN525C  
 C  
 C ONE CARD FOR EACH PROPORTIONAL LOAD REQUIRED MAN526C  
 C  
 C COL 1 TO 5 PROPORTIONAL LOAD TYPE, 1, 2 OR 3 MAN527C  
 C COL 6 TO 10 K, TABLE CONSTANT MAN528C  
 C COL 11 TO 20 TMIN, SMALLEST TIME LOADING IS VALID MAN529C  
 C COL 21 TO 30 TMAX, LARGEST TIME LOADING IS VALID MAN530C  
 C COL 31 TO 40 A0 MAN531C  
 C COL 41 TO 50 A1 MAN532C  
 C COL 51 TO 60 A2 MAN533C  
 C COL 61 TO 70 A3 MAN534C  
 C COL 71 TO 80 A4 MAN535C  
 C  
 C LOAD TYPE 1. 1 = TIME MAN536C  
 C  
 C  $PROP = A_0 + A_1 * T + A_2 * T^2 + A_3 * T^3 + A_4 * T^4$  MAN537C  
 C  
 C LOAD TYPE 2. MAN538C  
 C  
 C  $PROP = A_0 * (SIN(A_1 * T)) * K + A_2 * (COS(A_3 * T)) * K + A_5$  MAN539C  
 C  
 C MAN540C  
 C MAN541C  
 C MAN542C  
 C MAN543C  
 C MAN544C  
 C MAN545C  
 C MAN546C  
 C MAN547C  
 C MAN548C  
 C MAN549C  
 C MAN550C  
 C MAN551C  
 C MAN552C  
 C MAN553C  
 C MAN554C  
 C MAN555C  
 C MAN556C  
 C MAN557C  
 C MAN558C  
 C MAN559C

C LOAD TYPE 3. MAN560C  
 C PROP = USER DEFINED FUNCTION FROM SUBROUTINE EXPRLO(PROP,T) MAN561C  
 C MAN562C  
 C MAN563C  
 C \*\*NOTE\*\* PROPORTIONAL LOADS CAN BE ACCUMULATED FROM DIFFERENT MAN564C  
 C TYPES AT THE SAME TIME. MAN565C  
 C MAN566C  
 C 8.) INITIATION OF TIME INDEPENDENT SOLUTION (IS,IX,A6) MAN567C  
 C COL 1 TO 4 IOUT, OUTPUT CONTROL CODE. MAN568C  
 C IOUT = 0, ALL STRESSES AND DISP. PRINTED MAN569C  
 C IOUT = 1, 0, SELECTED PRINTOUT, MORE DATA INPUT MAN570C  
 C SEE SECTION 9 FOR DATA PREPARATION. MAN571C  
 C COL 7 TO 12 MUST CONTAIN SOLVE \*INDICATES ALL DATA INPUT\* MAN572C  
 C COMPLETE FORMULATION AND SOLUTION OF EQUATIONS. MAN573C  
 C COL 7 TO 12 MUST CONTAIN RESOLVE TO OBTAIN SUBSEQUENT MAN574C  
 C SOLUTIONS WHERE BOUNDARY CODES DO NOT CHANGE  
 C AND ALL PRESCRIBED DISPLACEMENTS ARE ZERO. MAN575C  
 C MAN576C  
 C 8.1) INITIATION OF DYNAMIC SOLUTION BY EXPLICIT INTEGRATION. MAN577C  
 C COL 1 TO 4 IOUT, OUTPUT CONTROL FOR DISPLACEMENT AND MAN578C  
 C STRESS PRINTOUT. SEE SECT. 9 FOR DATA INPUT.  
 C IF IOUT = 0, THE SPATIAL CONTROL DATA MAN579C  
 C COMES AT THE END OF THE DYNAMIC SEGMENT. MAN580C  
 C COL 7 TO 12 MUST CONTAIN EXPLICIT MAN581C  
 C SUBSEQUENT CARDS (2IS\*2F10+0\*2IS) MAN582C  
 C COL 1 TO 5 NUMBER OF TIME STEPS MAN583C  
 C COL 6 TO 10 PRINT INTERVAL MAN584C  
 C COL 11 TO 20 TIME INCREMENT MAN585C  
 C COL 21 TO 30 NERMARK DELTA-DAMPING TERM (GAMMA = .5) MAN586C  
 C COL 31 TO 35 NUMBER OF TIME EVOLUTION STRESS PLOTS. MAN587C  
 C COL 36 TO 40 NPROP, NUMBER OF PROPORTIONAL LOADS TO BE INPUT MAN588C  
 C COL 41 TO 45 NFORC, LAST NODE ON WHICH A FORCE IS CHANGED  
 C DURING EACH TIME STEP. MAN589C  
 C COL 46 TO 50 KKK, STABILITY CHECK OVERRIDE \*\* CAUTION USE  
 C ONLY WHEN A BETTER ESTIMATE OF THE STABLE TIME  
 C STEP IS AVAILABLE THAN CAN BE PERFORMED BY CODE  
 C KKK ZERO, USES INTERNAL STABILITY CHECK. MAN590C  
 C KKK NONZERO, DISREGARDS STABILITY CHECK. MAN591C  
 C SUBSEQUENT CARDS (3IS) ONE FOR EACH STRESS PLOT. MAN592C  
 C COL 1 TO 5 ELEMENT NUMBER CONTAINING STRESS TO BE PLOTTED. MAN593C  
 C COL 6 TO 10 LOCAL COORDINATE POINT CODE, 1 TO 7, AS  
 C PATTERNED AFTER, COL 11 TO COL 17, IN SOLVE. MAN594C  
 C COL 11 TO 15 STRESS COMPONENT CODE, 1 TO 6 FOR SIGMA(I,J),  
 C I.E., SIGMA(1,1)=1, SIGMA(1,2)=2, SIGMA(1,3)=3,  
 C SIGMA(2,2)=4, SIGMA(2,3)=5, SIGMA(3,3)=6. MAN595C  
 C MAN596C  
 C IF (NPROP.NE.0) READ PROPORTIONAL LOAD CARDS, SEE SECT. 7.3 MAN597C  
 C MAN598C  
 C IF (NFORC.NE.0) READ FORCE CARDS AT EACH TIME STEP. IF OUTPUT IS  
 C LIMITED BY IOUT NONZERO, THE FIRST FORCE CARD SET PRECEDES  
 C OUTPUT CARDS AND THE REMAINDER FOLLOW THE OUTPUT CARDS NO BLANK  
 C CARDS MAY BE USED BETWEEN SETS OF CARDS OTHER THAN THE USUAL  
 C BLANK TERMINATOR CARD FOR FORCE INPUT CARDS. MAN599C  
 C MAN600C  
 C IF (IOUT.NE.0) DATA FOR SPATIAL PRINTOUT CONTROL, SEE SECT.9. MAN601C  
 C MAN602C  
 C SPECIAL COMMENTS FOR DYNAMIC OPTION MAN603C  
 C (1) ONLY COLUMNS 1 TO 60 ARE AVAILABLE FOR PAGE HEADING. MAN604C  
 C (2) MAXIMUM ADVANTAGE OF ELEMENT REUSE OPTION SHOULD BE TAKEN. MAN605C  
 C (3) INITIAL CONDITIONS FOR DISPLACEMENT AND VELOCITY VECTORS,  
 C AS WELL AS STORAGE FOR ACCELERATION VECTOR, MAY BE MADE  
 C THROUGH INPUT OF AN INITIAL CONDITION CARD SET, WITHOUT  
 C MAN606C  
 C MAN607C  
 C MAN608C  
 C MAN609C  
 C MAN610C  
 C MAN611C  
 C MAN612C  
 C MAN613C  
 C MAN614C  
 C MAN615C  
 C MAN616C  
 C MAN617C  
 C MAN618C  
 C MAN619C  
 C MAN620C  
 C MAN621C  
 C MAN622C  
 C MAN623C  
 C MAN624C  
 C MAN625C  
 C MAN626C  
 C MAN627C  
 C MAN628C  
 C MAN629C  
 C MAN630C  
 C MAN631C

C SPECIFIED INITIAL CONDITIONS THEY ARE AUTOMATICALLY SET ZERO. MANA32C  
 C (4) SPATIAL LOADING IS INPUT THROUGH FORCE OR BOUNDARY MANA33C  
 C PRESSURE CARDS. ALL LOADS VARY PROPORTIONALLY WITH TIME MANA34C  
 C (5) EXTREME CAUTION ON ORDER OF DATA CARDS MUST BE OBSERVED. NO MANA35C  
 C EXTRA CARDS ARE PERMITTED AND STRICT COUNTS ARE OBSERVED MANA36C  
 C EXCEPT FOR THE NUMBER OF FORCE CARDS USED IN EACH TIME STEP. MANA37C  
 C MANA38C  
 C 8.2) INITIATION OF IMPLICIT TIME INTEGRATIONS (15,1X,A6) MANA39C  
 C COL 1 TO 5 NSEQ, NUMBER OF TIME SEQUENCES MANA40C  
 C COL 7 TO 12 MUST CONTAIN VISCOE FOR LINEAR VISCOELASTIC MANA41C  
 C QUASI-STATIC PROBLEMS (ONE INITIAL CONDITION MANA42C  
 C ONLY MUST BE USED) MANA43C  
 C COL 7 TO 12 MUST CONTAIN IMPLIC FOR DYNAMIC IMPLICIT MANA44C  
 C INTEGRATIONS (THREE INITIAL CONDITIONS ARE MANA45C  
 C REQUIRED, MORE CAN BE SPECIFIED WITHOUT ERROR) MANA46C  
 C MANA47C  
 C SUBSEQUENT CARDS, ONE SET FOR EACH TIME SEQUENCE MANA48C  
 C MANA49C  
 C CARD 1. (F10.0,815.2F10.0) MANA50C  
 C COL 1 TO 10 DT, TIME INCREMENT (NONZERO FOR IMPLIC) MANA51C  
 C COL 11 TO 15 NIS, NUMBER OF TIME STEPS IN SEQUENCE MANA52C  
 C COL 16 TO 20 INT, PRINT INTERVAL (DEFAULT 1) MANA53C  
 C COL 21 TO 25 NNI, FIRST NODE PRINTED MANA54C  
 C COL 26 TO 30 NNE, LAST NODE PRINTED MANA55C  
 C COL 31 TO 35 NEI, FIRST ELEMENT STRESS TO BE PRINTED MANA56C  
 C COL 36 TO 40 NEF, LAST ELEMENT STRESS TO BE PRINTED MANA57C  
 C COL 41 TO 45 NPROP, NUMBER OF PROPORTIONAL LOADS IN SEQUENCE MANA58C  
 C COL 46 TO 50 NFORC, LAST NODE FOR GENERALIZED FORCES TO BE MANA59C  
 C INPUT FOR EACH TIME IN SEQUENCE (SEE SECT.7. MANA60C  
 C FOR DATA PREPARATION FORMATS) MANA61C  
 C COL 51 TO 60 BETA, NEWMARK INTEGRATION PARAMETER (IMPLIC) MANA62C  
 C COL 61 TO 70 DEL = GAMMA = 0.5, NEWMARK INTEGRATION MANA63C  
 C PARAMETER (IMPLIC) MANA64C  
 C MANA65C  
 C CARD 2. (215.7F10.0) MANA66C  
 C ONE FOR EACH NPROP. SEE SECT.7.3 FOR DATA PREPARATION MANA67C  
 C MANA68C  
 C SUBSEQUENT CARDS FOR EACH TIME STEP IN THE SEQUENCE MANA69C  
 C MANA70C  
 C FORCE CARDS. SEE SECTION 7. FOR DATA PREPARATION FORMATS. MANA71C  
 C MANA72C  
 C MANA73C  
 C MANA74C  
 C MANA75C  
 C MANA76C  
 C \*\*\* SPECIAL USE OF SECTION 8.2 FOR STATIC RUNS WITH REPEATED  
 C LOAD CASES ...  
 C NSEQ = 1  
 C CONTROL WORD = VISCOE  
 C DT = -1. (MINUS)  
 C NIS = NUMBER OF LOAD CASES PLUS 1  
 C INT = 1  
 C NPROP = 0  
 C NNI, NNE, NEI, NEF AS DEFINED ABOVE  
 C NFORC = HIGHEST NODE OF ALL LOAD CASES WITH NON-ZERO LOAD  
 C \*\*\* USE FORCE DATA INPUT, SECTION 7., FOR EACH LOAD CASE  
 C BACK TO BACK, WHERE LAST NODE = NFORC. NOTE LAST NODE  
 C MAY OR MAY NOT HAVE NON-ZERO FORCES. THE LAST LOAD CASE  
 C IS A DUMMY CASE.  
 C  
 C 9.) OUTPUT CONTROL FOR LIMITED PRINTS MANA77C  
 C MANA78C  
 C DISPLACEMENT OUTPUT CONTROL, IF IOUT .NE. 0, MANA79C  
 C MANA80C  
 C CARD 1. (15) MANA81C  
 C COL 1 TO 5 NUPDIS = NUMBER OF DISPLACEMENT PRINT CARDS MANA82C  
 C MANA83C  
 C MANA84C  
 C MANA85C

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C          SUBSEQUENT CARDS (215) SKIP IF NUMDIS = 0
C
C          COL 1 TO 5    NODAL NUMBER TO BE OUTPUT.
C          COL 6 TO 10   HIGHER NODE NUMBER OF A GENERATED SEQUENCE.
C                          IF ZERO FIRST NODE IS COUNTED.
C          COL 11 TO 15  INCREMENT TO GENERATOR, DEFAULT = 1
C                          *** REPEAT UNTIL NUMDIS CARDS HAVE BEEN READ
C
C          STRESS OUTPUT CONTROL, IF IOUT.NE. 0.
C
C          CARD 1. (15,5X,11)
C
C          COL 1 TO 5    NUMSTR - NUMBER OF STRESS OUTPUT CARDS
C          COL 11 TO 17  NSIG(7) - PRINT PATTERN WITHIN AN ELEMENT.
C                          LOCAL POINTS OF EACH ELEMENT CAN BE
C                          SUPPRESSED BY NON-ZERO ENTRIES AS FOLLOWS:
C                              E.G.
C          COL 11      SUPPRESS PRINT AT LOCAL POINT 1. ( 0, 0, 0)
C          COL 12      SUPPRESS PRINT AT LOCAL POINT 2. (-1, 0, 0)
C          COL 13      SUPPRESS PRINT AT LOCAL POINT 3. ( 1, 0, 0)
C          COL 14      SUPPRESS PRINT AT LOCAL POINT 4. ( 0,-1, 0)
C          COL 15      SUPPRESS PRINT AT LOCAL POINT 5. ( 0, 1, 0)
C          COL 16      SUPPRESS PRINT AT LOCAL POINT 6. ( 0, 0,-1)
C          COL 17      SUPPRESS PRINT AT LOCAL POINT 7. ( 0, 0, 1)
C
C          SUBSEQUENT CARDS (215) SKIP IF NUMSTR = 0
C
C          COL 1 TO 5    ELEMENT NUMBER TO BE PRINTED.
C          COL 6 TO 10   HIGHER ELEMENT NUMBER OF A GENERATED SEQUENCE.
C                          IF ZERO ONLY FIRST ELEMENT IS COUNTED.
C          COL 11 TO 15  INCREMENT TO GENERATOR, DEFAULT = 1
C                          *** REPEAT UNTIL NUMSTR CARDS HAVE BEEN READ
C
C          .....
C
C          RETURN
C          END
C          *DECK FELMT
C          SUBROUTINE FELMT(UT,ISZDT,MHAND)
C
C          *** MAIN ASSEMBLY PROGRAM AND STORAGE ALLOCATION FOR VARIABLES
C
C          REAL LABL
C          LOGICAL CHECK,FLAG,NPL
C          DIMENSION CONWD(24),FLAG(7),TITLE(12),TYPE(7),UT(15ZDT),RE(2)
C          DIMENSION NEWHUF(15)
C          COMMON /TAPES/ ITP5,ITP6
C          COMMON /FOURIER/ NF(5),XN,F1,F2,F3,F4
C          COMMON /GAUSS/ LIM,SGAUSS(5,5),WGAUSS(5,5)
C          COMMON /LABELS/ LABL(6),XHED(3),XH,FHED(6),FH,UHED(6),UH,RHED(6),RH
C          X ,AWORD1,AWORD2,AWORD3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)
C          COMMON /SHAPE/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)
C          COMMON /TTLES/ XTIL(3),UTTL(6)
C          COMMON /VALUES/ NC1,CUN1,CUN2,CUN3(3)
C          COMMON /DYNAMO/ TIME,NSIG(7),NI,NSIEP,DS,NUMPLT,NEDATA(20,3),NPR,NPL
C          COMMON /PLOTS/ IPLOT
C
C          CXXXXXX
C          PUT IN COMMON TO HOLD NODE RENUMBERING ARRAY
C          CXXXXXX
C          COMMON /MINBW/ NEW(1500)
C          DATA CONWD/6H MATEI,6H NODAL,6H POLAR,6H ELEMEN,6H SOLVE,6H RESOLV,
C          X 6H ROUNDA,6H FORCE,6H ELUADS,6H ELUADS,6H VECTOR,6H MESH,6H PLOT,
C          X 6H FOURIER,6H REMARK,6H INITIA,6H GENERA,6H ITILE,6H BLOCK,6H EXPLIC,
C          X 6H ADUUP,6H VISCOS,6H IMPLIC,6H NEWTON/
C          DATA RE/4H NOT,4H /,4H BLANK/6H /
C          DATA FLAG,NLIST/7,24/,M1,M2,M3,M4,M5,M6,M7/1/,ITA/9/
C          DATA NEW/1/1/1/
C          RE*IND 30
C
C          C          FEL 17C
C          C          FEL 19C
C          C          FEL 20C
C          C          FEL 21C
C          C          FEL 22C
C
C          C          FEL 23C
C          C          FEL 24C
C          *** INITIATE SEARCH FOR START OF PROBLEM
C          100 READ (ITP5,1000) CC,HEAD,MHW

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	CC=HEAD(1)	FEL 26C
	IF(CC.EQ.START) GO TO 240	FEL 27C
	IF(CC.EQ.6HFEAP73) GO TO 240	
	IF(CC.EQ.CEASE) RETURN	FEL 28C
	GO TO 100	FEL 29C
120	CONTINUE	FEL 30C
	DO 122 I=1,12	FEL 31C
122	HEAD(I)=TITLE(I)	FEL 32C
CXXXX		
C	IF CC 79 OF FEAP73 CARD IS NON-ZERO THEN READ NODE	
C	CORRESPONDENCE ARRAY PUNCHED IN PREPROCESSOR	
CXXXX		
240	DO 600 I=1,1500	
600	NEW(I)=1	
	IF(MB*.NE.*MHW1) GO TO 650	
	J=0	
	K=0	
605	HEAD (5*601) I,NEWBUF	
601	FORMAT (1615)	
	J=J+1	
	IF(I.EQ.9999) GO TO 650	
	IF(I.EQ.J) GO TO 610	
	WRITE (5,602)	
	STOP 1	
602	FORMAT (50H0** ERROR ** NODE RESEQUENCING CARD OUT OF ORDER. )	
610	DO 615 I=1,15	
	K=K+1	
	IF(NEWBUF(I).NE.0) NEW(K)=NEWBUF(I)	
615	CONTINUE	
	GO TO 605	
C		
650	READ (11P5,1000) CC,TITLE	
	NDIM = CC	FEL 34C
	DO 150 I = 1,NDIM	FEL 35C
	CC = TITLE(I)	FEL 36C
	XHEAD(I) = CC	FEL 37C
	IF(CC.EQ.BLANK) XHEAD(I) = XTIL(I)	FEL 38C
150	CONTINUE	FEL 39C
	READ(11P5,1000) CC,TITLE	FEL 40C
	NDF = CC	FEL 41C
	DO 152 I = 1,NDF	FEL 42C
	CC = TITLE(I)	FEL 43C
	IF(CC.EQ.BLANK) CC = UTTL(I)	FEL 44C
	FHEAD(I) = CC	FEL 45C
	UHEAD(I) = CC	FEL 46C
152	FHEAD(I) = CC	FEL 47C
	NUMPL = 0	FEL 48C
	READ(11P5,1001) NEN,NEXTRA,IPEC,MHAN,IBUF,NC1,CON1,CUN2,CON3	FEL 49C
	DO 153 I=1,NDIM	
153	IF(CON3(I).EQ.0.0) CON3(I) = 1.0	
	IF(MHAN.LE.0) MHAN = MHAND	FEL 50C
	ISIZ = ISZDI	FEL 51C
	IDT = ISZDI/2	FEL 52C
	NICO = 0	FEL 53C
	NVEC = 0	FEL 54C
	NPL = .TRUE.	FEL 55C
	NUMPL = 0	FEL 56C
	NT = 1	FEL 57C
	IFLOT = 0	
C.... INITIALISE TIME AND CLOCK		
	CALL TICVUC(TIME,0)	FEL 58C
	DO 121 I=1,7	FEL 59C
121	TIME(I) = 0.0	FEL 60C
	NFLT = NEN * 2	FEL 61C
	NTERM = 0	FEL 62C
	DO 10 I=1,NFLAG	FEL 63C
10	FLAG(I)=.FALSE.	FEL 64C
C		FEL 65C
C.... STORAGE IS SET FOR A NSTF X NSTF ELEMENT MATRIX AND FORCE VECTOR		FEL 66C
C.... MAXIMUM SIZE FOR NSTF IS 120; THIS IS CONTROLLED BY DIMENSION ON LD		FEL 67C
C		FEL 68C
	NSTF = NDF*NEN + NEXTRA	FEL 69C
		FEL 70C
		FEL 71C



NR = (NSTF * 1	FEL 72C
NR = NSTF * (NSTF * 2) + 1	FEL 73C
IF (IM0.GT. 10) GO TO 710	FEL 74C
IF0 = 1	FEL 75C
IF (IR0F.LE.0) IR0F = ISZDT/20	FEL 76C
IF (IR0C.NE.0) IREC = 1	FEL 77C
WRITE (IIPA,2000) HEAD,IP0,NDIM,NDF,NEN,NSTF,RE (IREC*1),MBAN,IBUF,	FEL 78C
X NC1,CON1,CON2,CON3	FEL 79C
IF0 = IF0 * 1	FEL 80C
IF (NDIM.GT.3.OR.NDF.GT.6.OR.NEN.GT.20) GO TO 720	FEL 81C
C	FEL 82C
C... SEARCH THE LIST OF NAMES FOR A TRANSFER ADDRESS	FEL 83C
C	FEL 84C
125 HEAD (IIPS,1000) CC,TITLE,MBW	FEL 86C
I = CC	FEL 87C
CC = TITLE(1)	FEL 88C
DO 126 J = 1,NLIST	FEL 89C
IF (CC.EQ.CONW0(J)) GO TO 127	FEL 90C
126 CONTINUE	FEL 91C
IF (CC.EQ.START) GO TO 12C	FEL 92C
IF (CC.EQ.DMFAP73) GO TO 120	FEL 93C
IF (CC.EQ.CEASE) RETURN	FEL 94C
GO TO 125	FEL 95C
C	
C... NAMES (MAT,NOD,POL,ELE,SOL,RES,BCS,FOR,BLD,ELD,VEC,MES,PLU,FOU,REM,	FEL 97C
127 GO TO (210,220,225,230,200,270,228,260,250,255,245,200,350,200,131,	FEL 98C
X 345,221,133,220,200,200,200,200,200) ,J	FEL 99C
C... ICD,GEN,II,BLD,EXP,ADD,VIS,IMP,NEW)	FEL 100C
C	FEL 101C
131 WRITE (IIPA,2003) (TITLE(I),I=2,12)	FEL 102C
GO TO 125	FEL 103C
133 DO 134 I = 2,12	FEL 104C
134 HEAD(I) = TITLE(I)	FEL 105C
GO TO 125	FEL 106C
C	FEL 107C
C... INPUT THE MATERIAL CHARACTERIZATIONS.	FEL 108C
C	FEL 109C
210 J = 1	FEL 110C
IF (FLAG(1)) GO TO 215	FEL 111C
FLAG(1) = .TRUE.	FEL 112C
IF (I.LE.0) GO TO 700	FEL 113C
NUMMAT = 1	FEL 114C
M1 = 0	FEL 115C
M2 = 0	FEL 116C
M3 = 0	FEL 117C
M4 = 0	FEL 118C
M5 = 0	FEL 119C
215 IF (M0.GT.1520) GO TO 710	FEL 120C
CALL FSH(J,NUMNP,NUMEL,NUMMAT,NDIM,NDF,NEN,NEL),	FEL 121C
X DI,DI(M1),DI(M2),DI(M3),DI(M4),DI(M5),DI(M6))	FEL 122C
IF (IP0.LE.0) GO TO 100	FEL 123C
GO TO 125	FEL 124C
C	FEL 125C
C... INPUT THE NODAL LOCATIONS AND BOUNDARY CONDITION CODES.	FEL 126C
C	FEL 127C
220 IF (FLAG(2)) GO TO 225	FEL 128C
FLAG(2) = .TRUE.	FEL 129C
IF (I.LE.0) GO TO 700	FEL 130C
NUMNP = 1	FEL 131C
M3 = 0	FEL 132C
M4 = 0	FEL 133C
M5 = 0	FEL 134C
M6 = 0	FEL 135C
IF (M0.GT.1520) GO TO 710	FEL 136C
IF (.NOT.FLAG(3)) M6 = M0	FEL 137C
225 CONTINUE	FEL 138C
IF (CC.EQ.CONW0(19)) GO TO 227	FEL 139C
J = 2	FEL 140C
IF (CC.EQ.CONW0(17)) J = 4	FEL 141C
GO TO 215	FEL 142C
226 IF (.NOT.FLAG(2)) GO TO 730	
CALL CONVERT(NUMNP,NDIM,DI(M4))	
GO TO 125	
227 CONTINUE	

CALL HXAGEN(NUMNP,NUMEL,NDIM,NDF,NEN,NEL1,DT(M3),DT(M4),DT(M5),	FEL143C
X DT(M6))	FEL144C
IF(IPG.LE.0) GO TO 705	FEL145C
IF(.NOT.FLAG(3)) M0 = M0 + NEL1*NUMEL	FEL146C
FLAG(3) = .TRUE.	FEL147C
IF(M0.GT.ISZDT) GO TO 710	FEL148C
GO TO 125	FEL149C
C	FEL150C
C... BOUNDARY CODE OVERWRITE	FEL151C
C	FEL152C
228 IF(.NOT.FLAG(2)) GO TO 730	FEL153C
CALL HCODES(NUMNP,DT(M3))	FEL154C
GO TO 125	FEL155C
C	FEL156C
C... INPUT THE ELEMENT CONNECTION ARRAY.	FEL157C
C	FEL158C
230 J = 3	FEL159C
IF(FLAG(3)) GO TO 215	FEL160C
FLAG(3) = .TRUE.	FEL161C
IF(I.LE.0) GO TO 700	FEL162C
NUMEL = I	FEL163C
M6 = M0	FEL164C
M0 = M0 + NEL1*NUMEL	FEL165C
GO TO 215	FEL166C
C	FEL167C
C... INPUT THE VECTORS OF PRESCRIBED VALUE.	FEL168C
C	FEL169C
245 IF(.NOT.FLAG(2)) GO TO 730	FEL170C
IF(I.LE.0) GO TO 700	FEL171C
IF(FLAG(4)) GO TO 247	FEL172C
FLAG(4) = .TRUE.	FEL173C
READ(1,PS,1001) NSIZV,IPICK	FEL174C
NVEC = I	FEL175C
M8 = M0	FEL176C
MA = M8 + NSIZV*I	FEL177C
M0 = MA	FEL178C
IF(M0*2*I.GT.ISZDT) GO TO 710	FEL179C
247 CALL VECIN (NUMNP,NUMEL,NDF,NSIZV,IPICK,NVEC,DT(MA),DT(MB))	FEL180C
IF(IPG.LE.0) GO TO 100	FEL181C
GO TO 125	FEL182C
250 CONTINUE	FEL183C
IF(I.LE.0) GO TO 125	FEL184C
C	FEL185C
C... INPUT BOUNDARY LOADS	FEL186C
C	FEL187C
CALL HLOADS(I,NUMNP,NDIM,NDF,DT(M3),DT(M4),DT(M5))	FEL188C
IF(IPG.LE.0) GO TO 705	FEL189C
GO TO 125	FEL190C
C	FEL191C
C... INPUT ELEMENT LOADS	FEL192C
C	FEL193C
255 IF(I.LE.0) GO TO 125	FEL194C
CALL ELOADS(I,NDIM,NDF,NEN,NEL1,NSIF,NSIZV,NVEC,DT(M1),DT(M2),	FEL195C
1 DT(M3),DT(M4),DT(M6),DT(M5),DT(M9),DT(MB))	FEL196C
IF(IPG.LE.0) GO TO 705	FEL197C
GO TO 125	FEL198C
C	FEL199C
C... INPUT THE INITIAL CONDITIONS	FEL200C
C	FEL201C
345 IF(FLAG(6)) GO TO 347	FEL202C
FLAG(6) = .TRUE.	FEL203C
NICD = I	FEL204C
NSICD = NDF*NUMNP	FEL205C
M0 = M0	FEL206C
M0 = M0 + NICD*NSICD	FEL207C
MC = M0	FEL208C
IF(M0*2*I.GT.ISZDT) GO TO 710	FEL209C
347 CALL VECIN (NUMNP,NUMEL,NDF,NSICD,1,NICD,DT(MC),DT(MG))	FEL210C
IF(IPG.LE.0) GO TO 100	FEL211C
GO TO 125	FEL212C
C	
C... TWO AND THREE DIMENSIONAL PLOTTING PORTHOLE CONTROL.	

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C
350  CONTINUE
    IF (NDIM.EQ.2 .AND. NEN.NE.4) GO TO 707
    CHECK = .TRUE.
    WRITE (IIP6,2040)
    DO 351 K=1,3
    IF (.NOT. FLAG(K)) CHECK = .FALSE.
351  CONTINUE
    IF (.NOT. CHECK) GO TO 707
    KSTEPS = 1
    NVEC = 8
    NSIZV = NUMEL
    IF (NDIM.EQ.3) NVEC = 2
    IF (NDIM.EQ.3) NSIZV = 6
    IF (FLAG(4)) GO TO 352
    FLAG(4) = .TRUE.
    M0 = 0
    M0 = M0 + NVEC*NSIZV
    IF (M0.GT. ISZDT) GO TO 710
352  IPLOT = NDIM
    TDUM = NUMMAT
    GO TO 125

C
C... START THE SOLUTION FOR FIRST LOADING CASE
C
200  IF (FLAG(5)) GO TO 401
    CHECK=.TRUE.
    CALL ICTQC(TIME,1)
    DO 123 J=1,3
    IF (.NOT. FLAG(J)) CHECK=.FALSE.
123  CONTINUE
    IF (.NOT. CHECK) GO TO 707
    NTM = 0
    IF (CC.NE.CONWD(22).AND.CC.NE.CONWD(23)) GO TO 515
C... SET UP VISCOELASTIC/NONLINEAR SOLUTION CALL
    NTM = 1
    NSEQ = 1
    I=0
    ISIZ = ISZDT - IBUF
    IDT = ISIZ/2
    IF (NICO.NE.0) GO TO 515
    WRITE (IIP6,2038)
    NICO = 1
    IF (CC.EQ.CONWD(23)) NICO=3
    NSICD = NDF*NUMNP
    M0 = 0
    M0 = M0 + NICO*NSICD
    IF (M0.GT. ISIZ) GO TO 710
    DO 510 J = M0+1
510  DT(J) = 0.
515  M7 = 0
    M7A = 0
    IF (ID1.GE.M7) M7 = IDT + 1
    MU = M7
    ID1 = IDT + 1
    M0 = M7 + NDF*NUMNP
    IF (M0.GT. ISIZ) GO TO 710
    MB = M0-1

C
C... SAVE MESH DATA ON TAPE FOR PLOT2D/3D
C
    IF (IPLOT.GT.0) CALL PLOT2D(0,KSTEPS,HEAD,NUMNP,NUMEL,DT(M3),
    X DT(M4),DT(M6),DT(MB),NVEC,NSIZV,NEL1,NICO,TDUM,TDUM,NDF)

C
C... CHECK THE MESH FOR CONSISTENCY OF INPUT DATA
C
    CALL MESHCK(NUMNP,NUMEL,NUMMAT,NDIM,NDF,NEN,NEL1,IDT,ID1,MB,MAXHAN
    X ,DT,DT(M1),DT(M2),DT(M3),DT(M4),DT(M5),DT(M6),DT(M7),ISZA,NEQB,
    X IBLK,NUEG)
    MX = ID1 - NEQB

CAXXX
C OPEN TAPE11 QUICKIO FILE IF OUT-OF-CORE SOLN

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FEL213C
FEL214C
FEL215C
FEL216C
FEL217C
FEL218C
FEL219C
FEL220C
FEL221C
FEL222C
FEL223C
FEL224C
FEL225C
FEL226C
FEL227C
FEL228C
FEL229C
FEL230C
FEL231C
FEL232C
FEL233C
FEL234C
FEL235C
FEL236C
FEL237C
FEL238C
FEL239C
FEL240C
FEL241C
FEL242C
FEL243C
FEL244C
FEL245C
FEL246C
FEL247C
FEL248C

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FEL249C
FEL250C
FEL251C
FEL252C
FEL254C

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CXXXX
*01 IF (IBLK.GT.0) CALL QUICKIO (6LTAPE11,0,ISZA)
      IOUT = 1
      IF (I.NE.0) WRITE (IIP6,2039)
      CALL TICTOC (TIME,2)
      IF (CC.EQ.CONWD(24)) NSEQ=1
      IF (CC.EQ.CONWD(24)) GO TO 520
      IF (CC.EQ.CONWD(20)) GO TO *15
      IF (MAXBAN.GT.MBAN) GO TO 740
      IF (CC.EQ.CONWD(12).OR.CC.EQ.CONWD(13)) GO TO 125
      IF (IPG.LE.0) GO TO 705
      IF (CC.NE.CONWD(14)) GO TO 405
C
C.... FOURIER COMPOSITION
C
      READ (ITP5,1002) J,F1,F2,F3,F4
      WRITE (IIP6,2002) MEAU,IPG,J,F1,F2,F3,F4
      IPG = IPG + 1
      NTERM = NTERM + 1
      NF (NTERM) = J
      XN = J
C
C.... FORM THE STIFFNESS FOR THE ELEMENTS
C
*05 IF (FLAG(5)) GO TO 410
      LA = 1
      J = (MAXBAN+1)*NDEG + NDF*NUMNP*(1 + NTM)
      IF (ISIZ=J.LT.M7A) GO TO 410
C.... ONCORE SOLUTION IS POSSIBLE
      WRITE (IIP6,2031)
      NEQH = NDEG
      IBLK = 0
C.... CHECK TO SEE IF IUEST IS TO BE MOVED UP
      J = NUMNP*NDF-1
C.... SAVE FORCE VECTOR FOR NONLINEAR SOLUTION
      MX=M5+J
      REWIND 9
      WRITE (9) (DT(I),I=M5,MX)
      IF (M7.EQ.M7A) GO TO 409
      DO 408 K = 1,J
*08 DT (M7A+K) = DT (M7+K)
      DT (M7A) = DT (M7)
      M7 = M7A
*09 LA = M7 + J + 1
      MU = LA + NDEG*MAXBAN
      MX = MU + NUMNP*NDF
      M0 = MX + NDEG
      I01 = NDEG*MAXBAN + 1
      FLAG(5) = *TRUE*
*10 IF (NTM.EQ.1) GO TO 520
      CALL FORMST (NUMNP,NUMEL,NUMMAT,NDIM,NDF,NEN,NEL1,M8,DT,DT (M1),
      X DT (M2),DT (M3),DT (M4),DT (M5),DT (M6),DT,DT (M9),DT (M7),DT (M8),
      X DT (MU),DT (LA),NSIF,IBLK,NVEC,NSIZV,MAXBAN,IREF,NDEG)
      CALL TICTOC (TIME,3)
      IF (IPG.LE.0) GO TO 705
C
C.... FORM THE GLOBAL STIFFNESS AND SOLVE USING GAUSS ELIMINATION
C
      CALL SOLVEQ (NUMNP,NUMEL,NDF,I01,M8,MAXBAN,ITA,NSTF,ISZA,NEQB,IRLK,
      X DT (LA),DT (M5),DT (MU),DT (M7),DT,DT (M9),DT (MX),NDEG)
      CALL TICTOC (TIME,4)
      GO TO 420
C
C.... DYNAMIC SOLUTION BY EXPLICIT INTEGRATION
C
*15 IF (NICO.GE.3) GO TO 418
      WRITE (IIP6,2038)
      NICO = 3
      NSICO = NDF*NUMNP
      M0 = M0
      M0 = M0 + NICO*NSICO
      IF (M0.GE.ISZDT) GO TO 710

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FEL258C
FEL259C
FEL262C
FEL263C
FEL264C
FEL265C
FEL266C
FEL267C
FEL268C
FEL269C
FEL270C
FEL271C
FEL272C
FEL273C
FEL274C
FEL275C
FEL276C
FEL277C
FEL278C
FEL279C
FEL280C
FEL281C
FEL282C
FEL283C
FEL284C
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FEL286C
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FEL314C
FEL315C
FEL316C
FEL317C
FEL318C
FEL319C
FEL320C
FEL321C
FEL322C
FEL323C
FEL324C
FEL325C
FEL326C
FEL327C
FEL328C
FEL329C

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417 DO 417 J=MG,M0                                FEL330C
    DT(J) = 0.0                                    FEL331C
*18 MC = MG + NSICD                                FEL332C
    MD = MC + NSICD                                FEL333C
    ME = M0                                          FEL334C
    MF = ME + NSTF                                    FEL335C
    MO = 2*NSTF + MF                                FEL336C
    IF(MU.GT.1S2DT) GO TO 710                        FEL337C
    CALL EXPLCT(NUMNP,NUMEL,NUMMAT,NDIM,NDF,NEN,NEL1,NSTF,NVEC,NSIZV, FEL338C
    X DT(M1),DT(M2),DT(M3),DT(M4),DT(M5),DT(M6),DT(M7),DT(M8),DT(M9),
    X DT(M10),DT(M11),DT(M12),DT(M13),DT(M14),DT(M15),DT(M16),DT(M17),DT(M18),DT(M19),DT(M20),
    GO TO 125                                          FEL339C
270 IQUT = I                                          FEL340C
    CALL ICTOC(TIME,I)                                FEL341C
C                                                    FEL342C
C... FORM THE NEW LOAD AND DO A RESOLUTION ONLY      FEL343C
C                                                    FEL344C
    CALL HFSVEG(NUMNP,NDF,M7-1,MAXBAN,ISZA,NEGB,IBLK,DT(LA),DT(M5),
    X DT(MU),DT(M5),DT(M7),DT(MX),NDF*NUMNP)        FEL345C
    CALL ICTOC(TIME,5)                                FEL346C
420 CONTINUE                                          FEL347C
C                                                    FEL348C
C... COMPUTE AND OUTPUT THE NODAL DISPLACEMENTS AND ELEMENT STRESSES FEL349C
C                                                    FEL350C
    IF(IIRC.GT.0.AND.IHLK.EQ.0) REWIND 7             FEL351C
    CALL DISTR(NUMNP,NUMEL,NUMMAT,NDIM,NDF,NEN,NEL1,NSTF,NVEC,NSIZV, FEL352C
    X DT(M1),DT(M2),DT(M3),DT(M4),DT(M5),DT(M6),DT(M7),DT(M8),DT(M9),DT(M10),DT(M11),DT(M12),DT(M13),DT(M14),DT(M15),DT(M16),DT(M17),DT(M18),DT(M19),DT(M20),
    X IQUT,IIRC,IBLK)                                FEL353C
    CALL ICTOC(TIME,6)                                FEL354C
    IF(IHLK.GT.0) REWIND 7                            FEL355C
    IF(IHLK.GT.0) READ(7)(DT(I),I=1,M8)              FEL356C
    I = 0                                              FEL357C
    WRITE(IIP6,2030) IYME                             FEL358C
    DO 271 N=1,7                                       FEL359C
271 IYME(N) = 0.0                                     FEL360C
C                                                    FEL361C
C... INPUT THE FORCE VECTORS ON NODES (RESET TO ZERO AUTOMATICALLY FEL362C
C... AFTER EACH PROBLEM SOLUTION OR RESOLUTION).     FEL363C
C                                                    FEL364C
460 CONTINUE                                          FEL365C
    IF(.NOT.FLAG(2)) GO TO 730                        FEL366C
    CALL RESET(1,NUMNP,NDF,DT(M5))                  FEL367C
CXXXX                                                FEL368C
C CLOSE FILE 11 IF OUT-OF-CORE SOLN WAS DONE          FEL369C
CXXXX                                                FEL370C
    IF(IHLK.GT.0) CALL QUICKIO (6LTAPE11,4)          FEL371C
    GO TO 125                                          FEL372C
C                                                    FEL373C
C... TIME INTEGRATION SUBROUTINE FOR DYNAMIC AND VISCOELASTIC SOLUTIONS FEL374C
C                                                    FEL375C
520 IF(FLAG(7)) GO TO 540                            FEL376C
    FLAG(7) = .TRUE.                                  FEL377C
    I = M0 + NSICD = 1                                FEL378C
    IF(I.GT.ISIZ) GO TO 710                            FEL379C
    DO 530 J = M0,I                                    FEL380C
530 DT(J) = 0.                                         FEL381C
    IF(IHLK.EQ.0) GO TO 540                            FEL382C
    MU = M0 + NSICD                                    FEL383C
    IF(MU*NSICD.GT.ISIZ) GO TO 710                    FEL384C
540 ML=1
    IF((CC.EQ.CONWD(22)).OR.(CC.EQ.CONWD(23)))
    X CALL TSOLVE(NUMNP,NUMEL,NUMMAT,NDIM,NDF,NEN,NEL1,NSTF,NVEC,
    NSIZV,NICD,NSICD,IHLK,ISZA,NEGB,MAXBAN,NDEG,I01,M8,IBUF,DT,
    1 DT(M1),DT(M2),DT(M3),DT(M4),DT(M5),DT(M6),DT(M7),DT(M8),
    2 DT(M9),DT(M10),DT(M11),DT(M12),DT(M13),DT(M14),DT(M15),DT(M16),DT(M17),DT(M18),DT(M19),DT(M20),
    3 DT(MX),DT(1SIZ+1),DT(MU),NSEQ,TIME)
    4 NSEQ,TIME)
    IF(CC.EQ.CONWD(24))
    XCALL NEWTON (NUMNP,NUMEL,NUMMAT,NDIM,NDF,NEN,NEL1,NSTF,NVEC,NSIZV,
    IHLK,ISZA,NEGB,MAXBAN,NDEG,I01,M8,IBUF,DT,DT(M1),DT(M2),DT(M3),
    2DT(M4),DT(M5),DT(M6),DT(M7),DT(M8),DT(M9),DT(M10),DT(M11),DT(M12),DT(M13),DT(M14),DT(M15),DT(M16),DT(M17),DT(M18),DT(M19),DT(M20),
    3DT(MX),DT(1SIZ+1),DT(MU),NSEQ,TIME)
    I = 0

```

```

      GO TO 260
C.... PROGRAM EXECUTION TIME ARRAY, REFERENCE SUBROUTINE CLOCK
705 WRITE(IIP6,2030) IYME
      GO TO 100
707 WRITE(IIP6,2033) (FLAG(J),J=1,3)
      GO TO 100
700 WRITE(IIP6,2034) CC
      GO TO 100
710 WRITE(IIP6,2035) MD*ISIZ*CC
      GO TO 100
720 WRITE(IIP6,2036) NDIM,NDF,NEN
      GO TO 100
730 WRITE(IIP6,2037) CC
      GO TO 100
735 WRITE(IIP6,2038)
      GO TO 100
740 WRITE(IIP6,2032) MABAN
      GO TO 100

C....
C.... FORMATS
C....
1000 FORMAT (F5.0,1X,12A0,A1)
1001 FORMAT (G15.5F10.0)
1002 FORMAT (15,5X,4F10.0)
2000 FORMAT (1H1,12A6,30X,4HPAGE,I4////
X/19X,32H FINITE ELEMENT ANALYSIS PROGRAM///13X, 22(2H *) //
X 13X,2H *.1R,34H DIMENSIONAL PROBLEM * * * * * //
X 13X,2H *.1R,34H DEGREES OF FREEDOM PER NODE * * * * * //
X 13X,2H *.1R,34H NODES CONNECTED TO EACH ELEMENT * * * * * //
X 13X,2H *.1R,34H ELEMENT STIFFNESS SIZE * * * * * //13X,22(2H *)//
X 13X,2H *.HX,34H A GENERALIZED FORCE CHECK HAS * * * * * //
X 13X,2H *.9X,44,14HBEEN REQUESTED,14X,1H*//
X 13X,2H *.1R,34H IS MAXIMUM PERMISSIBLE HALF BAND * * * * * //
X 13X,2H *.1R,34H WORD BUFFER AREA RESERVED FOR H * * * * * //13X,22(2H *)//
X 13X,2H *.HX,18H CONSTANT 1---- =,1R,7X,1H* //
X 13X,2H *.HX,18H CONSTANT 2---- =,E12.4,4H * * * * * //
X 13X,2H *.HX,18H CONSTANT 3---- =,E12.4,4H * * * * * //
X 13X,2H *.HX,18H MULTIPLIER 1---- =,E12.4,4H * * * * * //
X 13X,2H *.HX,18H MULTIPLIER 2---- =,E12.4,4H * * * * * //
X 13X,2H *.HX,18H MULTIPLIER 3---- =,E12.4,4H * * * * * //13X,22(2H *)//
2002 FORMAT (1H1,12A6,30X,4HPAGE,I4//
X 5X,37HFOURIER COEFFICIENTS FOR HARMONIC NO. * * * * * I4//
X 10X, 25HRAPIAL COEFFICIENT = E12.4//
X 10X, 25HTANGENTIAL COEFFICIENT = E12.4//
X 10X, 25HAXIAL COEFFICIENT = E12.4//
X 10X, 25HTHERMAL COEFFICIENT = E12.4// )
2003 FORMAT (120X,11A6)
2030 FORMAT (1H1,21X,12HELAPSED TIME///10X,25HINPUT PROPERTIES AND MESH
,F10.3/10X,25HCHECK AND PLOT INPUT DATA,F10.3/10X,14HFORM STIFFNESS
,F21.3/10X,21HSOLUTION OF EQUATIONS,F14.3/10X,23HRESOLUTION OF EQUA
TIONS,F12.3/10X,14HOUTPUT ANSWERS,F21.3/10X,10HTOTAL TIME,F25.3 )
2031 FORMAT (// 5X,16HIN-CORE SOLUTION )
2032 FORMAT (27HOBANDWIDTH EXCEEDED, MRAND=15)
2033 FORMAT (55HALL INFORMATION FOR THIS PROBLEM HAS NOT BEEN PROVIDED/
X 15HOMATERIAL CARDS * L2/
X 15HONODAL CARDS * L2/
X 15HELEMENT CARDS * L2/ )
2034 FORMAT (5X,4X,16H TYPE CARD ERROR/)
2035 FORMAT (20HREQUIRED STORAGE = 17/20HAVAILABLE STORAGE = 17/
X 25HSTORAGE EXCEEDED DURING ,A6/)
2036 FORMAT (C0HINPUT SIZE ERROR, ,8H NDIM = ,15,8H, NDF = ,15,
C 8H, NEL = ,15)
2037 FORMAT (18HATTEMPT TO INPUT ,A6,20H BEFORE NODAL POINTS )
2038 FORMAT (5X, 48H***NON-FATAL ERROR*** INITIAL DYNAMIC VECTORS HAVE NOT
X BEEN INPUT ***INITIAL CONDITION ASSUMED ZERO )
2039 FORMAT (47HREQUEST MADE FOR LIMITED AMOUNT OF OUTPUT,
X ,25H ADDITIONAL DATA REQUESTED/1X)
2040 FORMAT (////33H REQUEST FOR PLOTS IS INITIATED. )
      END
*DECK
      HLKDATA
      SUBROUTINE HLKDAT
      REAL LABL

```

```

FEL397C
FEL398C
FEL399C
FEL400C
FEL401C
FEL402C
FEL403C
FEL404C
FEL405C
FEL406C
FEL407C
FEL408C
FEL409C
FEL410C
FEL411C
FEL412C
FEL413C
FEL414C
FEL415C
FEL416C
FEL417C
FEL419C
FEL420C
FEL421C
FEL422C
FEL423C
FEL424C
FEL425C
FEL426C
FEL427C
FEL428C
FEL429C
FEL430C
FEL431C
FEL432C
FEL433C
FEL436C
FEL437C
FEL438C
FEL439C
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FEL449C
FEL450C
FEL451C
FEL452C
FEL453C
FEL454C
FEL455C
FEL456C
FEL457C
FEL458C
FEL459C
FEL460C
FEL461C
FEL462C
FEL463C
FEL464C

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LOGICAL NPH,NPL
COMMON/UTNANO/ TIME,NSIG(7),NT,NSTEP,UT,NUMPLT,NELDATA(20,3),NPH,NPL
COMMON/DBLS/ LIM,SGAUSS(5,5),WGAUSS(5,5)
COMMON/LABELS/ LABL(6),XHED(3),XH,FHED(6),FH,UHED(6),UH,RHED(6),RH
X ,AWORD1,AWORD2,AWORD3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)
COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)
COMMON/ITILES/ XTIL(3),UTTL(6)
COMMON/ITAPES/ ITH5,ITP6
COMMON/ITAPL/ ITP13,ITP14,ITPHD,ITPRH
DATA SHAPE/8000.0/,START,CEASE/6HFEAP72,6HSTUP /
DATA AWORD1,SHELM01,SHELM02,SHELM03,SHELM04,SHELM05,SHELM06,SHELM07,
X SHELM08,SHELM09,SHELM10,SHELM11,SHELM12,SHELM13,SHELM14,SHELM15,
X SHELM16,SHELM17,SHELM18,SHELM19,SHELM20,SHELM21,SHELM22,SHELM23,
X SHELM24,SHELM25,SHELM26,SHELM27,SHELM28,SHELM29,SHELM30/
DATA SGAUSS/50.0,-.57735027,.57735027,3.0,-.77459667,0.0,.77459667
C ,2.0,0.0,-.33998104,.33998104,.86113631,0.0,-.93617985,
C -.53846931,0.0,.53846931,.90617985/
DATA WGAUSS/5.0,2.0,5.0,1.0,5.55555556,.88888889,.55555556,2.0,0.0,.34785485
C ,2.0,.5214515,.34785485,0.0,.23642689,.47862867,.56888889,.47862867
C .23642689/
DATA LABL(1)/BH (112) /
DATA XTIL/6H 1.0H 2.0H 3.0H XH/6H ORD. /
DATA UTTL/6H 1.0H 2.0H 3.6H 4.6H 5.6H 6/
DATA FH,RH,LH/ZH FORCE,6H FORCE,6H DISPL/
DATA AWORD1,AWORD2/BH F13.4,8H 6E13.4/,AWORD3/RH F12.2,/
DATA ITH5,ITP6 /5,6/ , NSIG/7,0, TIME/0.0/,NUMPLT,NPH,NPL/0,0,1/
DATA ITP13,ITP14/13,14/
RETURN
END
30C

*DECK TICTOC
SUBROUTINE TICTOC(TIME, I)
C
C SUPROUTINE TO TIME PROGRAM SEGMENTS AS MEASURE OF EFFICIENCY
C
C
C DIMENSION TIME(7)
C IF (I .LE. 6) GO TO 100
C CALL SECOND(T1)
C TIME (1) = T1 - T0
C TIME (7) = TIME(7) + TIME(1)
C T0 = T1
C RETURN
100 CALL SECOND(T0)
C RETURN
C END
C T1C 1C
C T1C 2C
C T1C 3C
C T1C 4C
C T1C 5C
C T1C 6C
C T1C 7C
C T1C 8C
C T1C 9C
C T1C 10C
C T1C 11C
C T1C 12C
C T1C 13C
C T1C 14C

*DECK MESH
SUBROUTINE MESH(MC,NUMNP,NUMEL,NUMMAT,NDIM,NDF,NEL,NEL1,
X D1,TYPE,0,ICOD,XYZ,F,IX)
C**** INPUT OF MESH AND MATERIAL PROPERTIES.
COMMON/ITAPES/ ITH5,ITP6
REAL LABL
DIMENSION TYPE(1),D(3,21,1),ICOD(1),XYZ(NDIM,1),F(NDF,1),
X IX(NEL1,1),IXD(20),DX(3),IXP(20),XWORD(11)
COMMON/DBLS/ LIM,SGAUSS(5,5),WGAUSS(5,5)
COMMON/LABELS/ LABL(6),XHED(3),XH,FHED(6),FH,UHED(6),UH,RHED(6),RH
X ,AWORD1,AWORD2,AWORD3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)
COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)
COMMON/VALUES/ NCL,CUN1,CUN2,CUN3(3)
DATA BLANK/6HBLANK /
GO TO(1,2,3,2),MC
CONTINUE
NCT = 0
DO 300 I = 1,NUMMAT
NCT = NCT + 1
IF(NCT.GT.3) GO TO 10
WRITE(ITP6,2001) HEAD,IPG,NUMMAT
IPG = IPG + 1
NCT = 4
CONTINUE
10 READ(ITP5,1002) M,DM,XWORD
IF(M.GT.NUMMAT,OR,M.LE.0) GO TO 301
TYPE(M) = DM
WRITE(ITP6,2002) UM,M,XWORD

```

	DO 280 J = 1,3	MES 280
	DO 280 K = 1,21	MES 290
280	D(J,K,M)=0.0	MES 300
C		MES 310
	K = 0	MES 320
	DO 290 J = 1,30	MES 330
	IF(DM.EQ.0) GO TO 400	MES 340
290	CONTINUE	MES 350
C....	ERROR IF EXIT ON LOOP 290	MES 360
301	WRITE(11P6,2032)M,DM	MES 370
	IP6 = 0	MES 380
	RETURN	MES 390
400	CALL ELMFIB(N,M,NDIM,NDF,NEL,NEL),NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	MES 400
	X IX,F,FORCE,ESTIF,U,VECT,1)	MES 410
300	CONTINUE	MES 420
	RETURN	MES 430
2	CONTINUE	MES 440
C		MES 450
C....	SET XYZ(1:N) TO BLANKS	MES 460
C		MES 470
	DO 20 N = 1,NUMNP	MES 480
	XYZ(1,N) = BLANK	MES 490
	DO 20 I = 1,NDF	MES 500
20	F(I,N) = 0.0	MES 510
C		MES 520
C....	START FILLING DATA	MES 530
C		MES 540
	N = 0	MES 550
	NXP = 0	MES 560
21	NP = N	MES 570
	IF(MC.EQ.2.AND.N.GE.NUMNP) GO TO 30	MES 580
	NX = NXP	MES 590
	READ(11P5,1006) N,NXP,I,DX	MES 600
	IF(N.LE.0.OR.N.GT.NUMNP) GO TO 30	MES 610
	IF(MC.EQ.2) NXP = 1	MES 620
	ICOD(N) = 1	MES 630
	DO 26 I = 1,NDIM	MES 640
26	XYZ(I,N) = LX(I)*CON3(I)	MES 650
	IF(NX.EQ.0) GO TO 21	MES 660
	IF((N-NP)*NX.GE.0) GO TO 25	MES 670
	NX = -NX	MES 680
	WRITE(11P6,2033) NP,N	MES 690
25	CONTINUE	MES 700
C		MES 710
C....	GENERATE THE NODES BETWEEN NP AND N IN INTERVALS OF NX	MES 720
C		MES 730
	LX = (IABS(N-NP) + IABS(NX) - 1)/IABS(NX)	MES 740
	DO 22 I = 1,NDIM	MES 750
22	DX(I) = (XYZ(1,N) - XYZ(1,NP))/LX	MES 760
23	NP = NP + NX	MES 770
	IF(NX.GT.0.AND.NP.GE.N) GO TO 21	MES 780
	IF(NX.LT.0.AND.NP.LE.N) GO TO 21	MES 790
	DO 24 I = 1,NDIM	MES 800
24	XYZ(I,NP) = XYZ(I,NP-NX) + DX(I)	MES 810
	ICOD(NP) = 0	MES 820
	IF(ICOD(NP-NX).EQ.ICOD(N)) ICOD(NP) = ICOD(N)	MES 830
	GO TO 23	MES 840
C		MES 850
C....	CHECK IF ALL DATA HAS BEEN INPUT OR GENERATED AND PRINT OUTPUT	MES 860
C		MES 870
30	CALL PRIMSH(1,NUMNP,NDIM,ICOD,XYZ)	MES 880
	RETURN	MES 890
C....	INPUT ELEMENT CONNECTION ARRAY	MES 900
3	NCT = 1	MES 910
	NSIDE = (NEL*1)/NDIM**2	MES 920
	N=0	MES 930
	NAP = 0	MES 940
C....	SET UP INCREMENT ARRAY IF NOT INPUT OFF CARDS	MES 950
	GO TO 132	MES 960
130	DO 151 L = 1,NEL	MES 970
151	IXU(L) = IXP(L)	MES 980
	IF(IXU(L).NE.0) GO TO 157	MES 990



```

DO 15 L = 1,NEL
150 IX(L) = NSIDE
    N1 = 4*NSIDE + 1
    N2 = 8*NSIDE - 4
    IF (N2.GT.NEL) GO TO 154
    DO 152 L = N1,N2
152 IX(L) = 1
154 N1 = 2*NSIDE
    N2 = N1 + 2 - NSIDE
    N4 = N1 + N2
    DO 156 L = N3,N1
    IX(L) = 1
    IX(L + N1) = 1
    IF (N2.GT.NEL) GO TO 156
    IX(L + N2) = 1
    IX(L + N4) = 1
156 CONTINUE
157 CONTINUE
132 READ(11PB,1003) M,MA,IR,IP,IAP
    IF (M.LE.0) RETURN
    READ(11PB,1004) (IX(I,M),I=1,NEL)
    IF (M.EQ.0) K = M - 1
    IX(NEL,K) = MA * 10*IP + 100*IR
    K1 = 1
    DO 133 I1 = 1,NEL
    IF (IX(I1,M),EQ.0) GO TO 134
    K1 = K1 + 1
133 CONTINUE
134 IF (K1.GT.NEL) GO TO 140
    DO 136 I1 = K1,NEL
136 IX(I1) = 0
140 M=N+1
    MCI = MCI - 1
    IF (MCI.GT.0) GO TO 142
    WRITE(11PB,2011) READ, IPG,NUMEL,(I,I=1,NEL)
    IPG = IPG + 1
    MCI = 50
142 CONTINUE
    IF (M.LE.0) GO TO 170
    DO 158 L = 1,NEL
158 IX(L,N) = IX(L,M-1) * IX(L)
    IX(NEL,N) = MAP
170 CONTINUE
    K = 0
    DO 34 J1=1,NEL
    IF (IX(J1,N),EQ.0) GO TO 141
    DO 325 J1=11,NEL
    IF (IX(J1,N),EQ.0) GO TO 340
    KK=1*NS*(IX(J1,N)-IX(J1,N)) + 1
    IF (K.LE.KK) K=KK
325 CONTINUE
340 CONTINUE
141 MAP= MOD(IX(NEL,N)+10)
    IR = IX(NEL,N)/100
    IP = MOD(IX(NEL,N),100)/10
    WRITE(11PB,2005) N,MAP,IR,IP,(IX(I,N),I=1,NEL),K
    IF (M=N) 155,150,140
155 WRITE(11PB,2031) N
    IPG = 0
    RETURN
180 MAP = MA
    IF (NUMEL.GT.N) GO TO 130
    RETURN
C**** FORMATS
1002 FORMAT(15,1X,A5,11A6)
1003 FORMAT(4I5,20I3)
1004 FORMAT(20I4)
1005 FORMAT(2I5,11U,4F10.0)
001 FORMAT(1H1,12AA,30X,4HPAGE,14//15,10H MATERIALS//
X 20H MATERIAL PROPERTIES //1X)
2002 FORMAT(5X,A5,1X, 8HMATHEIAL,15,5X,11A6/1X)
2005 FORMAT(25I5)

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```

MES100C
MES101C
MES102C
MES103C
MES104C
MES105C
MES106C
MES107C
MES108C
MES109C
MES110C
MES111C
MES112C
MES113C
MES114C
MES115C
MES116C
MES117C
MES118C
MES119C
MES120C
MES121C
MES122C
MES123C
MES124C
MES125C
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MES127C
MES128C
MES129C
MES130C
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MES148C
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MES150C
MES151C
MES152C
MES153C
MES154C
MES155C
MES156C
MES157C
MES158C
MES159C
MES160C
MES161C
MES162C
MES163C
MES164C
MES165C
MES166C
MES167C
MEU168
MES169C
MES170C
MES171C

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2011 FORMAT(1H),1246,3J,X,4HPAGE,14//15,9H ELEMENTS//
X 20H ELMT MATL RE PRNT ,3X,26HNODES CONNECTED TO ELEMENT/
X 20H NO. NO. USES ,1X,20(2H ,12,1H))
MES172C
MES173C
MES174C
C****
MES175C
C**** ERROR MESSAGES *
MES176C
C****
MES177C
2031 FORMAT (24H) ELEMENT CARD ERROR, N=15)
MES178C
2032 FORMAT(26H) MATERIAL CARD ERROR, M = ,13,2X,BH,TYPE = ,A5/)
MES179C
2033 FORMAT(
MES180C
X ,5H,58H**NON-FATAL ERROR** INCREMENT FOR GENERATION BETWEEN NODES,
MES181C
X ,15,4H AND,15,3/H IS OF INCORRECT SIGN ** SIGN CHANGED/1X)
MES182C
END
MES183C
*DECK MESHCK
SUBROUTINE MESHCK(NUMNP,NUMEL,NUMMAT,NDIM,NDF,NEL,NEL1,IUT,IOL,
MES 1C
X M7,M8,DT,TYPE,D,ICUD,XYZ,IF,IX,IDEST,ISZA,NEGB,IRLK,NDEG)
X INTEGER XYZ,X,BLANK
X REAL LAM
X DIMENSION TYPE(1),D(63,1),XYZ(NDIM,1),IX(NEL1,1),IDEST(NDF,1)
MES 3C
X ,ICUD(1),IF(NDF,1)
MES 4C
COMMON/LABELS/ LABL(6),XNED(3),XN,FHED(6),FN,UHED(6),UH,RHED(6),RH
MES 6C
X ,XNND(1),XNCPD(2),XNCPD3,HEAD(12),START,CEASE,IPG,NSTR,NORD(30)
MES 7C
COMMON/SHAP/ XJAC,SHAPF(4,20),SG(3,3),SA(3,3),X(3,20),LD(120)
MES 8C
COMMON /ITP6/ ITP5,ITP6
MES 9C
COMMON /MINBW/ NEW(1500)
DATA /LANK/EMHLANK /
MES 10C
CXXXX
C REVERSE ALL ARRAYS BASED ON MINIMUM BANDWIDTH
CXXXX
DO 10 N=1,NUMNP
NA=NEW(N)
10 IDEST(1,NA)=ICUD(N)
DO 15 N=1,NUMNP
15 ICUD(N)=IDEST(1,N)
DO 30 I=1,NDIM
DO 20 N=1,NUMNP
NA=NEW(N)
20 IDEST(1,NA)=XYZ(I,N)
DO 25 N=1,NUMNP
25 XYZ(I,N)=IDEST(1,N)
30 CONTINUE
DO 35 N=1,NUMEL
DO 35 I=1,NEL
K=IX(I,N)
IF(K.EQ.0) GO TO 35
IX(I,N)=NEW(K)
35 CONTINUE
36 CONTINUE
IOL = IUT + 1
C.... SET UP THE DESTINATION VECTOR
MES 11C
NDEG = 0
MES 12C
DO 14 N = 1,NUMNP
MES 13C
IC = ICUD(N)
MES 14C
IL = 100000
MES 15C
DO 12 I = 1,NDF
MES 16C
IDEST(I,N) = 0
MES 17C
IF(IC,LI,IL) GO TO 150
MES 18C
IC = IC - IL
MES 19C
IF(JC,GI,IL) GO TO 100
MES 20C
GO TO 120
MES 21C
100 NDEG = NDEG + 1
MES 22C
IDEST(I,N) = NDEG
MES 23C
120 IL = IL/10
MES 24C
140 CONTINUE
MES 25C
MCI = 0
MES 26C
DO 16 IJ=1,NUMNP
MES 27C
N=NEW(IJ)
MES 28C
MCI = MCI + 1
MES 29C
IF(MCI.GT.0) GO TO 160
MES 30C
MCI = 50
MES 31C
WRITE(ITP6,2033) HEAD,IPG,(UHED(I),I=1,NDF)
MES 32C
IPG = IPG + 1
MES 33C
160 WRITE (ITP6,2000) IJ,(IDEST(I,N),I=1,NDF)

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C.... COMPUTE THE BANDWIDTH OF THE MESH
MH = 0
DO 19 N = 1,NUMEL
  MM = 0
  NN = 0
  DO 17 I = 1,NEL
    K = IX(I,N)
    IF (K.EQ.0) GO TO 180
    DO 17 J = 1,NEL
      IC = IDEST(J,K)
      IF (IC.GT.MM) MM = IC
      IF (IC.EQ.0) GO TO 170
      IF (IC.LT.NN.OR.NN.EQ.0) NN = IC
170 CONTINUE
180 IF (MM-NN.GT.MH) MH = MM-NN
190 CONTINUE
  MH = MH + 1
  MH1 = MH + 2
  NEQH = 101/MH1
  IF (NEQH.GT.NDEG) NEQH = NDEG
  IRLK = (NDEG+NEQH - 1)/NEQH
  WRITE (ITP6,2031) HEAD,IPG,NDEG,MH,NEQH,IRLK
  IPG = IPG + 1
  ISZA = NEQH*(MH+1)
C.... CHECK MESH FOR CONSISTENCY OF DATA
DO 50 N = 1,NUMEL
  MA = MOD(IX(NEL1,N)+10)
  IF (MA.GT.NUMMAT.OR.MA.LE.0) GO TO 40
  DN = TYPE(MA)
  FLAG = 1.0
  DO 205 I=1,NEL
    K = IX(I,N)
    IF (K.GT.NUMAP.OR.N.LT.0) GO TO 40
    IF (K.EQ.0) GO TO 205
    NEN = I
    IF (XYZ(I,K) .NE. BLANK) GO TO 195
    FLAG = -1.
    WRITE (ITP6,2034) K,N
    IPG = 0
    GO TO 205
195 CONTINUE
DO 205 J = 1,NDIM
  X(J,I) = XYZ(J,K)
200 CONTINUE
205 IF (FLAG.GT. 0.0)
  X CALL ELMLIB(N,MA,NDIM,NEN,NEL1,NSIF,NSIZV,NVEC,MCT,DM,D,XYZ,
  X IX,F,FORCE,ESTIF,U,VECT,2)
  IF (DM.GT.0) GO TO 50
  WRITE (ITP6,2030) N,MA,DM,((IX(J,N),J=1,NEL)
  IPG = 0
50 CONTINUE
  IF (IPG.GT. 0) WRITE (ITP6,2032)
  RETURN
2000 FORMAT(10,5X,A16)
2030 FORMAT(10,10E15.10H, MATERIAL,15,10H, JACOBIAN,E12.4/
  X 9F 1X ARRAY,20I5)
2031 FORMAT(1H1,12A6,30X,4HPAGE,14//17H EQUATION SUMMARY//
  X 5X,21HNUMBER OF EQUATIONS =,16//5X,21HMAXIMUM HALF BAND =,16//
  X 5X,21HEQUATIONS PER BLOCK =,16//
  X 5X,21HNUMBER OF BLOCKS =,16//4X,2B(1H0)////1X)
2032 FORMAT(5X,46HNO ERRORS DETECTED DURING A CHECK OF THE MESH )
2033 FORMAT(1H1,12A6,30X,4HPAGE,14//19H DESTINATION VECTOR//
  X 5X,4HNODE,2X,3HDOF,5A6)
2034 FORMAT(21H ***FATAL ERROR***NOUE, 16, 34H NOT INPUT BUT IS USED IN E
  XLEMENT, 16 )
END
*DECK PRTFLM
SUBROUTINE PRTFLM(NF,MF,NEL,NEL1,IX)
  PRT LABL
  DIMENSION IX(NEL1,1)
  COMMON /ITP6S/ ITP5,ITP6
  COMMON /LABELS/ LABL(6),XH,FHED(6),FH,UHED(6),UH,RHED(6),RH

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X , AWORD1, AWORD2, AWORD3, HEAD(12), START, CEASE, IPG, NSTR, AWORD(30)      PRT 6C
MCT = 0                                                                    PRT 7C
DO 300 N = NE, ME                                                            PRT 8C
MCT = MCT + 1                                                                PRT 9C
IF (MCT.GT.0) GO TO 100                                                       PRT 10C
WRITE (IIP6,2011) HEAD,IPG,ME,(I,I=1,NEL)                                   PRT 11C
IPG = IPG + 1                                                                PRT 12C
MCT = 50                                                                    PRT 13C
100 CONTINUE                                                                PRT 14C
K = 0                                                                        PRT 15C
MA = MOD(IX(NEL1,N),10)                                                       PRT 16C
IR = IX(NEL1,N)/100                                                         PRT 17C
IP = MOD(IX(NEL1,N),100)/100                                                PRT 18C
DO 140 I1 = 1,NEL                                                           PRT 19C
IF (IX(I1,N),EQ.0) GO TO 150                                                 PRT 20C
DO 135 J1 = I1,NEL                                                         PRT 21C
IF (IX(J1,N),EQ.0) GO TO 140                                                 PRT 22C
KK = IABS(IX(I1,N)-IX(J1,N)) + 1                                           PRT 23C
IF (K.LT.KK) K = KK                                                         PRT 24C
135 CONTINUE                                                                PRT 25C
140 CONTINUE                                                                PRT 26C
150 WRITE (IIP6,2005) N,MA,IR,IP,(IX(I,N),I=1,NEL),K                     PRT 27C
300 CONTINUE                                                                PRT 28C
RETURN                                                                      PRT 29C
2005 FORMAT(25I5)                                                           PRT 30C
2011 FORMAT(1H1,12A6,3X,4HPAGE,14//15,9H ELEMENTS//                       PRT 31C
X 20H ELEM MATL RE PRINT ,3X,26HNODES CONNECTED TO ELEMENT/              PRT 32C
X 20H NO. NO. USES ,1X,20(2H *,12,1H*))                                   PRT 33C
END                                                                          PRT 34C
*DECK PRTM5H
SUBROUTINE PRTM5H(N1,NUMNP,NDIM,ICOD,XYZ)                                    PRT 1C
REAL LAML                                                                    PRT 2C
DIMENSION ICOD(1),XYZ(NDIM,1)                                              PRT 3C
COMMON /TAPES/ ITP5,ITP6                                                    PRT 4C
COMMON/LABELS/ LABL(6),XHED(3),XH,FHED(6),FH,UHED(6),UH,WHED(6),RH      PRT 5C
X , AWORD1, AWORD2, AWORD3, HEAD(12), START, CEASE, IPG, NSTR, AWORD(30)  PRT 6C
DATA BLANK/CHHLANK /                                                        PRT 7C
MCT = 0                                                                        PRT 8C
DO 33 N = N1,NUMNP                                                         PRT 9C
MCT = MCT + 1                                                                PRT 10C
IF (MCT.GT.0) GO TO 31                                                       PRT 11C
WRITE (IIP6,2010) HEAD,IPG,NUMNP,(XHED(I),XH,I=1,NDIM)                  PRT 12C
IPG = IPG + 1                                                                PRT 13C
MCT = 50                                                                    PRT 14C
31 IF (XYZ(1,N),EQ.BLANK) GO TO 32                                           PRT 15C
WRITE (IIP6,2006) N,ICOD(N),(XYZ(I,N),I=1,NDIM)                          PRT 16C
GO TO 33                                                                     PRT 17C
32 WRITE (IIP6,2007) N                                                       PRT 18C
33 CONTINUE                                                                PRT 19C
RETURN                                                                      PRT 20C
2006 FORMAT(21I2,7F13.4)                                                    PRT 21C
2007 FORMAT(1I2,5X,31HHAS NOT BEEN INPUT OR GENERATED )                 PRT 22C
2010 FORMAT(1H1,12A6,3X,4HPAGE,14//15,13H NODAL POINTS//                PRT 23C
X 12H NODAL POINT,6X,6H*H.C.,7(1X,246))                                   PRT 24C
END                                                                          PRT 25C
*DECK PRIMAT
SUBROUTINE PRIMAT(HEAD,IPG,NELM,NSTR,ESTIF,FORCE,LD,NT)                    PRT 1C
COMMON /TAPES/ ITP5,ITP6                                                    PRT 2C
DIMENSION ESTIF(N1,NT),FORCE(NT),LD(NT),HEAD(12)                         PRT 3C
MBLK = (NSTR+23)/24                                                         PRT 4C
NBLK = (NSTR + 8)/9                                                         PRT 5C
N1 = 1                                                                        PRT 6C
DO 200 NN = 1,NBLK                                                         PRT 7C
N2 = N1 + 8                                                                  PRT 8C
IF (N2.GT.NSTR) N2 = NSTR                                                  PRT 9C
M1 = 1                                                                        PRT 10C
DO 100 MM = 1,MBLK                                                         PRT 11C
M2 = M1 + 23                                                                PRT 12C
IF (M2.GT.NSTR) M2 = NSTR                                                  PRT 13C
WRITE (IIP6,2000) HEAD,IPG,NELM,(J,J=N1,N2)                              PRT 14C
IPG = IPG + 1                                                                PRT 15C
DO 50 I = M1,M2                                                            PRT 16C

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50  WRITE (IIPB,2002)  I,LD(1),(ESTIF(I,J),J=N1,N2)
100  M1 = -2 * J
    WRITE (IIPB,2001)  (FORCE(J),J=N1,N2)
200  N1 = N2 * I
    RETURN
2000  FORMAT (1H1,12A6,3LX,4HPAGE,14//5X,7HELEMENT,15,7H MATRIX//
X 2X,7HROW/COL,3X,7H112)
2001  FORMAT (/3X,5HFORCE,4X,1P9E12.3)
2002  FORMAT (/14,3H LD,15,1P9E12.3)
    END
*DECK BLOADS
SUBROUTINE HLOADS(NLD,NUMNP,NDIM,NDF,ICOD,XYZ,F)
COMMON/LABELS/ LABL(6),XHPD(3),XH,FHED(5),FH,UHED(5),UH,KHED(5),KH
X ,ANORD1,ANORD2,ANORD3,HEAD(12),START,CEASE,IPI,NSTR,NORD(30)
COMMON/SHAPE/ XJAC,SHAPE(4,20),SG(3,3),SN(3,3),X(3,20),LD(120)
COMMON /TAPES/ ITPS,IIPB
DIMENSION ICOD(1),XYZ(NDIM,1),F(NDF,1),IPRES(8),PR(8),SW(5),
X FS(5,4),INC(4)
DATA SW/5MSLD01,5MSLD02,5MSLD03,5MSLD04,5MSLD05/
MCT = 0
DO 92 J = 1,NLD
  READ (IIPB,100)  NDP,SLD,NRT,IPRES,INC,PR
  DO 200 I = 1,N
    IF (IPRES(I),EQ,0) GO TO 300
    JM = IPRES(I)
    IF (JM,GT,NUMNP,OR,JM,LT,0) GO TO 910
    JM = JM + NRT*INC(I)
    IF (JM,GT,NUMNP,OR,JM,LE,0) GO TO 920
200  NPRES = I
C... CHECK FOR MATCH OF LOAD TYPE
300  DO 400 JM = 1,5
    JMM = JM
    IF (SLD,EG,SW(JM)) GO TO 500
400  CONTINUE
    WRITE (IIPB,2000) SLD
    IPI = 0
    RETURN
500  DO 520 I = 1,NPRES
    J = IPRES(I)
    DO 510 K = 1,NDIM
      X(K,I) = XYZ(K,J)
    DO 520 K = 1,NDF
      FS(K,I) = 0
      GO TO (1,2,3,4,5),JMM
1    CALL SLD01(NDIM,NDF,NDP,NPRES,IPRES,PR,XYZ,FS)
      GO TO 600
2    CALL SLD02(NDIM,NDF,NDP,NPRES,IPRES,PR,XYZ,FS)
      GO TO 600
3    CALL SLD03(NDIM,NDF,NDP,NPRES,IPRES,PR,XYZ,FS)
      GO TO 600
4    CALL SLD04(NDIM,NDF,NDP,NPRES,IPRES,PR,XYZ,FS)
      GO TO 600
5    CALL SLD05(NDIM,NDF,NDP,NPRES,IPRES,PR,XYZ,FS)
600  MCT = MCT + NPRES = 2
    IF (MCT,GT,5) GO TO 610
    WRITE (IIPB,2000) HEAD,IPI,(FHED(I),FH,I=1,NDF)
    IPI = IPI + 1
    MCT = 5
610  WRITE (IIPB,2001) NDP,SLD
    DO 620 I = 1,NPRES
620  WRITE (IIPB,2002) IPRES(I),PR(I),(FS(J,I),J=1,NDF)
C... ADD TO THE GENERALIZED FORCES
    DO 740 I = 1,NPRES
      K = IPRES(I)
      IC = ICOD(K)
      IL = 10.000
      DO 730 J = 1,NDF
        IF (IC,LT,IL) GO TO 720
710  IC = IC - IL
        IF (IC,GE,IL) GO TO 710
        GO TO 730
720  F(J,K) = F(J,K) + FS(J,I)

```

730	IL = IL/10	HL0 620
740	CONTINUE	HL0 630
	IF (NRT.LF.0) GO TO 900	HL0 640
	NRT = NRT - 1	HL0 650
	DO 800 I = 1,8	HL0 660
800	IPRES(I) = IPRES(I) + INC(I)	HL0 670
	GO TO 500	HL0 680
900	CONTINUE	HL0 690
	RETURN	HL0 700
910	WRITE(IIP6,2031) NDP,SLD,IPRES	HL0 710
	GO TO 930	HL0 720
920	WRITE(IIP6,2031) NDP,SLD,IPRES,NRT,INC	HL0 730
930	IFG = 0	HL0 740
	RETURN	HL0 750
C....	FORMAT STATEMENTS	HL0 760
1000	FORMAT(1F,1X,A5,14,B15,B13/8F10.0)	HL0 770
2000	FORMAT(1H1,12A6,30X,4HPAGE,14//5X,26HDISTRIBUTED BOUNDARY LOADS//	HL0 780
	1 8X,4HNODE, 9X,4HLOAD,6(1X,2A6))	HL0 790
2001	FORMAT(1E,13H DIMENSIONAL ,A5,5H LOAD)	HL0 800
2002	FORMAT(1I0,1P7E13.4)	HL0 810
2030	FORMAT(17H **FATAL ERROR** ,A5,16H LOAD CARD ERROR/1X)	HL0 820
2031	FORMAT(21H **FATAL ERROR** NDP=,14,3X,A5,3X,5HNODES,B15/15X,15,	HL0 830
	1 21H INCREMENTS WITH INC=,B15,24H WILL CAUSE NODAL ERROR**)	HL0 840
	END	HL0 850
*DECK	SLD01	SLD 10
	SUBROUTINE SLD01(NDIM,NDF,NDP,NPRES,IPRES,PR,XZ,FS)	SLD 20
	DIMENSION IPRES(8),PR(8),FS(6,8)	SLD 30
	COMMON/GAUSS/ LIM,SGAUSS(5,5),*GAUSS(5,5)	SLD 40
	COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)	SLD 50
	NSIDE = NPRES - 1	SLD 60
	IF (NDP.EQ.2) NSIDE = NPRES/4	SLD 70
	LIM = NSIDE + 1	SLD 80
C....	COMPUTE NORMAL PRESSURE GENERALIZED FORCES	SLD 90
	LIM3 = LIM	SLD 100
	IF (NDP.EQ.1) LIM3 = 1	SLD 110
	NDP = NDP + 1	SLD 120
	DO 404 IT = 1,LIM3	SLD 130
	SS = SGAUSS(IT,LIM)	SLD 140
	WS = *GAUSS(IT,LIM)	SLD 150
	IF (LIM3.NE.1) GO TO 401	SLD 160
	SS = -1.0	SLD 170
	WS = 1.0	SLD 180
401	CONTINUE	SLD 190
	DO 404 JJ = 1,LIM	SLD 200
	TT = SGAUSS(JJ,LIM)	SLD 210
	WT = *GAUSS(JJ,LIM)*WS	SLD 220
	CALL BRICK2(TT,SS,-1.,NDIM,NPRES,NSIDE)	SLD 230
	PN = 0.	SLD 240
	DO 402 I = 1,NPRES	SLD 250
402	PK = PR(I) + PR(I)*SHAPE(4,I)	SLD 260
	PK = PK*WT	SLD 270
	DO 405 J=1,NDP	SLD 280
405	FS(J,I) = FS(J,I) + SG(NDP,J)*SHAPE(4,I)*PN	SLD 290
404	CONTINUE	SLD 300
	NDP = NDP - 1	SLD 310
	RETURN	SLD 320
	END	SLD 330
*DECK	SLD02	SLD 10
	SUBROUTINE SLD02(NDIM,NDF,NDP,NPRES,IPRES,PR,X,FS)	SLD 20
	DIMENSION PR(3,4),IPRES(8),PR(8),X(3,20),FS(6,8)	SLD 30
C....	DATA AR/-3.,-4.,1.,4.,0.,-4.,-1.,4.,3./	SLD 40
	ADD HORIZONTAL AND VERTICAL COMPONENT STRESS LOADS	SLD 50
	DO 600 K=1,3	SLD 60
	DO 600 L=1,3	SLD 70
	FS(1,K) = FS(1,K) + PR(K,L)*X(2,L)*PR(1)/6.0	SLD 80
600	FS(2,K) = FS(2,K) + PR(K,L)*X(1,L)*PR(2)/6.0	SLD 90
	RETURN	SLD 100
	END	
*DECK	SLD03	SLD 10
	SUBROUTINE SLD03(NDIM,NDF,NDP,NPRES,IPRES,PR,XZ,FS)	SLD 20
	DIMENSION IPRES(8),PR(8),FS(6,8)	

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COMMON/SHAPE/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)
NSIDE = NPRES = 1
LIM = NSIDE * 1
C.... COMPUTE NODAL PRESSURE GENERALIZED FORCES
DO 404 JJ = 1,LIM
  TT = SGAUSS(JJ,LIM)
  XT = XGAUSS(JJ,LIM)
  CALL BRICK2(TT,-1.,-1.,NDIM,NPRES,NSIDE)
  PN = 0.
  RR = 0.
  DO 405 I = 1,NPRES
    RR = RR + SHAPE(4,I)*X(1,I)
402  PN = PN + PN(I)*SHAPE(4,I)
    PA = PN*XT*RR
    DO 406 I = 1,NPRES
      RR = SHAPE(4,I)*PN
406  FS(J,I) = FS(J,I) + SG(2,J)*RR
404  CONTINUE
  RETURN
END
*DECK BRICKEN
SUBROUTINE BRICKEN(NUMP,NUMEL,NDIM,NDF,NEL,NEL1,ICOD,XYZ,F,IX)
  REAL LABL
  DIMENSION ICOD(1),XYZ(NDIM,1),IX(NEL1,1),F(NDF,1)
  COMMON /IAPES/ ITP5,ITP6
  COMMON /LABELS/ LABL(6),XHED(3),XH,FHED(6),FH,UHED(6),UH,RHED(6),RH
  X,AWOH1,AWOH2,AWOH3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)
  COMMON/SHAPE/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)
C
C.... GENERATION OF NODAL COORDINATES AND BOUNDARY CODES FOR BRICK SHAPE
C.... REGIONS USING ELEMENT SHAPE FUNCTIONS * * *
C
C.... INPUT THE REGION PROPERTIES
C
  READ(ITP5,1000) NN,NP,NS,NT,NI,NE,MA,NBC,IREUSE,IPRINT,IMSH,IELM
  READ(ITP5,1001) I1,I2,I3,I4,I5,I6
  READ(ITP5,1002) ((X(I,J),I=1,3),J=1,NN)
  IF(NS.LE.0) NS=1
  IF(NR.LE.0) NR=1
  IF(NT.LE.0) NT=1
  IF(NI.LE.0) NI=1
  IF(NE.LE.0) NE=1
  IF(MA.LE.0) MA=1
  WRITE(ITP6,2000) HEAD,IPG,NR,NS,NT,NI,NE,MA
  WRITE(ITP6,2100) I1,I2,I3,I4,I5,I6,(XHED(I),XH,I=1,NDIM)
  DO 10 N = 1,NN
10  WRITE(ITP6,2001) (X(I,N),I=1,NDIM)
C
C.... SET THE CONTROL CONSTANTS
C
  NSIDE = (NN+1)/NDIM**2
  DR = 2./FLOAT(NR)
  DS = 2./FLOAT(NS)
  DT = 2./FLOAT(NT)
  NR = NR + 1
  NS = NS + 1
  NT = NT + 1
  IF(NDIM.EQ.2) NT = 1
  NRS = NR*NS
  NF = NRS*NT + NI - 1
  IF(NF.GT.NUMP) GO TO 400
  IF(NBC.NE.0) GO TO 30
C
C.... SET THE BOUNDARY CONDITION CODES TO ZERO
C
  DO 20 I = 1,NF
    DO 15 J = 1,NDF
15  F(J,I) = 0.0
  20  ICOD(1) = 0
  30  CONTINUE

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      N = NI
      ME = NE - 1
      T = -1.0
      DO 300 K = 1,NT
      S = -1.0
      DO 200 J = 1,NS
      R = -1.0
      DO 100 I = 1,NR
      CALL BRICK2(R,S,T,NDIM,NN,NSIDE)
C
C.... COMPUTE THE NODAL COORDINATES OF THE N-TH NODE
C
      DO 50 L = 1,NDIM
      CC = 0.
      DO 40 M = 1,NN
      40 CC = CC + SHAPE(4,M)*X(L,M)
      50 XYZ(L,N) = CC
C
C.... SET THE SURFACE BOUNDARY CONDITIONS TO PRESCRIBED CONDITIONS
C
      IF(I.EQ.1) ICOD(N) = ICOD(N) + 11
      IF(I.EQ.NR) ICOD(N) = ICOD(N) + 12
      IF(J.EQ.1) ICOD(N) = ICOD(N) + 13
      IF(J.EQ.NS) ICOD(N) = ICOD(N) + 14
      IF(K.EQ.1) ICOD(N) = ICOD(N) + 15
      IF(K.EQ.NR) ICOD(N) = ICOD(N) + 16
      N = N + 1
      IF(K.EQ.NT.AND.K.NE.1) GO TO 100
      IF(I.EQ.NR.OR.J.EQ.NS) GO TO 100
      ME = ME + 1
      IX(NEL1,ME) = MA
      IX(1,ME) = N - 1
      IX(2,ME) = N
      IX(3,ME) = N + NR
      IX(4,ME) = N + NR - 1
      IF(NDIM.EQ.2) GO TO 100
      IX(5,ME) = N + NRS - 1
      IX(6,ME) = N + NRS
      IX(7,ME) = N + NRS + NR
      IX(8,ME) = N + NRS + NR - 1
      100 R = R + DR
      S = S + DS
      200 T = T + DT
      300 IF(IREUSE.EQ.0) GO TO 360
      J = (IREUSE - 1)*100 + MA
      DO 320 I = NE,ME,IREUSE
      320 IX(NEL1,I) = J
      360 IF(IPRINT.NE.0) IX(NEL1,NE) = IX(NEL1,NE) + 10
      IF(IMSH.EQ.0) CALL PRIMSH(NI,NF,NDIM,ICOD,XYZ)
      IF(IELM.EQ.0) CALL PRTELM(NE,ME,NEL,NEL1,IX)
      IF(IME.GT.NUMEL) NUMEL = ME
      RETURN
      400 WRITE(11P6,2030)
      1PG = 0
      RETURN
      1000 FORMAT(1A15)
      1001 FORMAT(6I10)
      1002 FORMAT(10X,3F10.0)
      2000 FORMAT(1H1,12A6,3X,4HPAGE,14//17H NODE GENERATIONS//
      X 10X,25HNUMBER OF R-INCREMENTS +15/
      X 10X,25HNUMBER OF S-INCREMENTS +15/
      X 10X,25HNUMBER OF T-INCREMENTS +15/
      X 10X,25HFIRST NODES NUMBER +15/
      X 10X,25HFIRST ELEMENT NUMBER +15/
      X 10X,25HELEMENT MATERIAL NUMBER +15/ 1X)
      2100 FORMAT(
      X 10X,25H1-FACE BOUNDARY CODE +110/
      X 10X,25H2-FACE BOUNDARY CODE +110/
      X 10X,25H3-FACE BOUNDARY CODE +110/
      X 10X,25H4-FACE BOUNDARY CODE +110/
      X 10X,25H5-FACE BOUNDARY CODE +110/
      X 10X,25H6-FACE BOUNDARY CODE +110//5X,3(2A6))

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HRK 50C
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HRK 53C
HRK 54C
HRK 55C
HRK 56C
HRK 57C
HRK 58C
HRK 59C
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HRK108C
HRK109C
HRK110C
HRK111C
HRK112C
HRK113C
HRK114C
HRK115C
HRK116C
HRK117C
HRK118C
HRK119C
HRK120C
HRK121C

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2001 FORMAT(5X,1P,F12.3)
2030 FORMAT(5X,47H **FATAL ERROR** INSUFFICIENT STORAGE FOR NODES)
END
*CHECK
ELOADS
SUBROUTINE ELOADS(NLD,NDIM,NDF,NEN,NEL1,NSTF,NSIZV,NVEC,TYPE,D,
1 ICOD,XYZ,IX,F,FORCE,ESTIF,VECT)
COMMON/LABELS/ LABL(6),XHED(3),XH,FHED(6),FH,UHED(6),UH,RHED(6),RH
X,AWORD1,AWORD2,AWORD3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)
COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)
COMMON /TAPES/ ITP5,ITP6
DIMENSION TYPE(1),ICOD(1),XYZ(NDIM,1),IX(NEL1,1),F(NDF,1),
1 FORCE(NDI,1),PR(6)
C.... LOADINGS ASSOCIATED WITH ELEMENTS
DO 500 M = 1,NLD
READ(ITP5,1000) IEL,ELO,INC,JEL,PR
INC = IABS(INC)
IF(INC.LE.0) INC = 1
IF(JEL.EQ.0) JEL = IEL
IF(JEL.GT.100) GO TO 120
IC = IEL
IEL = JEL
JEL = IC
120 N = IEL
DO 100 JM = 1,30
IF(ELO.EQ.WORD(JM)) GO TO 200
100 CONTINUE
DM = 0
WRITE(ITP6,2030) ELO,DM,N
IPG = 0
RETURN
200 DO 500 N = IEL,JEL,INC
MA = MOD(IX(NEL1,N),10)
DM = TYPE(MA)
IF(DM.NE.ELO) GO TO 110
DO 310 I = 1,NSTF
310 FORCE(I,1) = 0
DO 330 I = 1,NEN
K = IX(1,N)
IF(K.EQ.0) GO TO 330
LD(I) = K
NEL = I
DO 320 J = 1,NDIM
320 X(J,1) = XYZ(J,K)
330 CONTINUE
CALL ELMINH(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,UM,D,XYZ,
1 IX,F,FORCE,ESTIF,PR,VECT,5)
C.... ADD FORCES TO THE GENERALIZED FORCES
MCT = MCT - NEL - 2
IF(MCT.GT.0) GO TO 350
WRITE(ITP6,2000) HEAD,IPG,(FHED,FH,I=1,NDF)
IPG = IPG + 1
MCT = 50
350 WRITE(ITP6,2001) N,(I,PR(I),I=1,6)
DO 400 I = 1,NEL
K = LD(I)
IC = ICOD(K)
IL = 100000
DO 430 J = 1,NDF
IF(IC.LI.IL) GO TO 420
440 IC = IC - IL
IF(IC.GE.IL) GO TO 440
GO TO 430
420 F(J,K) = F(J,K) + FORCE(J,I)
430 IL = IL/10
410 CONTINUE
400 WRITE(ITP6,2002) K,(FORCE(J,1),J=1,NDF)
500 CONTINUE
600 CONTINUE
RETURN
1000 FORMAT(15,1X,A5,I4,I5,6F10.0)
2000 FORMAT(11H,12A6,30X,4HPAGE,14//5X,13HELEMENT LOADS//6X,4MNODE,
1 6(1X,2A6))

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2001 FORMAT(15X,7HELEMENT,15,6H LOADS,6(I3,1PE13.4,2H,))
2002 FORMAT(11J,1PAE13.4)
2030 FORMAT(17H **FATAL ERROR** ,A5,25H LOAD CARD ERROR, TYPE = , A5,
1 12H FOR ELEMENT, 15)
END
*DECK HCODES
SUBROUTINE HCODES(NUMNP,ICOD)
C
C.... BOUNDARY CODE GENERATOR
C
C.... ASSUME ICOD IS FILLED AND ONLY CHANGES ARE TO OCCUR
C.... IF NOT USER SHOULD INITIALIZE BY PLACING TWO CARDS WITH N=1 AND
C.... NX=1 ON FIRST CARD AND N= NUMNP AND NX=0 ON SECOND
C
COMMON/LABELS/ LABL(6),XHED(3),XH,FHED(6),FH,UMED(6),JH,MHED(6),RH
X ,AWORD1,AWORD2,AWORD3,HEAD(12),START,CEASE,IPG,NSIR,WORD(30)
COMMON/ITPES/ ITP5,ITP6
DIMENSION ICOD(1),IBC(6),NHC(6)
N = 0
NX = 0
100 NP = N
NXP = NX
READ(ITP5,1000) N,NX,IBC
IF(N.LE.(.04*N.GT.NUMNP)) GO TO 500
IT = IABS(IBC(1))
DO 200 I = 2,6
IT = 10*IT + IABS(IBC(I))
200 ICOD(N) = IT
IF(NXP.EQ.0) GO TO 100
IF((N-NP)*NXP.GE.0) GO TO 300
NXP = - NXP
WRITE(ITP6,2030) NP,N
300 NP = NP + NXP
IF(NXP.GT.0.AND.NP.GE.0) GO TO 100
IF(NXP.LT.0.AND.NP.LE.0) GO TO 100
IT = 0
IF(IBC(1).LT.0) IT = 1
DO 400 I = 2,6
IT = 10*IT
IF(IBC(I).LT.0) IT = IT + 1
400 CONTINUE
ICOD(NP) = IT
GO TO 300
C
C.... OUTPUT THE NONZERO CODES
C
500 MCT = 0
J = 0
DO 600 N = 1,NUMNP
IF(ICOD(N).EQ.0) GO TO 600
J = J + 1
NHC(J) = N
IBC(J) = ICOD(N)
IF(J.LT.4.AND.N.LT.NUMNP) GO TO 600
MCT = MCT + 1
IF(MCT.GT.6) GO TO 610
WRITE(ITP6,2030) HEAD,IPG
IPG = IPG + 1
MCT = 0
610 WRITE(ITP6,2001) (NHC(I),IBC(I),I=1,J)
J = 0
600 CONTINUE
RETURN
C
C.... FORMATS
C
1000 FORMAT(1A15)
2000 FORMAT(1H1,12A5,3U,4HPAGE,14//23H NONZERO BOUNDARY CODES//
1 6(5A,12HPNODE/H CODE,3X))
2001 FORMAT(1A(11,1H,17,3X))
2030 FORMAT(49H **NON FATAL ERROR** NODE INCREMENT BETWEEN NODES,15,
1 4H AND,15,34H OF INCORRECT SIGN, SIGN CHANGED**)

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END	HCU 67C
*DECK VECIN	
SUBROUTINE VECIN(NUMNP,NUMEL,NDF,NSIZV,IPICK,N ,PRIV,VECT)	VEC 1C
C	VEC 2C
C.... VECIN ALLOWS INPUT OF USER DEFINED VECTORS.	VEC 3C
C	VEC 4C
C.... IPICK IS A CODED PARAMETER AS FOLLOWS.	VEC 5C
C.... IPICK = 0, IMPLIES VECTORS ARE ASSOCIATED WITH NODES	VEC 6C
C.... IPICK = 1, IMPLIES VECTORS ARE ASSOCIATED WITH DEGREES FREEDOM	VEC 7C
C.... IPICK = 2, IMPLIES VECTORS ARE ASSOCIATED WITH ELEMENTS	VEC 8C
C	VEC 9C
DIMENSION DX(7),PRIV(2,N),VECT(NSIZV,N),VWORD(6)	VEC 10C
REAL LAM	VEC 11C
COMMON /TAPES/ ITP5,ITP6	VEC 12C
COMMON/LABELS/ LABL(6),XMHED(3),XMHED(6),FM,UMED(6),UMHED(6),RM	VEC 13C
X ,AWORD1,AWORD2,AWORD3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)	VEC 14C
DATA VWORD/CHNODAL ,6HPOINTS,6HDEG , F,6HFREEDOM,6HNO , EL,6HELEMENTS /	VEC 15C
DATA BLANK,FO,PRINT / 6H , 6H NO,6H PRINT /	VEC 16C
C	VEC 17C
C.... INPUT HEADER DESCRIPTIONS	VEC 18C
READ(1,TP5,1000) ((PRIV(I,J),I=1,2),J=1,N)	VEC 19C
IPRINT = 0	VEC 20C
DO 10 I=1,2	VEC 21C
DO 10 J=1,N	VEC 22C
IF(PRIV(I,J) .NE. BLANK) IPRINT = 1	VEC 23C
10 CONTINUE	VEC 24C
IF(IPRINT .EQ. 1) GO TO 11	VEC 25C
PRIV(1,1) = FO	VEC 26C
PRIV(2,1) = PRINT	VEC 27C
11 CONTINUE	VEC 28C
DO 12 I = 1,NSIZV	VEC 29C
DO 12 J = 1,N	VEC 30C
12 VECT(I,J) = 0.0	VEC 31C
MCT = 0	VEC 32C
L = 3	VEC 33C
IF(IPICK .LT. 0 .OR. IPICK .GT. 2) GO TO 450	VEC 34C
IP1 = 2*IPICK + 1	VEC 35C
IP2 = 2*IPICK + 2	VEC 36C
NSKIP = 1	VEC 37C
IF(IPICK .EQ. 1) NSKIP = NDF	VEC 38C
I = 0	VEC 39C
NXP = 0	VEC 40C
200 IP = 1	VEC 41C
NX = NXP	VEC 42C
HEAD(1,TP5,1000) I,NXP,(DX(J),J=1,N)	VEC 43C
IF(I.LE.0 .OR. I.GT.NSIZV) GO TO 290	VEC 44C
DO 21 J = 1,N	VEC 45C
210 VECT(I,J) = DX(J)	VEC 46C
IF(NX.EQ.0) GO TO 200	VEC 47C
IF((I-IP)*NX.GE.0) GO TO 220	VEC 48C
NX = -NX	VEC 49C
220 LX = (IABS(I-IP)+IABS(NX)-1)/IABS(NX)	VEC 50C
DO 23 J = 1,N	VEC 51C
230 DX(J) = (VECT(I,J)-VECT(IP,J))/LX	VEC 52C
240 IP = IP + NX	VEC 53C
IF(NX.GT.0 .AND. IP.GE.I) GO TO 200	VEC 54C
IF(NX.LT.0 .AND. IP.LE.I) GO TO 200	VEC 55C
DO 25 J = 1,N	VEC 56C
250 VECT(IP,J) = VECT(IP-NX,J) + DX(J)	VEC 57C
GO TO 240	VEC 58C
290 IF(IPRINT.EQ.0) GO TO 500	VEC 59C
C.... OUTPUT VECTORS	VEC 60C
IK = 0	VEC 61C
DO 300 K = 1,NSIZV	VEC 62C
KK = (K-1)/NSKIP	VEC 63C
KK = KK*NSKIP + 1	VEC 64C
IF(KK .EQ. K) IK = IK + 1	VEC 65C
MCT = MCT + 1	VEC 66C
IF(MCT.GT.0) GO TO 300	VEC 67C
WRITE(1,TP6,2000) HEAD,IPG,VWORD(IP1),VWORD(IP2),NSIZV,PRIV	VEC 68C
IPG = IPG + 1	VEC 69C
MCT = 50	VEC 70C

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300 WRITE(1TP6,2001)IK,(VECT(K,J),J=1,N)
GO TO 530
400 WRITE(1TP6,3000) 1
IPG = 0
GO TO 530
450 WRITE(1TP6,3001) IPICK
IPG = 0
500 CONTINUE
IF(IPHINT.EQ. 0)WRITE(1TP6,2000)HEAD,IPG,VWORD(IP1),VWORD(IP2),
X NSIZV,PRIV
RETURN
C.... FORMATS
1000 FORMAT(6X,2A6)
1001 FORMAT(2IS,7F10.0)
2000 FORMAT(1H1,12A6,30X,4HPAGE,14//5X,
X 31H INPUT VECTORS ASSOCIATED WITH ,2A6,5X,16H VECTOR LENGTH =15//
X 4X,6HNUMBER, 7(3X,2A6)/1X)
2001 FORMAT(110,7E15.5)
3000 FORMAT(25HVECTORS CARD ERROR, NODE=,I5)
3001 FORMAT(32H VECTOR PARAMETER ERROR, IPICK = 15)
END
*DECK PLUT2D
SUBROUTINE PLUT2D(ISX,KSTEPS,HED,NUMNP,NUMEL,ICOD,XYZ,IX,VECT,
X NVEC,NSIZV,NEL1,INICD,TIME,V,NDF)
COMMON/MJINH/NEW(1500)
COMMON/PLUTS/PL01
DIMENSION NSTEPS(2),PRSCAN(2),TPSCAN(2),IDUM(10),HED(12),ICOD(1),
X XYZ(NDF,1),IX(NEL1,1),VECT(1),V(NDF,1)
DATA INPUTCK,RA,NSTEPS,PRSCAN,TPSCAN,IDUM,DUM
X/0.0,0.0,0.0,0.0,0.0,0.0,1.0,1.0,10*0.0,0.0/
C
IF(PL01.EQ.3) GO TO 100
C
C PLUTING FOR TWO DIMENSIONAL PROBLEMS WITH FOUR NODE QUADS
C
IF(ISX.GE. 1) GO TO 50
C
C WRITE MESH DATA TO TAPE 3)
C
NUMMAT = TIME
NSTEPS(1)=KSTEPS
NSTEPS(2)=KSTEPS
DO 20 N=1,NUMNP
VECT(N)=0.0
IC=ICOD(N)
IL=100000
K=0
DO 10 J=1,2
5 IF(IC.LT.IL) GO TO 10
IC=IC-IL
IF(IC.GT.IL) GO TO 5
K=K+1
VECT(N)=J
IF(K.EQ.2) VECT(N)=3
10 IL=IL/10
20 CONTINUE
C
NF=NEL1-1
DO 30 N=1,NUMEL
RA=MOD(IX(NEL1,N),10)
30 IX(NEL1,N)=RA
C
WRITE(30) (HED(I),I=1,8),NUMNP,NUMEL,NUMMAT,INPUTCK,RA,NSTEPS,
XPRSCAN,TPSCAN,IDUM,DUM,DUM
WRITE(30) (VECT(N),XYZ(1,N),XYZ(2,N),N=1,NUMNP),((IX(I,N),I=1,5),
XN=1,NUMEL),DUM,DUM
C
DO 40 N=1,NUMEL
40 IX(NEL1,N)=0
C
MM=0
RETURN

```

VEC 71C  
VEC 72C  
VEC 73C  
VEC 74C  
VEC 75C  
VEC 76C  
VEC 77C  
VEC 78C  
VEC 79C  
VEC 80C  
VEC 81C  
VEC 82C  
VEC 83C  
VEC 84C  
VEC 85C  
VEC 86C  
VEC 87C  
VEC 88C  
VEC 89C  
VEC 90C  
VEC 91C



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C
C WRITE SYSTEM RESPONSE RECORDS FOR EACH TIME STEP
C
50 IF (ISW.EQ.1) GO TO 75
   NMM = NMM + 1
   NMM = NMM
   IF (NICO.EQ.3) GO TO 70
   NMM = -NMM
   WRITE(30) NMM,TIME,(V(1,N),V(2,N),N=1,NUMNP)
   GO TO 91
70 WRITE(30) NMM,TIME,(V(1,N),V(2,N),N=1,NUMNP),(V(1,NUMNP*N),
XV(2,NUMNP*N),N=1,NUMNP),(V(1,2*NUMNP*N),V(2,2*NUMNP*N),N=1,NUMNP)
   GO TO 91
75 CONTINUE
   WRITE(30) NMM,TIME,(VECT(N),VECT(NUMEL*N),VECT(2*NUMEL*N),
XVECT(3*NUMEL*N),N=1,NUMEL)
   WRITE(30) NMM,TIME,(VECT(4*NUMEL*N),VECT(5*NUMEL*N),
XVECT(6*NUMEL*N),VECT(7*NUMEL*N),N=1,NUMEL)
90 CONTINUE
   RETURN
C
100 CONTINUE
   NDIM = IPLOT
   NEL = NELL-2
C
C THREE DIMENSIONAL PLOTTING PORTHOLE....
C
C HED(12) ... TITLE OF PROBLEM
C NUMNP ... NUMBER OF NODAL POINTS
C NUMEL ... NUMBER OF ELEMENTS
C KSTEPS ... NUMBER OF TIME STEPS
C NICO ... NUMBER OF INITIAL CONDITIONS, 1=STATIC, 3=DYNAMIC.
C NEL ... NUMBER OF NODES PER ELEMENT
C NELL ... NEL * 2 ( MATERIAL LOCATION IN CONNECTIVITY ARRAY)
C NDIM ... SPATIAL DIMENSION OF PROBLEM (2 OR 3)
C NSIZV ... NUMBER OF STRESS/STRAIN COMPONENTS, 6.
C ICOD(NUMNP) ... BOUNDARY CODE, SIX DIGIT HOOLIAN NUMBER.
C XYZ(NDIM,NUMNP) ... CARTESIAN COORDINATES
C IX(NEL1,NUMEL) ... CONNECTIVITY ARRAY AND MATERIAL PROPERTY NO.
C V(NDIM,NUMNP*NICO) ... DISPLACEMENT VECTOR IF STATIC, ALSO
C VELOCITY AND ACCELERATION IF DYNAMIC.
C VECT(2,6) ... STRESS AND STRAIN AT ELEMENT CENTER.
C ORDER IS ... (XX),(XY),(XZ),(YY),(YZ),(ZZ)
C
IF (ISW.EQ.1) GO TO 200
C
C MESH DATA RECORDS FIRST.
C
   NUMMAT = TIME
   WRITE(30) (HED(I),I=1,12),NUMNP,NUMEL,NUMMAT,KSTEPS,NICO,NDIM,NELL
   WRITE(30) (ICOD(N),XYZ(K,N),K=1,NDIM),N=1,NUMNP)
X (IX(I,N),I=1,NEL),IX(NEL1,N),N=1,NUMEL),(NEW(I),I=1,1500)
   RETURN
C
200 IF (ISW.GT.1) GO TO 250
C
C RESPONSE DATA ... DISPLACEMENTS,AND (VELOCITIES AND ACCELERATIONS
C IF DYNAMIC).
C
   WRITE(30) TIME,(((V(I,N*(K-1)*NUMNP),I=1,NDIM),N=1,NUMNP),K=1,NICO)
   RETURN
C
C STRESS AND STRAIN AT ELEMENT CENTER FOR EACH ELEMENT RECORD
C
250 CONTINUE
   WRITE(30) (VECT(J),J=1,NSIZV),(VECT(J+NSIZV),J=1,NSIZV)
   RETURN
END
*DECK SOLVEQ
SUBROUTINE SOLVEQ(NUMNP,NUMEL,NDF,ID1,M7,MB,ITA,NSTF,ISZA,NEGB,
X IHLK,DT,F,U,IDEST,FORCE,ESTIF,FD,NDEG)
   DIMENSION DT(M7),U(1),IDEST(1),F(1),FD(NEGB)

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SOL 1C
SOL 2C
SOL 3C

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COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)
C**** ADD ELEMENT STIFFNESS TO GLOBAL STIFFNESS
IB = ID1 + ISZA - NEQB
IF (IBLK.GT.0)
XCALL MAKEA(DT(ID1),DT(IB),ISZA,NEQB,ITA,NUMEL,NSTF,NSTF,LD,FORCE,
X ESTIF,7,IBLK)
C**** SOLVE EQUATIONS
CALL USOL(DT,DT(ID1),FD,NEQB,MB,1,IBLK,ISZA,ITA,R,10,11,10)
C**** PUT SOLUTION VECTOR INTO U(I)
L = NDEG
N = NDEG
M = NDEG*NUMAP
IF (IBLK.EQ.0) GO TO 200
N = NEQB - IBLK*NEQB + NDEG
REWIND 7
REWIND 10
READ (7) DT
C**** PUT THE SOLUTION BACK IN ORDER
100 READ(10) (FU(I),I=1,N)
200 IF (IDEST(M).NE.L) GO TO 250
IF (IBLK.GT.0) U(M) = FU(N)
IF (IBLK.EQ.0) U(M) = U(N)
L = L - 1
IF (L.EQ.0) L = -1
N = N - 1
GO TO 300
U(M) = F(M)
300 M = M - 1
IF (M.LE.0) GO TO 400
IF (N.GT.0) GO TO 200
IF (IBLK.EQ.0) GO TO 250
N = NEQB
GO TO 100
400 CONTINUE
IF (IBLK.GT.0) REWIND 7
IF (IBLK.GT.0) READ(7) DT
RETURN
END
*DECK COREIN
SUBROUTINE COREIN(A,U,NEQ,ESTIF,FORCE,LD,NSTF)
DIMENSION A(NEQ,1),U(1),ESTIF(NSTF,NSTF),FORCE(NSTF,2),LD(1)
DO 200 I = 1,NSTF
K = LD(I)
IF (K.EQ.0) GO TO 200
U(K) = U(K) + FORCE(I,1)
DO 190 J = 1,NSTF
L = LD(J)*K+1
IF (L.GT.0) A(K,L) = A(K,L) + ESTIF(I,J)
190 CONTINUE
200 CONTINUE
RETURN
END
*DECK MAKEA
SUBROUTINE MAKEA(A,B,ISZA,NEQB,ITA, NELS,IELM,JELM,LD,FORCE,ESTIF,
X ITEL,IBLK)
C**** ASSEMBLY ROUTINE FOR ELEMENT STIFFNESS MATRICES * MODIFIED *
DIMENSION A(ISZA),LD(IELM),FORCE(IELM,2),ESTIF(IELM,IELM),R(1)
REWIND ITA
MB = SQRT(FLOAT(IBLK))
NEQB = NEQB*MB
MB = MB - 1
NR = 0
N10 = 0
N11 = 0
IL0 = 1
DO 400 N = 1,IBLK
C**** INITIALIZE RECORD
DO 20 I = 1,ISZA
20 A(I) = 0.0
IUPH = IL0 + NEQB - 1
ITAP = 10
NEL = N10

```

IF (MM.GI.0) GO TO 30	MAK 200
ITAP = ITEL	MAK 210
NFL = NELM	MAK 220
N10 = 0	MAK 230
REWIND 10	MAK 240
30 CONTINUE	MAK 250
REWIND ITAP	MAK 260
IF (ITAP.EQ.ITEL) READ(ITAP) ZIP	MAK 270
DO 290 K = 1,NFL	MAK 280
HEAD(ITAP) ESTIF,FORCE,LD	MAK 290
DO 200 I=1,JFLM	MAK 300
NR = LD(I)	MAK 310
IF (NR.LI.LLOW.OR.NR.GI.IUPR) GO TO 200	MAK 320
NR = NR - NII	MAK 330
B(NR) = B(NR) + FORCE(I,1)	MAK 340
DO 100 J=1,JELM	MAK 350
NC = LD(J)	MAK 360
IF (NC.LI.NR) GO TO 100	MAK 370
ICELL = (NC - NR)*NEGB + MR	MAK 380
A(ICELL) = A(ICELL) + ESTIF(I,J)	MAK 390
100 CONTINUE	MAK 400
200 CONTINUE	MAK 410
IF (MM.GI.0) GO TO 260	MAK 420
IF (MM.LE.0) GO TO 260	MAK 430
DO 220 I = 1,JELM	MAK 440
NR = LD(I) - NII	MAK 450
IF (NR.GI.NEGB.AND.NR.LE.NEMB) GO TO 240	MAK 460
220 CONTINUE	MAK 470
GO TO 260	MAK 480
240 WRITE(11) ESTIF,FORCE,LD	MAK 490
N10 = N10 + 1	MAK 500
260 CONTINUE	MAK 510
300 CONTINUE	MAK 520
XXXX	
C WRITE QUICKIO FILE	
XXXX	
CALL QUICKIO (6LTAPE11,1,A)	MAK 540
ILOW = IUPR + 1	MAK 550
MR = MM + 1	MAK 560
IF (MM.GI.MH) MR = 0	MAK 570
400 NII = NII + NEGB	MAK 580
RETURN	MAK 590
END	
*DECK	
USOL	USO 10
SUBROUTINE USOL (A,B,MAXB,NEGB,MH,LL,NBLOCK,NSR,NORG,NHRS,NII,	USO 20
X NT,NPSI)	USO 30
IMENSION A(NSH),B(NSR),MAXB(NEGB)	
COMMON /TIMING/ TRICK,NTRI,BACKCP,NHACK	
NTRI=NTRI+1	
CALL SECOND (TTT)	
C	USO 40
NH=NEGB*MH	USO 50
NH = 0	USO 60
IF (NBLOCK.EQ.0) GO TO 110	USO 70
NC=MH*LL	USO 80
NHR=(NR-1)/NEGB+1	USO 90
INC=NC/MH-1	USO 100
NEGB2 = NEGB*NEGB	USO 110
REWIND NORG	
IXNORG=0	
XXXX	
C MAXB IS SAVED ON FILE 9 (NORG)	
C EUN BLOCKS ARE RANDOMLY READ/WITTEN IN-PLACE ON FILE 11	
XXXX	
N=1	
100 IXNORG=IXNORG+1	
IF ((N.GI.1).AND.(NHR.EQ.1)) GO TO 110	
CALL QUICKIO (6LTAPE11+2*A,IXNORG)	
110 DO 300 I=1,NEGB	USO 270
D=A(I)	USO 280
IF (D.EQ.0) GO TO 300	USO 290
DO 130 J=1,NHR,NEGB	USO 300

	IF (A(J),NE,0.) MAX = J	US0 31C
130	CONTINUE	US0 32C
	MAXH(1) = MAX	US0 33C
	JL=1+1	US0 34C
	IF (JL.GT.NEGB) GO TO 300	US0 35C
	II=1	US0 36C
	DI=1./D	
	DO 200 J=JL,NEGB	US0 37C
	II=II+NEGB	US0 38C
	IF (II.GI.NMH) GO TO 200	US0 39C
	C=A(II)*DI	
	IF (C.EQ.0.0) GO TO 200	US0 41C
	KK = J	US0 42C
	CALL REDUC6 ((MAX-II)/NEGB+1,NEGB,NEGB,C,A(II),A(KK))	
	KK=J+NMH	US0 46C
	JJ=1+NMH	US0 47C
	A(KK)=A(KK)-C*A(JJ)	
200	CONTINUE	US0 52C
300	CONTINUE	US0 53C
	IF (NHBLOCK.EQ.0) GO TO 930	US0 54C
C		
	WRITE (NORG) MAXH	
	CALL QUICKIO (6LTAPE11,3,A,IANORG)	
	IF=1+NGRG	
C		US0 56C
C	SUBSTITUTE INTO REMAINING EQUATIONS	US0 57C
C		US0 58C
	DO 800 NN=1,NMH	US0 59C
	IF (NN.NI.NHBLOCK) GO TO 800	US0 60C
	IP=IP+1	
	CALL QUICKIO (6LTAPE11,2,H,IP)	
	IL = NN*NEGB2 + 1	US0 65C
	DO 700 I=1,NEGB	US0 66C
	D = A(II)	US0 67C
	IF (D.EQ.0.0) GO TO 700	US0 68C
	II=IL	US0 69C
	MAX = MAXH(1)	US0 70C
	DI=1./D	
	DO 69 K=1,NEGB	US0 71C
	IF (II.GI.NMH) GO TO 700	US0 72C
	C=A(II)*DI	
	IF (C.EQ.0.0) GO TO 690	US0 74C
	KK = K	US0 75C
	CALL REDUC6 ((MAX-II)/NEGB+1,NEGB,NEGB,C,A(II),B(KK))	
	KK = K + NMH	US0 79C
	JJ = 1 + NMH	US0 80C
	B(KK)=B(KK)-C*A(JJ)	
690	II = II + NEGB	US0 85C
700	IL = IL + INC	US0 86C
	IF (NHB.NF.1) GO TO 750	US0 87C
	DO 740 I=1,NSH	US0 88C
740	A(II)=B(I)	US0 89C
	GO TO 800	US0 90C
750	CALL QUICKIO (6LTAPE11,3,H,IP)	
800	CONTINUE	US0 92C
	N = N + 1	US0 96C
	IF (N.LE.NHBLOCK) GO TO 100	US0 97C
	CALL SECOND (TTTT)	
	TRICP=TRICP+(TTTT-TTT)	
	NBACK=NBACK+1	
C		US0 98C
C	BACKSUBSTITUTION - RESULTS ON TAPE NRST	US0 99C
C		US0 100C
	LS=NEGB	
	NEB=NEGB*(NEB+1)	US0 102C
	NUMB=NHB*NEGB	US0 103C
	MAX=NLR	
	DO 900 I=1,MAX	US0 105C
900	H(I) = .	US0 106C
	FEW[NR NRST	US0 107C
	N = 1	US0 108C
	NH=NHBLOCK+1	



```

907 N=NN-1
BACKSPACE NORG
READ (MUCH) MAXH
BACKSPACE NORG
CALL QUICKIO (6L14PE11,2,A,NR)
K=NEH
DO 41 J=1,NOM
I=K+NEGH
R(K)=R(1)
910 K=K+1
I=NH
K=0
DO 42 J=1,NEGH
I=I+1
K=K+1
920 R(K)=R(1)
930 DO 45 I=1,NEGH
J=NEGH+1-I
MAX = MAX(R(I))
IF (A(J).EQ.0.) GO TO 955
KK=J
JJ=KK+1
IL=J+NEGH
C=R(KK)
IF (MAX.LT.IL) GO TO 950
DO 44 I=1,MAX-NEGH
C=C-R(1)*R(JJ)
940 JJ=JJ+1
950 R(KK)=C/JJ
955 CONTINUE
IF (NHLOCK.FT.) RETURN
I=0
K=0
DO 46 J=1,NEGH
K=K+1
I=I+1
960 A(I)=R(K)
WRITE (NRST) (A(I),I=1,LS)
N = N + 1
IF (N.LE.NHLOCK) GO TO 907
CALL SECOND (TIT)
BACKSPACE ACPC (TIT-TIT)
RETURN
END
*DECK TSOLVE
SUBROUTINE TSOLVE (NOMP,NOMEL,NOMMAT,NDIM,NDF,NEN,NEL,NSTF,NVEC,
1 NSIZ,NICO,NBICO,IRLK,ISZA,NEWH,MAXBAN,NUEG,IO1,MR,IBUF,DS,
2 TYPE,N,ICG,XYZ,F,IX,IUEST,VECT,FORCE,ESTIF,LP,A,DU,MAXH,H,U,DF,
3 NSIG,TIME)
REAL LAHL
LOGICAL NPH,NPL
DIMENSION IPE(1),IC(1),XYZ(NDIM*1),F(NDF*1),IX(NEL*1),
1 IDLS(NDF*1),VECT(NSIZ*1),FORCE(NSTF*2),ESTIF(NSTF,NSTF),
2 A(NFGR*1),DU(NDF*1),H(IGUF),U(NDF*1),DF(1)*DS(MB)
COMMON/DYANUX/TIME,NSIG(7),N1,NSTEP,DT,NUMPL,NEDATA(20*3),NPR,NPL
COMMON/LABELS/ LAHL(6),XHED(3),XH,FHED(6),FH,UHED(6),RH
X,AROHDI,AROHDE,AROHU3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)
COMMON/SHAPE/ XJAC,SHAPE(4*20),S(3*3),SK(3*3),X(3*20),LD(120)
COMMON/TAPES/ ITP5,ITP6
COMMON/VISDAT/DIP,NH,ISZH,DUL(6*20),UL(6*20),UDL(6*20),C6
COMMON/VIAP/ ITP13,ITP14,ITR0,ITR
COMMON/PLUIS/IPL01
COMMON /MINH/ NEW(1500)
LOGICAL VFLAG
C*** MAIN VISCELASTIC SUBROUTINE FOR THE GENERAL PROGRAM
DO 2 N=1,NEH
DO 2 J=1,NDF
2 UDL(J,N)=0.
DO 3 I=2,7
NSIG(I)=1
3 IF (IBLK.EQ.0) GO TO 4
KU = NOMP*NBICO

```

	REWIND 7	TSU 26C
	WRITE(7) DS	TSU 27C
C....	SET THE OUTPUT LABELS FOR DISPLACEMENT PRINTS	TSU 28C
4	HEAD(12) = 6H TIME=	TSU 29C
	IFLG = -NSICU	TSU 30C
	I = 2	TSU 31C
	DO 5 N = 1,NDIM	TSU 32C
	LABL(1) = AWORD1	TSU 33C
5	I = I + 1	TSU 34C
	LABL(1) = AWORD2	TSU 35C
	DTP = 0.	TSU 36C
	TIME = 0.	TSU 37C
	MH = MAXHAN + 1	TSU 38C
	NEI = NEM + 1	TSU 39C
	DO 10 N = 1,NUMEL	TSU 40C
10	IX(NE1,N) = 0	TSU 41C
C....	PUT INITIAL DATA ON TAPE	TSU 42C
	ISZH = INCF	TSU 43C
	ITRD = ITR13	TSU 44C
	ITRW = ITR14	TSU 45C
	REWIND ITRD	TSU 46C
	IF (IRLK.GT.0) WRITE(ITRD) ((U(I,J),J=1,NDF),J=1,KU)	TSU 47C
	DO 11 N = 1,ISZH	TSU 48C
11	H(N) = 0.	TSU 49C
	NTB = (NH*ISZH-1)/ISZH	TSU 50C
	IF (INTL.E.1) GO TO 13	TSU 51C
	NT = NTB*2	TSU 52C
	DO 12 N = 1,NT	TSU 53C
12	WRITE(ITRD) H	TSU 54C
13	REWIND ITRD	TSU 55C
	IF (IRLK.GT.0) READ(ITRD) DM	TSU 56C
	NEP = NUMNP + NUMNP	TSU 57C
	CH = 0.	TSU 58C
	DO 900 K = 1,NSFW	TSU 59C
	READ(ITP6,1000) DT,NTS,INT,NNI,NNF,NEI,NEF,NPROP,NFORC,BETA,DEL	TSU 60C
	IF (INTL.E.0) INT = 1	TSU 61C
	GAM = .5*DEL	TSU 62C
	WRITE(ITP6,2000) HEAD,TIME,IPG,DT,NTS,INT,NNI,NNF,NEI,NEF	TSU 63C
	IF (NCO.NE.1) WRITE(ITP6,2002) BETA,GAM	TSU 64C
	IPG = IPG + 1	TSU 65C
	IF (NCO.EG.1) GO TO 1	TSU 66C
	IF (DT.EG.0) GO TO 901	TSU 67C
	C1 = 1 - GAM/BETA	TSU 68C
	C2 = (1 - GAM/BETA/2) * DT	TSU 69C
	C3 = GAM/BETA/DT	TSU 70C
	C4 = 1 - C3/BETA	TSU 71C
	C5 = 1/BETA/DT	TSU 72C
	CA = C5/DT	TSU 73C
1	IF (NPROP.GT.0) PROP = PROPLD(TIME,NPROP)	TSU 74C
	IF (NFORC.GT.0.AND.NPROP.GT.0) WRITE(ITP6,4001)	TSU 75C
	VFLAG = .FALSE.	TSU 76C
	PROP = 1.0	TSU 77C
	DO 400 IT = 1,NTS	TSU 78C
	IF (NPROP.GT.0) PROP = PROPLD(TIME+DT,0)	TSU 79C
	IF (NFORC.GT.0) CALL RESET(-NFORC,NUMNP,NDF,F)	TSU 80C
	IF (IRLK.EG.0) GO TO 15	TSU 82C
	IF (INTL.VFLAG) GO TO 20	TSU 84C
	DO 14 I = 1,NDEG	TSU 85C
14	DE(I) = 0.0	TSU 86C
	GO TO 20	TSU 87C
15	DO 19 N = 1,NUMNP	TSU 88C
	DO 18 K = 1,NDF	TSU 89C
	J = 10EST(K,N)	TSU 90C
	IF (J.EG.0) GO TO 18	TSU 91C
	IF (VFLAG) GO TO 17	TSU 92C
	DO 17 I = 1,MAXHAN	TSU 93C
16	A(J,I) = 1.0	TSU 94C
17	DE(J) = F(K,N)*PROP	TSU 95C
18	CONTINUE	TSU 96C
19	CONTINUE	TSU 97C
20	NH = 1	TSU 98C
C....	OUTPUT THE SOLUTION VECTOR FOR THE CURRENT TIME	TSU 100C

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      IF (IPLOT.EQ.3) CALL PLOT2D(1,NTS,HEAD,NUMNP,NUMEL,ICOD,XYZ,IX,
      X VECT,NVEC,NSIZV,NEL1,NICD,TIME,U,NDF)
      NSIG(I) = 1
      IF (NN1.EQ.3) OR (NI/INT)*INT.NE.NI) GO TO 40
      IF (N.EQ.1) AND (NI.EQ.1) GO TO 40
      NSIG(I) = 0
      MCT = 0
      DO 30 IJ=NI1,NNE
      N=NN*(IJ)
      MCT = MCT + 1
      IF (MCT.GT.3) GO TO 30
      WRITE (11PB,2001) HEAD,TIME,IPG,PROP,(XHE(I),XM,I=1,NDIM),
      X (UMED(I),UM,I=1,NDF)
      IPG = IPG + 1
      MCT = 0
30  WRITE (11PB,LABL) IJ,(XYZ(I,N),I=1,NDIM),(U(I,N),I=1,NDF)
40  IF (NICD.EQ.1) GO TO 44
C... UPDATE FOR DYNAMIC PROBLEMS
      DO 42 N = 1,NUMNP
      DO 42 I = 1,NDF
      TEMP = U(I,N*NUMNP)
      DM = U(I,N*NEP)
      U(I,N*NUMNP) = C1*TEMP + C2*DM
42  U(I,N*NEP) = C4*DM - C5*TEMP
44  MR = 0
      READ (11TW)
      IF (11BLK.GT.0) WRITE (11TW) ((U(I,J),I=1,NDF),J=1,KU)
      IF (NID.GT.1) READ (11WD) H
      TEMP = 0
      MCT = 0
      DO 46 N = 1,NUMEL
      NPR = .TRUE.
      IF (N.EQ.NEL1) AND (N.LE.NEF) NPR = .FALSE.
      IF (M.EQ.1) AND (NI.EQ.1) NPR = .TRUE.
      MA = MOD(IX(NEL1,N),10)
      IF (MR.LE.0) MRR = IX(NEL1,N)/100
      IF (MR.LE.0) MR = MRR
      DO 46 I = 1,NSTF
      FORCE(I,1) = 0
      FORCE(I,2) = 0
      LD(I) = 0
      IF (MR.NE.MRR) OR (VFLAG) GO TO 60
      DO 50 J = 1,NSTF
      ESTIF(I,J) = 0.0
50  CONTINUE
      L = 0
      DO 110 I = 1,NEN
      K = IX(I,N)
      IF (K.EQ.0) GO TO 120
      NEL = I
      DO 100 J = 1,NDIM
      X(J,I) = XYZ(J,K)
      DO 110 J = 1,NDF
      IF (NICD.NE.1) UDL(J,I) = U(J,K*NEP)
      DUL(J,I) = DU(J,K)
      UL(J,1) = U(J,K)
      L = L + 1
100  LG(L) = IDEST(J,K)
110  DM = TYPE(MA)
120  C (REMOVED PAGE CHANGER)
      TEMP = DM
C... COMPUTE ELEMENT STRESSES AND UPDATE FORCES
      CALL ELMLIH(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,
      X IX,H,FORCE,ESTIF,U,VECT,6)
      IF (IPLOT.EQ.3) CALL PLOT2D(2,NTS,HEAD,NUMNP,NUMEL,ICOD,XYZ,IX,
      X VECT,NVEC,NSIZV,NEL1,NICD,TIME,U,NDF)
C... FORM STIFFNESS IF NEEDED FOR THE NEXT TIME STEP
      IF ((.NOT.VFLAG) AND (MR.EQ.MRR)) OR (VFLAG AND (IX(NEL1,N).EQ.1))
      XCALL ELMLIB(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,
      X IX,H,FORCE,ESTIF,U,VECT,3)
C... MODIFY FOR THE DISPLACEMENT B.C.
      DO 130 L = 1,NEL

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J = IX(L,N)
IC = ICOD(J)
IL = 100000
NE = NDF*(L-1)
DO 125 K = 1,NDF
NE = NE + 1
IF (DT.LT.0.0) FORCE(NE,1) = 0.0
W = F(K,J)*PROP
IF (Q.EQ.0.0) GO TO 122
IF (IC.LT.IL.AND.IBLK*GT.0) F(K,J) = 0.0
IF (IC.LT.IL) GO TO 123
IX(NE,1,N) = 1
DO 121 I = 1,NSTF
121 FORCE(I,1) = FORCE(I,1) - ESTIF(I,NE)*W
122 IC = IC - IL
IF (IC*GT.IL) GO TO 122
123 IF (IBLK*GT.0) FORCE(NE,1) = FORCE(NE,1) + W
125 IL = IL/10
130 CONTINUE
IF (VFLAG) GO TO 300
IF (IBLK*EQ.0) CALL COREIN(A,DF,NEGH,ESTIF,FORCE,LD,NSTF)
IF (IBLK*GT.0) WRITE(7) ESTIF,FORCE,LD
GO TO 4.0
300 CONTINUE
C.... ADD THE FORCE TO THE SOLUTION FOR A RESOLVE
DO 310 K = 1,NSTF
J = LU(K)
IF (J*GT.0) UF(J) = UF(J) + FORCE(K,1)
310 CONTINUE
400 CONTINUE
IF (IPLT.EQ.2) CALL PLOT2D(2,NTS,HEAD,NUMNP,NUMEL,ICOD,XYZ,IX,
X VECT,NVEC,NSIZV,NEL1,ICD,TIME,U,NDF)
IF (ITR*GT.1) WRITE(11WR) H
IF (IBLK*GT.0) WRITE(11WR) (IX(NEL,N),N=1,NUMEL)
IF (.NOT.VFLAG) CALL SOLVEQ(NUMNP,NUMEL,NDF,101,MH,MAXHAN,9,NSTF,
1 ISZA,NEGH,IBLK,A,DU,DF,IOEST,FORCE,ESTIF,MAXH,NDEG)
IF (VFLAG) CALL RESVEQ(NUMNP,NDF,MH,MAXHAN,ISZA,NEGH,IBLK,A,DF,DF,
1 IOEST,IOEST,MAXH,IFLG)
C.... UPDATE THE SOLUTION
I = ITRD
ITRD = ITRW
ITWR = I
IF (IBLK*GT.0) BACKSPACE ITRD
IF (IBLK*GT.0) READ(ITRD) (IX(NEL,N),N=1,NUMEL)
REWIND ITRD
IF (IBLK*GT.0) READ(ITRD) ((U(I,J),I=1,NDF),J=1,KU)
K = 0
DO 700 N = 1,NUMNP
DO 700 I = 1,NDF
K = K + 1
DU(I,N) = UF(K)
IF (IOEST(I,K).EQ.0) DU(I,N) = F(I,N)*PROP - U(I,N)
TEMP=DU(I,N)
IF (DT.LT.0.0) U(I,N) = 0.0
IF (ICD.EQ.1) GO TO 700
U(I,N*NUMNP)=U(I,N*NUMNP)+C3*TEMP
U(I,N*NEP)=U(I,N*NEP)+C4*TEMP
700 U(I,N)=U(I,N)*TEMP
VFLAG = .TRUE.
DTP = DT
MR = MR - 1
800 TIME = TIME + DT
900 CONTINUE
RETURN
901 WRITE(11PB,2030)
IPB = 0
RETURN
C.... FORMATS
1000 FORMAT(F15.0,8I5,2F10.0)
2000 FORMAT(1H1,(2A6,E)3,5,17X,4HPAGE,14//5X,23HTIME DEPENDENT SOLUTION
X//1X,17HTIME INCREMENT = 1PE15.4/
X 1X,17HNUMBER OF STEPS = 15/

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TS0160C  
TS0161C  
TS0162C  
TS0163C  
TS0164C  
  
TS0165C  
TS0166C  
TS0167C  
TS0169C  
TS0170C  
TS0171C  
TS0172C  
TS0173C  
TS0174C  
TS0175C  
TS0176C  
TS0177C  
TS0178C  
TS0179C  
TS0180C  
TS0181C  
TS0182C  
TS0183C  
TS0184C  
TS0185C  
TS0186C  
TS0187C  
TS0188C  
  
TS0189C  
TS0190C  
TS0191C  
TS0192C  
TS0193C  
TS0194C  
TS0195C  
TS0197C  
TS0198C  
TS0199C  
TS0200C  
TS0201C  
TS0202C  
TS0203C  
TS0204C  
TS0205C  
TS0206C  
TS0207C  
TS0208C  
TS0209C  
TS0210C  
  
TS0211C  
TS0212C  
TS0213C  
TS0214C  
TS0215C  
TS0216C  
TS0217C  
TS0218C  
TS0219C  
TS0220C  
TS0221C  
TS0222C  
TS0223C  
TS0224C  
TS0225C  
TS0226C  
TS0227C  
TS0228C



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      X 10X,14HPRINT INTERVAL = 15/
      X 10X,14HPRINT NODES 15,3H 10,15/
      X 10X,14HPRINT ELEMENTS,15,3H 10,15/
2001 FORMAT(1H1,12A6,E13.5,17X,4HPAGE,14//5X,12HNODAL VALUES,
      X 50X,4H FROM,E13.5//12H NODAL POINT,9(1X,2A6))
2002 FORMAT(11X,30HNEWARK INTEGRATION PARAMETERS/
      C10X17HETA VALUE = F6.3/
      C10X17HGAMMA VALUE = F6.3)
2130 FORMAT(5X,53H**FATAL ERROR 33** TIME STEP ZERO FOR DYNAMIC PROBLEM
      X/1X)
4001 FORMAT(10H**WARNING** BOTH THE PROPORTIONAL LOADING AND FORCE ARE
      X BEING RESET ON EACH TIME STEP)
      END
      TS0229C
      TS0230C
      TS0231C
      TS0234C
      TS0235C
      TS0236C
      TS0237C
      TS0238C
      TS0240C
      TS0241C
*DECK RESVIEW
      SUBROUTINE RESVIEW(NUMBP,NDF,M7,MB,ISZA,NEQB,IBLK,DT,F,U,IDES,IDEST
      X ,MAXB,NES)
      DIMENSION U(M7),F(NDF,1),U(1),IDES(1),IDES(1)
      NE = 1AMSNES)
      IF (IBLK.GT.0) REWIND 7
      IF (NES.LE.0) GO TO 125
      REWIND 10
      C.... SAVE THE DESTINATION VECTOR FOR UNPACKING THE SOLUTION
      WRITE(10)(IDES(1),I=1,NE)
      C.... COMPACT THE FORCE VECTOR, F, INTO U.
      DO 100 I = 1,NE
      J = IDES(1)
      IF (J.EQ.0) GO TO 100
      U(I) = F(I)
      100 CONTINUE
      C.... USE RESOLVE TO GET THE NEW SOLUTION
      125 CALL RESOLVE (DT,U,ISZA,NEQB,MB,IBLK,B,MAXB)
      C.... REEXPAND THE SOLUTION VECTOR
      IF (IBLK.EQ.0.OR.NES.LE.0) GO TO 150
      REWIND 10
      READ(10)(IDES(1),I=1,NE)
      N = NE
      DO 200 I = 1,NE
      IF (IBLK.GT.0) J = IDES(I)
      IF (IBLK.EQ.0) J = IDES(N)
      IF (J.EQ.0) U(I) = 0.
      IF (J.NE.0) U(I) = U(J)
      200 N = N - 1
      IF (IBLK.GT.0) READ(7) DT
      RETURN
      END
*DECK RESOLVE
      SUBROUTINE RESOLVE (A,U,NSH,NEQB,MB,NHLOCK,NBNS,MAXB)
      DIMENSION A(NSH),U(1),MAXB(NEQB)
      COMMON /TIMING/ ITRICH,NITR,BACKCP,NHACK
      NHACK=NHACK+1
      CALL SECOND (TTT)
      C
      C.... RESOLVE USING DECOMPOSED MATRIX
      C
      CXXXX
      C NOTE MAXB WAS STORED ON FILE NORG (9) IN SUBT USOL
      C NORG IS SET TO 9 IN THIS ROUTINE
      CXXXX
      DATA NORG/9/
      REWIND NORG
      JJ =
      DO 300 N = 1,NHLOCK
      C
      IF (NHLOCK.EQ.0) GO TO 50
      READ (NORG) MAXB
      CALL QUICKIO (6LTAPE11,2,A,N)
      C
      50 DO 200 I=1,NEQB
      JJ=JJ+1
      D=A(I)
      IF (D.EQ.0.) GO TO 200
      UJ=U(JJ)

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      IF(UJ.EQ.0.) GO TO 200
      NMH=MAXH(I)
      II=NEQB+1
      CALL REDVC6 ((NMH-II)/NEQB+1,NEQB,1,UJ/D,A(II),U(JJ+1))
200  CONTINUE
300  CONTINUE
C
C**** BACKSUBSTITUTION
C
      NN=NBLOCK*1
      JJ = NEWH*NBLOCK
      IF(NBLOCK.EQ.0) JJ = NEQB
      DO 400 N = 1,NBLOCK
C
      IF(NBLOCK.EQ.0) GO TO 350
      NA=NN-1
      BACKSPACE NORG
      READ (NORG) MAXB
      BACKSPACE NORG
      CALL QUICKIO (6LTAPE11,2,A,NN)
C
      350 IL=NEWH*1
      DO 500 I=1,NEQB
      IL=IL-1
      D=A(IL)
      IF(D.EQ.0.) GO TO 500
      UJ=U(JJ)
      II=IL+NEQB
      KK=JJ
      NMH=MAXB(IL)
      DO 400 J=II,NMH,NEQB
      KK=KK+1
      UJ=UJ-A(JJ)*U(KK)
400  CONTINUE
      U(JJ)=UJ/D
      500 JJ=JJ-1
500  CONTINUE
      CALL SECOND (TTTT)
      BACKCH=BACKCP*(TTTT=TTT)
      RETURN
      END
      *DECK PROPLD
      FUNCTION PROPLD(T,N)
      DIMENSION TABLE(9,5),ILAB(3)
      COMMON /TAPES/ ITP5,ITP6
      DATA ILAB/10HPOLYNOMIAL,10HPERIODIC ,10HUSER INPUT/
      IF(N.EQ.0) GO TO 200
C.... INPUT TABLE
      NTERMS = N
      READ(ITP5,1000) ((TABLE(I,J),I=1,9),J=1,N)
      WRITE(ITP6,2001)
      DO 100 J = 1,N
      K = TABLE(1,J)
      IF(K.LT.1.OR.K.GT.3) GO TO 700
      L = TABLE(2,J)
100  WRITE(ITP6,2000) ILAB(K),L,(TABLE(I,J),I=3,9)
      RETURN
200  PROPLD = 0.
C.... INTERPOLATE THE TABLE
      DO 300 J = 1,NTERMS
      TMIN = TABLE(3,J)
      TMAX = TABLE(4,J)
      IF(T.LT.TMIN.OR.T.GT.TMAX) GO TO 300
      K = TABLE(1,J)
      GO TO (201,202,203)*K
201  TT = 1.0
      DO 211 I = 5,9
      PROPLD = PROPLD + TABLE(I,J)*TT
211  TT = TT*1
      RETURN
C82 LPOH
202  K = TABLE(2,J)

```

RES 28C  
RES 29C  
RES 30C  
RES 31C

RES 32C  
RES 33C  
RES 34C  
RES 35C  
RES 36C

RES 40C

RES 58C

RES 59C  
RES 60C

ROPL 1C  
ROPL 2C  
ROPL 3C  
ROPL 4C  
ROPL 5C  
ROPL 6C  
ROPL 7C  
ROPL 8C  
ROPL 9C  
ROPL 10C  
ROPL 11C  
ROPL 12C  
ROPL 13C  
ROPL 14C  
ROPL 15C  
ROPL 16C  
ROPL 17C  
ROPL 18C  
ROPL 19C  
ROPL 20C  
ROPL 21C  
ROPL 22C  
ROPL 23C  
ROPL 24C  
ROPL 25C  
ROPL 26C  
ROPL 27C

003 OT 0G  
ROPL 29C

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      IF (K.EQ.0) K = 1
      PROPLD = PROPLD + TABLE(9,J) + TABLE(5,J)*(SIN(TABLE(6,J)*T))**K
      X = TABLE(7,J)*(COS(TABLE(8,J)*T))**K
      RETURN
C33 LPOF                                003 UT OG
203 CALL EXPROD (PROP,1, TABLE(2,N))
      PROPLD = PROPLD
      RETURN
300 CONTINUE
      RETURN
700 WRITE (ITP6,2030) K
      IPG = 0
      RETURN
1000 FORMAT (2F5.0,7F10.3)
2000 FORMAT (2X,A10,15,7F14.5)
2001 FORMAT (//5X,23HPROPORTIONAL LOAD TABLE//
      X 3X,9HTYPE LOAD,5H EXP,5X,9HMIN, TIME,5X,9HMAX, TIME,5X,
      X 2HA0,12X,2HA1,12X,2HA2,12X,2HA3,12X,2HA4,1X)
2030 FORMAT (5H PROPORTIONAL LOAD INPUT TABLE ERROR, INPUT TYPE =,15/)
      END
*DECK RESET
      SUBROUTINE RESET (NN,NUMNP,NDF,F)
      COMMON /TAPES/ ITP5,ITP6
      REAL LABEL
      DIMENSION F (NDF,1)
      COMMON/LABELS/ LABEL(6),XHED(3),XH,FHED(6),FH,UHED(6),UH,RHED(6),RH
      X ,AWORD1,AWORD2,AWORD3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)
      COMMON /MINHW/ NEW(1500)
      DIMENSION FT(7)
      MCT = 0
      N=NN
      IF (N.GT.0) GO TO 200
      DO 100 I = 1,NUMNP
      DO 100 J = 1,NDF
100 F(J,I)=0.0
      IF (N.EQ.0) RETURN
      N = -N
200 CONTINUE
      READ (ITP5,1000) J,(FT(I),I=1,NDF)
      K=NEW(J)
      DO 210 I=1,NDF
210 F(I,K)=FT(I)
      MCT = MCT + 1
      IF (MCT.GT.0) GO TO 250
      WRITE (ITP6,2000) HEAD,IPG,NUMNP,(FHED(I),FH,I=1,NDF)
      IPG = IPG + 1
      MCT = 50
250 WRITE (ITP6,2001) J,(FT(I),I=1,NDF)
      IF (J.GE.NUMNP) RETURN
      IF (J.GE.N) RETURN
      GO TO 200
1000 FORMAT (15,5X,7F10.0)
2000 FORMAT (1H1,12A6,30X,4HPAGE,14//15,13H NODAL FORCES//
      X 12H NODAL POINT,7(1X,2A6))
2001 FORMAT (1I2,4F13.4)
      END
*DECK ELMLIB
      SUBROUTINE ELMLIB (NMA,NDIM,NDF,NEL,NELI,NSTF,NSIZ,NVEC,MCT,DM,D,
      X XYZ,IX,F,FORCE,ESTIF,U,VECT,ISW)
      REAL LABEL
      COMMON/LABELS/ LABEL(6),XHED(3),XH,FHED(6),FH,UHED(6),UH,RHED(6),RH
      X ,AWORD1,AWORD2,AWORD3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)
C
C..... ELEMENT LIBRARY FOR FEAP72
C
C..... IS* = 1 FOR MATERIAL CHARACTERIZATION
C..... IS* = 2 FOR CHECK ON MESH
C..... IS* = 3 FOR ELEMENT STIFFNESS FORMULATION
C..... IS* = 4 FOR ELEMENT OUTPUTS
C..... IS* = 5 FOR ELEMENT LOAD/VECTOR COMPUTATIONS
C..... IS* = 6 FOR NON-LINEAR LOAD VECTOR COMPUTATIONS
C
      DO 56 J = 1,30

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	IF(OM,EQ,NORD(J)) GO TO 57	FLM 17C
56	CONTINUE	FLM 18C
57	GO TO (1,2,3,4,5,6,7,8,9,10,11,12,13,14,15,16,17,18,19,20,21,22,	FLM 19C
	X 23,24,25,26,27,28,29,30) , J	FLM 20C
1	CALL ELMT01(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 21C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 22C
	GO TO 99	FLM 23C
2	CALL ELMT02(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 24C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 25C
	GO TO 99	FLM 26C
3	CALL ELMT03(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 27C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 28C
	GO TO 99	FLM 29C
4	CALL ELMT04(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 30C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 31C
	GO TO 99	FLM 32C
5	CALL ELMT05(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 33C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 34C
	GO TO 99	FLM 35C
6	CALL ELMT06(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 36C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 37C
	GO TO 99	FLM 38C
7	CALL ELMT07(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 39C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 40C
	GO TO 99	FLM 41C
8	CALL ELMT08(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 42C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 43C
	GO TO 99	FLM 44C
9	CALL ELMT09(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 45C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 46C
	GO TO 99	FLM 47C
10	CALL ELMT10(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 48C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 49C
	GO TO 99	FLM 50C
11	CALL ELMT11(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 51C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 52C
	GO TO 99	FLM 53C
12	CALL ELMT12(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 54C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 55C
	GO TO 99	FLM 56C
13	CALL ELMT13(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 57C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 58C
	GO TO 99	FLM 59C
14	CALL ELMT14(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 60C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 61C
	GO TO 99	FLM 62C
15	CALL ELMT15(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 63C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 64C
	GO TO 99	FLM 65C
16	CALL ELMT16(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 66C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 67C
	GO TO 99	FLM 68C
17	CALL ELMT17(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 69C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 70C
	GO TO 99	FLM 71C
18	CALL ELMT18(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 72C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 73C
	GO TO 99	FLM 74C
19	CALL ELMT19(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 75C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 76C
	GO TO 99	FLM 77C
20	CALL ELMT20(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 78C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 79C
	GO TO 99	FLM 80C
21	CALL ELMT21(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 81C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 82C
	GO TO 99	FLM 83C
22	CALL ELMT22(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 84C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 85C
	GO TO 99	FLM 86C
23	CALL ELMT23(N,MA,NDIM,NDF,NEL,NEL1,NSTF,NSIZV,NVEC,MCT,DM,D,XYZ,	FLM 87C
	X IX,F,FORCE,ESTIF,U,VECT,ISW)	FLM 88C



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      GO TO 99
24  CALL ELMT24(N,MA,NDIM,NDF,NEL,NEL1,NSIF,NSIZV,NVEC,MCT,DM,D,XYZ,
X IX,F,FORCE,ESTIF,U,VECT,ISW)
      GO TO 99
25  CALL ELMT25(N,MA,NDIM,NDF,NEL,NEL1,NSIF,NSIZV,NVEC,MCT,DM,D,XYZ,
X IX,F,FORCE,ESTIF,U,VECT,ISW)
      GO TO 99
26  CALL ELMT26(N,MA,NDIM,NDF,NEL,NEL1,NSIF,NSIZV,NVEC,MCT,DM,D,XYZ,
X IX,F,FORCE,ESTIF,U,VECT,ISW)
      GO TO 99
27  CALL ELMT27(N,MA,NDIM,NDF,NEL,NEL1,NSIF,NSIZV,NVEC,MCT,DM,D,XYZ,
X IX,F,FORCE,ESTIF,U,VECT,ISW)
      GO TO 99
28  CALL ELMT28(N,MA,NDIM,NDF,NEL,NEL1,NSIF,NSIZV,NVEC,MCT,DM,D,XYZ,
X IX,F,FORCE,ESTIF,U,VECT,ISW)
      GO TO 99
29  CALL ELMT29(N,MA,NDIM,NDF,NEL,NEL1,NSIF,NSIZV,NVEC,MCT,DM,D,XYZ,
X IX,F,FORCE,ESTIF,U,VECT,ISW)
      GO TO 99
30  CALL ELMT30(N,MA,NDIM,NDF,NEL,NEL1,NSIF,NSIZV,NVEC,MCT,DM,D,XYZ,
X IX,F,FORCE,ESTIF,U,VECT,ISW)
99  RETURN
END
*DECK INTEGRAL
SUBROUTINE INTEGRAL(LIM,NC1,NDIM,LINT,STOW)
      REAL LABEL
      DIMENSION STRI(7,3),ITRI(7,3),UTRI(7,3),WTRI(7,3),STOW(4,1)
      DIMENSION SIR(7),TIR(7),UIR(7),WB(2),SB(2),WC(2),SC(2)
      COMMON/GAUSS/ LZZ,SGAUSS(5,5),*GAUSS(5,5)
      COMMON/LABELS/ LABEL(6),XHED(3),XH,FHED(6),FH,UHED(6),UH,RHED(6),RH
X 1,AKOH01,AKOH02,AKOH03,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)
      DATA STRI/0.33333333,6*0.0,0.5,0.0,0.5,4*0.0,0.33333333,0.05971587,
      2*0.47014206,0.79742699,2*0.10128651/
      DATA ITRI/0.33333333,6*0.0,2*0.5,0.0,4*0.0,0.33333333,0.47014206,
      0.05971587,0.47014206,0.10128651,0.79742699,0.10128651/
      DATA UTRI/0.33333333,6*0.0,0.0,2*0.5,6*0.0,0.33333333,2*0.47014206
      0.05971587,2*0.10128651,0.79742699/
      DATA WTRI/0.1,0.7*0.33333333,0.225,3*0.13239415,3*0.12593918/
      DATA SIR,TIR,UIR/1.0,2*0.0,4*1.0,0.1,0.1,0.2*1.0,-1.0,-1.0,0.0,0.1,1.1,
      -1.0,-1.0,1.1/
      DATA WB,SB,WC,SC/1.3333333,0.886*26593,1.0,0.795822426,0.0,
      0.335180055,0.0,0.758786911/
C**** INTEGRATION TABLE CONSTRUCTION
      IF(NC1.LT.0) NC1 = 0
      IF(NC1.GT.2) NC1 = 2
      NC = NC1 + 1
      GO TO (241,244,249),NC
C**** NC1 = 0: GAUSS INTEGRATION
241  I1 = 0
      IF(LIM.LT.1) LIM = 2
      IF(LIM.GT.5) LIM = 5
      LIM3 = LIM
      IF(NDIM.EQ.2) LIM3 = 1
      DO 243 I = 1,LIM3
      UU = SGAUSS(I,LIM)
      WU = *GAUSS(I,LIM)
      IF(NDIM.NE.2) GO TO 242
      UU = -1.0
      WU = 1.0
242  CONTINUE
      DO 243 J = 1,LIM
      IT = SGAUSS(J,LIM)
      WT = *GAUSS(J,LIM)*WU
      DO 243 K = 1,LIM
      II = I1 + 1
      STOW(1,II) = SGAUSS(K,LIM)
      STOW(2,II) = IT
      STOW(3,II) = UU
      STOW(4,II) = *GAUSS(K,LIM)*WT
243  CONTINUE
      LINT = I1
      GO TO 244

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FLM 89C
FLM 90C
FLM 91C
FLM 92C
FLM 93C
FLM 94C
FLM 95C
FLM 96C
FLM 97C
FLM 98C
FLM 99C
ELM100C
ELM101C
ELM102C
ELM103C
ELM104C
ELM105C
ELM106C
FLM107C
ELM108C
ELM109C
ELM110C
ELM111C
INT 1C
INT 2C
INT 3C
INT 4C
INT 5C
INT 6C
INT 7C
INT 8C
INT 9C
INT 10C
INT 11C
INT 12C
INT 13C
INT 14C
INT 15C
INT 16C
INT 17C
INT 18C
INT 19C
INT 20C
INT 21C
INT 22C
INT 23C
INT 24C
INT 25C
INT 27C
INT 28C
INT 29C
INT 30C
INT 31C
INT 32C
INT 33C
INT 34C
INT 35C
INT 36C
INT 37C
INT 38C
INT 39C
INT 40C
INT 41C
INT 42C
INT 43C
INT 44C
INT 45C
INT 46C
INT 47C
INT 48C

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C**** NCI = 1, IRONS INTEGRATION
244 IF (NDIM.NE.3.OR.LIM.EQ.1) GO TO 241
    LINT = 6
    IF (LIM.GE.2) LINT = 14
    HS = SR(LIM)
    BW = SB(LIM)
    DO 245 I = 1,7
    I2 = 2*I
    I1 = I2 - 1
    IF (I.NE.4) GO TO 245
    HS = SC(LIM)
    BW = AC(LIM)
245 CONTINUE
    STW(1,I1) = SIR(1)*HS
    STW(1,I2) = -STW(1,I1)
    STW(2,I1) = IIR(1)*HS
    STW(2,I2) = -STW(2,I1)
    STW(3,I1) = OIR(1)*HS
    STW(3,I2) = -STW(3,I1)
    STW(4,I1) = HW
    STW(4,I2) = HW
246 GO TO 248
C**** NCI = 2, TRIANGULAR COORDINATE INTEGRATION
249 LINT = 3
    IF (LIM.GE.3) LIM = 3
    IF (LIM.EQ.2) LINT = 3
    IF (LIM.GE.3) LINT = 7
    DO 247 I = 1,LINT
    STW(1,I) = STR(1,LIM)
    STW(2,I) = TTR(1,LIM)
    STW(3,I) = UTR(1,LIM)
247 STW(4,I) = WTR(1,LIM)
248 CONTINUE
    RETURN
    END
*DECK COMPACT
SUBROUTINE COMPACT(ESTIF,FORCE,ISIZE,NSIZE,JSIZE,ITAPE)
C
C**** REDUCE LAST EQUATIONS OF THE NSIZE X NSIZE STIFFNESS MATH TO A
C**** FINAL JSIZE X JSIZE STIFFNESS MATRIX * STORE ON ITAPE
C
    DIMENSION ESTIF(ISIZE,ISIZE)*FORCE(ISIZE)
    I1=NSIZE-JSIZE
    IF (I1.LE.0) RETURN
    DO 700 II=1,I1
    KK=NSIZE-II
    K1=KK+1
    C = ESTIF(K1,K1)
    IF (C.EQ.0.) GO TO 700
    DO 650 JJ=1,KK
    ESTIF(K1,JJ)=ESTIF(K1,JJ)/C
    CC=ESTIF(K1,JJ)
    FORCE(JJ)=FORCE(JJ)-CC*FORCE(K1)
    DO 650 LL=1,KK
    ESTIF(LL,JJ)=ESTIF(LL,JJ)-ESTIF(LL,K1)*CC
    FORCE(K1)=FORCE(K1)/C
650 CONTINUE
700 IF (ITAPE.GT.0) WRITE(ITAPE) ESTIF*FORCE
    RETURN
    END
*DECK CKQUAD
SUBROUTINE CKQUAD(FLAG,K,NEL)
COMMON /SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LO(120)
    FLAG = 1.
    SS = 1.0
    DO 30 I = 1,2
    SS = -SS
    TT = 1.0
    DO 30 J = 1,2
    TT = -TT
    CALL SHAP(SS,TT,K,XXX,YYY,NEL)
    IF (K.GE.1) CALL THEIA(SG,SK,1.0)

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	IF (XJAC*LE,0,0) GO TO 40	CKG 12C
30	CONTINUE	CKG 13C
	RETURN	CKG 14C
40	FLAG = XJAC	CKG 15C
	RETURN	CKG 16C
	END	CKG 17C
*DECK	CKHRIN	
	SUBROUTINE CKHRIN(NDIM,NEN,FLAG)	CKH 1C
	COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)	CKH 2C
	FLAG = 1.	CKH 3C
	NSIDE = (NEN-1)/NDIM**2	CKH 4C
	L = 1	CKH 5C
	IF (NDIM*EG,2) L = 2	CKH 6C
	SS = 1.0	CKH 7C
	DO 30 I = 1,2	CKH 8C
	SS = -SS	CKH 9C
	TT = 1.0	CKH 10C
	DO 30 J = 1,2	CKH 11C
	TT = -TT	CKH 12C
	UU = 1.0	CKH 13C
	DO 30 K = 1,2	CKH 14C
	UU = -UU	CKH 15C
	CALL BRICK2(UU,TT,SS,NDIM,NEN,NSIDE)	CKH 16C
	IF (XJAC*LE,0,0) GO TO 40	CKH 17C
30	CONTINUE	CKH 18C
	RETURN	CKH 19C
40	FLAG = XJAC	CKH 20C
	RETURN	CKH 21C
	END	CKH 22C
*DECK	PACKD	
	SUBROUTINE PACKD(U,1,C11,C12,C33,M)	PAC 1C
	DIMENSION U(3,21,1)	PAC 2C
	GO TO (291,293,294),1	PAC 3C
291	CONTINUE	PAC 4C
	D(1,1,M) = C11	PAC 5C
	D(1,2,M) = C12	PAC 6C
	D(2,2,M) = C11	PAC 7C
	D(2,1,M) = C12	PAC 8C
	D(3,3,M) = 4.0*C33	PAC 9C
	RETURN	PAC 10C
293	CONTINUE	PAC 11C
	D(1,3,M) = C12	PAC 12C
	D(2,3,M) = C12	PAC 13C
	D(3,3,M) = C11	PAC 14C
294	CONTINUE	PAC 15C
	D(1,1,M) = C11	PAC 16C
	D(1,2,M) = C12	PAC 17C
	D(1,3,M) = C33	PAC 18C
	D(2,1,M) = C33	PAC 19C
	D(2,2,M) = C33	PAC 20C
	D(2,3,M) = C33	PAC 21C
	D(3,1,M) = C11	PAC 22C
	D(3,2,M) = C33	PAC 23C
	D(3,3,M) = C33	PAC 24C
	D(3,1,M) = C33	PAC 25C
	D(3,2,M) = C33	PAC 26C
	D(3,3,M) = C33	PAC 27C
	RETURN	PAC 28C
	END	PAC 29C
*DECK	BRICK2	
	SUBROUTINE BRICK2(R,S,T,NDIM,NEL,NSIDE)	HRI 1C
	DIMENSION N1(12),N2(12),N3(12),NODE(12),S2ORD(12),S3ORD(12),SS(3)	HRI 2C
	DIMENSION CUM(3)	HRI 3C
	COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)	HRI 4C
	DATA NODE/2,4,6,8,9,10,11,12,14,16,18,20/	HRI 5C
	DATA N1/1,2,1,2,3,3,3,3,1,2,1,2/,N2/2,3,2,3,1,1,1,1,2,3,2,3/	HRI 6C
	DATA N3/3,1,3,1,1,2,2,2,2,3,1,3,1/	HRI 7C
	DATA S2ORD/-0.5,-0.5,0.5,-0.5,-0.5,0.5,0.5,-0.5,-0.5,0.5,0.5,0.5/	HRI 8C
	DATA S3ORD/-0.5,0.5,-0.5,-0.5,-0.5,-0.5,0.5,0.5,0.5,0.5,-0.5,0.5/	HRI 9C
	SS(1) = R	HRI 10C
	SS(2) = S	HRI 11C
	SS(3) = T	HRI 12C

K1 = 4	RR1 13C
L1 = 1	RR1 14C
IF(NSIDE,EQ,1) GO TO 110	RR1 15C
K1 = 12	RR1 16C
L1 = 2	RR1 17C
C**** FORM MIDSIDE SHAPE FUNCTIONS	RR1 18C
DO 100 L=1,12	RR1 19C
N = NODE(L)	RR1 20C
I = N1(L)	RR1 21C
J = N2(L)	RR1 22C
K = N3(L)	RR1 23C
SJ = SZORD(L)	RR1 24C
SL = SJORD(L)	RR1 25C
RP = 1. -SS(I)**2	RR1 26C
SP = .5 + SJ*SS(J)	RR1 27C
TP = .5 + SL*SS(K)	RR1 28C
SHAPE(I,N) = -2.0*SS(I)*SP*TP	RR1 29C
SHAPE(J,N) = SJ*RP*TP	RR1 30C
SHAPE(K,N) = SL*RP*SP	RR1 31C
100 SHAPE(4,N) = RP*SP*TP	RR1 32C
C**** FORM CORNER SHAPE FUNCTIONS	RR1 33C
110 SJ = -1.5	RR1 34C
K = 0	RR1 35C
DO 250 J = 1,2	RR1 36C
L = 1	RR1 37C
DO 200 I = 5,8	RR1 38C
RP = .5 + SZORD(I)*K	RR1 39C
SP = .5 + SJORD(I)*S	RR1 40C
TP = .5 + SJ*T	RR1 41C
SHAPE(1,L + K) = SZORD(I)*SP*TP	RR1 42C
SHAPE(2,L + K) = RP*SZORD(I)*TP	RR1 43C
SHAPE(3,L + K) = RP*SP*SJ	RR1 44C
SHAPE(4,L + K) = RP*SP*TP	RR1 45C
200 L = L + 1	RR1 46C
K = K + 1	RR1 47C
250 SJ = 1.5	RR1 48C
IF(NSIDE,EQ,1) GO TO 360	RR1 49C
C**** CORRECT BASIC CORNER FUNCTIONS BY PROPORTIONS OF MIDSIDE FUNCTIONS	RR1 50C
K = 8	RR1 51C
L = 9	RR1 52C
DO 350 I = 1,8,2	RR1 53C
DO 300 J = 1,4	RR1 54C
SHAPE(J,I) = SHAPE(J,I) - 0.5*(SHAPE(J,I+1)+SHAPE(J,K)+SHAPE(J,L))	RR1 55C
300 SHAPE(J,I+12) = SHAPE(J,I+12) - 0.5*(SHAPE(J,I+13)+SHAPE(J,K+12)	RR1 56C
X + SHAPE(J,L))	RR1 57C
K = I + 1	RR1 58C
L = L + 1	RR1 59C
350 C**** FORM THE JACOBIAN DETERMINANT	RR1 60C
DO 370 I = 1,3	RR1 61C
DO 370 J = 1,3	RR1 62C
370 SK(I,J) = 0.0	RR1 63C
DO 400 I = 1,NDIM	RR1 64C
DO 400 J = 1,NDIM	RR1 65C
DO 400 K = 1,NEL	RR1 66C
400 SK(I,J) = SK(I,J) + SHAPE(J,K)*X(I,K)	RR1 67C
IF(NDIM,EQ,2) SK(3,3) = 1.0	RR1 68C
SG(1,1) = SK(2,2)*SK(3,3) - SK(2,3)*SK(3,2)	RR1 69C
SG(2,2) = SK(1,1)*SK(3,3) - SK(1,3)*SK(3,1)	RR1 70C
SG(3,3) = SK(1,1)*SK(2,2) - SK(1,2)*SK(2,1)	RR1 71C
SG(1,2) = -SK(1,2)*SK(3,3) + SK(1,3)*SK(3,2)	RR1 72C
SG(1,3) = -SK(1,2)*SK(2,3) - SK(1,3)*SK(2,2)	RR1 73C
SG(2,1) = -SK(2,1)*SK(3,3) + SK(2,3)*SK(3,1)	RR1 74C
SG(2,3) = -SK(1,1)*SK(2,3) + SK(2,1)*SK(1,3)	RR1 75C
SG(3,1) = -SK(2,1)*SK(3,2) - SK(2,2)*SK(3,1)	RR1 76C
SG(3,2) = -SK(1,1)*SK(3,2) + SK(1,2)*SK(3,1)	RR1 77C
XJAC = XJ	RR1 78C
DO 400 I = 1,NDIM	RR1 79C
600 XJAC = XJAC + SG(I,1)*SK(I,1)	RR1 80C
DO 400 J = 1,NEL	RR1 81C
DO 710 I = 1,NDIM	RR1 82C
TEMP = 0.	RR1 83C
DO 700 K = 1,NDIM	RR1 84C



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700 TEMP = TEMP + SG(K,1)*SHAPE(K,J)
710 COLM(I) = TEMP
DO 800 I=1,NDIM
800 SHAPE(I,J) = COLM(I)
RETURN
END
*DECK QSHPR
SUBROUTINE QSHPR(SS,I1,Iw,XXX,YYY,NEL)
C**** SHAPE FUNCTION SUBROUTINE FOR 8 POINT ISOPARAMETRIC ELEMENT.
DIMENSION SQRD(8),TORD(8),ST(2)
COMMON /ZSHAP/ XJAC,SHAPE(4,20),SG(3,3),SA(3,3),X(3,20),L0(120)
COMMON /ASTI/ XS,XI,YS,YI,ZS,ZI
DATA SQRD/-1.,0.,0.,1.,0.,-1.,1.,0./,TORD/-1.,-1.,-1.,0.,0.,3*1.,0./
IF(NEL,GT,4) GO TO 50
C**** FOUR POINT ELEMENT SHAPE FUNCTIONS
J = 1
DO 30 I = 1,3,2
SI = SQRD(I)/2.
TI = TORD(I)/2.
SP = .5 + SI*SS
TP = .5 + TI*TI
SHAPE(I,J) = SI*TP
SHAPE(2,J) = SP*TI
SHAPE(3,J) = SP*TP
30 J = J + 1
GO TO 30
C**** EIGHT POINT ELEMENT SHAPE FUNCTIONS
50 DO 100 I = 1,8,2
SI = SQRD(I)
TI = TORD(I)
SP = SI*SS
TP = TI*TI
SP = 1.+SP
TP = 1.+TP
STU = (SI+TI-1.)*.25
SHAPE(1,I) = SI*TP*(STU+SP*.25)
SHAPE(2,I) = TI*SP*(STU+TP*.25)
SHAPE(3,I) = SP*TP*STU
100 CONTINUE
DO 200 I = 2,8,4
TI = TORD(I)
TP = 1.+TI*TI
SS = 1.-SS*SS
SHAPE(1,I) = -SS*TP
SHAPE(2,I) = 0.5*SS*TI
SHAPE(3,I) = 0.5*SS*TP
200 CONTINUE
DO 300 I = 4,8,4
SI = SQRD(I)
SP = 1.+SI*SS
TI = 1.-TI*TI
SHAPE(1,I) = 0.5*SI*TIM
SHAPE(2,I) = -TI*SP
SHAPE(3,I) = 0.5*SP*TIM
300 CONTINUE
C**** FORM JACOBIAN AND TRANSFORMATION MATRIX FOR PLATE ELEMENT
350 XXX=0.
YYY=0.
XS=0.
YS=0.
ZS = .
XT=0.
YT=0.
ZI = .
DO 400 I = 1,NEL
SP=SHAPE(1,I)
TP=SHAPE(2,I)
Fp=SHAPE(3,I)
XJ=X(I,I)
YI=Y(I,I)
XXX=XXX+XJ*Fp
YYY=YYY+YI*Fp

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      IAB2(2) = T6000                                FLM 44C
      D(1,MA) = NG * .01                              FLM 45C
      D(2,MA) = NK * .01                              FLM 46C
      NG = 5 * 2*AG                                    FLM 47C
      IF (NG.EQ.0) GO TO 100                          FLM 48C
      READ(1,1001) (D(1,MA),I=5,NU)                 FLM 49C
      WRITE(1,1002) (D(1,MA),I=5,NU)                 FLM 50C
100  IF (NK.EQ.0) GO TO 200                          FLM 51C
      NL = NU * 3                                       FLM 52C
      NL = NU * 2*NK                                    FLM 53C
      READ(1,1003) (D(1,MA),I=NL,NU)                FLM 54C
      WRITE(1,1002) (D(1,MA),I=NL,NU)                FLM 55C
200  IF (NU.LT.54) RETURN                              FLM 56C
      WRITE(1,1003) MA,NG,NK,NU                     FLM 57C
      IP5 = 0                                           FLM 58C
      RETURN                                           FLM 59C
C**** CHECK MESH FOR NEGATIVE JACOBIANS AT ELEMENT CORNERS
2  CALL LKSHK(NDIM,NEL,DM)                            FLM 61C
      NG = D(1,MA)                                    FLM 62C
      NK = D(2,MA)                                    FLM 63C
      NK = NK * (LINT+1)*(NK+NG*((NDIM+1)*NDIM)/2)   FLM 64C
      RETURN                                           FLM 65C
C**** COMPUTE THE VISCOELASTIC STIFFNESS MATRICES FOR SHEAR AND RULK
C**** SET UP THE MATERIAL PROPERTIES
3  NL = 4                                              FLM 66C
      NG = D(1,MA)                                    FLM 67C
      GG = D(4,MA)                                    FLM 68C
      IF (GG.LE.0) GO TO 150                          FLM 69C
      NL = 2*NG * NL                                  FLM 70C
      DO 14 I = 6,NL,2                                FLM 71C
140  GG = GG + D(1,MA)*HIST(DT/D(I+1,MA))           FLM 72C
150  NK = D(2,MA)                                    FLM 73C
      XK = D(5,MA)                                    FLM 74C
      IF (NK.LE.0) GO TO 170                          FLM 75C
      IL = NL * 2                                     FLM 76C
      NL = NL + 2*NK                                  FLM 77C
      DO 16 I = IL,NL,2                                FLM 78C
160  XK = XK + D(1,MA)*HIST(DT/D(I+1,MA))           FLM 79C
C**** COMPUTE THE INTEGRALS FOR THIS ELEMENT
170 CONTINUE                                           FLM 80C
      CALL VSTIF(NDIM,NDF,NEL,NSIDE,ESTIF(NDF+1,1),FORCE(1,2),STOR,LINT) FLM 81C
      SS=D(3,MA)*CA                                    FLM 82C
      CALL ASSEPH(NDIM,NDF,NEL,NSIDE,ESTIF(NDF+1,1),ESTIF,FORCE(1,2),GG, FLM 83C
      (XK,SS))                                         FLM 84C
      RETURN                                           FLM 85C
C**** STRESS COMPUTATION FOR THE VISCOELASTIC ELEMENT
C**** VISCOELASTIC LOAD HISTORY COMPUTATION AND ELEMENT ASSEMBLY FROM IP
4  CONTINUE                                           FLM 86C
C**** COMPUTATION AND OUTPUT OF STRESS AND HISTORY OF VISCOELASTIC SOLN. FLM 87C
C**** DO THE INTEGRATION AND OUTPUT THE STRESS
      NG = D(1,MA)                                    FLM 88C
      NK = D(2,MA)                                    FLM 89C
      GG = D(4,MA)                                    FLM 90C
      XK = D(5,MA)                                    FLM 91C
C**** CHECK FOR H STORAGE
      LL = NK * (LINT+1)*(NK+NG*((NDIM+1)*NDIM)/2)   FLM 92C
      IF (LL.GT.ISZH) CALL VROFF(ITRD,IFAR,H,NH,ISZH) FLM 93C
      NL = LINT * 1                                    FLM 94C
      DO 500 LL = 1,NL                                FLM 95C
      NOPRNT = NPH                                     FLM 96C
      IF (LL.NE.1) NOPRNT = .TRUE.                   FLM 97C
      NPL = .FALSE.                                    FLM 98C
      IF (NUMPLT.LE.0) GO TO 210                      FLM 99C
      DO 205 NP = 1,NUMPLT                            FLM 100C
205  IF (NEDATA(NP,1).EQ.N.AND.NEDATA(NP,2).EQ.LL) NPL = .TRUE. FLM 101C
210  CONTINUE                                           FLM 102C
      IF (LL.NE.1) GO TO 220                          FLM 103C
      SS = 0.0                                         FLM 104C
      TT = 0.0                                         FLM 105C
      UU = 0.0                                         FLM 106C
      IF (NDIM.EQ.2) UU = -1.0                       FLM 107C
      GO TO 230                                         FLM 108C

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220  SS = STW(1,LL=1)
    TT = STW(2,LL=1)
    UU = STW(3,LL=1)
    WW = STW(4,LL=1)
230  CALL HRICA2(SS,TT,UU,NDIM,NEL,NSIDE)
C... COMPUTE THE PRESENT STRAIN AND STRAIN INCREMENT
    DTH = 0.
    THP = 0.
    DO 330 I = 1,NDIM
    CC = 0.0
    DO 300 K = 1,NEL
300  CC = CC + SHAPE(4,K)*X(I,K)
    XX(I) = CC
    DO 320 J = 1,NDIM
    CC = 0.
    DD = 0.
    DO 310 K = 1,NEL
    CC = CC + SHAPE(1,K)*DUL(J,K) + SHAPE(J,K)*DUL(1,K)
310  DD = DD + SHAPE(1,K)*DUL(J,K) + SHAPE(J,K)*DUL(1,K)
    DEFS(1,J) = DD/XJAC
    DEFS(J+1,1) = CC/XJAC
    XIG(1,J) = GG*DEFS(J+1,1)
320  XIG(J+1,1) = XIG(1,J)
    DTH = DTH + DEFS(1,1)
330  THP = THP + DEFS(1+1,1)
    DO 340 I = 1,NDIM
    DEFS(1,I) = DEFS(1,1) - DTH/3.0
340  DEFS(1+1,1) = DEFS(1+1,1)/2.0
    XIK = THP*(XK/2.-GG/3.)
    YIK = XIK
    DTH = DTH/2.
C... UPDATE THE SHEAR HISTORY
    IG = 0
    IF(NG.LE.0) GO TO 410
    DO 400 I = 1,NG
    IG = IG + 2
    DD = U(IG+1,MA)
    CC = U(IG,MA)*HIST(DIP/DD)
    DD = EXP(-DT/DD)
    DO 400 J = 1,NDIM
    DO 400 K = J,NDIM
    EE = H(NH) + CC*DEFS(J,K)
    XIG(J,K) = XIG(J,K) + EE
    EE = LE*DD
    XIG(K+1,J) = XIG(K+1,J) + EE
    H(NH) = EE
400  NH = NH + 1
400  CONTINUE
C... UPDATE THE BULK HISTORY
410  IF(NA.LE.0) GO TO 430
    DO 420 I = 1,NA
    IG = IG + 2
    DD = U(IG+1,MA)
    EE = H(NH) + U(IG,MA)*HIST(DIP/DD)*DTH
    XIK = XIK + EE
    EE = LE*EXP(-DT/DD)
    YIK = YIK + EE
    H(NH) = EE
420  NH = NH + 1
430  CONTINUE
    DO 430 I = 1,NDIM
    XIG(1,I) = XIG(1,I) + XIK
    XIG(1+1,1) = XIG(1+1,1) + YIK
435  OUTPUT THE STRESSES FOR THE CURRENT TIME STEP
C... IF(NOPRNT) GO TO 640
    MCT = MCT + 1
    IF(MCT.GT.0) GO TO 630
    WRITE(IIP6,2004) HEAD,TIME,IPG,(XHED(I),XHI,I=1,NDIM),
    X ((SHEH(I),J),SH,I=J,NDF),J=1,NDF)
    WRITE(IIP6,2003) (BLANK,HLANK,I=1,NDIM),
    X ((SHEH(I),J),EH,I=J,NDF),J=1,NDF)
    IPG = IPG + 1

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FLM116C
FLM117C
FLM118C
FLM119C
FLM120C
FLM121C
FLM122C
FLM123C
FLM124C
FLM125C
FLM126C
FLM127C
FLM128C
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FLM179C
FLM180C
FLM181C
FLM182C
FLM183C
FLM184C
FLM185C
FLM186C
FLM187C

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        MCI = 16
630  WRITE (1,PE6,1AR1) A, (XX(I), I=1,NDIM), ((XIG(I,J), J=1,NDIM), I=1,NDIM) FLM188C
        WRITE (1,PE6,1AR2) UM, MA, ((DEPS(J+1,I), J=1,NDIM), I=1,NDIM) FLM189C
640  IF (NPL) CALL PLDATA(NUMEL,NDIM,N,LL,SHEU,XX,XIG,FORCE) FLM190C
        IF (LL.NE.1 .OR. IPLOT.EQ. 0.0) GO TO 650 FLM191C
        IF (IPLOT.EQ.3) GO TO 645
        VECT(N,1) = XIG(1,1)
        VECT(N,2) = XIG(2,2)
        VECT(N,3) = 0.0
        VECT(N,4) = XIG(1,2)
        VECT(N,5) = DEPS(2,1)
        VECT(N,6) = DEPS(3,2)
        VECT(N,7) = 0.0
        VECT(N,8) = DEPS(3,1)
        GO TO 650
645  KK = 0
        DO 646 I=1,NDIM
        DO 646 J=1,NDIM
        KK = KK + 1
        VECT(KK,1) = XIG(1,J)
        VECT(KK,2) = DEPS(J+1,I)
C
C      CALCULATE PRINCIPLE STRAINS
C
650  IF (NUPRINT) GO TO 550
        IF (MA.EQ.1) GO TO 550
        IF (NDIM.NE.3) GO TO 550
        CALL PRINC(DEPS)
        WRITE (6,3400) DEPS(1,1),DEPS(1,2),DEPS(1,3)
        MCI = MCI+2
        FORMAT(5A,22H PRINCIPAL STRAINS= 3E15.5 /)
3000  IF (ISX.EQ.1 .OR. LL.EQ.1) GO TO 500
C.... COMPUTE THE LOADS FOR THE NEXT TIME STEP FLM193C
        DO 44 I = 1,NDIM FLM194C
        DO 44 J = 1,NDIM FLM195C
        XIG(1,J) = XIG(J+1,I)*XX FLM196C
        XIG(J,I) = XIG(I,J) FLM197C
        I1 = 0 FLM198C
        XX = XJAC*XX*U(3,MA) FLM199C
        DO 47 I = 1,NEL FLM200C
        EE = SHAPE(4,I)*XX FLM201C
        DO 46 J = 1,NDF FLM202C
        CC = 0.0 FLM203C
        DO 45 K = 1,NDIM FLM204C
        CC = CC + SHAPE(K,I)*XIG(J,K) FLM205C
        FORCE(I1+J,1) = FORCE(I1+J,1) - CC*EE*UOL(J,I) FLM206C
        I1 = I1 + NDF FLM207C
        CONTINUE FLM208C
        RETURN FLM209C
C.... ELEMENT LOAD COMPUTATION FOR BODY TYPE FORCES, ETC. FLM210C
        CONTINUE FLM211C
        RETURN FLM212C
4000  IFG = 0 FLM213C
        RETURN FLM214C
C.... FORMAT STATEMENTS FLM215C
1000  FORMAT(2I5,3F10.0) FLM216C
1001  FORMAT(F10.0) FLM217C
2000  FORMAT(23H VISCOELASTIC MATERIAL//I10,12H SHEAR TERMS// FLM218C
        X I10,12H BULK TERMS//I10,12H G-INFINITY=,1PE12.4/I0X, FLM219C
        X I2H K-INFINITY=,1PE12.4/I0X,12H DENSITY =, 1PE12.4/I0X) FLM220C
2001  FORMAT(10X,14H SHEAR MODULUS,6X,4H TIME/(10X,1PE12.4)) FLM221C
2002  FORMAT(10X,14H BULK MODULUS,6X,4H TIME/(10X,1PE12.4)) FLM222C
2003  FORMAT(10X,4(2A6)) FLM223C
2004  FORMAT(10H,12A6,1E13.5,20X,4HPAGE,13//5A,16HELEMENT STRESSES// FLM224C
        I 1X,7HELEMENT,9(2A6)) FLM225C
2030  FORMAT(4H MATERIAL,13,17H EXCEEDS STORAGE,15,13H SHEAR TERMS,15, FLM226C
        X 16H BULK TERMS NEED,14,7H WORDS. ) FLM227C
        END FLM228C
        SUBROUTINE PRINC(STRAIN)
C
C      CALCULATE PRINCIPLE STRAIN
C

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C      DIMENSION STRAIN(4,3)
C      S11 = STRAIN(2,1)
C      S12 = STRAIN(3,1)/2.
C      S13 = STRAIN(4,1)/2.
C      S22 = STRAIN(3,2)
C      S23 = STRAIN(4,2)/2.
C      S33 = STRAIN(4,3)
C      R1 = -(S11*S22*S33)
C      R2 = S11*S22*S11*S33+S22*S33*(S13*S13+S12*S12+S23*S23)
C      R3 = S13*S13*S22+S12*S12*S33+S23*S23*S11
C      I = -(S11*S22*S33 + 2.*S12*S23*S13)
C      IF (R1.EQ.0.0.AND.B2.EQ.0.0.AND.R3.EQ.0.0) GO TO 10
C      CALL CUBIC(R1,R2,R3,ITELL)
10  STRAIN(1,1) = R1
C      STRAIN(1,2) = R2
C      STRAIN(1,3) = R3
C      RETURN
C      END
C      SUBROUTINE CUBIC(R1,R2,R3,ITELL)
C      DATA ZC,ZU,ZH,ZP,ZR,ZD,ZE,ZF,ZG,ZH,ZI,ZJ,ZK,ZL,ZM,ZN,ZO,ZP,ZQ,ZR,ZS,ZT,ZU,ZV,ZW,ZX,ZY,ZZ
C      THIS ROUTINE DETERMINES THE THREE ROOTS OF A RATIONAL CUBIC
C      EQUATION OF FORM:
C      
$$X^3 + R1X^2 + R2X + R3 = 0$$

C      THE USER SUPPLIES THE COEFFICIENTS (R1,R2,R3) THROUGH THE
C      SUBROUTINE CALL LIST. THE SUBROUTINE RETURNS THE ROOTS THROUGH
C      THE SAME VARIABLE LIST, ALONG WITH THE INTEGER, ITELL, WHICH TELLS
C      THE FORM OF THE ROOTS. THERE ARE ONLY TWO POSSIBLE FORMS FOR
C      ROOTS OF A CUBIC EQUATION...
C      FORM=1, (ITELL=1) THE ROOTS ARE ALL REAL NUMBERS.
C      FORM=2, (ITELL=2) THE FIRST ROOT,R1, IS REAL AND THE
C      OTHER TWO ROOTS ARE COMPLEX CONJUGATE
C      PAIRS, WHERE R2 IS THE REAL PART AND R3
C      IS THE IMAGINARY PART.
C      T3 = 1./3.
C      Q = 2.*(R1*T3)**3 - R1*R2*T3 + R3
C      Q2 = Q**2
C      P = R2 - R1*R1*T3
C      R = Q**3 + (P*T3)**3
C      R2 = SQRT(ABS(R))
C      A = Q2 + R2
C      B = Q2 - R2
C      TERM = -R1*T3
C      IF (R.GT.0.0) GO TO 50
C      PATH FOR REAL ROOT DETERMINATION (FORM=1)
C      ITELL = 1
C      THETA = ATAN2(Q2,R2)
C      RC = (SQRT(Q2**2+R2**2))**T3
C      ZA = RC*COS(THETA*T3)
C      ZH = RC*SIN(THETA*T3)
C      R1 = 2.*ZA + TERM
C      R2 = 2.*(ZC*ZC-ZH*ZU) + TERM
C      R3 = 2.*(ZC*ZC + ZH*ZU) + TERM
C      RETURN
C      PATH FOR ONE REAL ROOT AND TWO COMPLEX, (FORM=2)
C      ITELL = 2
C      AT3 = (ABS(A))**T3

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      HT3 = (AMST(4)+HT3)
      AT3 = SIGN(AT3+A)
      HT3 = SIGN(HT3+H)
      H1 = AT3 * HT3 * IERM
      H2 = ZC*(AT3+HT3) * IERM
      H3 = ZD*(A13-HT3)
C
      RETURN
      END
*DECK VISCQ
      SUBROUTINE VSTIF(NDIM,NDF,NEL,NSIDE,ST,FORCE,STOR,LINT)
      DIMENSION STOR(4,27),ST(1),FORCE(NDF,1)
      COMMON/SHAP/XJAC,SHAP(4+2C),XJUNK(18),X(3,20),LD(120)
C... INITIALIZE THE ARRAYS FOR STIFFNESS AND INTEGRALS
      DO 20 J1 = 1,LINT
      SS = STOR(1,11)
      TT = STOR(2,11)
      UU = STOR(3,11)
      WW = STOR(4,11)
      CALL BWTCK2(SS,TT,UU,NDIM,NEL,NSIDE)
      DVOL = WU*XJAC
C... FORM THE INDEPENDENT INTEGRALS
      NS =
      DO 100 J = 1,NEL
      TEMP=SHAP(4,J)*XJAC*WU
      DO 10 J1 = 1,NDIM
      FORCE(J1,J)=TEMP*FORCE(J1,J)
      CC = SHAP(4,J)*DVOL
      DO 10 J1 = 1,J
      L = NDIM
      IF(1.EQ.J1) L = N
      DO 10 M = 1,L
      NS = NS + 1
100   ST(NS) = ST(NS) + CC*SHAP(M+1)
200   CONTINUE
      RETURN
      END
      SUBROUTINE ASSEMB(NDIM,NDF,NEL,NSIF,ST,ESTIF,FORCE,GG,XX,HO)
      DIMENSION ESTIF(NSIF,1),ST(1),FORCE(1)
C... ASSEMBLY OF ELEMENT INTO THE STIFFNESS
      NU = NEL*NDIM
      NS = (NU*(NU+1))/2
      J1 = NDF*(NEL-1)
      MD1 = MD1P + 1
      XX = XK - ZC*GG/3.
      DO 23 J = 1,NEL
      DO 22 K = 1,NDIM
      M1 = MD1 - K
      I1 = J1
      DO 21 I = J,NEL
      L = 1
      IF(1.EQ.J1) L = N
      DO 20 M = L,NDIM
      M1 = MD1 - M
      CC = ST(NS)
      ST(NS) = 3.0
      ESTIF(I1*M1,J1+M1) = ESTIF(I1*M1,J1+M1) + XX*CC
      CC = GG*CC
      ESTIF(I1*M1,J1+M1) = ESTIF(I1*M1,J1+M1) + CC
      IF(M1.NE.N1) GO TO 200
      DO 100 K = 1,NDIM
      ESTIF(I1*K,J1+K) = ESTIF(I1*K,J1+K) + CC
100   NS = NS + 1
200   I1 = J1 - NDF
210   J1 = J1 - NDF
220   CONTINUE
230   J1 = J1 - NDF
      NU = NEL*NDF
      DO 300 J = 2,NU
      K = J - 1
      DO 30 I = 1,K
      ESTIF(I,J) = ESTIF(I,J) + ESTIF(J,I)
300   ESTIF(J,I) = ESTIF(I,J)

```

```

      DO 400 I=1,N1
      400 ESTIF(I,I)=ESTIF(I,I)+RU*FORCE(I)
      RETURN
      END
      FUNCTION HIST(C)
      C.... HISTORY TERM FOR THE INTEGRATION OF VISCOELASTIC EQUATIONS
      C.... HIST = (1. - EXP(-C))/C
      IF(C.GT.0.01) GO TO 200
      HIST = 1.
      DO 100 I = 1,3
      J = 5 - I
      100 HIST = 1.-C/FLUAT(J)*HIST
      RETURN
      200 HIST = (1. - EXP(-C))/C
      RETURN
      END
      SUBROUTINE VRUFF(ITRU,ITWR,H,IN,ISZH)
      C.... SUPPLY THE HISTORY VECTORS AND RESET THE POINTER
      DIMENSION H(ISZH)
      WRITE(ITRU) H
      READ(ITRU) H
      IN = 1
      RETURN
      END
      *DECK FLMT22
      SUBROUTINE FLMT22(N,MA,NDIM,NDF,NEL,NELI,NSTF,NSIZV,NVEC,MCT,DM,D,
      X XYZ,IX,N,FORCE,ESTIF,U,VECT,IS*)
      REAL LABL
      LOGICAL NPH,NPL,NPHNT
      DIMENSION D(63,1),ESTIF(NSTF,NSTF),FORCE(NSTF,2),STW(4,N),XX(3),
      X IX(NELI,1),U(NDF,1),DEPS(4,3),XIG(4,3),TAB1(11),TAB2(11)
      X XSHL(3,3),H(1),IG(6),JG(6)
      DIMENSION VECT(NSIZV,NVEC)
      COMMON/UYKAMD/IME,NSIG(7),NI,NSIEP,DI,NUMPLT,NEDATA(20,3),NPH,NPL
      COMMON/LABELS/ LABL(6),XHED(3),XH,FHED(6),FH,UHED(6),UH,RHED(6),RH
      X ,AWORD1,AWORD2,AWORD3,HEAD(12),START,CEASE,IPG,NSTH,WORD(30)
      COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)
      COMMON/TAPES/ ITP5,ITP6
      COMMON/VISDAT/UTP,RH,ISZH,DUL(6,20),UL(6,20),UDL(6,20),C6
      COMMON/VTAPES/ ITP13,ITP14,ITR0,ITWR
      COMMON/PLOTS/IFLOT
      DATA (TAB1(I),TAB2(I))/RH(1)G,17 ,BH ( RH /YSWORD/1H)/
      DATA SHE(20H XX=,6H XY=,6H XZ=,6H YY=,6H YZ=,
      X RH XG=,6H YZ=,6H ZZ= /
      DATA FWORD,EWORD,XWORD/RH,DPF12,3,RH,1PE12,3,RH ,12X /
      DATA SH,FH,PLANN/6MSTRESS,6MSTRAIN,6M /YWORD/RH,AS,17 /
      DATA 10/1,1,2,1,2,3/UG/1,2,2,3,3,3/
      C
      C GENERAL 2D,3D, FLUID ELEMENT (ONE POINT QUADRATURE)
      C
      IF(ISX.GT.6) GO TO 4000
      NSIE = (NEL+1)/NUMP2
      GO TO (1,2,3,4,5,6), IS*
      C.... CHARACTERIZE THE MATERIAL FOR ISOTROPIC LINEAR VISCOELASTICITY
      I READ(ITP5,100) NG,NK,D(4,MA),D(5,MA),D(3,MA)
      WRITE(ITP6,2000) NG,NK,D(4,MA),D(5,MA),D(3,MA)
      CALL INTREL(1,0,NULH,LINT,STW)
      C6 = .
      DT = .
      BTP = .
      NH = 1
      C.... SET UP HEADS FOR PRINTOUTS
      I = 2
      DO 60 J=1,NDIM
      TAB1(I) = FWORD
      TAB2(I) = XWORD
      I = I + 1
      DO 61 J = 1,6
      TAB1(I) = EWORD
      TAB2(I) = EWORD
      610 I = I + 1
      TAB1(I) = SWORD

```

```

      ASS 36C
      ASS 37C
      ASS 38C
      ASS 39C
      IST 1C
      IST 2C
      IST 3C
      IST 4C
      IST 5C
      IST 6C
      IST 7C
      IST 8C
      IST 9C
      IST 10C
      IST 11C
      IST 12C
      VHU 1C
      VHU 2C
      VHU 3C
      VHU 4C
      VHU 5C
      VHU 6C
      VHU 7C
      VHU 8C
      FLN 1C
      FLN 2C
      FLN 3C
      FLN 4C
      FLN 5C
      FLN 6C
      FLN 7C
      FLN 8C
      FLN 9C
      FLN 10C
      FLN 11C
      FLN 12C
      FLN 13C
      FLN 14C
      FLN 15C
      FLN 16C
      FLN 17C
      FLN 18C
      FLN 19C
      FLN 20C
      FLN 21C
      FLN 22C
      FLN 23C
      FLN 24C
      FLN 25C
      FLN 26C
      FLN 28C
      FLN 29C
      FLN 30C
      FLN 31C
      FLN 32C
      FLN 33C
      FLN 34C
      FLN 35C
      FLN 36C
      FLN 37C
      FLN 38C
      FLN 39C
      FLN 40C
      FLN 41C
      FLN 42C

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	1AMP(1) = SWORD	FLM 43C
	1AMP(2) = TWORD	FLM 44C
	D(1,MA) = DG * .01	FLM 45C
	D(2,MA) = DK * .01	FLM 46C
	NL = 0 * 20NG	FLM 47C
	IF (NG.EQ.0) GO TO 100	FLM 48C
	READ(11P5,1001) (D(I,MA),I=6*NU)	FLM 49C
	WRITE(11P5,2001) (D(I,MA),I=6*NU)	FLM 50C
100	IF (NK.EQ.0) GO TO 200	FLM 51C
	NL = NU * 1	FLM 52C
	NU = NU * 20NK	FLM 53C
	READ(11P5,1001) (D(I,MA),I=NL*NU)	FLM 54C
	WRITE(11P5,2002) (D(I,MA),I=NL*NU)	FLM 55C
200	IF (NU.LT.64) RETURN	FLM 56C
	WRITE(11P5,2030) MA,NK,NK,NU	FLM 57C
	IRG = 0	FLM 58C
	RETURN	FLM 59C
C....	CHECK MESH FOR NEGATIVE JACOBIANS AT ELEMENT CORNERS	FLM 60C
2	CALL CMBRIA(NDIM,NEL,DM)	FLM 61C
	NG = D(1,MA)	FLM 62C
	NK = D(2,MA)	FLM 63C
	NR = NR * (LINT+1)*(NK*NG*((NDIM+1)*NDIM)/2)	FLM 64C
	RETURN	FLM 65C
C....	COMPUTE THE VISCOELASTIC STIFFNESS MATRICES FOR SHEAR AND BULK	FLM 66C
C....	SET UP THE MATERIAL PROPERTIES	FLM 67C
3	NL = 4	FLM 68C
	DG = D(1,MA)	FLM 69C
	DK = D(2,MA)	FLM 70C
	IF (NG.LT.0) GO TO 100	FLM 71C
	NL = 20NG * NL	FLM 72C
	DO 14 I = 6*NL+2	FLM 73C
140	GG = GG + D(I,MA)*HIST(DT/D(I+1,MA))	FLM 74C
150	KK = D(3,MA)	FLM 75C
	KK = D(5,MA)	FLM 76C
	IF (NK.LT.0) GO TO 170	FLM 77C
	IL = NL * 2	FLM 78C
	NL = NL * 20NK	FLM 79C
	DO 16 I = 1L,NL+2	FLM 80C
160	KK = KK + D(I,MA)*HIST(DT/D(I+1,MA))	FLM 81C
C....	COMPUTE THE INTEGRALS FOR THIS ELEMENT	FLM 82C
170	CONTINUE	FLM 83C
	CALL VSTIF(NDIM,NDF,NEL,NSIDE,ESTIF(NDF+1,1),FORCE(1,2),STOW,LINT)	FLM 84C
	SS=D(3,MA)*CA	FLM 85C
	CALL ASSEMB(NDIM,NDF,NEL,NSTF,ESTIF(NDF+1,1),ESTIF,FORCE(1,2),GG,	FLM 86C
	(KK,SS)	FLM 87C
	RETURN	FLM 88C
C....	STRESS COMPUTATION FOR THE VISCOELASTIC ELEMENT	FLM 89C
C....	VISCOELASTIC LOAD HISTORY COMPUTATION AND ELEMENT ASSEMBLY FROM TP	FLM 90C
4	CONTINUE	FLM 91C
C....	COMPUTATION AND OUTPUT OF STRESS AND HISTORY OF VISCOELASTIC SOLN.	FLM 92C
C....	DO THE INTEGRATION AND OUTPUT THE STRESS	FLM 93C
	NG = D(1,MA)	FLM 94C
	NK = D(2,MA)	FLM 95C
	GG = D(4,MA)	FLM 96C
	KK = D(5,MA)	FLM 97C
C....	CHECK FOR H STORAGE	FLM 98C
	LL = NR * (LINT+1)*(NK*NG*((NDIM+1)*NDIM)/2)	FLM 99C
	IF (LL.GT.ISZH) CALL VSHUFF(ITRD,ITWR,H,NH,ISZH)	FLM100C
	PL = LINT	FLM102C
	DO 500 LL = 1,LL	FLM103C
	NOPRNT = NPH	FLM105C
	IF (LL.NE.1) NOPRNT = .TRUE.	FLM106C
	NPL = .FALSE.	FLM107C
	IF (NUMPLI.LL+1) GO TO 210	FLM108C
	DO 205 NP = 1,NUMPLI	FLM109C
205	IF (NEGATA(NP,1).EQ.N.AND.NEGATA(NP,2).EQ.LL) NPL = .TRUE.	FLM110C
210	CONTINUE	FLM111C
	IF (LL.NE.1) GO TO 220	FLM112C
	SS = 0.0	FLM113C
	TT = 0.0	
	UU = 0.0	
	KK = STOR(4,1)	

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      IF (NDIM.EQ.2) UU = -1.0
      GO TO 230
220  SS = STOK(1,LL-1)
      TT = STOK(2,LL-1)
      UU = STOK(3,LL-1)
      WW = STOK(4,LL-1)
230  CALL BRICK2(SS,TT,UU,NDIM,NEL,NSIDE)
C.... COMPUTE THE PRESENT STRAIN AND STRAIN INCREMENT
      DTH = 0.
      THP = 0.
      DO 33 I = 1,NDIM
      CC = 0.0
      DO 30 K = 1,NEL
300  CC = CC + SHAPE(4,K)*X(I,K)
      XX(I) = CC
      DO 32 J = 1,NDIM
      CC = 0.
      DO 31 K = 1,NEL
      CC = CC + SHAPE(1,K)*UL(J,K) + SHAPE(J,K)*UL(I,K)
310  DD = DD + SHAPE(1,K)*DUL(J,K) + SHAPE(J,K)*DUL(I,K)
      DEPS(I,J) = DD/XJAC
      DEPS(J+1,I) = CC/XJAC
      XIG(I,J) = GG*DEPS(J+1,I)
320  XIG(J+1,I) = XIG(I,J)
      DTH = DTH + DEPS(I,I)
      THP = THP + DEPS(I+1,I)
330  DO 34 I = 1,NDIM
      DEPS(I,I) = DEPS(I,I) - DTH/3.0
340  DEPS(I+1,I) = DEPS(I+1,I)/2.0
      XIK = THP*(XK/2.-GG/3.)
      YIK = XIK
      DTH = DTH/2.
C.... UPDATE THE SHEAR HISTORY
      IG = 4
      IF (NG.LE.0) GO TO 410
      DO 400 I = 1,NG
      IG = IG + 2
      DD = U(IG+1,MA)
      CC = U(IG,MA)*HIST(DTH/DD)
      DD = EXP(-DTH/DD)
      DO 402 J = 1,NDIM
      DO 400 K = J,NDIM
      EE = H(NH) + CC*DEPS(J,K)
      XIG(J,K) = XIG(J,K) + EE
      EE = EE*DD
      XIG(K+1,J) = XIG(K+1,J) + EE
      H(NH) = EE
402  NH = NH + 1
400  CONTINUE
C.... UPDATE THE BULK HISTORY
410  IF (NK.LE.0) GO TO 430
      DO 42 I = 1,NK
      IG = IG + 2
      DD = U(IG+1,MA)
      EE = H(NH) + U(IG,MA)*HIST(DTH/DD)*DTH
      XIK = XIK + EE
      EE = EE*EXP(-DTH/DD)
      YIK = YIK + EE
      H(NH) = EE
420  NH = NH + 1
430  CONTINUE
      DO 435 I = 1,NDIM
      XIG(I,I) = XIG(I,I) + XIK
435  XIG(I+1,I) = XIG(I+1,I) + YIK
C.... OUTPUT THE STRESSES FOR THE CURRENT TIME STEP
      IF (NPRINT) GO TO 640
      MCT = MCT - 1
      IF (MCT.GT.0) GO TO 630
      WRITE(IIP6,2004) HEAD,TIME,IPG,(XHEC(I),XK,I=1,NDIM),
      X((SHEC(I,J),SH,I=J,NDF),J=1,NDF)
      WRITE(IIP6,2003) (BLANK,HLANK,I=1,NDIM),

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FLM114C
FLM115C
FLM116C
FLM117C
FLM118C
FLM119C
FLM120C
FLM121C
FLM122C
FLM123C
FLM124C
FLM125C
FLM126C
FLM127C
FLM128C
FLM129C
FLM130C
FLM131C
FLM132C
FLM133C
FLM134C
FLM135C
FLM136C
FLM137C
FLM138C
FLM139C
FLM140C
FLM141C
FLM142C
FLM143C
FLM144C
FLM145C
FLM146C
FLM147C
FLM148C
FLM149C
FLM150C
FLM151C
FLM152C
FLM153C
FLM154C
FLM155C
FLM156C
FLM157C
FLM158C
FLM159C
FLM160C
FLM161C
FLM162C
FLM163C
FLM164C
FLM165C
FLM166C
FLM167C
FLM168C
FLM169C
FLM170C
FLM171C
FLM172C
FLM173C
FLM174C
FLM175C
FLM176C
FLM177C
FLM178C
FLM179C
FLM180C
FLM181C
FLM182C
FLM183C
FLM184C
FLM185C

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      X ((SHED(I,J),EM,I=J,NDF),J=1,NDF)
      IPG = IPG + 1
      MCT = 16
630  WRITE(IIP6,IAH1) N, (XX(I),I=1,NDIM), ((XIG(I,J),J=1,NDIM),I=1,NDIM)
      WRITE(IIP6,IAH2) UM,MA,((DEPS(J+1,I),J=1,NDIM),I=1,NDIM)
640  IF (NPL) CALL PLDATA(NUMEL,NDIM,N,LL,SHED,XX,XIG,FORCE)
      IF (LL.NE.1 .OR. IPLU1.EQ. 0.0) GO TO 650
      IF (IPLU1.EQ.1) GO TO 645
      VECT(N,1) = XIG(1,1)
      VECT(N,2) = XIG(2,2)
      VECT(N,3) = 0.0
      VECT(N,4) = XIG(1,2)
      VECT(N,5) = DEPS(2,1)
      VECT(N,6) = DEPS(3,2)
      VECT(N,7) = 0.0
      VECT(N,8) = DEPS(3,1)
      GO TO 650
645  KK = 1
      DO 646 I=1,NDIM
      DO 646 J=1,NDIM
      KK = KK + 1
      VECT(KK,1) = XIG(1,J)
      VECT(KK,2) = DEPS(J+1,1)
650  IF (IS*EQ*4) GO TO SUB
C.... COMPUTE THE LOADS FOR THE NEXT TIME STEP
      DO 446 I = 1,NDIM
      DO 44 J = 1,NDIM
      XIG(I,J) = XIG(J+1,I)*WWW
440  XIG(J,I) = XIG(I,J)
      II = J
      KK=XJAC*WWW*U(3,MA)
      DO 47 I = 1,NEL
      FE=SHAPE(4,I)*WWW
      DO 46 J = 1,NDF
      CC = 1.0
      DO 45 K = 1,NDIM
      CC = CC + SHAPE(K,I)*XIG(J,K)
450  FORCE(II+J,1)=FORCE(II+J,1)+CC*EE*U(II,J,I)
460  FORCE(II+J,1)=FORCE(II+J,1)+CC*EE*U(II,J,I)
470  II = II + NDF
500  CONTINUE
      RETURN
C.... ELEMENT LOAD COMPUTATION FOR BODY TYPE FORCES, ETC.
5  CONTINUE
      RETURN
4000 IPG = 0
      RETURN
C.... FORMAT STATEMENTS
1000 FORMAT(2I5,3F10.0)
1001 FORMAT(8F10.0)
2000 FORMAT(23H VISCOELASTIC MATERIAL//110,12H SHEAR TERMS/
X 110,12H BULK TERMS//10X,12H G-INFINITY=,1PE12.4/10X,
X 12H K-INFINITY=,1PE12.4/10X,12H DENSITY =,1PE12.4/1X)
2001 FORMAT(10X,14H SHEAR MODULUS*6X,4HTIME/(10X,1PE12.4))
2002 FORMAT(10X,14H BULK MODULUS*6X,4HTIME/(10X,1PE12.4))
2003 FORMAT(8X,9(2A6))
2004 FORMAT(1H1,12A6,E13.5,21X,4HPAGE,13//5X,16HELEMENT STRESSES//
1 1X,7HELEMENT,9(2A6))
2030 FORMAT(9H MATERIAL,13,17H EXCEEDS STORAGE.,15,13H SHEAR TERMS.,15,
X 16H BULK TERMS,14,7H WORDS. )
      END
*DECK ELMT23
SUBROUTINE ELMT23(N,MA,NDIM,NDF,NEL,NELI,NSTF,NSIZV,NVEC,MCT,UM,D, FLM 1C
X XYZ,IX,M,FORCE,ESTIF,U,VECT,ISA) FLM 2C
      REAL LABEL FLM 3C
      LOGICAL NPH,NPL,NUPNT FLM 4C
      DIMENSION U(43,1),ESTIF(NSTF,NSTF),FORCE(NSTF,2),STOW(4,4),XX(2), FLM 5C
X IX(NELI,1),U(NDF,1),DEPS(3,2),XIG(3,2),H(1),SHED(6) FLM 6C
      DIMENSION VECT(NSIZV,NVEC)
      COMMON/DYNAROT/ TIME,NSIG(7),NT,NSTEP,DT,NUMPLI,NEDATA(20,3),NPH,NPL FLM 7C
      COMMON/LABELS/ LABEL(6),XHEP(3),XHFHEP(6),FHFHEP(6),UH,RHED(6),RH FLM 8C
X ,VWURD1,AWURD2,AWURD3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30) FLM 9C
      COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120) FLM 10C

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	COMMON/VTAPES/ ITP5,ITP6	FLM 11C
	COMMON/VISDAT/ DTP,NM,IS2H,DUL(6,20),UL(6,20),UDL(6,20),C6	FLM 12C
	COMMON/VTAPES/ ITP13,ITP14,ITRD,ITWR	FLM 13C
	COMMON/PLDIS/ IPL120	
	DATA SHED/3H RR,3H RZ,3H ZZ,3H TT,6HPRINC1,6HPRINC2/	FLM 14C
	DATA =Y,TT/1.3333333333333333,0.6666666666666667/	FLM 15C
	IF (ISW.GT.0) GO TO 4000	FLM 16C
	NSIDE = (NEL+1)/4	FLM 17C
	GO TO (1,2,3,4,5,4), ISW	FLM 18C
C....	CHARACTERIZE THE MATERIAL FOR ISOTROPIC LINEAR VISCOELASTICITY	FLM 19C
1	READ(ITP5,1000) NG,NK,D(4,MA),D(5,MA),D(3,MA)	FLM 20C
	WRITE(ITP6,2000) NG,NK,D(4,MA),D(5,MA),D(3,MA)	FLM 21C
	CALL INTEGL(2,0,NDIM,LINT,STUX)	FLM 22C
	CA = 0.0	FLM 23C
	DT = 0.0	FLM 24C
	DTP = 0.0	FLM 25C
	NH = 1	FLM 26C
	D(1,MA) = NG + .01	FLM 27C
	D(2,MA) = NK + .01	FLM 28C
	NU = 5 + 2*NG	FLM 29C
	IF (NG.EQ.0) GO TO 100	FLM 30C
	READ(ITP5,1001) (D(1,MA),I=6,NU)	FLM 31C
	WRITE(ITP6,2001) (D(1,MA),I=6,NU)	FLM 32C
100	IF (NK.EQ.0) GO TO 200	FLM 33C
	NL = NU + 1	FLM 34C
	NU = NU + 2*NK	FLM 35C
	READ(ITP5,1001) (D(1,MA),I=NL,NU)	FLM 36C
	WRITE(ITP6,2002) (D(1,MA),I=NL,NU)	FLM 37C
200	IF (NU.LT.64) RETURN	FLM 38C
	WRITE(ITP6,2030) NA,NG,NK,NU	FLM 39C
	IPG = 0	FLM 40C
	RETURN	FLM 41C
C....	CHECK MESH FOR NEGATIVE JACOBIANS AT ELEMENT CORNERS	FLM 42C
2	CALL CKHPRK(NDIM,NEL,DM)	FLM 43C
	NG = D(1,MA)	FLM 44C
	NK = D(2,MA)	FLM 45C
	NH = NH + (LINT+1)*(NK+NU*((NDIM+1)*NDIM)/2+NG)	FLM 46C
	CONT = 45./ATAN(1.)/2.	FLM 47C
	RETURN	FLM 48C
C....	COMPUTE THE VISCOELASTIC STIFFNESS MATRICES FOR SHEAR AND BULK	FLM 49C
C....	SET UP THE MATERIAL PROPERTIES	FLM 50C
3	NL = 4	FLM 51C
	NG = D(1,MA)	FLM 52C
	GG = D(4,MA)	FLM 53C
	KK = D(3,MA)	FLM 54C
	IF (NG.LE.0) GO TO 150	FLM 55C
	NL = 2*NG + NL	FLM 56C
	DO 140 I = 0,NL-2	FLM 57C
140	GG = GG + D(I,MA)*HIST(DT/D(I+1,MA))	FLM 58C
150	NK = D(2,MA)	FLM 59C
	KK = D(5,MA)	FLM 60C
	IF (NK.LE.0) GO TO 170	FLM 61C
	IL = NL + 2	FLM 62C
	NL = NL + 2*NK	FLM 63C
	DO 160 I = IL,NL-2	FLM 64C
160	KK = KK + D(I,MA)*HIST(DT/D(I+1,MA))	FLM 65C
170	C11 = KK + 1/3*GG	FLM 66C
	C12 = KK - 1/3*GG	FLM 67C
	C13 = C12	FLM 68C
	C14 = 0.0	FLM 69C
	C22 = C11	FLM 70C
	C23 = C12	FLM 71C
	C33 = C11	FLM 72C
	C34 = 0.0	FLM 73C
	C44 = GG	FLM 74C
	DO 190 KK = 1,LINT	FLM 75C
	S = STUX(1,KK)	FLM 76C
	T = STUX(2,KK)	FLM 77C
	WW = STUX(4,KK)	FLM 78C
	CALL BRICK2(S,T,=1.,2*NEL,NSIDE)	FLM 79C
	PR = 0.0	FLM 80C
	DO 180 I = 1,NEL	FLM 81C



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180  RH = RH + X(1,1)*SHAPE(4,1)
    DVOL = RH*RH*XJAC
    II = I
    DO 385 J = 1,NEL
C.... COMPUTE LUMPED MASS
    TEMP = DVOL*SHAPE(4,1)*RO
    FORCE(II,1,2) = FORCE(II,1,2) + TEMP
    FORCE(II+1,2) = FORCE(II+1,2) + TEMP
C.... COMPUTE ELEMENT STIFFNESS
    RN = SHAPE(1,1)/XJAC*DVOL
    ZN = SHAPE(2,1)/XJAC*DVOL
    ET = SHAPE(4,1)/RH*DVOL
    A11 = C11*RN + C12*ET + C14*ZN
    A12 = C13*ZN + C14*RN
    A21 = C12*RN + C22*ET
    A22 = C23*ZN
    A31 = C13*RN + C23*ET + C34*ZN
    A32 = C33*ZN + C34*RN
    A41 = C14*RN + C44*ZN
    A42 = C34*ZN + C44*RN
    J1 = II
    DO 375 J = 1,NEL
    RN = SHAPE(1,J)/XJAC
    ZN = SHAPE(2,J)/XJAC
    ET = SHAPE(4,J)/RH
    ESTIF(II,J1) = ESTIF(II,J1) + RN*A11 + ET*A21 + ZN*A41
    ESTIF(II,J1+1) = ESTIF(II,J1+1) + ZN*A31 + RN*A41
    IF(1.NE.J)
    *ESTIF(II+1,J1) = ESTIF(II+1,J1) + RN*A12 + ET*A22 + ZN*A42
    ESTIF(II+1,J1+1) = ESTIF(II+1,J1+1) + ZN*A32 + RN*A42
375  J1 = J1 + NDF
380  JJ = II + NDF
390  CONTINUE
C.... CONSTRUCT SYMMETRIC ESTIF
    DO 395 I = 2,NSTF
    K = I-1
    DO 395 J = 1,K
    ESTIF(I,J) = ESTIF(I,J) + ESTIF(J,I)
395  ESTIF(J,I) = ESTIF(I,J)
    IF(16.EQ.J) RETURN
    DO 19 I = 1,NSTF
190  ESTIF(I,1) = ESTIF(I,1) + C6*FORCE(I,2)
    RETURN
C.... STRESS COMPUTATION FOR THE VISCOELASTIC ELEMENT
C.... VISCOELASTIC LOAD HISTORY COMPUTATION AND ELEMENT ASSEMBLY FROM TP
4    CONTINUE
C.... COMPUTATION AND OUTPUT OF STRESS AND HISTORY OF VISCOELASTIC SOLN.
C.... DO THE INTEGRATION AND OUTPUT THE STRESS
    RG = U(1,MA)
    RK = U(2,MA)
    GG = U(4,MA)
    XA = U(5,MA)
C.... CHECK FOR H STORAGE
    LL = RH + (LINT+1)*(NK+NG*(NDIM+1)*NDIM)/2*NG
    IF(LL.GT.ISZH) CALL VBUFF(ITRU,ITWR,H,NH,ISZH)
    NL = LINT + 1
    DO 205 LL = 1,NL
    NDRHI = NPH
    IF(NSIG(LL).NE.0) NDRHI = .TRUE.
    NPL = .FALSE.
    IF(NUMPLT.LE.0) GO TO 210
    DO 205 NP = 1,NUMPLT
205  IF(NEDATA(NP,1).EQ.N.AND.NEDATA(NP,2).EQ.LL) NPL = .TRUE.
210  CONTINUE
    IF(LL.NE.1) GO TO 220
    S = 0.0
    T = 0.0
    GO TO 230
220  S = STOW(1,LL-1)
    T = STOW(2,LL-1)
    W = STOW(4,LL-1)
230  CALL BRICK2(S,T,-1.,2,NEL,NSIDE)

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FLM 42C
FLM 43C
FLM 44C
FLM 45C
FLM 46C
FLM 47C
FLM 48C
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FLM 150C
FLM 151C
FLM 152C
FLM 153C
FLM 154C

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C...	COMPUTE THE PRESENT STRAIN AND STRAIN INCREMENT	FLM155C
	DTH = 0.	FLM156C
	THP = 0.	FLM157C
	DO 33 I = 1,2	FLM158C
	CC = 0.	FLM159C
	DO 30 J, K = 1,NEL	FLM160C
300	CC = CC + SHAPE(4,K)*X(I,K)	FLM161C
	XX(I) = CC	FLM162C
	DO 32 J = 1,2	FLM163C
	CC = 0.	FLM164C
	DO = 0.	FLM165C
	DO 31 K = 1,NEL	FLM166C
	CC = CC + SHAPE(1,K)*DUL(J,K) + SHAPE(J,K)*DUL(I,K)	FLM167C
310	DO = DO + SHAPE(I,K)*DUL(J,K) + SHAPE(J,K)*DUL(I,K)	FLM168C
	DEPS(I,J) = DO/XJAC	FLM169C
	DEPS(J,I) = CC/XJAC	FLM170C
	XIG(I,J) = GG*DEPS(J+1,1)	FLM171C
320	XIG(J+1,1) = XIG(I,J)	FLM172C
	DTH = DTH + DEPS(1,1)	FLM173C
330	THP = THP + DEPS(1+1,1)	FLM174C
	RR = XX(I)	FLM175C
	ET = 0.	FLM176C
	DET = 0.	FLM177C
	DO 35 K = 1,NEL	FLM178C
	CC = SHAPE(4,K)	FLM179C
	ET = ET + CC*DUL(1,K)	FLM180C
350	DET = DET + CC*DUL(1,K)	FLM181C
	ST = GG*ET*2.0	FLM182C
	STF = ST	FLM183C
	XIK = (THP + ET*2./RR)*(XX/2. - GG/3.)	FLM184C
	YIK = XIK	FLM185C
	DTH = DTH/2. + DET/RR	FLM186C
	DET = DET*2.0 - DTH*RR*TT	FLM187C
	DO 34 I = 1,2	FLM188C
	DEPS(1,I) = DEPS(1,1) - DTH*TT	FLM189C
340	DEPS(1+1,1) = DEPS(1+1,1)/2.0	FLM190C
C...	UPDATE THE SHEAR HISTORY	FLM191C
	IG = 4	FLM192C
	IF(NG.LE.0) GO TO 410	FLM193C
	DO 400 I = 1,NG	FLM194C
	IG = IG + 2	FLM195C
	DD = D(IG+1,MA)	FLM196C
	CC = D(IG,MA)*HIST(UTP/DD)	FLM197C
	DD = EXP(-D1/DD)	FLM198C
	DO 402 J = 1,2	FLM199C
	DO 402 K = J,2	FLM200C
	EE = H(NH) + CC*DEPS(J,K)	FLM201C
	XIG(J,K) = XIG(J,K) + EE	FLM202C
	EE = EF*DD	FLM203C
	XIG(K+1,J) = XIG(K+1,J) + EE	FLM204C
	H(NH) = EE	FLM205C
402	NH = NH + 1	FLM206C
	EE = H(NH) + CC*DET	FLM207C
	ST = ST + EE	FLM208C
	EE = EF*DD	FLM209C
	H(NH) = EE	FLM210C
	STF = STF + EE	FLM211C
	NH = NH + 1	FLM212C
400	CONTINUE	FLM213C
C...	UPDATE THE WULK HISTORY	FLM214C
410	IF(NK.LE.0) GO TO 430	FLM215C
	DO 42 I = 1,NK	FLM216C
	IG = IG + 2	FLM217C
	DD = D(IG+1,MA)	FLM218C
	EE = H(NH) + D(IG,MA)*HIST(UTP/DD)*DTH	FLM219C
	XIK = XIK + EE	FLM220C
	EE = EE*EXP(-D1/DD)	FLM221C
	YIK = YIK + EE	FLM222C
	H(NH) = EE	FLM223C
420	NH = NH + 1	FLM224C
430	CONTINUE	FLM225C
	DO 435 I = 1,2	FLM226C

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435  XIG(I,1) = XIG(I,1) * XIK
    XIG(I,2) = XIG(I,2) * YIK
    FT = F/XX(I)
    ST = ST/XX(I) * XIK
C.... OUTPUT THE STRESSES FOR THE CURRENT TIME STEP
    IF (NORMAT) GO TO 644
    MCT = MCT + 1
    IF (MCT,1) GO TO 630
    WRITE (IPE,2004) HEAD,TIME,IPG
    IPG = IPG + 1
    MCT = 16
630  WRITE (IPE,2003) N,(XX(I),I=1,2),((XIG(I,J),J=1,2),I=1,2),ST,
    X      DM,MA,((DEPS(J,1),J=1,2),I=1,2),FT
640  IF (NPL) CALL PLDATA(NUMEL,NDIM,N,LL,SHED,XX,XIG,FORCE)
    IF (IPLIG) KE = 1 GO TO 641
    IF (LL,NE,1) GO TO 641
    VECT(N,1) = XIG(1,1)
    VECT(N,2) = XIG(2,2)
    VECT(N,3) = ST
    VECT(N,4) = XIG(1,2)
    VECT(N,5) = DEPS(2,1)
    VECT(N,6) = DEPS(3,2)
    VECT(N,7) = FT
    VECT(N,8) = DEPS(3,1)
641  CONTINUE
    IF (IS*EG*4,OR*LL*EW*1) GO TO 500
C.... COMPUTE THE LOADS FOR THE NEXT TIME STEP
    DD = X*WHR
    XIG(1,1) = XIG(2,1)*DD
    XIG(1,2) = XIG(3,1)*DD
    XIG(2,2) = XIG(3,2)*DD
    XIG(2,1) = XIG(1,2)
    STF = (STF/WR + YIK)*X*WHR*JAC
    WW = XJAC*DD*DD(3,MA)
    II = 0
    DO 47, I = 1,NEL
    CC = SHAPE(4,I)
    FORCE(II+1,1) = FORCE(II+1,1) - STF*CC
    EE = CC*WW
    DO 46, J = 1,2
    CC = 1.0
    DO 45, K = 1,2
    CC = CC + SHAPE(K,I)*XIG(J,K)
450  FORCE(II+J,1) = FORCE(II+J,1) - CC*EE*UDL(J,I)
460  II = II + NDF
470  II = II + NDF
500  CONTINUE
    RETURN
C.... ELEMENT LOAD COMPUTATION FOR BODY TYPE FORCES, ETC.
5  CONTINUE
    RETURN
4000 IPG = 0
    RETURN
C.... FORMAT STATEMENTS
1000 FORMAT(2I5,4F10.0)
1001 FORMAT('H10.0)
2000 FORMAT(5X,57H LINEAR VISCOELASTIC MATERIAL, AXISYMMETRIC SOLID ELE
XMENT //10,12H SHEAR TERMS//
X 110,12H BULK TERMS//10X,12H G-INFINITY=,1PE12.4/10X,
X 12H K-INFINITY=,1PE12.4/10X,12H DENSITY =,1PE12.4/1X)
2001 FORMAT(10X,14H SHEAR MODULUS,6X,4HTIME/(10X,1PE12.4))
2002 FORMAT(10X,14H BULK MODULUS,6X,4HTIME/(10X,1PE12.4))
2003 FORMAT(10X,17,0P2F12.3,1P4E12.3/8X,A5,17,12X,1P4E12.3)
2004 FORMAT(10X,12A6,E13.5,20X,4HPAGE,13//5X,16HELEMENT STRESSES//
1 1X,7HELEMENT,4X,6HH ORD.,6X,6HZ ORD.,5X,9HHH=STRESS,3X,
2 9HHZ=STRESS,3X,9HZZ=STRESS,3X,9HIT=STRESS/35X,9HHH=STRAIN,3X,
3 9HZZ=STRAIN,3X,9HZZ=STRAIN,3X,9HIT=STRAIN)
2030 FORMAT(9H MATERIAL,13,17H EXCEEDS STORAGE.,15,13H SHEAR (KMPA),10X
X 16H BULK TERMS NEEDED,14,7H WORDS.)
END
*DECK ELMT24
SUBROUTINE ELMT24(N,MA,NDIM,NDF,NEL,NEL1,KSTF,NB12,NB2C,NCT,NDIM,
X XYZ,XH,FORCE,ESTIF,U,VECT,IS*)

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AD-A043 582

CIVIL ENGINEERING LAB (NAVY) PORT HUENEME CALIF  
A FINITE ELEMENT HEAD INJURY MODEL. VOLUME II. COMPUTER PROGRAM--ETC(U)  
JUL 77 T A SHUGAR  
CEL-R-854-2

F/G 6/2

UNCLASSIFIED

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2 OF 2  
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	REAL LABEL	FLM 3C
	LOGICAL NPH,NPL,NOPRNT	FLM 4C
	DIMENSION D(63,1),FSIIF(NSIF,NSIF),FORCE(NSIF,2),STUW(4,4),XX(2),	FLM 5C
	X IX(NEL,1),U(NDF,1),UEPS(3,2),XIG(3,2),H(1),SHED(6)	FLM 6C
	DIMENSION VECT(NSIZV,NVEC)	
	COMMON/DYNAMO/TIME,NSIG(7),NT,NSIEP,DT,NUMPLI,NEDATA(20,3),NPH,NPL	FLM 7C
	COMMON/LABELS/ LABEL(6),XHED(3),XM,FHED(6),FH,UHED(6),UH,RHED(6),RM	FLM 8C
	X AWORD1,AWORD2,AWORD3,HEAD(12),START,CEASE,IPG,NSTR,WORD(30)	FLM 9C
	COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)	FLM 10C
	COMMON/TAPE/ ITP5,ITP6	FLM 11C
	COMMON/VISUAL/ DTF,NH,ISZH,DUL(6,20),UL(6,20),UDL(6,20),C6	FLM 12C
	COMMON/VTAPE/ ITP13,ITP14,ITR0,ITWR	FLM 13C
	COMMON/PLOTS/ IPL120	
	DATA SHED/3H RR,3H HZ,3H ZZ,3H TT,6HPRINC1,6HPRINC2/	FLM 14C
	DATA FT,TT/1,333333333333333,6666666666666667/	FLM 15C
C		
C	AXISYMMETRIC FLUID ELEMENT (ONE POINT QUADRATURE)	
C		
	IF (ISW.GT.6) GO TO 4000	FLM 16C
	NSIDE = (NEL+1)/4	FLM 17C
	GO TO (1,2,3,4,5,4), ISW	FLM 18C
C....	CHARACTERIZE THE MATERIAL FOR ISOTROPIC LINEAR VISCOELASTICITY	FLM 19C
1	READ(ITP5,100) NG,NK,D(4,MA),D(5,MA),D(3,MA)	FLM 20C
	WRITE(ITP6,200) NG,NK,D(4,MA),D(5,MA),D(3,MA)	FLM 21C
	CALL INTEGL(1,0,NDIM,LINT,STUW)	
	C6 = 0.0	FLM 23C
	DT = 0.0	FLM 24C
	DTW = 0.0	FLM 25C
	NH = 1	FLM 26C
	D(1,MA) = NG * .01	FLM 27C
	D(2,MA) = NK * .01	FLM 28C
	NU = 3 * 2*NG	FLM 29C
	IF (NG.EQ.0) GO TO 100	FLM 30C
	READ(ITP5,1001) (D(I,MA),I=6,NU)	FLM 31C
	WRITE(ITP6,2001) (D(I,MA),I=6,NU)	FLM 32C
100	IF (NK.EQ.0) GO TO 200	FLM 33C
	NL = NU + 1	FLM 34C
	NU = NU * 2*NG	FLM 35C
	READ(ITP5,1001) (D(I,MA),I=NL,NU)	FLM 36C
	WRITE(ITP6,2002) (D(I,MA),I=NL,NU)	FLM 37C
200	IF (NU.LT.64) RETURN	FLM 38C
	WRITE(ITP6,2030) MA,NG,NK,NU	FLM 39C
	IPG = 0	FLM 40C
	RETURN	FLM 41C
C....	CHECK MESH FOR NEGATIVE JACOBIANS AT ELEMENT CORNERS	FLM 42C
2	CALL CBBRIK(NDIM,NEL,DM)	FLM 43C
	NG = D(1,MA)	FLM 44C
	NK = D(2,MA)	FLM 45C
	NH = NH + (LINT+1)*(NK+NG*((NDIM+1)*NDIM)/2*NG)	FLM 46C
	CONJ = 45./ATAN(1.)/2.	FLM 47C
	RETURN	FLM 48C
C....	COMPUTE THE VISCOELASTIC STIFFNESS MATRICES FOR SHEAR AND BULK	FLM 49C
C....	SET UP THE MATERIAL PROPERTIES	FLM 50C
3	NL = 4	FLM 51C
	NG = D(1,MA)	FLM 52C
	GG = D(4,MA)	FLM 53C
	RO = D(3,MA)	FLM 54C
	IF (NG.LE.0) GO TO 150	FLM 55C
	NL = 2*NG + NL	FLM 56C
	DO 14, I = 5,NL,2	FLM 57C
140	GG = GG + D(I,MA)*HIST(DT/D(I+1,MA))	FLM 58C
150	NK = D(2,MA)	FLM 59C
	XK = D(5,MA)	FLM 60C
	IF (NK.LE.0) GO TO 170	FLM 61C
	IL = NL * 2	FLM 62C
	NL = NL + 2*NK	FLM 63C
	DO 16, I = IL,NL,2	FLM 64C
160	XK = XK + D(I,MA)*HIST(DT/D(I+1,MA))	FLM 65C
170	C11 = XK + FT*GG	FLM 66C
	C12 = XK - FT*GG	FLM 67C
	C13 = C12	FLM 68C
	C14 = 0.0	FLM 69C

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C22 = C11
C23 = C12
C33 = C11
C34 = 0.0
C44 = GG
DO 39, KK = 1, LNI
S = STOK(1, KK)
T = STOK(2, KK)
ww = STOK(4, KK)
CALL HPCK2(S, T, -1.0, 2, NEL, NSIDE)
RR = 0.0
DO 18, I = 1, NEL
RR = RR + X(I, 1) * SHAPE(4, I)
DVOL = ww * RR * XJAC
I1 = 1
DO 38, I = 1, NEL
C.... COMPUTE LUMPED MASS
TEMP = DVOL * SHAPE(4, I) * RU
FORCE(I1, 2) = FORCE(I1, 2) + TEMP
FORCE(I1+1, 2) = FORCE(I1+1, 2) + TEMP
C.... COMPUTE ELEMENT STIFFNESS
RN = SHAPE(1, I) / XJAC * DVOL
ZN = SHAPE(2, I) / XJAC * DVOL
ET = SHAPE(4, I) / RR * DVOL
A11 = C11 * RN + C12 * ET + C14 * ZN
A12 = C13 * ZN + C14 * RN
A21 = C12 * RN + C22 * ET
A22 = C23 * ZN
A31 = C13 * RN + C23 * ET + C34 * ZN
A32 = C33 * ZN + C34 * RN
A41 = C14 * RN + C44 * ZN
A42 = C34 * ZN + C44 * RN
J1 = 1
DO 375, J = 1, NEL
RN = SHAPE(1, J) / XJAC
ZN = SHAPE(2, J) / XJAC
ET = SHAPE(4, J) / RR
ESTIF(I1, J1) = ESTIF(I1, J1) + RN * A11 + ET * A21 + ZN * A41
ESTIF(I1, J1+1) = ESTIF(I1, J1+1) + ZN * A31 + RN * A41
IF (I1.NE.J1)
*ESTIF(I1+1, J1) = ESTIF(I1+1, J1) + RN * A12 + ET * A22 + ZN * A42
ESTIF(I1+1, J1+1) = ESTIF(I1+1, J1+1) + ZN * A32 + RN * A42
375 J1 = J1 + NUF
380 I1 = I1 + NUF
390 CONTINUE
C.... CONSTRUCT SYMMETRIC ESTIF
DO 395, I = 2, NSTF
K = I-1
DO 395, J = 1, K
ESTIF(I, J) = ESTIF(I, J) + ESTIF(J, I)
395 ESTIF(J, I) = ESTIF(I, J)
IF (C6.EQ.0) RETURN
DO 190, I = 1, NSTF
ESTIF(I, I) = ESTIF(I, I) + C6 * FORCE(I, 2)
190 RETURN
C.... STRESS COMPUTATION FOR THE VISCOELASTIC ELEMENT
C.... VISCOELASTIC LOAD HISTORY COMPUTATION AND ELEMENT ASSEMBLY FROM TP
* CONTINUE
C.... COMPUTATION AND OUTPUT OF STRESS AND HISTORY OF VISCOELASTIC SOLN.
C.... DO THE INTEGRATION AND OUTPUT THE STRESS
NG = 0(1, MA)
NR = 0(2, MA)
GG = 0(4, MA)
XR = 0(5, MA)
C.... CHECK FOR H STORAGE
LL = NH * (LINT * 1) * (NR * NG * ((NDIM * 1) * NDIM) / 2 * NG)
IF (LL.GT.152H) CALL VHPFF(IITRD, I1 * R, H, NH, 152H)
NL = LINT
DO 500, LL = 1, NL
NOPRNT = NR
IF (NSIG(LL).NE.0) NOPRNT = .TRUE.
NPL = .FALSE.

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FLM 70C
FLM 71C
FLM 72C
FLM 73C
FLM 74C
FLM 75C
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FLM137C
FLM139C
FLM140C
FLM141C
FLM142C

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	IF (NUMPLT, LE, 0) GO TO 210	FLM143C
	DO 205 NP = 1, NUMPLT	FLM144C
205	IF (NEDATA(NP, 1) .EQ. N .AND. NEDATA(NP, 2) .EQ. LL) NPL = .TRUE.	FLM145C
210	CONTINUE	FLM146C
	IF (LL, NE, 1) GO TO 220	FLM147C
	S = 0.0	FLM148C
	T = 0.0	FLM149C
	GO TO 230	FLM150C
220	S = STOW(1, LL-1)	FLM151C
	T = STOW(2, LL-1)	FLM152C
	XX = STOW(4, LL-1)	FLM153C
230	CALL MUCK2(S, T, -1, 2, NEL, NSIDE)	FLM154C
C...	COMPUTE THE PRESENT STRAIN AND STRAIN INCREMENT	FLM155C
	DTH = 0.	FLM156C
	THP = 0.	FLM157C
	DO 33 I = 1, 2	FLM158C
	CC = 0.0	FLM159C
	DO 300 K = 1, NEL	FLM160C
300	CC = CC + SHAPE(4, K) * X(I, K)	FLM161C
	XX(I) = CC	FLM162C
	DO 320 J = 1, 2	FLM163C
	CC = 0.	FLM164C
	DD = 0.	FLM165C
	DO 310 K = 1, NEL	FLM166C
	CC = CC + SHAPE(1, K) * UL(J, K) + SHAPE(J, K) * UL(I, K)	FLM167C
310	DD = DD + SHAPE(1, K) * DUL(J, K) + SHAPE(J, K) * DUL(I, K)	FLM168C
	DEPS(1, J) = DD / XJAC	FLM169C
	DEPS(J+1, I) = CC / XJAC	FLM170C
	XIG(1, J) = GG * DEPS(J+1, I)	FLM171C
320	XIG(J+1, I) = XIG(1, J)	FLM172C
	DTH = DTH + DEPS(1, I)	FLM173C
330	THP = THP + DEPS(1+1, I)	FLM174C
	HR = XX(I)	FLM175C
	ET = 0.	FLM176C
	DET = 0.	FLM177C
	DO 350 K = 1, NEL	FLM178C
	CC = SHAPE(4, K)	FLM179C
	ET = ET + CC * UL(1, K)	FLM180C
350	DET = DET + CC * DUL(1, K)	FLM181C
	ST = GG * ET * 2.0	FLM182C
	STF = ST	FLM183C
	XIK = (THP + ET * 2. / HR) * (XK / 2. - GG / 3.)	FLM184C
	YIK = XIK	FLM185C
	DTH = DTH / 2. + DET / HR	FLM186C
	DET = DET * 2.0 - DTH * HR * IT	FLM187C
	DO 34 I = 1, 2	FLM188C
	DEPS(1, I) = DEPS(1, I) - DTH * IT	FLM189C
340	DEPS(1+1, I) = DEPS(1+1, I) / 2.0	FLM190C
C...	UPDATE THE SHEAR HISTORY	FLM191C
	IG = 4	FLM192C
	IF (NG, LE, 3) GO TO 410	FLM193C
	DO 400 I = 1, NG	FLM194C
	IG = IG + 2	FLM195C
	DD = U(IG, 1, MA)	FLM196C
	CC = U(IG, MA) * HIST(UTP / DD)	FLM197C
	DD = EXP(-DT / DD)	FLM198C
	DO 402 J = 1, 2	FLM199C
	DO 402 K = J, 2	FLM200C
	EE = H(NH) + CC * DEPS(J, K)	FLM201C
	XIG(J, K) = XIG(J, K) + EE	FLM202C
	EE = LE * DD	FLM203C
	XIG(K+1, J) = XIG(K+1, J) + EE	FLM204C
	H(NH) = EE	FLM205C
402	NH = NH + 1	FLM206C
	EE = H(NH) + CC * DET	FLM207C
	ST = ST + EE	FLM208C
	EE = LE * DD	FLM209C
	H(NH) = EE	FLM210C
	STF = STF + EE	FLM211C
	NH = NH + 1	FLM212C
400	CONTINUE	FLM213C
C...	UPDATE THE BULK HISTORY	FLM214C

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410 IF(NK,LE,3) GO TO 430
DO 420 I = 1,NK
IG = IG + 2
DD = D(IG,1,MA)
EE = H(NH) * P(IG,MA)*HIST(DT/UDU)*DTH
XIK = XIK + EE
EE = LE*EXP(-DT/UDU)
YIK = YIK + EE
H(NH) = EE
420 NH = NH + 1
430 CONTINUE
DO 435 I = 1,2
XIG(I,1) = XIG(I,1) + XIK
435 XIG(I+1,1) = XIG(I+1,1) + YIK
ET = ET/XX(1)
ST = ST/XX(1) + XIK
C.... OUTPUT THE STRESSES FOR THE CURRENT TIME STEP
IF(NOMRNT) GO TO 640
MCT = MCT + 1
IF(MCT.GT,3) GO TO 630
WRITE(ITPE,2004) HEAD,TIME,IPG
IPG = IPG + 1
MCT = 16
630 WRITE(ITPE,2003) N,(XX(1),I=1,2),((XIG(I,J),J=1,2),I=1,2),ST,
DM,MA,((DEPS(J+1,I),J=1,2),I=1,2),ET
440 IF(NPL) CALL PLDATA(NUMEL,NDIM,N,LL,SMED,XX,XIG,FORCE)
IF(IPLT2D.NE.1) GO TO 641
IF(ILL.NE.1) GO TO 641
VFCT(N,1) = XIG(1,1)
VFCT(N,2) = XIG(2,2)
VFCT(N,3) = ST
VFCT(N,4) = XIG(1,2)
VFCT(N,5) = DEPS(2,1)
VFCT(N,6) = DEPS(3,2)
VFCT(N,7) = ET
VFCT(N,8) = DEPS(3,1)
641 CONTINUE
IF(IJSEW.4) GO TO 500
C.... COMPUTE THE LOADS FOR THE NEXT TIME STEP
DO = ***KH
XIG(1,1) = XIG(2,1)*DD
XIG(1,2) = XIG(3,1)*DD
XIG(2,2) = XIG(3,2)*DD
XIG(2,1) = XIG(1,2)
STF = (STF/KR + YIK)***XJAC
** = XJAC*DD*D(3,MA)
I1 = 0
DO 470 I = 1,NEL
CC = SHAPE(4,I)
FORCE(I1+1,1) = FORCE(I1+1,1) - STF*CC
EE = CC**
DO 460 J = 1,2
CC = **
DO 450 K = 1,2
CC = CC + SHAPE(K,I)*XIG(J,K)
450 FORCE(I1+J,1) = FORCE(I1+J,1) - CC*EE*UDL(J,I)
460 I1 = I1 + NDF
470 I1 = I1 + NDF
500 CONTINUE
RETURN
C.... ELEMENT LOAD COMPUTATION FOR BODY TYPE FORCES, ETC.
5 CONTINUE
RETURN
4000 IPG = 0
RETURN
C.... FORMAT STATEMENTS
1000 FORMAT(2I5,4F10.0)
1001 FORMAT(5X,10.0)
2000 FORMAT(5X,57H LINEAR VISCOELASTIC MATERIAL, AXISYMMETRIC SOLID ELE
XMENT //10,12H SHEAR TERMS/
X 110,12H BULK TERMS//10X,12H G-INFINITY=,1PE12.4/10X,
X 12H K-INFINITY=,1PE12.4/10X,12H DENSITY =,1PE12.4/1X)
2001 FORMAT(10X,14H SHEAR MODULUS,6X,4H TIME/(10X,1PE12.4))

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FLM215C
FLM216C
FLM217C
FLM218C
FLM219C
FLM220C
FLM221C
FLM222C
FLM223C
FLM224C
FLM225C
FLM226C
FLM227C
FLM228C
FLM229C
FLM230C
FLM231C
FLM232C
FLM233C
FLM234C
FLM235C
FLM236C
FLM237C
FLM238C
FLM239C
FLM240C
FLM242C
FLM243C
FLM244C
FLM245C
FLM246C
FLM247C
FLM248C
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FLM257C
FLM258C
FLM259C
FLM260C
FLM261C
FLM262C
FLM263C
FLM264C
FLM265C
FLM266C
FLM267C
FLM268C
FLM269C
FLM270C
FLM271C
FLM272C
FLM273C
FLM274C
FLM275C

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2002	FORMAI(10X,14H BULK MODULUS,6X,4MTIME/(10X,1P2E12.4))	FLM276C
2003	FORMAI(1HU,17,0P2F12.3,1P4E12.3/8X,A5,17,12X,1P4E12.3)	FLM277C
2004	FORMAI(1H1,12A6,E13.5,20X,4HPAGE,13//5X,16MELEMENT STRESSES//	FLM278C
	1 1X,7HELEMENT,4X,6HM ORD.,6X,6HZ ORD.,5X,9HRR-STRESS,3X,	FLM279C
	2 9HRZ-STRESS,3X,9HZZ-STRESS,3X,9HTT-STRESS/35X,9HRR-STRAIN,3X,	FLM280C
	3 9HRZ-STRAIN,3X,9HZZ-STRAIN,3X,9HTT-STRAIN)	FLM281C
2030	FORMAI(9H MATERIAL,13,17H EXCEEDS STORAGE.,15,13H SHEAR TERMS.,15,	FLM282C
	X 16H BULK TERMS NEED,14,7H #OROS. )	FLM283C
	END	FLM284C
*WEUH		



Appendix G

SAMPLE INPUT DATA

PRECEDING PAGE BLANK NOT FILMED

50000

FEAPTS

1	653	667	671	671	682	696	703	704	651	695	713	715	647	694	712
2	716	645	649	709	714	638	680	699	700	617	658	661	662	650	666
3	668	649	649	693	701	702	648	692	707	710	646	691	706	711	644
4	689	715	708	637	679	697	698	616	657	659	660	643	663	664	655
5	642	676	681	682	641	675	685	687	649	674	684	688	639	673	683
6	686	636	672	677	678	615	654	655	656	611	618	619	620	610	625
7	628	629	609	624	632	634	608	623	631	635	607	622	630	633	606
8	621	626	627	625	612	613	614	531	535	536	537	530	542	545	546
9	529	541	544	551	528	540	548	552	527	539	547	550	526	538	543
10	544	525	532	533	514	595	600	603	604	594	599	601	602	593	596
11	597	598	589	590	591	592	521	522	523	524	572	571	566	567	516
12	436	439	440	576	569	565	556	515	435	443	445	568	567	564	555
13	514	434	442	446	563	562	561	554	513	433	441	444	560	559	558
14	553	512	432	437	431	517	518	519	520	573	574	575	576	577	580
15	581	562	578	583	585	586	579	584	587	588	502	507	510	511	501
16	506	508	509	500	503	504	505	496	497	498	499	488	489	490	491
17	495	494	493	492	487	429	430	431	461	460	459	452	449	425	345
18	347	458	457	456	451	448	424	344	348	455	454	453	450	447	423
19	343	446	470	469	468	467	462	426	427	428	463	464	465	466	471
20	472	473	474	475	478	479	480	476	481	483	484	477	482	485	486
21	473	478	471	422	412	417	419	420	411	414	415	416	407	408	409
22	410	399	400	401	402	406	405	404	403	398	390	391	392	397	396
23	395	394	393	369	341	342	354	353	352	351	350	349	338	258	363
24	362	361	360	359	355	339	340	372	371	370	369	364	356	357	358
25	365	366	367	368	373	374	375	376	377	380	381	382	378	383	385
26	386	379	384	387	388	328	333	336	337	327	332	334	335	326	329
27	330	331	322	323	324	325	314	315	316	317	321	320	319	318	313
28	305	306	307	312	311	310	304	308	304	296	297	303	302	301	300
29	299	298	295	297	269	268	267	266	265	261	259	260	278	277	276
30	275	276	262	263	264	271	272	273	274	279	280	281	282	283	286
31	287	288	284	289	291	292	285	290	293	294	247	252	255	256	246
32	251	252	254	245	248	249	250	241	242	243	244	233	234	235	236
33	240	239	238	237	232	224	225	226	231	230	229	228	227	223	215
34	216	222	221	220	219	218	217	214	206	213	212	211	210	209	208
35	207	215	188	187	186	185	180	177	178	179	181	182	183	184	189
36	190	191	192	193	196	197	198	194	199	201	202	195	200	203	204
37	153	154	155	156	152	144	145	146	147	143	135	136	138	137	134
38	126	129	128	127	125	120	119	118	117	97	98	99	100	161	162
39	163	164	157	17	78	79	148	71	63	64	139	65	62	56	130
40	58	57	55	121	51	50	49	101	102	103	104	165	168	169	170
41	154	82	87	88	149	72	33	34	140	66	22	18	131	59	19
42	17	122	52	43	44	105	108	109	110	166	171	173	175	159	83
43	89	93	150	73	35	37	141	67	23	7	132	60	20	15	123
44	53	45	47	106	111	113	114	167	172	174	176	160	84	90	94
45	151	74	36	38	142	68	24	8	133	61	21	16	124	54	46
46	44	117	112	115	116	81	86	92	95	80	85	91	96	76	75
47	70	69	30	32	40	41	29	31	39	42	28	27	26	25	6
48	10	12	13	5	9	11	14	6	3	2	1				

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3

3

R

350

A MATER

1 ELM21

HAND BONE--OUTER TABLE

335570. 487600. .00027

2 ELM21

POROUS BONE--DIPOLE

335570. 487600. .00002

3 ELM21

HAND BONE--INNER TABLE

335570. 487600. .00027

4 ELM22

SHEARLESS SOLID--SUBARACHNOID SPACE

3.227 305. .0000937

5 ELM22

SHEARLESS SOLID--HRAIN

3.227 305000. .0000937

6 ELM21

NECK COVER

355. 487. .00027

716 NODAL

1

0 2.226 -1.455 3.469 5

2

0 2.503 -1.017 3.535 5

3

0 2.634 -.581 3.593 5

4

1 2.717 -0.000 3.587 5

5	0	1.600	-1.723	3.617	5
6	0	1.648	-1.297	3.982	5
7	0	1.691	-.751	4.218	5
8	1	1.690	-0.000	4.378	5
9	0	.792	-2.096	3.803	5
10	0	.775	-1.523	4.212	5
11	0	.780	-.878	4.581	5
12	1	.774	-0.000	4.692	5
13	0	.081	-2.228	3.876	5
14	0	-.033	-1.619	4.346	5
15	0	-.038	-.819	4.687	5
16	1	-.033	-0.000	4.928	5
17	0	-.815	-2.243	3.839	5
18	0	-.835	-1.610	4.366	5
19	0	-.827	-.772	4.734	5
20	1	-.834	-0.000	4.936	5
21	0	-1.514	-1.961	3.785	5
22	0	-1.597	-1.439	4.236	5
23	0	-1.608	-.761	4.589	5
24	1	-1.587	-0.000	4.694	5
25	0	-2.072	-1.579	3.680	5
26	0	-2.287	-1.101	3.825	5
27	0	-2.323	-.594	3.993	5
28	1	-2.311	-0.000	4.083	5
29	0	2.204	-1.429	3.439	5
30	0	2.480	-1.000	3.504	5
31	0	2.611	-.573	3.564	5
32	1	2.695	-0.000	3.563	5
33	0	1.580	-1.692	3.582	5
34	0	1.623	-1.267	3.939	5
35	0	1.668	-.736	4.176	5
36	1	1.667	-0.000	4.336	5
37	0	.774	-2.057	3.765	5
38	0	.761	-1.497	4.172	5
39	0	.765	-.861	4.528	5
40	1	.760	-0.000	4.647	5
41	0	.078	-2.183	3.832	5
42	0	-.037	-1.589	4.297	5
43	0	-.041	-.801	4.641	5
44	1	-.040	-0.000	4.871	5
45	0	-.801	-2.190	3.792	5
46	0	-.826	-1.575	4.312	5
47	0	-.820	-.752	4.678	5
48	1	-.823	-0.000	4.875	5
49	0	-1.481	-1.909	3.744	5
50	0	-1.568	-1.404	4.190	4
51	0	-1.576	-.743	4.529	4
52	1	-1.553	-0.000	4.633	4
53	0	-2.029	-1.535	3.642	4
54	0	-2.242	-1.076	3.789	4
55	0	-2.275	-.583	3.955	4
56	1	-2.268	-0.000	4.045	4
57	0	2.161	-1.376	3.380	4
58	0	2.433	-.967	3.441	4
59	0	2.564	-.558	3.508	4
60	1	2.652	-0.000	3.514	4
61	0	1.540	-1.631	3.514	4
62	0	1.574	-1.209	3.853	4
63	0	1.623	-.705	4.092	4
64	1	1.622	-0.000	4.252	4
65	0	.738	-1.979	3.688	4
66	0	.734	-1.446	4.091	4
67	0	.735	-.827	4.423	4
68	1	.733	-0.000	4.556	4
69	0	.073	-2.094	3.744	4
70	0	-.046	-1.530	4.199	4
71	0	-.047	-.767	4.547	4
72	1	-.053	-0.000	4.757	4
73	0	-.774	-2.083	3.697	4
74	0	-.809	-1.505	4.204	4
75	0	-.807	-.711	4.566	4
76	1	-.801	-0.000	4.752	4

77	0	-1.414	-1.803	3.661	4
78	0	-1.509	-1.335	4.097	4
79	0	-1.512	-1.708	4.410	4
80	1	-1.484	-0.000	4.511	4
81	0	-1.443	-1.445	3.567	4
82	0	-2.154	-1.025	3.716	4
83	0	-2.178	-1.561	3.878	4
84	1	-2.183	-0.000	3.968	4
85	0	2.140	-1.350	3.350	4
86	0	2.410	-1.950	3.410	4
87	0	2.540	-1.550	3.480	4
88	1	2.630	-0.000	3.490	4
89	0	1.520	-1.600	3.480	4
90	0	1.550	-1.180	3.810	4
91	0	1.600	-1.690	4.050	4
92	1	1.600	-0.000	4.210	4
93	0	.726	-1.940	3.650	4
94	0	.726	-1.420	4.050	4
95	0	.726	-1.910	4.370	4
96	1	.726	-0.000	4.510	4
97	0	.076	-2.050	3.700	4
98	0	-1.050	-1.500	4.150	4
99	0	-1.050	-1.750	4.500	3
100	1	-1.050	-0.000	4.700	3
101	0	-1.760	-2.030	3.650	3
102	0	-1.800	-1.470	4.150	3
103	0	-1.800	-1.690	4.510	3
104	1	-1.790	-0.000	4.690	3
105	0	-1.340	-1.750	3.620	3
106	0	-1.480	-1.300	4.050	3
107	0	-1.480	-1.690	4.350	3
108	1	-1.450	-0.000	4.450	3
109	0	-1.400	-1.400	3.530	3
110	0	-2.110	-1.000	3.680	3
111	0	-2.130	-1.550	3.840	3
112	1	-2.140	-0.000	3.930	3
113	0	2.119	-1.324	3.320	3
114	0	2.347	-1.933	3.379	3
115	0	2.516	-1.542	3.452	3
116	1	2.608	-0.000	3.466	3
117	0	1.500	-1.569	3.446	3
118	0	1.526	-1.151	3.767	3
119	0	1.577	-1.675	4.008	3
120	1	1.578	-0.000	4.168	3
121	0	.702	-1.901	3.612	3
122	0	.706	-1.394	4.009	3
123	0	.705	-1.793	4.317	3
124	1	.707	-0.000	4.464	3
125	0	.067	-2.006	3.656	3
126	0	-1.054	-1.470	4.101	3
127	0	-1.053	-1.733	4.453	3
128	1	-1.067	-0.000	4.643	3
129	0	-1.746	-1.977	3.603	3
130	0	-1.791	-1.435	4.096	3
131	0	-1.793	-1.664	4.454	3
132	1	-1.779	-0.000	4.628	3
133	0	-1.346	-1.697	3.579	3
134	0	-1.451	-1.265	4.003	3
135	0	-1.448	-1.672	4.290	3
136	1	-1.416	-0.000	4.389	3
137	0	-1.857	-1.355	3.493	3
138	0	-2.066	-1.975	3.644	3
139	0	-2.082	-1.539	3.802	3
140	1	-2.097	-0.000	3.892	3
141	0	2.317	-1.751	3.066	3
142	0	2.685	-1.191	3.166	3
143	0	2.932	-1.535	3.184	3
144	1	3.019	-0.000	3.153	3
145	0	2.288	-1.718	3.042	3
146	0	2.656	-1.174	3.144	3
147	0	2.899	-1.526	3.163	3
148	1	2.981	-0.000	3.132	2

149	0	2.229	-1.653	2.994	2
150	0	2.599	-1.138	3.101	2
151	0	2.833	-.509	3.121	2
152	1	2.907	-0.000	3.091	2
153	0	2.200	-1.620	2.970	2
154	0	2.570	-1.120	3.080	2
155	0	2.800	-.500	3.100	2
156	1	2.870	-0.000	3.070	2
157	0	2.171	-1.587	2.946	2
158	0	2.541	-1.102	3.059	2
159	0	2.767	-.491	3.079	2
160	1	2.833	-0.000	3.049	2
161	0	1.683	-2.194	3.111	2
162	0	1.662	-2.158	3.083	2
163	0	1.621	-2.086	3.028	2
164	0	1.600	-2.050	3.000	2
165	0	1.579	-2.014	2.972	2
166	0	1.674	-1.301	3.161	2
167	0	1.768	-.648	3.287	2
168	1	1.797	-0.000	3.337	2
169	0	.777	-2.589	3.121	2
170	0	.763	-2.547	3.098	2
171	0	.734	-2.462	3.053	2
172	0	.720	-2.420	3.030	2
173	0	.706	-2.378	3.007	2
174	0	.817	-1.479	3.266	2
175	0	.863	-.735	3.443	2
176	1	.877	-0.000	3.514	2
177	0	.073	-2.712	3.220	2
178	0	.072	-2.659	3.197	2
179	0	.071	-2.553	3.152	2
180	0	.070	-2.500	3.130	2
181	0	.069	-2.447	3.108	2
182	0	.039	-1.531	3.335	2
183	0	.040	-.745	3.523	2
184	1	.040	-0.000	3.604	2
185	0	-.791	-2.622	3.217	2
186	0	-.776	-2.567	3.193	2
187	0	-.745	-2.456	3.144	2
188	0	-.730	-2.400	3.120	2
189	0	-.715	-2.344	3.096	2
190	0	-.732	-1.476	3.334	2
191	0	-.750	-.715	3.516	2
192	1	-.755	-0.000	3.593	2
193	0	-1.504	-2.328	3.159	2
194	0	-1.473	-2.279	3.137	2
195	0	-1.411	-2.180	3.092	2
196	0	-1.380	-2.130	3.070	2
197	0	-1.349	-2.080	3.048	1
198	0	-1.470	-1.315	3.271	1
199	0	-1.522	-.665	3.413	1
200	1	-1.543	-0.000	3.475	1
201	0	-1.934	-1.606	3.028	1
202	0	-2.274	-1.054	3.151	1
203	0	-2.339	-.598	3.184	1
204	1	-2.422	-0.000	3.276	1
205	0	-1.980	-1.650	3.050	1
206	0	-2.330	-1.080	3.170	1
207	0	-2.400	-.610	3.210	1
208	1	-2.480	-0.000	3.300	1
209	0	-2.026	-1.694	3.072	1
210	0	-2.386	-1.106	3.189	1
211	0	-2.461	-.622	3.236	1
212	1	-2.538	-0.000	3.324	1
213	0	-2.119	-1.782	3.116	1
214	0	-2.499	-1.157	3.228	1
215	0	-2.584	-.645	3.288	1
216	1	-2.653	-0.000	3.373	1
217	0	-2.165	-1.826	3.139	1
218	0	-2.556	-1.183	3.247	1
219	0	-2.645	-.656	3.313	1
220	1	-2.711	-0.000	3.397	1



221	0	2.409	-2.008	2.582	1
222	0	2.915	-1.380	2.676	1
223	0	3.181	-0.630	2.704	1
224	1	3.307	-0.000	2.682	1
225	0	2.377	-1.969	2.562	1
226	0	2.886	-1.360	2.657	1
227	0	3.151	-0.623	2.688	1
228	1	3.260	-0.000	2.664	1
229	0	2.312	-1.890	2.521	1
230	0	2.829	-1.320	2.619	1
231	0	3.090	-0.608	2.656	1
232	1	3.167	-0.000	2.628	1
233	0	2.280	-1.850	2.500	1
234	0	2.800	-1.300	2.600	1
235	0	3.060	-0.600	2.640	1
236	1	3.120	-0.000	2.610	1
237	0	2.248	-1.810	2.479	1
238	0	2.771	-1.280	2.581	1
239	0	3.030	-0.592	2.624	1
240	1	3.073	-0.000	2.592	1
241	0	1.771	-2.400	2.538	1
242	0	1.753	-2.362	2.529	1
243	0	1.718	-2.287	2.510	1
244	0	1.700	-2.250	2.500	1
245	0	1.682	-2.213	2.490	1
246	0	1.811	-1.413	2.613	5
247	0	1.929	-0.686	2.694	5
248	1	1.962	-0.000	2.715	5
249	0	.854	-2.737	2.448	5
250	0	.846	-2.690	2.448	5
251	0	.829	-2.597	2.449	5
252	0	.820	-2.550	2.450	5
253	0	.811	-2.503	2.451	4
254	0	.913	-1.538	2.642	4
255	0	.973	-0.744	2.753	4
256	1	.990	-0.000	2.792	4
257	0	.965	-2.800	2.520	4
258	0	.966	-2.737	2.520	4
259	0	.969	-2.612	2.520	4
260	0	.970	-2.550	2.520	3
261	0	.971	-2.488	2.520	3
262	0	.991	-1.565	2.679	3
263	0	.102	-0.758	2.786	3
264	1	.104	-0.000	2.826	3
265	0	-.733	-2.754	2.597	3
266	0	-.719	-2.695	2.593	3
267	0	-.693	-2.578	2.584	2
268	0	-.680	-2.520	2.580	2
269	0	-.667	-2.462	2.576	2
270	0	-.703	-1.518	2.692	2
271	0	-.735	-0.736	2.782	2
272	1	-.747	-0.000	2.817	2
273	0	-1.490	-2.422	2.576	2
274	0	-1.463	-2.371	2.575	1
275	0	-1.408	-2.270	2.572	1
276	0	-1.380	-2.220	2.570	1
277	0	-1.352	-2.170	2.568	1
278	0	-1.493	-1.352	2.674	1
279	0	-1.585	-0.686	2.742	1
280	1	-1.618	-0.000	2.765	1
281	0	-1.934	-1.768	2.544	5
282	0	-2.345	-1.064	2.661	4
283	0	-2.518	-0.675	2.689	3
284	1	-2.585	-0.000	2.669	2
285	0	-1.980	-1.810	2.550	1
286	0	-2.410	-1.100	2.670	0
287	0	-2.580	-0.690	2.710	0
288	1	-2.609	-0.000	2.690	0
289	0	-2.026	-1.852	2.556	0
290	0	-2.475	-1.136	2.679	0
291	0	-2.642	-0.705	2.731	1
292	1	-2.735	-0.000	2.711	2

293	0	-2.117	-1.937	2.568	3
294	0	-2.615	-1.208	2.697	4
295	0	-2.767	-.734	2.773	5
296	1	-2.884	-0.000	2.753	5
297	0	-2.162	-1.980	2.574	4
298	0	-2.670	-1.245	2.705	3
299	0	-2.830	-.748	2.795	2
300	1	-2.958	-0.000	2.774	1
301	0	2.415	-2.157	2.101	0
302	0	3.064	-1.546	2.197	0
303	0	3.374	-.723	2.191	0
304	1	3.438	-0.000	2.187	0
305	0	2.391	-2.118	2.088	0
306	0	3.036	-1.524	2.185	1
307	0	3.343	-.717	2.183	2
308	1	3.384	-0.000	2.180	3
309	0	2.344	-2.039	2.063	4
310	0	2.979	-1.481	2.162	5
311	0	3.281	-.706	2.168	5
312	1	3.275	-0.000	2.167	4
313	0	2.320	-2.000	2.050	3
314	0	2.950	-1.460	2.150	2
315	0	3.250	-.700	2.160	1
316	1	3.220	-0.000	2.160	0
317	0	2.296	-1.561	2.037	0
318	0	2.921	-1.439	2.138	0
319	0	3.219	-.694	2.152	0
320	1	3.165	-0.000	2.153	0
321	0	1.766	-2.424	1.991	1
322	0	1.755	-2.380	1.993	2
323	0	1.732	-2.293	1.998	3
324	0	1.720	-2.250	2.000	4
325	0	1.708	-2.207	2.002	5
326	0	1.897	-1.460	2.113	5
327	0	2.031	-.719	2.169	4
328	1	2.051	-0.000	2.185	3
329	0	.898	-2.563	1.868	2
330	0	.891	-2.517	1.878	1
331	0	.877	-2.426	1.899	0
332	0	.870	-2.380	1.910	0
333	0	.863	-2.334	1.921	0
334	0	.977	-1.523	2.089	0
335	0	1.039	-.748	2.164	0
336	1	1.054	-0.000	2.188	1
337	0	.089	-2.749	1.979	2
338	0	.084	-2.687	1.982	3
339	0	.075	-2.562	1.987	4
340	0	.070	-2.500	1.990	5
341	0	.065	-2.438	1.993	5
342	0	.125	-1.558	2.098	4
343	0	.137	-.758	2.153	3
344	1	.137	-0.000	2.171	2
345	0	-.690	-2.752	2.030	1
346	0	-.675	-2.689	2.035	0
347	0	-.645	-2.563	2.045	0
348	0	-.630	-2.500	2.050	0
349	0	-.615	-2.437	2.055	0
350	0	-.681	-1.518	2.108	0
351	0	-.729	-.739	2.138	1
352	1	-.744	-0.000	2.152	2
353	0	-1.494	-2.381	2.043	3
354	0	-1.465	-2.326	2.050	4
355	0	-1.408	-2.215	2.063	5
356	0	-1.380	-2.160	2.070	1
357	0	-1.352	-2.105	2.077	1
358	0	-1.503	-1.357	2.108	1
359	0	-1.625	-.690	2.130	1
360	1	-1.661	-0.000	2.147	1
361	0	-1.940	-1.754	2.014	1
362	0	-2.369	-1.107	2.141	1
363	0	-2.655	-.666	2.169	2
364	1	-2.718	-0.000	2.236	2

365	0	-1.940	-1.800	2.010	2
366	0	-2.430	-1.150	2.150	2
367	0	-2.730	-.690	2.180	2
368	1	-2.780	-0.000	2.240	2
369	0	-2.020	-1.841	2.006	2
370	0	-2.491	-1.193	2.159	3
371	0	-2.805	-.714	2.191	3
372	1	-2.842	-0.000	2.244	3
373	0	-2.100	-1.923	1.997	3
374	0	-2.613	-1.278	2.177	3
375	0	-2.956	-.762	2.214	3
376	1	-2.967	-0.000	2.251	3
377	0	-2.140	-1.965	1.993	4
378	0	-2.674	-1.321	2.186	4
379	0	-3.032	-.785	2.228	4
380	1	-3.070	-0.000	2.254	4
381	0	2.475	-2.163	1.680	4
382	0	3.155	-1.581	1.685	4
383	0	3.419	-.719	1.731	4
384	1	3.470	-0.000	1.790	5
385	0	2.459	-2.133	1.678	5
386	0	3.124	-1.561	1.681	5
387	0	3.382	-.714	1.731	5
388	1	3.407	-0.000	1.790	5
389	0	2.426	-2.071	1.673	5
390	0	3.061	-1.520	1.674	5
391	0	3.307	-.705	1.730	5
392	1	3.282	-0.000	1.790	5
393	0	2.410	-2.040	1.670	5
394	0	3.030	-1.500	1.670	5
395	0	3.270	-.700	1.730	5
396	1	3.220	-0.000	1.790	5
397	0	2.394	-2.009	1.667	5
398	0	2.499	-1.480	1.666	4
399	0	3.233	-.695	1.730	4
400	1	3.157	-0.000	1.790	4
401	0	1.758	-2.367	1.554	4
402	0	1.748	-2.325	1.568	4
403	0	1.729	-2.242	1.596	4
404	0	1.720	-2.200	1.610	4
405	0	1.711	-2.158	1.624	3
406	0	1.935	-1.458	1.668	3
407	0	2.049	-.726	1.703	3
408	1	2.062	-0.000	1.717	3
409	0	.983	-2.526	1.447	3
410	0	.972	-2.469	1.448	3
411	0	.951	-2.356	1.449	3
412	0	.940	-2.300	1.450	2
413	0	.929	-2.244	1.451	2
414	0	1.025	-1.501	1.596	2
415	0	1.061	-.744	1.635	2
416	1	1.067	-0.000	1.650	2
417	0	.164	-2.749	1.500	2
418	0	.158	-2.687	1.500	2
419	0	.146	-2.562	1.500	1
420	0	.140	-2.500	1.500	1
421	0	.134	-2.438	1.500	1
422	0	.164	-1.544	1.568	1
423	0	.145	-.746	1.560	1
424	1	.138	-0.000	1.556	1
425	0	-.695	-2.689	1.538	1
426	0	-.679	-2.629	1.546	5
427	0	-.646	-2.510	1.562	4
428	0	-.630	-2.450	1.570	3
429	0	-.614	-2.390	1.578	2
430	0	-.664	-1.502	1.554	1
431	0	-.722	-.732	1.507	0
432	1	-.737	-0.000	1.499	0
433	0	-1.476	-2.286	1.569	0
434	0	-1.452	-2.239	1.574	0
435	0	-1.404	-2.146	1.585	0
436	0	-1.380	-2.100	1.590	1

437	0	-1.356	-2.054	1.595	2
438	0	-1.495	-1.369	1.520	3
439	0	-1.621	-.689	1.475	4
440	1	-1.633	-0.000	1.471	5
441	0	-1.834	-1.741	1.531	5
442	0	-2.429	-1.210	1.499	4
443	0	-2.719	-.638	1.478	3
444	1	-2.628	-0.000	1.506	2
445	0	-1.880	-1.800	1.530	1
446	0	-2.480	-1.250	1.500	0
447	0	-2.780	-.650	1.470	0
448	1	-2.730	-0.000	1.480	0
449	0	-1.926	-1.859	1.529	0
450	0	-2.531	-1.290	1.501	0
451	0	-2.841	-.662	1.462	1
452	1	-2.832	-0.000	1.454	2
453	0	-2.017	-1.978	1.526	3
454	0	-2.634	-1.370	1.502	4
455	0	-2.962	-.687	1.447	5
456	1	-3.035	-0.000	1.403	5
457	0	-2.063	-2.038	1.524	4
458	0	-2.685	-1.410	1.503	3
459	0	-3.023	-.699	1.440	2
460	111	-3.137	-0.000	1.378	1
461	0	2.475	-2.134	1.369	0
462	0	3.112	-1.566	1.198	0
463	0	3.505	-.784	1.113	0
464	111	3.638	-0.000	1.182	0
465	0	2.461	2.100	1.379	0
466	0	3.089	-1.551	1.236	1
467	0	3.441	-.776	1.172	2
468	1	3.533	-0.000	1.224	3
469	0	2.434	-2.033	1.400	4
470	0	3.043	-1.517	1.312	5
471	0	3.314	-.759	1.291	5
472	1	3.324	-0.000	1.308	4
473	0	2.420	-2.000	1.410	3
474	0	3.020	-1.500	1.350	2
475	0	3.250	-.750	1.350	1
476	1	3.220	-0.000	1.350	0
477	0	2.406	-1.967	1.420	0
478	0	2.997	-1.483	1.388	0
479	0	3.186	-.741	1.409	0
480	1	3.116	-0.000	1.392	0
481	0	1.779	-2.184	1.212	1
482	0	1.764	-2.143	1.237	2
483	0	1.735	-2.061	1.286	3
484	0	1.720	-2.020	1.310	4
485	0	1.705	-1.979	1.334	5
486	0	1.927	-1.424	1.309	5
487	0	1.974	-.739	1.297	4
488	1	1.998	-0.000	1.271	3
489	0	1.016	-2.502	1.114	2
490	0	1.005	-2.464	1.118	1
491	0	.982	-2.388	1.126	0
492	0	.970	-2.350	1.130	0
493	0	.958	-2.312	1.134	0
494	0	1.082	-1.497	1.162	0
495	0	1.040	-.742	1.139	0
496	1	1.030	-0.000	1.166	1
497	0	.070	-2.658	1.104	2
498	0	.070	-2.618	1.110	3
499	0	.070	-2.539	1.123	4
500	0	.070	-2.500	1.130	5
501	0	.070	-2.461	1.137	1
502	0	.217	-1.517	1.099	1
503	0	.092	-.702	.941	1
504	1	.070	-0.000	.899	1
505	0	-.802	-2.585	1.018	1
506	0	-.784	-2.526	1.046	1
507	0	-.748	-2.409	1.102	1
508	0	-.730	-2.350	1.130	2

509	0	-.712	-2.291	1.158	2
510	0	-.633	-1.459	1.042	2
511	0	-.724	-.714	.814	2
512	1	-.740	-0.000	.803	2
513	0	-1.485	-2.298	1.106	2
514	0	-1.459	-2.236	1.110	2
515	0	-1.406	-2.112	1.117	3
516	0	-1.380	-2.090	1.120	3
517	0	-1.354	-1.988	1.123	3
518	0	-1.398	-1.405	.890	3
519	0	-1.534	-.709	.743	3
520	1	-1.532	-0.000	.723	3
521	0	-1.837	-1.795	.920	3
522	0	-2.269	-1.260	.786	4
523	0	-2.461	-.745	.688	4
524	1	-2.271	-0.000	.705	4
525	0	-1.880	-1.850	.920	4
526	0	-2.330	-1.300	.770	4
527	0	-2.530	-.750	.660	4
528	1	-2.380	-0.000	.650	4
529	0	-1.923	-1.905	.920	5
530	0	-2.391	-1.340	.754	5
531	0	-2.599	-.755	.632	5
532	1	-2.489	-0.000	.595	5
533	0	-2.010	-2.015	.920	5
534	0	-2.514	-1.420	.721	5
535	0	-2.738	-.766	.575	5
536	1	-2.708	-0.000	.484	5
537	0	-2.054	-2.070	.920	5
538	0	-2.575	-1.460	.704	5
539	0	-2.807	-.771	.547	5
540	1	-2.817	-0.000	.429	5
541	0	-2.402	-1.869	1.197	5
542	0	2.4769	-1.2446	1.08	5
543	0	2.639	-.737	1.104	4
544	1	2.818	-0.000	.938	4
545	0	1.805	-1.825	1.111	4
546	0	1.870	-1.391	1.006	4
547	0	1.645	-.800	.950	4
548	1	1.834	-0.000	.760	4
549	0	1.007	-2.175	.782	4
550	0	1.326	-1.483	.695	3
551	0	.999	-.771	.636	3
552	1	.963	-0.000	.896	3
553	0	.004	-2.356	.827	3
554	0	.528	-1.440	.744	3
555	0	-.194	-.492	.134	3
556	1	-.194	-0.000	-.015	3
557	0	-.853	-2.008	.889	2
558	0	-.519	-1.323	.740	2
559	0	-.809	-.683	-.152	2
560	1	-.794	-0.000	.069	2
561	0	-1.474	-1.937	.573	2
562	0	-1.102	-1.646	.122	2
563	0	-1.468	-.698	-.130	2
564	1	-1.477	-0.000	-.128	1
565	0	-1.843	-1.755	.542	1
566	0	-2.060	-1.261	.105	1
567	0	-2.117	-.700	-.063	1
568	1	-1.936	-0.000	.044	1
569	0	2.426	-1.880	1.160	1
570	0	2.490	-1.250	1.050	1
571	0	2.650	-.750	1.070	5
572	1	2.850	-0.000	.870	4
573	0	1.820	-1.850	1.080	3
574	0	1.920	-1.400	.920	2
575	0	1.670	-.810	.880	1
576	1	1.840	-0.000	.700	0
577	0	1.020	-2.200	.750	0
578	0	1.350	-1.500	.600	0
579	0	1.060	-.750	.500	0
580	1	1.040	-0.000	.690	0



581	0	0.000	-2.380	.780	1
582	0	.520	-1.500	.580	2
583	0	-.180	-.500	.100	3
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585	0	-.900	-2.000	.800	5
586	0	-.480	-1.400	.650	5
587	0	-.780	-.720	-.240	4
588	1	-.780	-0.000	-.030	3
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592	1	-1.480	-0.000	-.240	0
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594	0	-2.080	-1.280	.080	0
595	0	-2.140	-.700	-.090	0
596	1	-1.980	-0.000	0.000	1
597	0	2.438	-1.891	1.123	2
598	0	2.503	-1.255	1.02	3
599	0	2.661	-.763	1.036	4
600	1	2.882	-0.000	.802	5
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602	0	1.970	-1.409	.834	4
603	0	1.695	-.820	.810	3
604	1	1.846	-0.000	.640	2
605	0	1.033	-2.225	.718	1
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607	0	1.121	-.729	.364	0
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609	0	-.004	-2.404	.733	0
610	0	.512	-1.560	.416	0
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613	0	-.898	-2.046	.711	3
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622	0	-2.100	-1.299	.055	0
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624	1	-2.024	-0.000	-.044	0
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626	0	2.529	-1.266	0.95	1
627	0	2.682	-.788	.969	2
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 313 324 325 317 393 404 405 397  
 287 1  
 321 329 330 322 401 409 410 402  
 288 2  
 322 330 331 323 402 410 411 403  
 289 3  
 323 331 332 324 403 411 412 404  
 290 4  
 324 332 333 325 404 412 413 405  
 291 1  
 329 337 338 330 409 417 418 410  
 292 2



330 338 339 331 410 418 419 411  
293 3  
331 339 340 332 411 419 420 412  
294 4  
332 340 341 333 412 420 421 413  
295 1  
337 345 346 338 417 425 426 418  
296 2  
338 346 347 339 418 420 427 419  
297 3  
339 347 348 340 419 427 428 420  
298 4  
340 348 349 341 420 428 429 421  
299 1  
345 353 354 346 425 433 434 426  
300 2  
346 354 355 347 426 434 435 427  
301 3  
347 355 356 348 427 435 436 428  
302 4  
348 356 357 349 428 436 437 429  
303 1  
353 377 373 354 433 451 453 434  
304 2  
354 373 369 355 434 453 449 435  
305 3  
355 369 365 356 435 449 445 436  
306 4  
356 365 361 357 436 445 441 437  
307 5  
317 325 420 318 397 405 406 398  
308 5  
318 326 327 319 398 406 407 399  
309 5  
319 327 328 320 399 407 408 400  
310 5  
325 333 334 326 405 413 414 406  
311 5  
326 334 335 327 406 414 415 407  
312 5  
327 335 336 328 407 415 416 408  
313 5  
333 341 342 334 413 421 422 414  
314 5  
334 342 343 335 414 422 423 415  
315 5  
335 343 344 336 415 423 424 416  
316 5  
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317 5  
342 350 351 343 422 430 431 423  
318 5  
343 351 352 344 423 431 432 424  
319 5  
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320 5  
350 358 359 351 430 438 439 431  
321 5  
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322 5  
357 361 362 358 437 441 442 438  
323 5  
358 362 363 359 438 442 443 439  
324 5  
359 363 364 360 439 443 444 440  
325 4  
361 365 366 362 441 445 446 442  
326 4  
362 366 367 363 442 446 447 443  
327 4  
363 367 368 364 443 447 448 444  
328 3

365 369 37 366 445 449 450 446  
329 3  
366 370 371 367 446 450 451 447  
330 3  
367 371 372 368 447 451 452 448  
331 2  
369 373 374 371 449 453 454 450  
332 2  
370 374 375 371 450 454 455 451  
333 2  
371 375 376 372 451 455 456 452  
334 1  
373 377 378 374 453 457 458 454  
335 1  
374 378 379 375 454 456 459 455  
336 1  
375 379 380 376 455 459 460 456  
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338 1  
382 386 387 383 462 466 467 463  
339 1  
383 387 388 384 463 467 468 464  
340 2  
385 389 39 386 465 469 470 466  
341 2  
386 390 391 387 466 470 471 467  
342 2  
387 391 392 388 467 471 472 468  
343 3  
389 393 394 390 469 473 474 470  
344 3  
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345 3  
391 395 396 392 471 475 476 472  
346 4  
393 397 398 394 473 477 478 474  
347 4  
394 398 399 395 474 478 479 475  
348 4  
395 399 400 396 475 479 480 476  
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350 2  
385 402 403 389 465 482 483 469  
351 3  
389 403 404 393 469 483 484 473  
352 4  
393 404 405 397 473 484 485 477  
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354 2  
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356 4  
404 412 413 405 484 492 493 485  
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358 2  
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359 3  
411 419 420 412 491 499 500 492  
360 4  
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417 425 426 418 497 505 506 498  
362 2  
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363 3  
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364 4

420 428 429 421 500 508 509 501  
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 366 2  
 426 434 435 427 506 514 515 507  
 367 3  
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 370 2  
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 372 4  
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 396 3  
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 398 2  
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 399 2  
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 400 1

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 411 3  
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 474 478 479 475 570 542 543 571  
 414 4  
 475 479 480 476 571 543 544 572  
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 418 4  
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 425 3  
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 426 4  
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 428 2  
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 430 4  
 500 508 509 501 581 585 557 553  
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 432 2  
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 433 3  
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471 4  
543 547 548 544 571 575 576 572  
472 4



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475 4  
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 611 615 616 612 639 643 644 640  
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 518 2  
 614 618 619 615 642 646 647 643  
 519 5  
 615 619 620 616 643 647 648 644  
 520 2  
 617 621 622 618 645 649 650 646  
 521 2  
 618 622 623 619 646 650 651 647  
 522 2  
 619 623 624 620 647 651 652 648  
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 625 629 630 626 653 657 658 654  
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 626 630 631 627 654 658 659 655  
 525 1  
 627 631 632 628 655 659 660 656  
 526 1  
 629 633 634 630 657 661 662 658  
 527 1  
 630 634 635 631 658 662 663 659  
 528 1  
 631 635 636 632 659 663 664 660  
 529 1  
 633 637 638 634 661 665 666 662  
 530 1  
 634 638 639 635 662 666 667 663  
 531 1  
 635 639 640 636 663 667 668 664  
 532 1  
 637 641 642 638 665 669 670 666  
 533 1  
 638 642 643 639 666 670 671 667  
 534 6  
 639 643 644 640 667 671 672 668  
 535 1  
 641 645 646 642 669 673 674 670  
 536 1  
 642 646 647 643 670 674 675 671  
 537 6  
 643 647 648 644 671 675 676 672  
 538 1  
 645 649 650 646 673 677 678 674  
 539 1  
 646 650 651 647 674 678 679 675  
 540 1  
 647 651 652 648 675 679 680 676  
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 542 1  
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 543 1  
 655 659 660 656 683 687 688 684  
 544 1

657 661 662 668 685 689 690 686  
 545 1  
 661 665 666 682 689 691 692 690  
 546 1  
 681 685 686 682 693 697 698 694  
 547 1  
 682 686 687 683 694 698 699 695  
 548 1  
 683 687 688 684 695 699 700 696  
 549 1  
 685 689 69 686 697 701 702 698  
 550 1  
 689 691 692 690 701 703 704 702  
 551 1  
 693 697 698 694 705 709 710 706  
 552 1  
 694 698 699 695 706 710 711 707  
 553 1  
 695 699 700 696 707 711 712 708  
 554 1  
 697 701 702 698 709 713 714 710  
 555 1  
 701 703 704 702 713 715 716 714

4 FORCE  
 4 -.9659 -.2588  
 12 PLOT3D  
 1 INPLIC11  
 0.001 2 11 0.01 716 1 555 314. .25  
 2  
 STOP

Appendix H

OTHER ASSOCIATED CODE LISTINGS

```

*DECK SKMSH
PROGRAM SKMSH(INPUT,OUTPUT,TAPES=INPUT,TAPE7=OUTPUT,TAPE6=OUTPUT,
1 PUNCH)
COMMON /JAC/ NJAC, TSA,NSA
COMMON /X/ XA(3,20)
DIMENSION DEN(12)
DIMENSION NHIG(15,15,15),NC(8),IPT(1215)
DIMENSION XI(1215),YI(1215),ZI(1215),CT(1215)
COMMON ICONN(1120,10),X(1290),Y(1290),Z(1290),H(1290),HC(1290)
DIMENSION ICORRC(8)
INTEGER HC
INTEGER P
REAL MASS
DATA ICORRC/1,2,4,3,5,6,8,7/
A=0.51735J26*189626
WRITE(6,2008)
2008 FORMAT(1H,45A40HWELCOME TO SKULL MESH ///)
HEAD(5,200) NSIZE
READ(5,207) NOIM, NEL,NSA,NJAC,IHALF,JACON,MASSD
* *IFACE
207 FORMAT(15)
WRITE(6,2012) NOIM,NEL,NSA,NJAC,IHALF,JACON,MASSD
* *IFACE
2012 FORMAT(40H NUMBER OF DIMENSIONS.....,15,/,
X 40H NUMBER OF NODES FOR EACH ELEMENT.....,15,/,
X 40H SUBMACHNOID SPACE SPECIFIED.....,15,/,
X 40H CONSTANT THICKNESS TO BE COMPUTED.....,15,/,
X 40H HALF SKULL SPECIFIED.....,15,/,
X 40H JACOBIAN CHECK ONLY.....,15,/,
X 40H MASS DISTRIBUTION PROPERTIES.....,15,/,
X 40H FACIAL BONES TO BE SPECIFIED.....,15,/)
HEAD(5,4999) NUMMAT
DO 5000 IM = 1,NUMMAT
HEAD(5,5001) DEN(IM)
WRITE(6,5002) IM,DEN(IM)
5000 CONTINUE
4999 FORMAT(15)
5001 FORMAT(1,29X,E11.4)
5002 FORMAT(19H MATERIAL,15,9H DENSITY=,E11.4,20H LB-SEC**2 PER IN**4)
C
C CHECK JACOBIANS FOR COMPLETELY PRESCRIBED MESH
C
DO 7002 I=1,1215
7002 IPT(I)=0
IF (JACON.EQ.0) GO TO 2
READ(5,3000) NPT
3000 FORMAT(15)
DO 3 N1=1,NPT
READ(5,91) N,BC(N),X(N),Y(N),Z(N), IPT(N)
3 CONTINUE
READ(5,3000) NELEM
DO 4 N1=1,NELEM
READ(5,3001) N,IM,(ICONN(N,M),M=1,8)
3001 FORMAT(215,/,8I4)
ICONN(N,10) = IM
4 CONTINUE
GO TO 670
2 READ(5,2010) SCALX, SCALY, SCALZ, SCALT
2010 FORMAT(4F10.3)
WRITE(6,2009) NSIZE
2009 FORMAT(230H ORDER OF BRAIN DISCRETIZATION,15,2X,5HCURED//)
WRITE(6,2011) SCALX,SCALY,SCALZ,SCALT
2011 FORMAT(40H X COORDINATE SCALE FACTOR.....,F5.3,/,
X 40H Y COORDINATE SCALE FACTOR.....,F5.3,/,
X 40H Z COORDINATE SCALE FACTOR.....,F5.3,/,
X 40H THICKNESS SCALE FACTOR.....,F5.3,/)
NTOT=(NSIZE+1)**3+4*((NSIZE+1)**3-(NSIZE-1)**3)
NPT=NTOT
N1=1
N4=4
N5=5
N6=6

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```

      NR=NSIZE*4
      N9=NSIZE*5
      N10=NSIZE*6
      N11=NSIZE*7
      N12=NSIZE*8
      N13=NSIZE*9
C
C   BUILD A BASIC ARRAY OF NODES WITHIN NBIG
C
      NODE=
      DO 11 I=1,5
      DO 11 J=5,N9
      DO 11 K=5,N9
      NODE = NODE*1
11  NBIG(I,J,K)= NODE
      DO 12 I=6,N8
      DO 12 J=1,5
      DO 12 K=5,N9
      NODE = NODE*1
12  NBIG(I,J,K) = NODE
      DO 13 J=N8,N9
      DO 13 K=1, N13
      NODE = NODE*1
13  NBIG(I,J,K) = NODE
      DO 14 J=N9,N13
      DO 14 K=5,N9
      NODE = NODE*1
14  NBIG(I,J,K) =NODE
      DO 15 I=N9,N13
      DO 15 J=5,N9
      DO 15 K=5,N9
      NODE =NODE*1
15  NBIG(I,J,K)=NODE
      N101=NODE
C
C   FOLD IN THE EDGES
C
      DO 22 I=1,4
      IM=N13+1-I
C   IM RUNS FROM 13 TO 10 AS I GOES FROM 1 TO 4
      IP= N9+1
C   IP RUNS FROM 10 TO 13 AS I GOES FROM 1 TO 4
      DO 21 J=5,N9
      FRONT TOP
      NBIG(5,J,1)= NBIG(I,J,5)
C
      FRONT BOTTOM
      NBIG(5,J,10)= NBIG(I,J,N9)
C
      BACK TOP
      NBIG(N9,J,1)= NBIG(IM,J,5)
C
      BACK BOT
21  NBIG(N9,J,10)= NBIG(IP,J,N9)
      DO 22 K=5,N9
C
C   LEFT FRONT
      NBIG(5,I,K)= NBIG(I,5,K)
C
C   RIGHT FRONT
      NBIG(5,IM,K) = NBIG(I,N9,K)
C
C   LEFT REAR
      NBIG(N9,I,K)= NBIG(IM,5,K)
C
C   RIGHT REAR
      NBIG(N9,IP,K) = NBIG(IP,N9,K)
C
22  CONTINUE
      DO 25 I=6,N8
      DO 25 J=1,4
      JM=N13+1-J

```

```

      JP=N9+J
C
C LEFT TOP
      NBIG(1,5,J) = NBIG(1,J,5)
C
C LEFT BOTTOM
      NBIG(1,5,JP) = NBIG(1,J,N9)
C
C RIGHT TOP
      NBIG(1,N9+J) = NBIG(1,JP,5)
C
C RIGHT BOTTOM
      NBIG(1,N9,JP) = NBIG(1,JP,N9)
      25 CONTINUE
C

      HSUM = 0
      KOUNT = 0
      70 READ(5,1000) XX,YY,ZZ,IP,NODE,HH
      1000 FORMAT(SX,3E10.0,IS,19,E10.0)
      WRITE(6,1000) XX,YY,ZZ,IP,NODE,HH
      IF(NODE.EQ.0) GO TO 71
      KOUNT = KOUNT + 1
      X(NODE) = XX * SCALX
      Y(NODE) = YY * SCALY
      Z(NODE) = ZZ * SCALZ
      H(NODE) = HH * SCALH
      IF (NJAC .NE. 0) HSUM = HSUM + H(NODE)
      IF(NODE) = IP
      GO TO 70
      71 CONTINUE
      ISA = (HSUM/KOUNT) * .25
      IF (NJAC .NE. 0) WRITE (6,1001) ISA
      1001 FORMAT (41H CONSTANT SUBARACHNOID SPACE THICKNESS IS, F5.3, 2X, 6H1
      XINCHES//)

C
C
C BUILD AND WRITE THE ICONN ARRAY
C
      NLEME = 1
      DO 33 I = 1,4
      DO 33 J = 5,NR
      DO 33 K = 5,NR
      IMAT = 1
      NELEM = NELEM + 1
      CALL ISTORE(NELEM,I,J,K,NBIG,ICONN,IMAT)
      33 CONTINUE
      DO 37 I = 5,NR
      DO 34 J = 1,4
      DO 34 K = 5,NR
      IMAT = J
      NELEM = NELEM + 1
      CALL ISTORE(NELEM,I,J,K,NBIG,ICONN,IMAT)
      34 CONTINUE
      DO 35 J = 5,NR
      DO 35 K = 1,4
      IMAT = K
      NELEM = NELEM + 1
      CALL ISTORE(NELEM,I,J,K,NBIG,ICONN,IMAT)
      35 CONTINUE
      DO 135 J = 5,NR
      DO 135 K = 5,NR
      IMAT = 5
      NELEM = NELEM + 1
      CALL ISTORE(NELEM,I,J,K,NBIG,ICONN,IMAT)
      135 CONTINUE
      DO 235 J = 5,NR

```

```

      DO 235 K=N9,N12
      IMAT=N13-K
      NELEM=NELEM+1
      CALL ISTORE (NELEM,I,J,K,NHIG,ICONN,IMAT)
235  CONTINUE
      DO 36 J=N9,N12
      DO 36 K=S+NB
      IMAT=N13-J
      NELEM=NELEM+1
      CALL ISTORE (NELEM,I,J,K,NHIG,ICONN,IMAT)
36  CONTINUE
      DO 37 J=N9,N12
      DO 38 K=S+NB
      DO 38 K=S+NB
      IMAT=N13-I
      NELEM=NELEM+1
      CALL ISTORE (NELEM,I,J,K,NHIG,ICONN,IMAT)
38  CONTINUE
200  FORMAT (915)

```

```

C
C
C FIND SURFACE NODE AND NEIGHBORS. CODE=1 FOR CORNERS, 2 FOR EDGES, 3 FACES.
C
      CALL POSTS(5,5,4, 5,6,4, 5,4,6, 6,5,4, 1,1,1, 1
1  5, 5, 5, 5, 5, 3, 5, 5, 2, 5, 5, 1, 1,NBIG,IPT) 1
      CALL POSTS(4,5,N9, 4,6,N9, 6,4,N9, 4,5,N8, 1,1,1, 2
1  5, 5, N9, 5, 5,N11, 5, 5,N12, 5, 5,N13, 1,NBIG,IPT) 2
      CALL POSTS(5,N9,4, 6,N9,4, 4,N9,6, 4,N8,5, 1,1,1, 3
1  5, N9, 5, 5, N9, 3, 5, N9, 2, 5, N9, 1, 1,NBIG,IPT) 3
      CALL POSTS(4,N9,N9, 6,N9,N10, 4,N8,N9, 4,N9,N8, 1,1,1, 4
1  5, N9, N9, 5, N9,N11, 5, N9,N12, 5, N9,N13, 1,NBIG,IPT) 4
      CALL POSTS(N9,4,5, N8,4,5, N9,4,6, N9,6,4, 1,1,1, 5
1  N9, 5, 5, N9, 5, 3, N9, 5, 2, N9, 5, 1, 1,NBIG,IPT) 5
      CALL POSTS(N9,4,N9, N8,4,N9, N10,6,N9, N9,4,N8, 1,1,1, 6
1  N9, 5, N9, N9, 5,N11, N9, 5,N12, N9, 5,N13, 1,NBIG,IPT) 6
      CALL POSTS(N9,N9,4, N9,N8,4, N10,N9,6, N8,N9,4, 1,1,1, 7
1  N9, N9, 5, N9, N9, 3, N9, N9, 2, N9, N9, 1, 1,NBIG,IPT) 7
      CALL POSTS(N9,N9,N10, N9,N8,N10, N8,N9,N10, N10,N9,N8, 1,1,1, 8
1  N9, N9, N9, N9, N9,N11, N9, N9,N12, N9, N9,N13, 1,NBIG,IPT) 8
      DO 40 L=6,N8
      M=L-1
      P=L+1
      CALL POSTS(4,5,L, 4,6,L, 4,5,P, 6,4,L, 4,5,M, 9
1  5, 5, L, 3, 5, L, 2, 5, L, 1, 5, L, 2,NBIG,IPT) 9
40  CONTINUE
      DO 41 L=6,N8
      M=L-1
      P=L+1
      CALL POSTS(4,N9,L, 6,N10,L, 4,N9,P, 4,N8,L, 4,N9,M, 10
1  5, N9, L, 3, N9, L, 2, N9, L, 1, N9, L, 2,NBIG,IPT) 10
41  CONTINUE
      DO 42 L=6,N8
      M=L-1
      P=L+1
      CALL POSTS(N9,4,L, N8,4,L, N9,4,P, N10,6,L, N9,4,M, 11
1  N9, 5, L, N11, 5, L,N12, 5, L, N13, 5, L, 2,NBIG,IPT) 11
42  CONTINUE
      DO 43 L=6,N8
      P=L+1
      M=L-1
      CALL POSTS(N10,N9,L, N10,N8,L, N10,N9,P, N8,N10,L, N10,N9,M, 12
1  N9, N9, L, N11, N9, L,N12, N9, L, N13, N9, L, 2,NBIG,IPT) 12
43  CONTINUE
      DO 44 L=6,N8
      M=L-1
      P=L+1
      CALL POSTS(L,4,5, M,4,5, L,4,6, P,4,5, L,6,4, 13
1  L, 5, 5, L, 5, 3, L, 5, 2, L, 5, 1, 2,NBIG,IPT) 13
44  CONTINUE

```

```

DO 45 L=A,NG
M=L-1
P=L+1
CALL POSTS(L,N10,5, P,N10,5, L,N10,6, M,N10,5, L,N8,4,
1 L, N9, 5, L, N9, 3, L, N9, 2, L, N9, 1, 2,NBIG,IPT) 14
45 CONTINUE
DO 46 L=A,NG
M=L-1
P=L+1
CALL POSTS(L,L,5, 4,P,5, 4,L,6, 4,M,5, 6,L,4,
1 5, L, 5, 5, L, 3, 5, L, 2, 5, L, 1, 2,NBIG,IPT) 15
46 CONTINUE
DO 47 L=A,NG
M=L-1
P=L+1
CALL POSTS(N10,L,5, N10,M,5, N10,L,6, N10,P,5, N8,L,4,
1 N9, L, 5, N9, L, 3, N9, L, 2, N9, L, 1, 2,NBIG,IPT) 16
47 CONTINUE
DO 48 L=A,NG
M=L-1
P=L+1
CALL POSTS(L,4,N9, M,4,N9, L,6,N10, P,4,N9, L,4,N8,
1 L, 5, N9, L, 5,N11, L, 5,N12, L, 5,N13, 2,NBIG,IPT) 17
48 CONTINUE
DO 49 L=A,NG
M=L-1
P=L+1
CALL POSTS(L,N9,N10, P,N9,N10, L,N8,N10, M,N9,N10, L,N10,N8,
1 L, N9, N9, L, N,N11, L, N,N12, L, N,N13, 2,NBIG,IPT) 18
49 CONTINUE
DO 50 L=A,NG
M=L-1
P=L+1
CALL POSTS(L,L,N9, 4,P,N9, 6,L,N10, 4,M,N9, 4,L,N8,
1 5, L, N9, 5, L,N11, 5, L,N12, 5, L,N13, 2,NBIG,IPT) 19
50 CONTINUE
DO 51 L=A,NG
M=L-1
P=L+1
CALL POSTS(N9,L,N10, P,M,N10, N8,L,N10, N9,P,N10, N10,L,N8,
1 N9, L, N9, N9, L,N11, N9, L,N12, N9, L,N13, 2,NBIG,IPT) 20
51 CONTINUE
DO 52 J=A,NG
DO 52 K=A,NG
JP=J+1
JM=J-1
KPK=K+1
KMK=K-1
CALL POSTS(4,J,K, 4,JP,K, 4,J,KP, 4,JM,K, 4,J,KM,
1 5, J, K, 3, J, K, 2, J, K, 1, J, K, 3,NBIG,IPT) 21
52 CONTINUE
DO 53 J=A,NG
DO 53 K=A,NG
JP=J+1
JM=J-1
KPK=K+1
KMK=K-1
CALL POSTS(N10,J,K, N10,JM,K, N10,J,KP, N10,JP,K, N10,J,KM,
1 N9, J, K, N11, J, K,N12, J, K, N13, J, K, 3,NBIG,IPT) 22
53 CONTINUE
DO 54 I=A,NG
DO 54 K=A,NG
IP=I+1
IM=I-1
KP=K+1
KM=K-1
CALL POSTS(I,4,K, IM,4,K, I,4,KP, IP,4,K, I,4,KM,
1 I, 5, K, 1, 3, K, 1, 2, K, 1, 1, K, 3,NBIG,IPT) 23
54 CONTINUE
DO 55 I=A,NG
IP=I+1

```

```

      IM= I-1
      KP= K-1
      KM= K-1
      CALL POSTS(I,N10,K, IP,N10,K, I,N10,KP, IM,N10,K, I,N10,KM,
1      1, 09, K, I,N11, K, I,N12, K, I,N13, K, 3,NBIG,IPT) 24
55 CONTINUE 24
      DO 56 I=K,NK
      DO 56 J=K,NK
      IP= I+1
      IM= I-1
      JP= J+1
      JM= J-1
      CALL POSTS(I,J,4, I,JP,4, IM,J,4, I,JM,4, IP,J,4,
1      1, J, 5, 1, J, 3, 1, J, 2, 1, J, 1, 3,NBIG,IPT) 25
56 CONTINUE 25
      DO 57 I=K,NK
      DO 57 J=K,NK
      IP= I+1
      IM= I-1
      JP= J+1
      JM= J-1
      CALL POSTS(I,J,N10, I,JP,N10, IP,J,N10, I,JM,N10, IM,J,N10,
1      1, J, N9, 1, J,N11, 1, J,N12, 1, J,N13, 3,NBIG,IPT) 26
57 CONTINUE 26

```

```

C
C   GENERATE COORDINATES FOR UNSPECIFIED INTERIOR NODES
C

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```

600 NMIS=
      IF (NELEM .EQ. 0) GO TO 663
      DO 635 K=3,NPT
      IF (IPT(K) .NE. 0) GO TO 635
      X(K)=(X(1)+X(NPT))*0.5
      Y(K)=(Y(1)+Y(NPT))*0.5
      Z(K)=(Z(1)+Z(NPT))*0.5
      NMIS=NMIS+1
635 CONTINUE
      IF (NMIS .EQ. 0) GO TO 663
      DO 655 NN=1,NMIS
      I0=0
      DO 637 K=1,NPT
      XT(K)=0.0
      YT(K)=0.0
      ZT(K)=0.0
637 CT(K)=0.0
      DO 641 IE=1,NELEM
      DO 641 JJ=1,8
      N=N+JJ
      M=NOD(M)
      IF (IPT(J) .NE. 0) GO TO 641
      I=JJ-1
      L=I/4
      IF (4*L .EQ. I) I=4+I
      M=N+I
      I=NOD(M)
      K=JJ+1
      L=JJ/4
      IF (4*L .EQ. JJ) K=K-4
      M=N+K
      K=NOD(K)
      L=JJ+4
      IF (L .GT. 8) L=L-8
      M=N+L
      L=NOD(L)
      XT(J)=XT(J)+X(I)+X(K)+X(L)
      YT(J)=YT(J)+Y(I)+Y(K)+Y(L)
      ZT(J)=ZT(J)+Z(I)+Z(K)+Z(L)
640 CT(J)=CT(J)+3.0
641 CONTINUE

```



```

DO 650 K=1,NPT
D4=CT(K)
IF (PI(K) .GT. 0 .OR. D4 .EQ. 0.0) GO TO 650
D1=X(K)
D2=Y(K)
D3=Z(K)
X(K)=X1(K)/D4
Y(K)=Y1(K)/D4
Z(K)=Z1(K)/D4
D1=ABS (X(K)-D1)
D2=ABS (Y(K)-D2)
D3=ABS (Z(K)-D3)
IF (D1+D2+D3 .GT. .0001) IOT=1
650 CONTINUE
ICOUNT=ICOUNT+1
IF (IOT .EQ. 0) GO TO 663
655 CONTINUE
C
663 CONTINUE
C
C READ IN FACIAL BONES IF SPECIFIED
C
IF (IFACE.NE.1) GO TO 680
C
C READ IN THE FACIAL NODAL POINTS
C
READ(5,3000) NPTF
WRITE(6,3000) NPTF
DO 681 I=1,NPTF
READ(5,91) N,HC(N),X(N),Y(N),Z(N),IPT(N)
WRITE(6,91) N,HC(N),X(N),Y(N),Z(N),IPT(N)
681 CONTINUE
C
C READ THE FACIAL CONNECTIVITY
C
READ(5,3000) NELEMF
WRITE(6,3000) NELEMF
DO 682 I=1,NELEMF
READ(5,3001) N,ICUNN(N,10),(ICUNN(N,J),J=1,8)
WRITE(6,3001) N,ICUNN(N,10),(ICUNN(N,J),J=1,8)
682 CONTINUE
C
C UPDATE THE NUMBER OF NODES AND ELEMENTS
C
NPT=NPT+NPTF
NELEMF=NELEMF+NELEMF
680 CONTINUE
C
C IF SO SPECIFIED, HALVE THE MESH
C
IF (HALF .EQ. 1) CALL HALF(NPT*NELEMF, NEL)
C
C REFERENCE CO-ORDINATES TO FRANKFORT PLANE
C
DO 669 N=1,NPT
X(N)=5.52-X(N)
Y(N) = -Y(N)
Z(N)=Z(N)-5.75
669 CONTINUE
C
C CHECK MESH FOR NEGATIVE JACOBIANS
C
670 DO 209 N = 1,NELEMF
DO 905 I=1,NEL
K = ICUNN(N,I)
IF (K.GT.NPT .OR. K.LT.0) GO TO 905
IF (K.EQ.0) GO TO 905
XA(1,I) = X(K)
XA(2,I) = Y(K)
XA(3,I) = Z(K)
905 CONTINUE
K1=ICUNN(N,1)
K2=ICUNN(N,2)

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K3=ICUNN(N,3)
K4=ICUNN(N,4)
K5=ICUNN(N,5)
K6=ICUNN(N,6)
K7=ICUNN(N,7)
KH=ICUNN(N,8)
NSIDE = (NEL+1)/NOIM**2
L=1
IF (NDIM*EQ.2) L=2
LJK = 0
EVOL = 0.
SS = 1.
DO 30 I = 1,2
SS = -SS
II = 1.
DO 30 JJ = 1,2
II = -II
UU = 1.
DO 30 K = 1,2
UU = -UU
CALL CHKJAC (UU,II,SS,NOIM,NEL,NSIDE,XJAC)
EVOL=EVOL*XJAC
C
C COMPUTE MASS PROPERTIES OF SKULL
C
LJK = LJK + 1
IF (XJAC.GT.0.0) GO TO 30
WRITE(6,208) N,XJAC,(ICUNN(N,LL),LL=1,NEL)
208 FORMAT (/5X ELEMENT,15,20X, 8HJACOBIAN,E11.4/6H ICUNN,8I4)
JAC = -1
WRITE(6,802) X(K1), X(K2), X(K3), X(K4), X(K5), X(K6), X(K7), X(K8)
WRITE(6,803) Y(K1), Y(K2), Y(K3), Y(K4), Y(K5), Y(K6), Y(K7), Y(K8)
WRITE(6,804) Z(K1), Z(K2), Z(K3), Z(K4), Z(K5), Z(K6), Z(K7), Z(K8)
802 FORMAT (6H X ,E14.4)
803 FORMAT (6H Y ,E14.4)
804 FORMAT (6H Z ,E14.4)
LKJM=ICORRC(LJK)
WRITE(6,209) ICUNN(N,LKJM)
209 FORMAT (/5H NODE,15,12H IS INVERTED)
30 CONTINUE
VOL = VOL + EVOL
IM = ICUNN(N,10)
FMASS = EVOL*DEN(IM)
MASS = MASS+FMASS
IF (MASSD.NE.0) CALL MASSDIS(K1,K2,K3,K4,K5,K6,K7,KH,FMASS,N)
XNELEM,MASS,IMHALF)
209 CONTINUE
WRITE (6,99) VOL,MASS
99 FORMAT (/9H VOLUME =,E11.4,10H INCHES**3/7H MASS =,E11.4,
X 19H LB*SEC**2 PER INCH)
IF (JACON.EQ. 1) GO TO 94
IF (JAC.LT.0) GO TO 94
WRITE(6,942)
942 FORMAT (/55H MESH HAS BEEN CHECKED-NO NEGATIVE JACOBIANS WERE FOUND)
91 FORMAT (15,10X, 11, 4X, 3F10.3,15)
DO 700 N=1,NPT
PUNCH 91, N,RC(N),X(N),Y(N),Z(N),IPT(N)
700 WRITE (6,91) N,RC(N),X(N),Y(N),Z(N),IPT(N)
PUNCH 93, (N,ICUNN(N,10),(ICUNN(N,M),M=1,8),N=1,NELEM)
WRITE (6,93) (N,ICUNN(N,10),(ICUNN(N,M),M=1,8),N=1,NELEM)
93 FORMAT (215,8I4)
GO TO 94
940 WRITE (6,941) K
941 FORMAT (5H NODE,15,2X,11H EXCEEDS NPT)
94 STOP
END
*DECK POSTS
SUBROUTINE POSTS(LA,LR,LC,LD,LE,LF,LG,LH,LI,LJ,LK,LL,LM,LN,LO,LP,
1 LU,LR,LS,LT,LU,LV,LW,LX,LY,LZ,L1,ICODE,NBIG,IPT)
COMMON /JAC/ NJAC, 15A,NSA
COMMON ICUNN(1120*10),X(1290),Y(1290),Z(1290),H(1290),RC(1290)

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```

      INTEGER HC
      DIMENSION IPT(1215),NRIG(15,15,15),NS(4),NA(4)
      NODE=NRIG(LA,LH,LC)
      IPT(NODE)=2
      NS(1)=NRIG(LD,LE,LF)
      K=NS(1)
      NS(2)=NRIG(LG,LH,LI)
      I=NS(2)
      NS(3)=NRIG(LJ,LK,LL)
      J=NS(3)
      3 NA(1)=NRIG(LP,LQ,LM)
      NA(2)=NRIG(LS,LT,LU)
      NA(3)=NRIG(LV,LW,LX)
      NA(4)=NRIG(LY,LZ,LI)
      N=NA(1)
      IPT(N)=1
      DO 40 I=2,4
      N=NA(I)
      IP=I+1
      40 IPT(N)=IP
      IF(ICODE-1) 4,4,2
      4 CONTINUE
      L=J
      NTHI=3
      GO TO 103
      2 NS(4)=NRIG(LM,LN,LO)
      L=NS(4)
      NTHI=4
      103 CONTINUE
      N=NODE
C
C      I,J,K,L ARE THE AUXILIARY NODES FOR NODE N
C
      104 C1 = 0
      C2 = 0
      C3 = 0
      DO 110 I1 = 1,NTHI
      XX = X(N)
      YY = Y(N)
      ZZ = Z(N)
      GO TO (105,106,107,108),I1
      105 A1 = X(J) - XX
      A2 = Y(J) - YY
      A3 = Z(J) - ZZ
      H1 = X(I) - XX
      H2 = Y(I) - YY
      H3 = Z(I) - ZZ
      GO TO 109
      106 A1 = X(I) - XX
      A2 = Y(I) - YY
      A3 = Z(I) - ZZ
      H1 = X(K) - XX
      H2 = Y(K) - YY
      H3 = Z(K) - ZZ
      GO TO 109
      107 A1 = X(K) - XX
      A2 = Y(K) - YY
      A3 = Z(K) - ZZ
      H1 = X(L) - XX
      H2 = Y(L) - YY
      H3 = Z(L) - ZZ
      GO TO 109
      108 A1 = X(L) - XX
      A2 = Y(L) - YY
      A3 = Z(L) - ZZ
      H1 = X(J) - XX
      H2 = Y(J) - YY
      H3 = Z(J) - ZZ
      109 C1 = (A2*H3 - A3*H2)/NTHI + C1
      C2 = (A3*H1 - A1*H3)/NTHI + C2
      C3 = (A1*H2 - A2*H1)/NTHI + C3
      110 CONTINUE

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```

C
C DIRECTION COSINES
C
      CL = (C1*C1 + C2*C2 + C3*C3)**.5
      COAL = (C1/CL)
      COHE = (C2/CL)
      COGA = (C3/CL)
      COSCH = (COAL*COAL+COHE*COHE+COGA*COGA)**.5
C
C IF SO SPECIFIED-INCLUDE THE SUBARACHNOID SPACE
C
      IF (NSA .EQ. 0) GO TO 116
C
C CORRECT FOR NEG JACOBIANS BY SPECIFYING AVERAGE THICKNESS TSA
C
      IF (NJAC .NE. 0) GO TO 917
      TSA=H(NODE)*.25
917 CONTINUE
      N=NA(1)
      X(N) = XX - TSA*COAL
      Y(N) = YY - TSA*COHE
      Z(N) = ZZ - TSA*COGA
116 N=NODE
C
C DIPLOE LAYER THICKNESS IS ASSUMED ONE HALF THE TOTAL SKULL THICKNESS
C
117 T = H(N)*.25
      N=NA(2)
      X(N)=XX + T*COAL
      Y(N)=YY + T*COHE
      Z(N)=ZZ + T*COGA
      N=NA(3)
      X(N)=XX + 3.*T*COAL
      Y(N)=YY + 3.*T*COHE
      Z(N)=ZZ + 3.*T*COGA
      N=NA(4)
      X(N)=XX + 4.*T*COAL
      Y(N)=YY + 4.*T*COHE
      Z(N)=ZZ + 4.*T*COGA
      RETURN
      END
      NO
*DECK FUNCTION NOD(M)
      COMMON ICUNN(1120,10),X(1290),Y(1290),Z(1290),H(1290),HC(1290)
      INTEGER HC
      NM1=(M-1)/8
      N=NM1+1
      JM1=M-NM1*8
      J=JM1
      NOD=ICUNN(N,J)
      RETURN
      END
*DECK ISTORE
      SUBROUTINE ISTORE(NELEM,I,J,K,NBIG,ICUNN,IMAT)
      DIMENSION NBIG(15,15,15),ICUNN(1120,10)
      ICUNN(NELEM,1)=NBIG(I,J,K)
      ICUNN(NELEM,2)=NBIG(I,J+1,K)
      ICUNN(NELEM,3)=NBIG(I,J+1,K+1)
      ICUNN(NELEM,4)=NBIG(I+1,J,K)
      ICUNN(NELEM,5)=NBIG(I+1,J,K)
      ICUNN(NELEM,6)=NBIG(I+1,J+1,K)
      ICUNN(NELEM,7)=NBIG(I+1,J+1,K+1)
      ICUNN(NELEM,8)=NBIG(I+1,J,K+1)
      ICUNN(NELEM,9)=0
      ICUNN(NELEM,10)=IMAT
      RETURN
      END
*DECK HALF
      SUBROUTINE HALF (NPT,NELEM,NEL)
      COMMON ICUNN(1120,10),X(1290),Y(1290),Z(1290),H(1290),HC(1290)
      INTEGER HC
      DIMENSION ICROS(1215)

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      DO 5 N = 1, NPT
      5 ICRUS(N) = 0
      WRITE (6,130) NPT,NELEM,NEL
130 FORMAT (315)
      N = 0
C
C SET TO ZERO ALL Y COORDINATES ON MIDSAAGITTAL PLANE AND
C DELETE NODES WITH NEGATIVE Y COORDINATES
C
      DO 10 I = 1,NPT
      YI = Y(I)
      YI = ABS(Y(I))
      IF (YI .LT. .001) GO TO 7
      IF (Y(I) .LT. 0.) GO TO 10
      GO TO 8
      7 YI = 0.
      N = N + 1
      ICRUS(I) = N
      X(N) = X(I)
      Y(N) = YI
      Z(N) = Z(I)
      IF (YI .EQ. 0.) BC(N) = 1
10 CONTINUE
      N = N + 1
      DO 20 I = N,NPT
      X(I) = 0.
      Y(I) = 0.
      Z(I) = 0.
20 NPT = N-1
C
C FORM CONNECTIVITY FOR HALF SKULL
C
      KOUNT = 0
      DO 30 J = 1,NELEM
      DO 25 K = 1,NEL
      JK = ICONN(J,K)
      IF (ICRUS(JK) .EQ. 0) GO TO 30
      ICONN(J,K) = ICRUS(JK)
25 CONTINUE
      KOUNT = KOUNT + 1
      DO 27 K = 1,NEL
27 ICONN(KOUNT,K) = ICONN(J,K)
30 CONTINUE
      NELEM = KOUNT
      WRITE (6,130) NPT,NELEM,NEL
      RETURN
      END
*DECK CHKJAC
      SUBROUTINE CHKJAC(W,S,T,NDIM,NEL,NSIDE,XJAC)
      COMMON /X/ XA(3,20)
      DIMENSION SHAPE(4,20),SG(3,3),SK(3,3)
      DIMENSION N1(12),N2(12),N3(12),NODE(12),S2ORD(12),S3ORD(12),SS(3)
      DIMENSION COLM(3)
      DATA NODE/2,4,6,8,9,10,11,12,14,16,18,20/
      DATA N1/1,2,1,2,3,3,3,3,1,2,1,2/,N2/2,3,2,3,1,1,1,1,2,3,2,3/
      DATA N3/3,1,3,1,2,2,2,2,3,1,3,1/
      DATA S2ORD/-.5,-.5, .5,-.5,-.5, .5, .5,-.5,-.5, .5, .5, .5/
      DATA S3ORD/-.5, .5,-.5,-.5,-.5,-.5, .5, .5, .5, .5, .5,-.5/
      SS(1) = W
      SS(2) = S
      SS(3) = T
      KI = 4
      LI = 1
      IF(NSIDE.EQ.1) GO TO 110
      KI = 12
      LI = 2
C**** FORM MIDSIDE SHAPE FUNCTIONS
      DO 100 L=1,12
      N = NODE(L)
      I = N1(L)
      J = N2(L)
      K = N3(L)

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```

      RRI 2C
      RRI 3C
      RRI 5C
      RRI 9C
      RRI 8C
      RRI 9C
      RRI 10C
      RRI 11C
      RRI 12C
      RRI 13C
      RRI 14C
      RRI 15C
      RRI 16C
      RRI 17C
      RRI 18C
      RRI 19C
      RRI 20C
      RRI 21C
      RRI 22C
      RRI 23C

```



```

      SJ = SZORD(L)
      SL = SZORD(L)
      RP = 1. - SS(1)**2
      SP = .5 + SJ*SS(J)
      TP = .5 + SL*SS(K)
      SHAPE(1,N) = -2.0*SS(1)*SP*TP
      SHAPE(J,N) = SJ*RP*TP
      SHAPE(K,N) = SL*RP*SP
      SHAPE(4,N) = RP*SP*TP
100  C**** FORM CORNER SHAPE FUNCTIONS
110  SJ = .0.5
      K = 8
      DO 250 J = 1,2
        L = 1
        DO 200 I = 5,8
          RP = .5 + SZORD(I)*R
          SP = .5 + SZORD(I)*S
          TP = .5 + SJ*T
          SHAPE(1,L + K) = SZORD(I)*SP*TP
          SHAPE(2,L + K) = RP*SZORD(I)*TP
          SHAPE(3,L + K) = RP*SP*SJ
          SHAPE(4,L + K) = RP*SP*TP
200  L = L + L1
      K = K1
250  SJ = .0.5
      IF(NSIDE.EQ.1) GO TO 360
C**** CORRECT BASIC CORNER FUNCTIONS BY PROPORTIONS OF MIDSIDE FUNCTIONS
      K = 8
      L = 9
      DO 350 I = 1,8,2
        DO 300 J = 1,4
          SHAPE(J,I) = SHAPE(J,I) - 0.5*(SHAPE(J,I+1) + SHAPE(J,K) + SHAPE(J,L))
300  SHAPE(J,I+12) = SHAPE(J,I+12) - 0.5*(SHAPE(J,I+13) + SHAPE(J,K+12)
      X + SHAPE(J,L))
      K = I + 1
350  L = L + 1
C**** FORM THE JACOBIAN DETERMINANT
360  DO 370 I = 1,3
      DO 370 J = 1,3
370  SK(I,J) = 0.0
      DO 400 I = 1,NDIM
        DO 400 J = 1,NDIM
          DO 400 K = 1,NEL
            SK(I,J) = SK(I,J) + SHAPE(J,K)*XA(I,K)
            IF(NDIM.EQ.2) SK(3,3) = 1.0
            SG(1,1) = SK(2,2)*SK(3,3) - SK(2,3)*SK(3,2)
            SG(2,2) = SK(1,1)*SK(3,3) - SK(1,3)*SK(3,1)
            SG(3,3) = SK(1,1)*SK(2,2) - SK(1,2)*SK(2,1)
            SG(1,2) = -SK(1,2)*SK(3,3) + SK(1,3)*SK(3,2)
            SG(1,3) = SK(1,2)*SK(2,3) - SK(1,3)*SK(2,2)
            SG(2,1) = -SK(2,1)*SK(3,3) + SK(2,3)*SK(3,1)
            SG(2,3) = -SK(1,1)*SK(2,3) + SK(2,1)*SK(1,3)
            SG(3,1) = SK(2,1)*SK(3,2) - SK(2,2)*SK(3,1)
            SG(3,2) = -SK(1,1)*SK(3,2) + SK(1,2)*SK(3,1)
            XJAC = 0.0
            DO 400 I = 1,NDIM
              XJAC = XJAC + SG(I,1)*SK(I,1)
600  XJAC = XJAC + SG(1,1)*SK(1,1)
      RETURN
      END
C**** DECK MASSDIS
      SUBROUTINE MASSDIS(K1,K2,K3,K4,K5,K6,K7,K8,EMASS,N,NEL,ELEM,MASS,
      X1HALF)
C
C MASS MOMENT OF INERTIA TENSOR IS COMPUTED VIA PARALLEL AXIS THEOREM
C
      COMMON ICONK(1120*10)*X(1290)*Y(1290)*Z(1290)*H(1290)*RC(1290)
      REAL MASS
C
C COMPUTE ELEMENT CG COORDINATES
C
      XCG = (X(K1)+X(K2)+X(K3)+X(K4)+X(K5)+X(K6)+X(K7)+X(K8))/8.0
      YCG = (Y(K1)+Y(K2)+Y(K3)+Y(K4)+Y(K5)+Y(K6)+Y(K7)+Y(K8))/8.0

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      ZCG = (Z(K1)+Z(K2)+Z(K3)+Z(K4)+Z(K5)+Z(K6)+Z(K7)+Z(K8))/8.0
      RHUX = YCG*YCG + ZCG*ZCG
      RHUY = XCG*XCG + ZCG*ZCG
      RHUZ = XCG*XCG + YCG*YCG
C
C   ACCUMULATE FIRST MOMENT OF ELEMENT MASS WRT REFERENCE AXES
C
      FMUY = XCG*EMASS+FMUY
      FMOX = ZCG*EMASS+FMOX
C
C   ACCUMULATE SECOND MOMENT OF ELEMENT MASS WRT REFERENCE AXES
C
      SMOXX = RHUX*EMASS+SMOXX
      SMOYY = RHUY*EMASS+SMOYY
      SMOZZ = RHUZ*EMASS+SMOZZ
      SMOXY = 0.
      SMOXZ = XCG*ZCG*EMASS+SMOXZ
      SMOYZ = 0.
C
      IF (N.LT.NELEM) RETURN
C
C   COMPUTE CG LOCATION WRT REFERENCE AXES
C
      WRITE (6,93)
93  FORMAT (1H1,29H MASS DISTRIBUTION PROPERTIES//)
      IF (IHALF.EQ.1) WRITE (6,94)
94  FORMAT (1/30H VALUES ARE FOR A HALVED SKULL/)
      XBAR = FMUY/MASS
      YBAR = 0.
      ZBAR = FMOX/MASS
      WRITE (6,95) XBAR,YBAR,ZBAR
      WRITE (6,96) SMOXX,SMOXY,SMOAZ,SMOXY,SMOYY,SMOYZ,SMOAZ,SMOYZ,SMOZZ
C
C   COMPUTE MASS MOMENT OF INERTIA WRT TO CG REFERENCE
C
      SMOXX = SMOXX - MASS*ZBAR*ZBAR
      SMOYY = SMOYY - MASS*(XBAR*XBAR + ZBAR*ZBAR)
      SMOZZ = SMOZZ - MASS*XBAR*XBAR
      SMOXY = 0.
      SMOXZ = SMOXZ - MASS*XBAR*ZBAR
      SMOYZ = 0.
      WRITE (6,96) SMOXX,SMOXY,SMOAZ,SMOXY,SMOYY,SMOYZ,SMOAZ,SMOYZ,SMOZZ
C
C   PRINCIPAL MASS MOMENTS OF INERTIA WRT TO CG REFERENCE
C
      AVE = (SMOXX+SMOZZ)*.5
      BASE = (SMOXX-SMOZZ)*.5
      RAD = (BASE*BASE + SMOXZ*SMOAZ)**.5
      SMOXX = AVE + RAD
      SMOZZ = AVE - RAD
      DUM = SMOAZ
      SMOXZ = 0.
      WRITE (6,95) SMOXX,SMOXY,SMOAZ,SMOXY,SMOYY,SMOYZ,SMOAZ,SMOYZ,SMOZZ
      THETA = .5*ATAN(DUM/BASE)
      THETA = 180.*THETA/3.1415926
      WRITE (6,97) THETA
97  FORMAT (1/55H PRINCIPAL INERTIA AXIS ORIENTATION WRT FRANKFORD PLAN
      XE/PA.1,4H DEGREES)
98  FORMAT (1/21H CG LOCATION (INCHES)/6H XBAR=F10.3/
      XAH YBAR=F10.3/6H ZBAR=F10.3)
99  FORMAT (1/34H INERTIA TENSOR WRT REFERENCE AXES/
      X3E11.3/3E11.3.5X.13H LH=SEC**2-IN/3E11.3)
96  FORMAT (1/52H INERTIA TENSOR WRT PARALLEL AXES PASSING THROUGH CG/
      X3E11.3/3E11.3.5X.13H LH=SEC**2-IN/3E11.3)
95  FORMAT (1/32H PRINCIPAL INERTIA TENSOR WRT CG/
      X3E11.3/3E11.3.5X.13H LH=SEC**2-IN/3E11.3)
      RETURN
      END

```

```

      PROGRAM HANDS (INPUT,OUTPUT,PUNCH,TAPE5=INPUT,TAPE6=OUTPUT,
      * TAPE7=PUNCH)
C
CXXXX
C PREPROCESSOR FOR FEAP (BANDWIDTH MINIMIZING)
C
C INPUT
C
C CARD) (CIE) NUMNP,IPUNCH,IPRINT
C ELEMENT CARDS
C EOF CARD)
C
C IF IPUNCH AND/OR IPRINT FLAGS ARE ON, THE NODE RESEQUENCING
C ARRAY IS PUNCHED AND/OR PRINTED. THIS ARRAY IS THEN INPUT
C IN THE FEAP PROGRAM IMMEDIATELY FOLLOWING THE FEAP CARD,
C A 1 IN CC 79 OF THE FEAP CARD SIGNALS THAT THE RESEQUENCING
C ARRAY IS INCLUDED IN THE FEAP INPUT.
C
C THE FORMAT FOR THE RESEQUENCING PUNCH IS *
C
C CC 1-5: A SEQUENCE NUMBER (TO MAINTAIN CARD ORDER)
C CC 11-16: 15 ENTRIES IN THE RESEQUENCING ARRAY (1515)
C THE PUNCH IS TERMINATED WITH A 9999 IN CC 1-5
CXXXX
C
C
C COMMON /FILES/ I11,I12,I13
C COMMON /CONTRL/ JNK(7),INUM,INEX,IENUM,NTUTD
C COMMON /FRRS/ IERF(33)
C COMMON /OPUT/ MED(21)
C COMMON /SIG1/ MSEP(91)
C COMMON /MISC/ IASTER,ZERO,RAU
C COMMON /KRRCOM/ KRR,KANT
C COMMON /TPULGY/ I1,J1(3),NEXT,LAST,KORIG,KNEW
C COMMON /S/ NR,MM,IR,IR
C COMMON /P/ IHO,IHE
C COMMON /TIME/ STIME,NCM
C
C COMMON IA(1)
CXX
C READ (5,50) INUM,IPUNCH,IPRINT
C 50 FORMAT (315)
CXX
C INEX=INUM
C IENUM=2
C NTUTD=0
C NR=INUM
C MM=0
C LAST=J*INUM
C NEXT=INUM+1
C
C N1=1
C N2=N1+INEX
C N3=N2+INUM
C N4=N3+INUM
C N5=N4+INUM
C N6=N5+INUM
C N7=N6+INUM
C N8=N7+INUM
C N9=N8+LAST
C
C CALL FIELD (IA(N9))
C DO 10 I=1,INEX
C 10 IA(I)=1
C DO 20 I=N2,N9
C 20 IA(I)=0
C
C CALL LLIN (IA(N1),IA(N7),IA(N8),INUM*4)
C
C N9=N8+LAST+2*MM+10
C CALL FIELD (IA(N9))
C CALL SCHEME (IA(N8),IA(N1),IA(N2),IA(N3),IA(N4),IA(N5),IA(N6),

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      * IA(N7),IA(N8*LAST))
C      WRITE (6,100) KORIG*KNEW
100  FORMAT (1H1/15H ORIGINAL HW =,IS/15H FINAL HW =,IS )
C      IF(IPRINT.NE.0) WRITE (6,800)
800  FORMAT (2CH10LD NODE      NEW NODE    //)
      N7=N7+INUM-1
      J=0
      IH=N7-1
      K=0
      DO 40 I=N7,N7H
      J=J+1
      IF(IPUNCH.EQ.0) GO TO 30
      IF(MOD(J,15).NE.1) GO TO 30
      IL=IH+1
      IH=IH+15
      IF(IH.GT.N7H) IH=N7H
      K=K+1
      WRITE (7,60) K,(IA(IL),L=IL,IH)
30  IF(IPRINT.NE.0) WRITE (6,70) J,IA(I)
40  CONTINUE
60  FORMAT (16IS)
70  FORMAT (1X,1B,5X,1B)
      J=9999
      IF(IPUNCH.NE.0) WRITE (7,60) J
      STOP
      END
      SUBROUTINE FIELD (IA)
      DIMENSION IA(1)
      DATA LAST/,/,MAX/,J20000H/,MIN/,55000H/
      NOW=((LUCF(IA(1))/100H)+1)*100H
      IF(NOW.GT.MAX) GO TO 100
      IF(NOW.LT.MIN) NOW=MIN
      IF(NOW.GT.LAST) GO TO 50
      IF(NOW.LT.LAST-10000) GO TO 50
      GO TO 75
50  CALL XHFL (NOW)
      LAST=NOW
75  RETURN
100  WRITE (6,200) NOW
200  FORMAT (16H0*** ERROR EXCESSIVE CORE REQUESTED /
      * 11X,12H00H REQUESTS, 17, 7H WORDS )
      STOP
      END
      IDENT  LOAD  (I*H*LL*IN*IF)
      ENTRY  LOAD
      VFI)  24/4HLOAD,3H/5
LOAD      HSS4  1
          SB1  1
          SA2  A1+H1
          SA3  A2+H1
          SA4  A3+H1
          SA5  A4+H1
          MA3  1
          SA2  X2
          SA3  X3
          SA4  X4
          SA1  X1
          SA5  X5
          SB3  X2
          IA5  X2-X3      LH TO B3
          SB1  X3-1      LH-LL (MASK WIDTH LESS 1)
          SB2  X5        NO OF LFT SHFTS FOR IN
                        NO OF RHT SHFTS TO BUILD MASK
          LA4  H1+X4     POSN IN
          SA4  60        H4=60
          SA6  H2+X6     FORM MASK
          SA4  B4-H1     LFT SHFTS TO PUT (IF) IN LO BITS
          LA6  H3+X6     POSN MASK
          BA7  X6+X1     GET OLD VALUE
          HA1  X1-X7     ZERO-OUT OLD VALUE
          LA7  B4+X7     POSN (IF) IN LO BITS

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      MA6      X1*X4      MASK IN NEW VALUE
      SA7      B5        STORE IF
      SA6      A1        STORE I
      EQ       LOAD      ALL DONE
      END
      IDENT    UNLOAD (I,LM,LL,IN)
      ENTRY    UNLOAD
      VFD      36/6HUNLOAD,24/4
UNLOAD      BSSZ      1
            S01      1
            SA2      A1*B1
            SA3      A2*B1
            SA4      A3*B1
            MX1      1
            SA2      X2
            SA3      X3
            SA1      X1
            SD4      X4
            SD3      X2
            IAS      X2-X3      LM-LL (MASK WIDTH LESS 1)
            SD2      X5        NO OF RIGHT SHIFTS TO BUILD MASK-
            IAS      X5-X2      -LL TO X7
            AA6      B2,X0      FORM MASK
            SM1      X7*B1      NO OF LEFT SHIFTS TO PACK IN RIGHT
            LA6      B3,X6      POSN MASK
            BA7      X6*X1      EXTRACT VALUE
            LA6      B1,X7      SHIFT IN TO LO ORDER POSN
            SA6      B4        STORE IN
            EQ       UNLOAD      ALL DONE
      END
      SUBROUTINE ELIN (NODIN,MAXJ,ID,NMNP,NSEC)
      COMMON /FILES/  I11,I12,I13
      COMMON /CONTRL/  NLOAD,NCOMB,IMODE,IPLT,INSTR1,ICORE,INSYS,INUM,
1      INEX,IENUM,INTOTU
      COMMON /FRRS/    IERF(25),KARD,KARDS,NSECT,KOLS(2),IERR,NERRS,
1      MAIN1
      COMMON /OPUT/    HED1(8),HE02(8),NLINE,NPAGE,LSTOFF,LSPS,LSP
      COMMON /STG1/    MSEP(25),LSEP(25),IORD(25),ICARD(8),NSEP,NSEP1,
1      LUSEP,MXERR,NSEP,NK,LBLNK,IT
      COMMON /MISC/     IASTER,ZERO,RAD
      COMMON /KRA/COM/  KKK,KNT
      DIMENSION        NODIN(1),MXJ(1),ID(1),IJ(8),JI(24)
      DIMENSION        IAT(8),IX(8)
      NODH=NMP-NINTOTU
      KARD=1
      MAXN=8
      IEL=0
15  READ (5,115) IELN,IAT
115 FORMAT (15,15X,8I3)
      IF (EOF(5)) 99,20
20  IF (IELN.LE.0) GO TO 99
      IF (IELN.GT.IEL+1) GO TO 60
      IF (IELN.LE.IEL) GO TO 98
25  READ (5,120) IJ
120 FORMAT (8I4)
      IEL=IELN
      IH=0
      DO 30 I=1,MAXN
      IF (IJ(I).LE.0) GO TO 35
      IH=IH+1
      IX(IH)=IAT(IH)
      IF (IX(IH).EQ.0) IX(IH)=1
30  CONTINUE
35  M=0
      DO 45 I=1,IH
      L=IJ(I)
      IF (L.GT.NMNP) GO TO 97
      M=M+1
      JI(M)=L
45  CONTINUE
      IF (M.GT.0) CALL PACK (JI,MXJ,ID,M)
      GO TO 15

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60 IF (IEL.EQ.0) GO TO 96
   NG=IELN-IEL-1
   DO 80 N=1,NG
     M=0
     IEL=IEL+1
     DO 70 I=1,IM
       IJ(I)=IJ(I)+IX(I)
       L=IJ(I)
       IF (L.GT.NMNP) GO TO 97
       M=M+1
       JI(M)=L
70 CONTINUE
   IF (M.GT.0) CALL PACK (JI,MXJ,IO,M)
80 CONTINUE
   GO TO 25
96 WRITE (6,600)
600 FORMAT (50H0** ERROR ** FIRST ELEMENT OMITTED. )
   STOP 1
97 WRITE (6,610) IEL
610 FORMAT (50H0** ERROR ** NODE NUMBER OUT OF RANGE. ELEM NO. ,IS )
   STOP 2
98 WRITE (6,620) IELN
620 FORMAT (50H0** ERROR ** ELEMENT OUT OF SEQUENCE. ELEM NO. ,IS )
   STOP 3
99 RETURN
END
SUBROUTINE PACK (JI,MXJ,IJ,N)
COMMON /TPOLGY/ I1,JJ,J2,J3,NEXT,LAST
COMMON /CONTROL/ IUMMY(7),INUM
DIMENSION      JI(1),MXJ(1),IJ(1)
NN=N
DO 20 J=1,N
  J=JI(1)
  IF (MXJ(J).EQ.0) GO TO 20
  MXJJ=MXJ(J)
  CALL UNLOAD (MXJJ,45,3,KI)
  CALL UNLOAD (MXJJ,30,16,KK)
  CALL UNLOAD (MXJJ,15,1,KJ)
  L=I
  IF (KJ.EQ.0) GO TO 10
  NN=NN+1
  L=NN
10 JI(L)=KI
  IF (KK.EQ.0) GO TO 20
  NN=NN+1
  JI(NN)=KK
20 CONTINUE
30 IF (NEXT>NN>NN.LI.LAST) GO TO 50
  LAST=LAST+INUM
  CALL FIELD (IJ(LAST))
  IL=LAST-INUM+1
  DO 40 I=IL,LAST
    IJ(I)=0
40 CONTINUE
  GO TO 30
50 DO 100 I=1,NN
  II=JI(I)
  DO 90 J=1,NN
    JJ=JI(J)
    IF (II.EQ.JJ) GO TO 90
    CALL PK (IJ(II))
90 CONTINUE
100 CONTINUE
  RETURN
END
      IDENT  PK      (IJ(I))
      ENTRY  PK
PR      VFD      12/2HPR,48/1
      BSS      1
      USE      /TPOLGY/
I      BSS      1
JI      BSS      1

```

J2	BSS	1	
J3	BSS	1	
NEXT	BSS	1	
	USE		
	USE	/S/	
NUMY	BSS	1	
MM	BSS	1	
	USE		
	S <sub>H</sub> 2	1	1 TO A2
	S <sub>H</sub> 4	NEXT	NEXT TO X4
	S <sub>H</sub> 3	J1	J1 TO X3
	S <sub>H</sub> 1	X1	
	S <sub>H</sub> 1	A1	
	L <sub>A</sub> 6	14	14 BIT LO MSK IN X0
	S <sub>D</sub> 3	46	INITIAL SHFT VALUE
	S <sub>D</sub> 2	X2	1 TO H2
	S <sub>H</sub> 4	14	H4=14
	S <sub>H</sub> 1	H1-H2	AD(IJ(0)) TO H1
	S <sub>H</sub> 7	H0	INIT LOOP COUNTER
	S <sub>H</sub> 5	1	
	S <sub>H</sub> 2	MM	MM TO X2
	S <sub>A</sub> 7	X4+1	FORM NEW NEXT IN X7
	S <sub>H</sub> 2	H3	INIT H2
	S <sub>H</sub> 6	X3	J1 TO H6
LOOP	L <sub>X</sub> 6	H4,X1	SHFT IJ INTO X6
	S <sub>H</sub> 7	H7,A5	
	H <sub>A</sub> 5	X6,X0	EXTRACT J
	S <sub>D</sub> 5	X5	MOVE J TO H5
	Z <sub>H</sub>	X5,PACK12	IS J ZERO
	E <sub>H</sub>	H5,H6,PK	DONE IF J=J1
	L <sub>A</sub> 6	14	SHFT IJ
	S <sub>D</sub> 2	H2-H4	DEC SHFT REG
	H <sub>A</sub> 5	X6,X0	EXTRACT J
	S <sub>H</sub> 5	X5	MOVE J TO H5
	Z <sub>H</sub>	X5,PACK12	IS J ZERO
	E <sub>H</sub>	H5,H6,PK	DONE IF J=J1
	L <sub>A</sub> 6	14	SHFT IJ
	S <sub>D</sub> 2	H2-H4	DEC SHFT REG
	H <sub>A</sub> 5	X6,X0	EXTRACT J
	S <sub>H</sub> 5	X5	MOVE J TO H5
	Z <sub>H</sub>	X5,PACK3	IS J ZERO
	E <sub>H</sub>	H5,H6,PK	DONE IF J=J1
	S <sub>H</sub> 1	H1,X1	FETCH NEXT IJ TO X1
	S <sub>H</sub> 2	H3	INIT SHFT REG
	E <sub>H</sub>	LOOP	BACK TO TOP TO CONSIDER NEXT IJ
PACK3	L <sub>A</sub> 4	42	SHFT CURRENT NEXT
	S <sub>H</sub> 7	44	STORE NEW NEXT
PACK12	H <sub>A</sub> 6	X6,X4	PACK CURRENT NEXT INTO ADDRESS PORTION
	H <sub>A</sub> 6	X6,X3	PACK J1 INTO IJ
	L <sub>A</sub> 6	H2,X6	REPUSH IJ
	S <sub>H</sub> 6	X2	MOVE MM TO H6
	S <sub>A</sub> 6	A1	STORE UPDATED IJ
	G <sub>T</sub>	H6,H7,PK	DONE IF MM.GT.LOOP COUNT
	S <sub>A</sub> 7	H7	NEW MM TO X7
	S <sub>A</sub> 7	A2	STORE NEW MM
	E <sub>H</sub>	PK	DONE
	END		
	IDENT	UNPK (IJ(I))	
	ENTRY	UNPK	
	PRD	24/4HUNPK,36/1	
UNPK	BSS	1	
	USE	/TPOLGT/	
1	BSS	1	
J1	BSS	1	
J2	BSS	1	
J3	BSS	1	
NEXT	BSS	1	
	USE		
	M <sub>A</sub> 0	14	BUILD MASK
	S <sub>A</sub> 1	X1	
	L <sub>A</sub> 0	14	14 BIT LO MASK

```

      LX1      14      SHFT IJ(1)
      HA6      X1*X0    GET J1
      LX1      14      SHFT IJ(1)
      SA6      J1      STORE J1
      HA7      X1*X0    GET J2
      LA1      14      SHFT IJ(1)
      SA7      J2      STORE J2
      HA6      X1*X0    GET J3
      LA1      18      SHFT IJ(1)
      SA6      J3      STORE J3
      SH7      X1      GET NEXT
      SA7      H7      PUT NEXT IN X7
      SA7      NEXT     STORE NEXT
      EQ       UNPK     DONE
      END

SUBROUTINE SCHEME (IG,IC,IDEG,IDIS,IW,NEW,ICC,ILD,IPP)
  DIMENSION IG(1),IC(1),IDEG(1),IDIS(1),IW(1),NEW(1),ICC(1),ILD(1),
  * IPP(1),NODESL(100)
  COMMON /TPOLOGY/ I1,J1(3),NEXT,LAST,KORIG,KNEW
  COMMON /S/ AN,MM,IM,IH
  COMMON /P/ IH0,IHE
  COMMON /TIME/ STIME,NCM
  EQUIVALENCE (IH,ATIME)
  NT=MIN0(80,AN)
  NUM=1
  NCM=2
  IO=2
  IP=0
  CALL DEGREE (IG,IDEG)
  MODD=MODF(IDEG,IPP)
  NCM=COMPHI(IG,IC,IDEG,IW,ICC)
  MAXD=MAXDGR(0,IC,IDEG)
  MM=MAXD
  DO 30 I=1,NN
    NEW(I)=I
  30 ILD(I)=I
    IS=MAXBND(0,IG,IC,IDEG,NEW,ILD)
    KORIG=IS
    IH0=IH
    DO 35 I=1,NN
      NEW(I)=J
  35 ILD(I)=J
      DO SUB NC=1,NCM
        MI=MINDEG(NC,IC,IDEG)
        MAD=MI
        IF (NUM.EQ.0) GO TO 91
        MA=MAXDGR(NC,IC,IDEG)
  40 MAD=MI+((MA-MI)*NUM)/NUM
        MAD=MIN0(MAD,MODD-1)
        MAD=MAX(MAD,MI)
  41 CALL D1AM(NC,MAD,NL,NODESL,MAXLEV,IG,IC,IDEG,IDIS,IW,ICC)
        JMAX=MIN0(NT,NL)
        IMM=JMAX
        DO 400 J=1,JMAX
          CALL RELABL(1,NODESL(J),IG,IC,IDEG,IDIS,IW,NEW,ICC,ILD)
          IH=MAXBND(NC,IG,IC,IDEG,NEW,ILD)
          IE=ICC(NC+1)-1
          IF (IH-IE) 400,350,300
  300 IMM=IH
          IMM=IH
          IJ=J
          GO TO 400
  350 IF (IMM.LE.IH) GO TO 400
          IMM=IH
          IJ=J
  400 CONTINUE
          CALL RELABL(1,NODESL(IJ),IG,IC,IDEG,IDIS,IW,NEW,ICC,ILD)
  500 CONTINUE
          CALL STACK(IDEG,NEW,ILD,IW)
          IH=MAXBND(0,IG,IC,IDEG,NEW,ILD)
          IF (IH-IE) 715,742,744

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```

742 IF (IM.LI.IH) GO TO 715
744 DO 712 I=1,NN
    ILD(I)=1
712 NEW(I)=1
    CALL STACK(IDEG,NEW,ILD,IM)
    IH=IS
    IM=IM
715 IM=IM
    CALL REVERS(NEW,ILD)
    IM=MAXND(0,IG,IC,IDEG,NEW,ILD)
717 IF (IM.LI.IH) GO TO 720
    CALL REVERS(NEW,ILD)
    IM=MAXND(0,IG,IC,IDEG,NEW,ILD)
720 IM=IM
    KNEW=IH
600 RETURN
END
SUBROUTINE UFGREE (IG,IDEG)
    DIMENSION IG(1),IDEG(1)
    COMMON /TPOLGY/ I1,J1,J2,J3,NEXT
    COMMON /S/ NN,MM,IM,IH
    DO 50 I=1,NN
        NEXT=1
        J=0
10    CALL UNPK (IG(NEXT))
        IF (J3.EQ.0) GO TO 20
        J=J+3
        GO TO 10
20    IF (J2.EQ.0) GO TO 30
        J=J+2
        GO TO 40
30    IF (J1.EQ.J) GO TO 40
        J=J+1
40    IDEG(I)=J
        MM=MAX(MM,J)
50    CONTINUE
    RETURN
END
FUNCTION MODE(IDEG,MODE)
    COMMON /S/ NN,MM
    DIMENSION IDEG(1),MODE(1)
    DO 10 I=1,MM
10    MODE(I)=0
        DO 20 I=1,NN
            K=IDEG(I)
            IF (K.EQ.0) GO TO 20
            MODE(K)=MODE(K)+1
20    CONTINUE
        MODE=0
        MAX=0
        DO 30 I=1,MM
            K=MODE(I)
            IF (K.LE.MAX) GO TO 30
            MAX=K
30    CONTINUE
        MODE=1
    RETURN
END
FUNCTION COMPNT (IG,IC,IDEG,IW,ICC)
    DIMENSION IG(1),IC(1),IDEG(1),IW(1),ICC(1)
    COMMON /S/ NN,MM,IM,IH
    COMMON /TPOLGY/ I1,JJ(3),NEXT
    DO 100 I=1,NN
        ICC(I)=0
        IC(I)=0
100    CONTINUE
        NC=0
        ICC(1)=1
105    DO 110 I=1,NN
            IF (IC(I).EQ.0) GO TO 120
            COMPNT=NC
110    CONTINUE

```



```

      RETURN
120 NC=NC+1
      KI=0
      KO=1
      IW(1)=I
      IC(1)=NC
      IF (NC.LE.0) GO TO 130
      IS=ICC(NC)+1
      ICC(NC+1)=IS
130 KI=KI+1
      II=I*(KI)
      N=IDEG(II)
      IF (N.EQ.0) GO TO 105
      NEXT=II
140 CALL UNPK (IG(NEXT))
      DO 15 J=1,3
      IA=JJ(J)
      IF (IA.EQ.0) GO TO 200
      IF (IC(IA).NE.0) GO TO 180
      IC(IA)=NC
      KO=KO+1
      IW(KO)=IA
      IS=ICC(NC+1)+1
      ICC(NC+1)=IS
180 CONTINUE
      GO TO 140
200 IF (KO.LE.KI) GO TO 105
      GO TO 130
      END
      FUNCTION MAXDGR(NC,IG,IC,IDEG)
      DIMENSION IC(1),IDEG(1)
      COMMON /S/ NN,MM,IH,IR
      M=0
      IF (NC.NE.0) GO TO 40
      DO 30 I=1,NN
      IF (IDEG(I).LE.M) GO TO 30
      M=IDEG(I)
30 CONTINUE
      GO TO 100
40 DO 50 I=1,NN
      IF (IC(I).NE.NC) GO TO 50
      IF (IDEG(I).LE.M) GO TO 50
      M=IDEG(I)
50 CONTINUE
100 MAXDGR=M
      RETURN
      END
      FUNCTION MAXRND(NC,IG,IC,IDEG,NEW,ILO)
      DIMENSION IG(1),IC(1),IDEG(1),NEW(1),ILO(1)
      COMMON /S/ NN,MM,IH,IR
      COMMON /TPOLGY/ II,JJ(3),NEXT
      IH=0
      M=0
      DO 100 I=1,NN
      MX=0
      IA=NEW(I)
      IF (NC.EQ.0) GO TO 50
      IF (IA.EQ.0) GO TO 100
      IF (NC.NE.IC(IA)) GO TO 100
50 N=IDEG(IA)
      IF (N.LE.0) GO TO 100
      NEXT=IA
60 CALL UNPK (IG(NEXT))
      DO 80 J=1,3
      II=JJ(J)
      IF (II.EQ.0) GO TO 90
      IB=MAX0(I-1,ILO(II))
      IF (IB.GT.MX) MX=IB
80 CONTINUE
      GO TO 40
90 IF (MX.GT.M) M=MX
      IH=IH+MX

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```

100 CONTINUE
  MAXDEG=M
  RETURN
END
FUNCTION MINDEG(NC,IC,IDEG)
  DIMENSION IC(1),IDEG(1)
  COMMON /S/ NN,MM,IH,IH
  M=10000
  IF(NC,NE,1) GO TO 40
  DO 30 I=1,NN
    IF(M,LE,IDEG(I)) GO TO 30
    M=IDEG(I)
  30 CONTINUE
  GO TO 100
  40 DO 60 I=1,NN
    IF(IC(I),NE,NC) GO TO 60
    IF(M,LE,IDEG(I)) GO TO 60
    M=IDEG(I)
  60 CONTINUE
100 MINDEG=M
  RETURN
END
SUBROUTINE DIAM(NC,MAXDEG,NL,NODESL,MAXLEV,
  * IG,IC,IDEG,IOIS,IW,ICC)
  DIMENSION IG(1),IOIS(1),IW(1),ICC(1),IC(1),IDEG(1)
  DIMENSION NODESL(1)
  COMMON /S/ NN,MM,IH,IH
  NL=0
  MAXLEV=1000
  DO 100 I=1,NN
    IF(NC,NE,IC(I)) GO TO 100
    IF(MAXDEG,LT,IDEG(I)) GO TO 100
  105 MD=IDIST(1,ML,MAXLEV,IG,IC,IDEG,IOIS,IW,ICC)
    IF(MD,LE,0) GO TO 115
  56 IF(ML-MAXLEV)58,64,100
  58 MAXLEV=ML
    NL=NL+1
    NODESL(NL)=I
    GO TO 100
  64 IF(NL,GE,100) GO TO 100
    NL=NL+1
    NODESL(NL)=I
  100 CONTINUE
  GO TO 110
  115 ML=I
    NODESL(1)=I
    MAXLEV=0
  110 RETURN
END
FUNCTION IDIST(NS,ML,MAXLEV,IG,IC,IDEG,IOIS,IW,ICC)
  DIMENSION IG(1),IC(1),IDEG(1),IOIS(1),IW(1),ICC(1)
  COMMON /S/ NN,MM,IH,IH
  COMMON /T/OLGY/ II,JJ(3),NEXT
  ICN=IC(NS)
  NAC=ICC(ICN+1)-ICC(ICN)
  DO 50 I=1,NN
    IF(IC(I),NE,IC(NS)) GO TO 50
    IOIS(I)=0
  50 CONTINUE
  LL=1
  L=0
  KI=0
  KO=1
  ML=0
  I*(1)=NS
  IDIS(NS)=-1
  130 KI=KI+1
    IF(KI,NE,LL) GO TO 135
  132 L=L+1
    LL=KO+1
    K=KO+KI+1
    IF(K,LE,ML) GO TO 135

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133 ML=K
    IF (ML.GT.MAXLEV) GO TO 220
135 II=IW(KI)
    N=IDEG(II)
    IF (N.EQ.0) GO TO 215
    NEXT=II
140 CALL UNPK (IG(NEXT))
    DO 143 I=1,3
    IA=JJ(I)
    IF (IA.EQ.0) GO TO 200
    IF (IDIS(IA).NE.0) GO TO 190
    IDIS(IA)=L
    KO=KO+1
    IA*(KO)=IA
190 CONTINUE
    GO TO 143
200 IF (KO.LI.NNC) GO TO 130
205 IDIST=L
    IDIS(NS)=0
    K=KO-KI
    IF (K.GT.ML) ML=K
    GO TO 230
215 L=0
    GO TO 205
220 IDIST=1
230 RETURN

END
SUBROUTINE RELAHL(NS,NODES,IG,IC,IDEG,IDIS,IW,NEW,ICC,ILD)
    INTEGER X
    DIMENSION IG(1),IC(1),IDEG(1),IDIS(1),IW(1),NEW(1),ICC(1)
    DIMENSION ILD(1)
    DIMENSION NODES( 1),IAJ(50)
    COMMON /S/ AN,MM,IH,IB
    COMMON /TPOLGY/ II,JI(3),NEXT
    I=NODES(1)
    ICN=IC(I)
    NT=ICC(ICN)-1
    DO 50 I=1,NN
    IF (IC(I).NE.ICN) GO TO 50
40 IDIS(I)=0
50 CONTINUE
    DO 100 J=1,NS
    JJ=NODES(J)
    IDIS(JJ)=-1
    JT=J*NT
    NEW(JJ)=JJ
100 ILD(JJ)=JT
    KI=NT
    KO=NS+NT
    LL=KO
    L=1
    J=KO
    NAC=ICC(ICN+1)-1
130 KI=KI+1
    IF (KI.NE.LL) GO TO 135
132 L=L+1
    LL=KO+1
135 II=NEW(KI)
    N=IDEG(II)
    IF (N.EQ.0) GO TO 255
    IJ=0
    NEXT=II
140 CALL UNPK(IG(NEXT))
    DO 143 I=1,3
    IA=JI(I)
    IF (IA.EQ.0) GO TO 200
    IF (IDIS(IA).NE.0) GO TO 190
150 IJ=IJ+1
    IDIS(IA)=L
    KO=KO+1
    IAJ(IJ)=IA
    IW(IJ)=IDEG(IA)

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190 CONTINUE
GO TO 140
200 IF (IJ-1) 250,210,220
210 J=K0
    IZ=IAJ(I)
    NEW(K0)=IZ
    ILD(I2)=K0
    GO TO 250
220 X=0
221 DO 230 I=2,IJ
    IF (IW(I).GE.IW(I-1)) GO TO 230
    X=IW(I)
    IW(I)=IW(I-1)
    IW(I-1)=X
225 X=IAJ(I)
    IAJ(I)=IAJ(I-1)
    IAJ(I-1)=X
230 CONTINUE
    IF (X.GT.0) GO TO 220
235 DO 240 I=1,IJ
    J=J+1
    IZ=IAJ(I)
    NEW(J)=IZ
    ILD(I2)=J
240 CONTINUE
250 IF (K0+L1-KN) GO TO 130
255 RETURN
END
SUBROUTINE STACK(IDEG,NEW,ILD,IW)
DIMENSION IDEG(1),NEW(1),ILD(1),IW(1)
COMMON /S/ NN
COMMON /ZER0/ KI
KI=0
NN=NN-1
DO 10 I=1,NN
    IF (ILD(I).GT.0) GO TO 10
    KI=KI+1
    IW(KI)=ILD(I)
10 CONTINUE
    IF (KI.LE.0) GO TO 70
    CALL SORT(IW,KI)
    DO 40 L=1,KI
    I=IW(L)-L+1
    K=NEW(I)
    IF (I.GE.NN) GO TO 30
    DO 20 J=I+NN
20 NEW(J)=NEW(J+1)
30 NEW(NN)=K
40 CONTINUE
70 DO 80 I=1+NN
    K=NEW(I)
80 ILD(K)=1
    RETURN
END
SUBROUTINE SORT(LIST,NL)
DIMENSION LIST(1)
IF (NL.LE.1) RETURN
NL=NL-1
DO 20 I=1,NL
    K=NL-1
    KFLAG=0
    DO 10 J=1,K
    IF (LIST(J).LE.LIST(J+1)) GO TO 10
    KFLAG=1
    L=LIST(J)
    LIST(J)=LIST(J+1)
    LIST(J+1)=L
10 CONTINUE
    IF (KFLAG.EQ.0) RETURN
20 CONTINUE
RETURN
END

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SUBROUTINE HEVFRS(NEW,ILO)
DIMENSION NEW(1):ILO(1)
COMMON /S/ KN
COMMON /ZERO/ KT
J=(NN-KT)/2
LL=NN-KT+1
DO 10 I=1,J
L=LL-I
K=NEW(L)
NEW(L)=NEW(I)
10 NEW(I)=K
DO 20 I=1,NN
K=NEW(I)
20 ILO(I)=I
RETURN
END

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	DO 402 J = 1,NDF	FLM 73C
	FORCE(K,1) = 0.0	FLM 74C
	UDL(K,1) = UL(J,1)	FLM 75C
402	K = K + 1	FLM 76C
	CALL INCOMP (AR(1,NSTP),AH,CC,ESTIF,UDL,MM,NSIF,2)	FLM 77C
	K = 1	FLM 78C
	DO 403 I = 1,NDIM	FLM 79C
	DO 403 J = 1,NDF	FLM 80C
	UL(J,1+4FL) = CC(K)	FLM 81C
403	K = K + 1	FLM 82C
	NSTP = NEL + NDIM	FLM 83C
	NL = (NDIM*(NDIM+1))/2	FLM 84C
	XX = D(4,MA)	FLM 85C
	GG = 2.0*U(2,MA)	FLM 86C
	DO 413 II = 1,LINI	FLM 87C
	CALL BRICK2(STUW(1,II),STUW(2,II),STUW(3,II),NDIM,NEL,NSIDE)	FLM 88C
	CALL INSHAP(STUW(1,II),SG,SHAPE(1,NEL+1),NDIM)	FLM 89C
	DO 411 K = 1,NL	FLM 90C
	UDL(K,II) = SIG(K)	FLM 91C
411	UDL(K,II+LINT) = EPS(K)	FLM 92C
	IF (IS*EQ.4) GO TO 413	FLM 93C
C****	FORM DISPLACEMENT GRADIENTS	FLM 94C
	DO 405 I = 1,NDF	FLM 95C
	DO 405 J = 1,NDIM	FLM 96C
	TEMP = 0.	FLM 97C
	DO 404 K = 1,NSTP	FLM 98C
404	TEMP = TEMP + SHAPE(J,K)*UL(I,K)	FLM 99C
405	DV(I,J) = TEMP/XJAC	FLM100C
C****	FORM STRESS - STRAINS	FLM101C
	DO 407 I = 1,NDIM	FLM102C
	XX(I) = 0.	FLM103C
	DO 406 J = 1,NEL	FLM104C
406	XX(I) = XX(I) + X(I,J)*SHAPE(4,J)	FLM105C
407	DUL(I,II) = XX(I)	FLM106C
	EPS(1) = DV(1,1)	FLM107C
	EPS(2) = 0.5*(DV(1,2)+DV(2,1))	FLM108C
	EPS(3) = DV(2,2)	FLM109C
	TEMP = EPS(1) + EPS(3)	FLM110C
	IF (NDIM*EQ.2) GO TO 408	FLM111C
	EPS(4) = 0.5*(DV(2,3)+DV(3,2))	FLM112C
	EPS(5) = DV(3,3)	FLM113C
	EPS(6) = 0.5*(DV(1,3)+DV(3,1))	FLM114C
	TEMP = TEMP + EPS(5)	FLM115C
408	DO 409 I = 1,NL	FLM116C
409	SIG(I) = GG*EPS(I)	FLM117C
	DO 411 I = 1,NL,2	FLM118C
410	SIG(I) = SIG(I) + XX*TEMP	FLM119C
	WU = STUW(4,II)	FLM120C
	TEMP = WU*U(3,MA)*XJAC	FLM121C
	II = 1	FLM122C
	DO 412 I = 1,NEL	FLM123C
	SX = SHAPE(1,I)*WU	FLM124C
	SY = SHAPE(2,I)*WU	FLM125C
	SM = SHAPE(4,I)*TEMP	FLM126C
	FORCE(II, 1) = FORCE(II, 1) - SIG(1)*SX - SIG(2)*SY - SM*UDL(1,1)	FLM127C
	FORCE(II+1,1) = FORCE(II+1,1) - SIG(2)*SX - SIG(3)*SY - SM*UDL(2,1)	FLM128C
	IF (NDIM*EQ.2) GO TO 412	FLM129C
	SZ = SHAPE(3,I)*WU	FLM130C
	FORCE(II, 1) = FORCE(II, 1) - SZ*SIG(6)	FLM131C
	FORCE(II+1,1) = FORCE(II+1,1) - SZ*SIG(4)	FLM132C
	FORCE(II+2,1) = FORCE(II+2,1) - SZ*SIG(5) - SY*SIG(4) - SX*SIG(6)	FLM133C
	SM*UDL(3,1)	FLM134C
412	II = II + NDF	FLM135C
413	CONTINUE	FLM136C
	NSTP = NDIM + NDIM + 1	FLM137C
	DO 421 J = 1,NSTP	FLM138C
	NOFINT = NPF	FLM139C
	IF (NSIG(J).GT.0) NOFINT = .TRUE.	FLM140C
	NPL = .FALSE.	FLM141C
	IF (NOMPLT*LE.0) GO TO 41	FLM142C
	DO 40 NP = 1,NOMPLT	FLM143C
40	IF (NEDATA(NP,1).EQ.N.AND.NEDATA(NP,2).EQ.J) NPL = .TRUE.	FLM144C

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41 IF (NUPRNT.AND..NOT.NPL) GO TO 421
DO 414 I = 1,NDIM
414 XX(I) = 0.0
DO 415 I = 1,NL
415 SIG(I) = 0.0
EPS(I) = 0.0
DO 417 L = 1,LINT
E = (1.+S1(J)/STW(1,L))* (1.+S2(J)/STW(2,L))/4.0
IF (NDIM.EQ.3) E = E*(1.+S3(J)/STW(3,L))/2.0
DO 416 I = 1,NDIM
416 XX(I) = XX(I) + E*UDL(I,L)
DO 417 I = 1,NL
SIG(I) = SIG(I) + E*UDL(I,L)
417 EPS(I) = EPS(I) + E*UDL(I,L*LINT)
IF (NUPRNT) GO TO 420
C.... COMPUTE INVARIANTS FOR TWO D PROBLEMS
IF (NDIM.NE.2) GO TO 418
CALL PSTRES(NDIM,SIG,SIG(4),SIG(5),SIG(6))
CALL PSTRES(NDIM,EPS,EPS(4),EPS(5),EPS(6))
418 MCT = MCT + 1
IF (MCT.GT.10) GO TO 419
WRITE (IIP6,2002) U,HEAD,TIME,IPI,(XHED(I),XH,I=1,NDIM),
X (THED(I),SH,I=1,NL)
WRITE (IIP6,2001) (BLANK,BLANK,I=1,NDIM),(THED(I),EH,I=1,NL)
IF (NDIM.EQ.2) WRITE (IIP6,2004)
IPI = IPI + 1
MCT = 15
419 WRITE (IIP6,IAR1) N,(XX(I),I=1,NDIM), SIG
WRITE (IIP6,IAR2) U,M,MA,EPS
420 IF (NPL) CALL PLDATA(N,NDIM,N,J,THED,XX,SIG,FORCE)
421 CONTINUE
5 RETURN
C.... FORMAT STATEMENTS
1000 FORMAT(4F10.0,I10)
2000 FORMAT(5X,54HLINEAR ISOTROPIC ELASTIC TWO/THREE DIMENSIONAL ELEMEN
17/5X,3HE =,E15.5,5X,4HNU =,F10.5,5X,SHRMO =,F10.5/5X,4HAA =,F10.5
2,2X51H(1.0 FOR PLANE STRESS, 2.0 FOR PLANE STRAIN AND 3D)/5X)
2001 FORMAT(1X,9(2A6))
2002 FORMAT(4I,12A6,E13.5,17X,4HPAGE,13//5X,16HELEMENT STRESSES//
1 1X,7HELEMENT,9(2A6))
2003 FORMAT(5X,9H*****FATAL ERROR 12** ATTEMPT TO USE INCOMPATIBLE MODE E
1ELEMENT WITH WRONG NUMBER OF CONNECTED NODES)
2004 FORMAT(72X,8H1-VALUES,4X,8H2-VALUES,7X,5HANGLE )
END
SUBROUTINE VSTIF(NDIM,NDF,NEL,NSIDE,ST,FORCE,STW,LINT)
C..... VSTIF REWRITTEN.....7/20/73.....
DIMENSION STW(4,27),ST(1),FORCE(NDF,1),SHP(60)
COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SK(3,3),X(3,20),LD(120)
EQUIVALENCE(SHP,SHAPE)
N=NEL*NDIM
DO 200 I=1,LINT
SS=STW(1,I)
TT=STW(2,I)
UU=STW(3,I)
VV=STW(4,I)
CALL H1CR2(SS,TT,UU,NDIM,NEL,NSIDE)
DVOL=UU/XJAC
XJAC=XJAC*U
NS=0
DO 50 J=1,NEL
CC=SHAPE(4,J)*XJAC
DO 50 I=1,NDIM
NS=NS+1
SHP(NS)=SHAPE(I,J)
50 FORCE(I,J)=FORCE(I,J)+CC
NS=0
DO 150 J=1,N
CC=SHP(J)*DVOL
DO 100 I=1,J
ST(I+NS)=ST(I+NS)+CC*SHP(I)
100 NS=NS+J
150
200 CONTINUE

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FLM145C
FLM146C
FLM147C
FLM148C
FLM149C
FLM150C
FLM151C
FLM152C
FLM153C
FLM154C
FLM155C
FLM156C
FLM157C
FLM158C
FLM159C
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FLM184C
FLM185C
FLM186C
FLM187C
FLM188C
VST 1C
VST 2C
VST 3C
VST 4C
VST 5C
VST 6C
VST 7C
VST 8C
VST 9C
VST 10C
VST 11C
VST 12C
VST 13C
VST 14C
VST 15C
VST 16C
VST 17C
VST 18C
VST 19C
VST 20C
VST 21C
VST 22C
VST 23C
VST 24C
VST 25C
VST 26C
VST 27C
VST 28C

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RETURN	VST 29C
END	VST 30C
SUBROUTINE ASSEMB(NDIM,NDF,NEL,NSTF,ST,ESTIF,FORCE,GG,XX,R0)	ASS 1C
C..... ASSEMB ***** 7/03/73 *****	ASS 2C
DIMENSION ESTIF(NSTF,NSTF),ST(1),FORCE(1)	ASS 3C
C.... ASSEMBLY OF ELEMENT INTO THE STIFFNESS	ASS 4C
NU = NEL*NDIM	ASS 5C
NS = (NU*(NU+1))/2	ASS 6C
J1 = NDF*(NEL-1)	ASS 7C
ND1 = NDIM + 1	ASS 8C
DO 23 J = 1,NEL	ASS 9C
DO 22 N = 1,NDIM	ASS 10C
N1 = ND1 - N	ASS 11C
I1 = J1	ASS 12C
DO 21 I = J,NEL	ASS 13C
L = 1	ASS 14C
IF(I.EQ.J) L = N	ASS 15C
DO 200 M = L,NDIM	ASS 16C
M1 = ND1 - M	ASS 17C
CC = ST(NS)	ASS 18C
ST(NS) = 0.0	ASS 19C
ESTIF(I1+M1,J1+M1) = ESTIF(I1+M1,J1+M1) + XX*CC	ASS 20C
CC = GG*CC	ASS 21C
ESTIF(I1+N1,J1+M1) = ESTIF(I1+N1,J1+M1) + CC	ASS 22C
IF(M1.NE.N1) GO TO 200	ASS 23C
DO 190 K = 1,NDIM	ASS 24C
ESTIF(I1+K,J1+K) = ESTIF(I1+K,J1+K) + CC	ASS 25C
200 NS = NS - 1	ASS 26C
210 I1 = I1 - NDF	ASS 27C
220 CONTINUE	ASS 28C
230 J1 = J1 - NDF	ASS 29C
NU = NEL*NDF	ASS 30C
DO 300 J = 2,NU	ASS 31C
K = J - 1	ASS 32C
DO 300 I = 1,K	ASS 33C
ESTIF(I,J) = ESTIF(I,J) + ESTIF(J,I)	ASS 34C
ESTIF(J,I) = ESTIF(I,J)	ASS 35C
DO 400 I=1,NU	ASS 36C
400 ESTIF(I,I) = ESTIF(I,I)+R0*FORCE(I)	ASS 37C
RETURN	ASS 38C
END	ASS 39C
SUBROUTINE INSHAP(S,SG,SHAPE,NDIM)	INS 1C
DIMENSION SHAPE(4,1),SG(3,3),S(3)	INS 2C
DO 101 I = 1,NDIM	INS 3C
SHAPE(4,I) = 1. - S(I)**2	INS 4C
DO 101 J = 1,NDIM	INS 5C
101 SHAPE(I,J) = -2.0*SG(J,I)*S(J)	INS 6C
RETURN	INS 7C
END	INS 8C
SUBROUTINE INCOMP(A,H,C,S,F,ME,NE,ISW)	INC 1C
DIMENSION A( 9,ME),B( 9,NE),C(ME),S(NE,NE),F(NE)	INC 2C
GO TO (100,200,100), ISW	INC 3C
C.... REDUCE ARRAYS	INC 4C
100 DO 102 N=1,ME	INC 5C
P = A(N,1)	INC 6C
IF(P.EQ.0.0) GO TO 105	INC 7C
IF(N.EQ.ME) GO TO 103	INC 8C
NP = N+1	INC 9C
DO 102 I=NP,ME	INC 10C
AP = A(I,N)/P	INC 11C
IF(AP.EQ.0.0) GO TO 103	INC 12C
C(I) = C(I) - AP*C(N)	INC 13C
DO 101 J=1,NE	INC 14C
A(I,J) = A(I,J) - A(N,J)*AP	INC 15C
101 A(J,I) = A(I,J)	INC 16C
DO 102 J=1,NE	INC 17C
102 B(I,J) = B(I,J) - B(N,J)*AP	INC 18C
103 CONTINUE	INC 19C
IF(ISA.NE.1) GO TO 105	INC 20C
DO 104 I = 1,NE	INC 21C
AP = B(N,I)/P	INC 22C
F(I) = F(I) - AP*C(N)	INC 23C

	DO 104 J = 1,NE	INC 24C
	S(I,J) = S(I,J) - H(N,J)*AP	INC 25C
104	S(J,1) = S(I,J)	INC 26C
105	CONTINUE	INC 27C
	RETURN	INC 28C
C...	RECOVER INCOMPATIBLE DISPLACEMENTS	INC 29C
200	DO 201 I = 1,NE	INC 30C
	DO 201 J = 1,NE	INC 31C
201	C(I) = C(I) - B(I,J)*F(J)	INC 32C
C...	BACKSUBSTITUTE	INC 33C
	N = NE	INC 34C
202	C(N) = C(N)/A(N,N)	INC 35C
	IF(N*LE+1) RETURN	INC 36C
	I = N	INC 37C
	N = N-1	INC 38C
	DO 203 J = 1,NE	INC 39C
203	C(N) = C(N) - A(N,J)*C(J)	INC 40C
	GO TO 202	INC 41C
	END	INC 42C
	SUBROUTINE INSTIF(FORCE,SGO,STUW,LINT,NDF,NDIM,NEL,NSIDE,MM,	INS 1C
	1 C11,C12,C33)	INS 2C
	DIMENSION ESTIF( 4,1),FORCE(1),SGO(3,3),STUW(4,1)	INS 3C
	COMMON/SHAP/ XJAC,SHAPE(4,20),SG(3,3),SN(3,3),X(3,20),LD(120)	INS 4C
	ZN = 0.0	INS 5C
	A42 = 0.0	INS 6C
	DO 305 L = 1,LINT	INS 7C
	CALL BRICK2(STUW(1,L),STUW(2,L),STUW(3,L),NDIM,NEL,NSIDE)	INS 8C
	CALL INSHAP(STUW(1,L),SGO,SHAPE(1,NEL*1),NDIM)	INS 9C
	DV = STUW(4,L)/XJAC	INS 10C
	J1 = 1	INS 11C
	NELP = NEL + NDIM	INS 12C
	DO 304 J = 1,NELP	INS 13C
	I1 = 1	INS 14C
	XN = SHAPE(1,J)*DV	INS 15C
	YN = SHAPE(2,J)*DV	INS 16C
	A11 = C11*XN	INS 17C
	A12 = C12*YN	INS 18C
	A21 = C33*YN	INS 19C
	A22 = C33*XN	INS 20C
	A31 = C12*XN	INS 21C
	A32 = C11*YN	INS 22C
	IF(NDIM*EG+2) GO TO 301	INS 23C
	ZN = SHAPE(3,J)*DV	INS 24C
	A13 = C12*ZN	INS 25C
	A42 = C33*ZN	INS 26C
	A53 = C11*ZN	INS 27C
301	DO 303 I = 1,NDIM	INS 28C
	XN = SHAPE(1,I*NEL)	INS 29C
	YN = SHAPE(2,I*NEL)	INS 30C
	IF(NDIM*EG+3) ZN = SHAPE(3,I*NEL)	INS 31C
	ESTIF(I1,J1) = ESTIF(I1,J1) + XN*A11 + YN*A21 + ZN*A42	INS 32C
	ESTIF(I1,J1+1) = ESTIF(I1,J1+1) + XN*A12 + YN*A22	INS 33C
	ESTIF(I1+1,J1) = ESTIF(I1+1,J1) + XN*A21 + YN*A31	INS 34C
	ESTIF(I1+1,J1+1) = ESTIF(I1+1,J1+1) + XN*A22 + YN*A32 + ZN*A42	INS 35C
	IF(NDIM*EG+2) GO TO 303	INS 36C
	ESTIF(I1,J1+2) = ESTIF(I1,J1+2) + XN*A13 + ZN*A22	INS 37C
	ESTIF(I1+1,J1+2) = ESTIF(I1+1,J1+2) + YN*A13 + ZN*A21	INS 38C
	ESTIF(I1+2,J1) = ESTIF(I1+2,J1) + ZN*A31 + XN*A42	INS 39C
	ESTIF(I1+2,J1+1) = ESTIF(I1+2,J1+1) + YN*A42 + ZN*A12	INS 40C
	ESTIF(I1+2,J1+2) = ESTIF(I1+2,J1+2) + YN*A21 + ZN*A53 + XN*A22	INS 41C
303	I1 = I1 + NDF	INS 42C
304	J1 = J1 + NDF	INS 43C
305	CONTINUE	INS 44C
	RETURN	INS 45C
	END	INS 46C
	SUBROUTINE PSTRES(NDIM,SIG,P1,P2,P3)	PST 1C
	DIMENSION SIG(6)	PST 2C
	DATA RT2/1.414213562373097,PI23/2.09439510239321/	PST 3C
C...	STRESSES MUST BE STORED IN ARRAY SIG(6) IN THE ORDER	PST 4C
C	TAU-XX,TAU-XY,TAU-XZ,TAU-YY,TAU-YZ,TAU-ZZ FOR 3-D PROBLEMS	PST 5C
	IF(NDIM*LT+2*OR*NDIM*GT+3) RETURN	PST 6C
	IF(NDIM*EG+3) GO TO 100	PST 7C



C****	COMPUTE PRINCIPLE STRESSES FOR 2-D PROBLEMS	PST 8C
	TEMP = 22.5/ATAN(1.0)	PST 9C
	X11 = (SIG(1) + SIG(3))/2.	PST 10C
	X12 = (SIG(1) - SIG(3))/2.	PST 11C
	RHO = SQRT(X12**2 + SIG(2)*SIG(2))	PST 12C
	P1 = X11 + RHO	PST 13C
	P2 = X11 - RHO	PST 14C
	P3 = 45.0	PST 15C
	IF(X12.NE.0.0) P3 = TEMP*ATAN2(SIG(2),X12)	PST 16C
	RETURN	PST 17C
C****	COMPUTE PRINCIPLE STRESSES FOR 3-D PROBLEMS.	PST 18C
Y00	RHO = 0.	PST 19C
	X11 = (SIG(1) + SIG(4) + SIG(6))/3.0	PST 20C
	X12 = SIG(1)*(SIG(4)+SIG(6))+SIG(4)*SIG(6)-SIG(2)**2-SIG(3)**2	PST 21C
	-SIG(5)**2	PST 22C
	X13 = SIG(1)*SIG(4)*SIG(6)+2.*SIG(2)*SIG(3)*SIG(5)	PST 23C
	-SIG(1)*SIG(5)**2-SIG(4)*SIG(3)**2-SIG(6)*SIG(2)**2	PST 24C
	D12 = 3.*X11*X11 - X12	PST 25C
	TAU0 = 0.	PST 26C
	IF(D12.EQ.0.0) GO TO 200	PST 27C
	TAU0 = SQRT(2.*D12/3.)	PST 28C
	TEMP = (X13 + (D12 - X11*X11)*X11)*RT2/TAU0**3	PST 29C
	RHO = SQRT(ABS(1.-TEMP*TEMP))	PST 30C
	RHO = ATAN2(RHO,TEMP)/3.	PST 31C
C****	COMPUTE AND RETURN PRINCIPAL STRESSES	PST 32C
200	P1 = X11 + TAU0*R12*COS(RHO)	PST 33C
	P2 = X11 + TAU0*R12*COS(RHO + PI/3)	PST 34C
	P3 = X11 + TAU0*R12*COS(RHO + PI/3)	PST 35C
	RETURN	PST 36C
	END	PST 37C